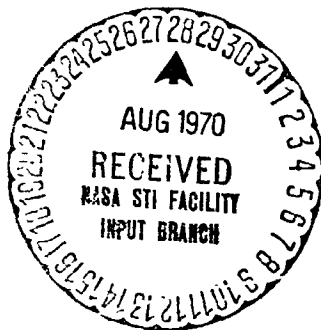


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TRANSLUNAR, LUNAR PARKING ORBIT, AND TRANSEARTH PROCEDURES

C-PRIME MISSION

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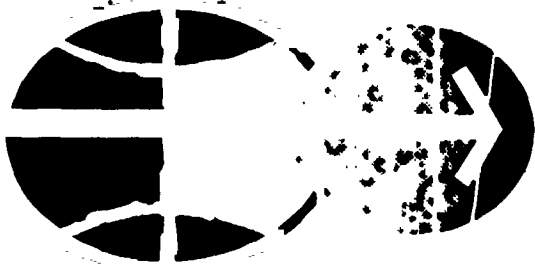
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FACILITY FORM 602

(ACCESSION NUMBER)
79
(PAGES)
TMX-65076
(NASA CR OR TMX OR AD NUMBER)

(THRU)
1
(CODE)
21
(CATEGORY)

MANNED SPACECRAFT CENTER
HOUSTON, TEXAS

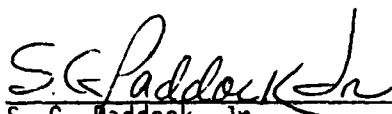


TRANSLUNAR, LUNAR PARKING ORBIT, AND
TRANSEARTH PROCEDURES

C-PRIME MISSION

AS-503/CSM-103
12 November 1968

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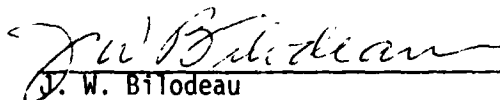
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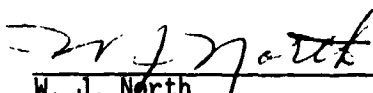
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1.0 PURPOSE

This document contains the nominal primary crew procedures for the CSM-103 spacecraft during the lunar Mission C-Prime. The specific time period covered by these procedures is from CSM/S-IVB separation at 3:20 GET through the last TEMCC (MCC7) at 144:50 GET.

The purpose of the Translunar, Lunar Parking Orbit and Transearth procedures document is to provide a single source of guidance, navigation and control procedures information for use in flight planning, in crew training, and in preparation of onboard checklists.

This document is a control document for this phase of the Mission C-Prime nominal crew procedures. Revisions to this document are subject to approval by the Procedures Configuration Control Board. Comments should be directed to Mr. Duane K Mosel, Flight Procedures Branch, Extension 5340 or to Stephen G. Paddock, Jr., Apollo Flight Crew Support Group, Houston Operations, McDonnell Douglas Astronautics Company, Extension 6101.

2.0 LIST OF ACRONYMS AND ABBREVIATIONS

AOS	Acquisition of Signal
ATT	Attitude
BEF	Blunt End Forward
CB	Circuit Breaker
CDR	Commander
CM	Command Module
CMC	Command Module Computer
CMP	Command Module Pilot
COAS	Crew Optical Alignment Sight
CSM	Command and Service Module
DAP	Digital Autopilot
DB	Deadband
DSKY	Display and Keyboard
DV	Delta Velocity
EMS	Entry Monitor System
ET	Event Timer
FDAI	Flight Director Attitude Indicator
FPS	Feet Per Second
GDC	Gyro Display Coupler
GETI	Ground Elapsed Time of Ignition
GMBL	Gimbal
GND	Ground
GPI	Gimbal Position Indicator
HA	Apogee Altitude
HOR	Horizon
HP	Perigee Altitude
IMU	Inertial Measurement Unit
LEB	Lower Equipment Bay

LOS	Loss of Signal
LM	Lunar Module
LMK	Landmark
LMP	Lunar Module Pilot
LOI	Lunar Orbit Injection
LV	Launch Vehicle
MDC	Main Display Console
MGA	Middle Gimbal Angle
MTVC	Manual Thrust Vector Control
OPT	Optics
ORDEAL	Orbital Rate Drive Earth and Lunar
OSS	Optical Subsystem
PAD	Data Voiced to Crew From Ground
PB	Pushbutton
PGNCS	Primary Guidance, Navigation, and Control System
PIPA	Pulse Integrating Pendulous Accelerometers
R	Range
R DOT	Range Rate
REFSMMAT	Reference Stable Member Matrix
RCS	Reaction Control System
RHC	Rotation Hand Controller
S-IVB	Saturn S-IVB Stage
SCS	Stabilization and Control System
SCT	Scanning Telescope
SECS	Sequence Events Control System
SEF	Small End Forward
SEP	Separation
SM	Service Module
SPS	Service Propulsion System
SC	Spacecraft

S/U	Setup
SXT	Sextant
TEMCC	Transearch Midcourse Correction
TFI	Time From Ignition
THC	Translation Hand Controller
THETA	Angle Between SC +X Axis and Local Horizontal
TIGN	Time of Ignition
TLI	Translunar Injection
TEI	Transearch Injection
TLM	Telemetry
TLMCC	Translunar Midcourse Correction
TRUN	Trunnion
TVC	Thrust Vector Control
VG	Velocity to be Gained
(XX:XX)	Indicates ground elapsed time from liftoff in hours: minutes

3.0 NOMINAL MISSION PROCEDURES

The nominal Mission C-Prime Translunar, Lunar Parking Orbit, and Transearth flight crew procedures presented in this document are divided into two sections, a major activities timeline and standard procedures. The timeline identifies all guidance, navigation, and control related activities as a function of GET. The procedures necessary to successfully perform an activity required only once during the mission, e.g. CSM/S-IVB separation, are presented in detail within the timeline. Whereas, the procedures for an activity reoccurring periodically throughout the mission, e.g. an IMU alignment, are presented in detail in an addendum to avoid repetition. The timeline identifies how and when the standard procedure should be applied and the standard procedure is sufficiently flexible to cover any option required during the mission.

3.1 Major Activities

3.1.1 Introduction

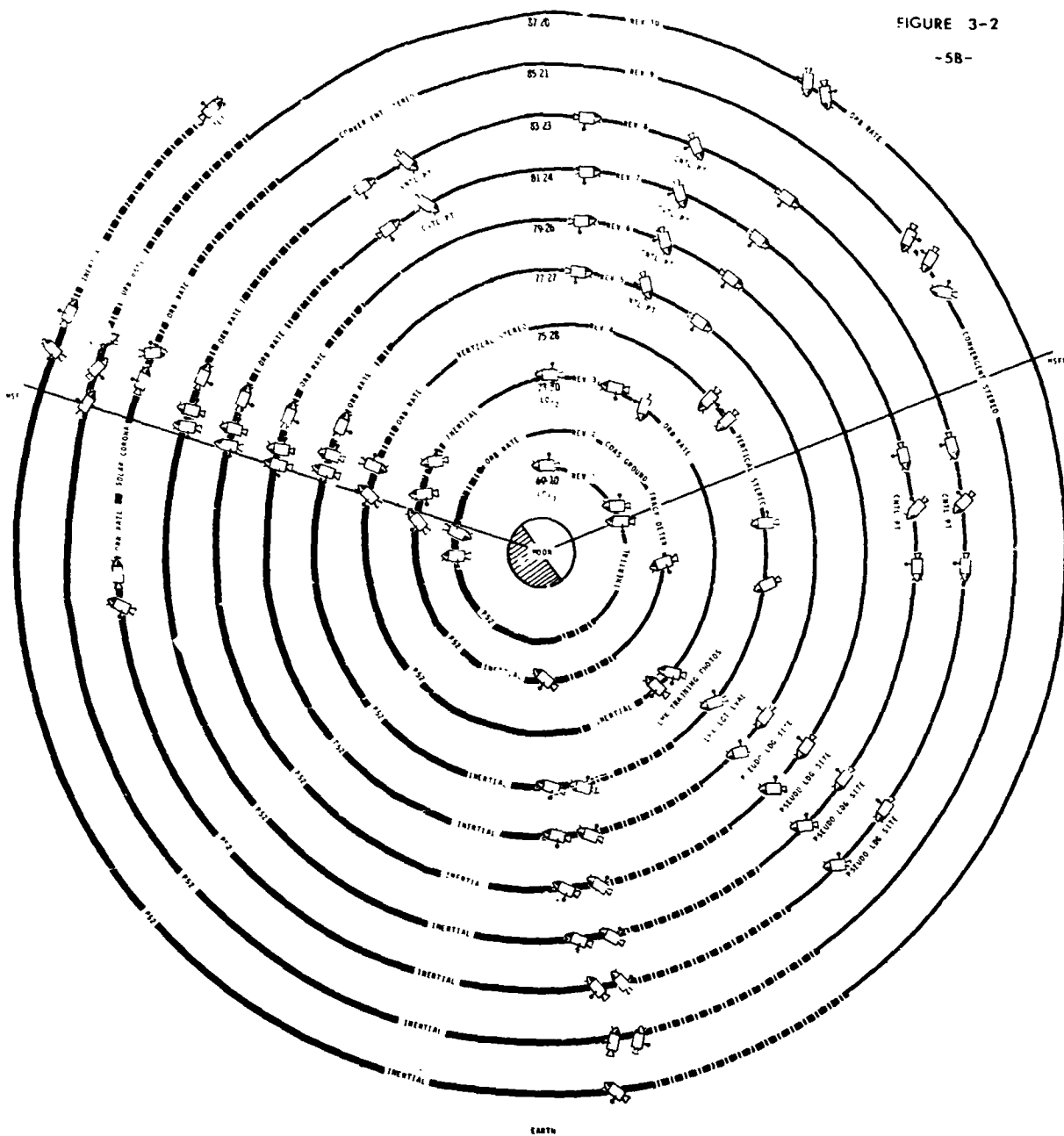
The major procedures timeline for the Mission C-Prime includes activities from after the TLI (2:51 GET) through the last Transearth midcourse correction (144:50 GET). Figure 3-1 shows the locations in time and approximate positions in space of selected major activities during the Translunar and Transearth phases. Figure 3-2 is a CSM attitude profile for the Lunar Parking Orbit phase. The nominal Mission C-Prime timeline assumes a 21 December 1968 launch, a 72 degree launch azimuth, and a nominal TLI ignition time of 02:50:31 GET.

The procedures presented in this document include primarily guidance, navigation, and control functions. Management of other systems, such as electrical power and environmental control are recommended as identified in Reference (1).

LUNAR PARKING ORBIT CSM ATTITUDE PROFILE

FIGURE 3-2

-5B-



Section 3.1.2

 MISSION C-PRIME PROCEDURES
 TLI TO CM/SM SEP
 MAJOR PROCEDURE TIMELINE

P47 F 16 83 CDR SET FT=58.00(COUNT UP) 1
 FDAI SEL=1/2
 MAN ATT(3)-RATE CMD 1
 ATT DR-MIN
 RATE=LOW
 CDR TRANS CONT PWR-ON(UP) 1
 RHC PWR DIRECT(BOTH)-MNA/MNR 1
 SC CONT-CMC
 CMC MODE=FREE 1
 RMAG MODE(3)-RATE2 8
 AUTO RCS SEL(16)-MNA 8
 VERIFY CR EDS(ALL)-CLOSED 8
 VERIFY CR RCS LOGIC(2)-CLOSED 8
 TVC SERVO PWR 1-ACI/MNA 7
 EDS PWR-ON(UP) 7
 VERIFY LV ENG LT-ON 1
 FDAI/SM(ROTH)(ORDEAL)-INER 13

D T S P L A R F V
 P R A C R F V
 E O Y F V
 T G S V
 P A N E L

(00+00) LIFT OFF

(02+51) TRANSLUNAR INJECTION

ASSUMED PRE-CSM/S-IVB SEPARATION
 SWITCH POSITIONS AND DAP
 CONFIGURATION
 (ADDENDUM 1)

(03+07) CMP LOAD N17 WITH SEP ATTITUDE
 KEY V63E
 CDR MONITOR S-IVH MANUEVER TO
 INERTIAL SEP ATTITUDE

(03+14)

-2MTN CDR START ET COUNT UP
 THC-ARMED
 RHC-ARMED 250
 CB PYRO A SEG A-CLOSED 250
 CB PYRO R SEG B-CLOSED 8
 CB SFCS ARM(ROTH)-CLOSED 8
 SECS LOGIC(BOTH)-ON(UP)
 LMP REPORT LOGIC ARM TO GND
 RECEIIVE GO FROM GND
 NONESS BUS-MNA
 FLT RCDR-RECORD
 TLM INPUTS PCM-HIGH
 UP TLM CMD-RESET THFN NORM
 TAPE RCDR PCM-PCM/ANLG

		TAPE RCDR RCD=RCD		CB EDS(ALL)=OPEN	0
		TAPE RCDR FWD=FWD		CB PYRO A SEQ A=OPEN	250
		TAPE MOTION TB=RP		CB PYRO R SEQ R=OPEN	250
		CMP COPY POST TLI DVS FROM DS-VY			
		KEY V32E			
		VERIFY DSKY DVS ZERO			
		CMR SECS PYRO ARM(ROTH)=ON(UP)	8		
-1MIN		CM RCS LOGIC=ON(UP)	1		
		RCS CMD=ON	2		
		CMP KEY V60E (N17=N20)	1		
		CMR EMS FUNCTION=DV	1		
		FMS MODE=AUTC	1		
		CMC MODE=HOLD			
				THC=THRUST(-)X FOR 5SEC	1
				EMS MODE=STBY	1
				EMS FUNCTION=OFF	1
				PRO	
			F 37 BB	KEY 00E	
			P00		
				CDR CMC MODE=HOLD	1
-2SEC		THC=THRUST(-)X AND HOLD		HMAG MODE(3)=RATE2	1
		(PRESEP TENSION)		MAN ATT(3)=RATE CMD	1
(03+20)				KEY V16 N20E	1
-05SEC		CSM/LV SEP PB=ON AND HOLD	1		
		VERIFY LV ENG 1 LT-OFF	1	KEY V16 N20E	
		CSM/LV SEP PB=RELEASE	1	COPY CURRENT R,P,Y	
				COMPUTE COMMANDED R,P,Y FOR	
				180 DEG PITCH AND 60 DEG	
				ROLL LEFT ATT MANEUVER	
				PRO	
		THC=RELEASE			
		VERIFY EMS DV IND READS 1FPS	1		
		(APPROXIMATE)	1		
		MAINTAIN INERTIAL SEP ATT	7		
		USING FDAI	7	KEY V49E	
		RHC PWR DIRECT(ROTH)=OFF	1	(COMMANDED R,P,Y)	
		CM RCS LOGIC=OFF	1	LOAD COMPUTED R,P,Y	
		EDS PWR=OFF	7	KEY V62E	
		TVC SERVO PWR 1-OFF	7	PRO	
		SECS LOGIC(BOTH)=OFF	8	(COMMANDED R,P,Y)	
		SECS PYRO ARM(ROTH)=SAFE	8	PRO	
		CB SECS ARM(ROTH)=OPEN	8	(COMMANDED R,P,Y)	
		CB SECS LOGIC(ROTH)=OPEN	8	MONITOR ATT MANEUVER	
		CH RCS LOGIC=OPEN	8	OBSERVE S-IVB IN CMD WINDOW	

F 50 18 (COMMANDED R.P.O.Y)
 KEY ENTER
 KEY V48E
 (DAP CONFIGURATION)
 F 04 46 LOAD 11102 (MIN DR,LOW RATE)
 11111
 B
 PRO
 F 06 47 (CSM AND LM WT)
 VERIFY PAD DATA
 PHO
 F 06 47 (SPS GMBL TRIM)
 VERIFY PAD DATA
 PRO
 (03,24) CDR PERFORM FORMATION FLYING
 AT 50 TO 100 FT
 TAKE PHOTOGRAPHS OF S-IVR
 (03,30) PROCEDURES FOR CSM FVASIVE
 MANEUVER
 P00 CMP WHEN COMP LT-OUT(NO INTEG,
 KEY V96E
 KEY V89E
 F 16 54 (R,ROOT,THETA)
 CDR MAN ATT(3)-ACC CMD
 MANEUVER SC TO LOCAL
 VERTICAL(+X AXIS TOWARD
 EARTH) WHILE TRANSLATING TO
 MAINTAIN S-IVR IN CMD WINDOW
 ATT 09-MIN 1
 RATE-LOW 1
 MAN ATT(3)-RATE CMD 1
 HOLD LOCAL VERTICAL ATTITUDE
 (DSKY THETA APPROX 270 DEG)
 WITH S-IVB BETWEEN SC
 AND EARTH
 CME PRO
 P67 KEY V37E47E
 F 16 83 (DVX,DVY,DVZ)
 CDR 8MAG MODE(3)-ATT1/RATE2
 EMS FUNCTION=DV
 EMS MODE=AUTO
 (03,35)
 THC=THRUST AFT (-X) UNTIL
 DSKY READS 1.5FPS
 VERIFY S-IVB RANGE OPENING
 VISIALLY
 THC=LOCKED
 PITCH UP FOR MSFN COVERAGE
 CMP PRO
 F 37 88
 P00 KEY 00E
 KEY V66E (TRANSFER CSM SV
 FROM CSM SLOTS TO LM SLOTS)
 KEY V48E
 F 04 46 (DAP CONFIGURATION)
 LOAD 11112 (MAX DB,LOW RATE)
 11111

B

NO. 1 (SIRIUS) (FAR HOR)
NO. 15 (PROCYON) (FAR HOR)
NO. 16 (PROCYON) (FAR HOR)

F 06 47 (CSM AND LM WT)
VERIFY PAD DATA

(05+00)

LMP RECEIVE BLOCK DATA PAD
FROM GND

F 06 48 (SPS GMBL TRIM)
VERIFY PAD DATA

(05+20)

PROCEDURES FOR LOI POSITION
DETERMINATION
(ADDENDUM 5) (P21)

CNR MONITOR SC INERTIAL ATTITUDE
HOLD UNTIL APPROX (04+00)

(04+00)

PROCEDURES FOR IMU REALIGN
TO REFSMMAT
(ADDENDUM 2) (P52)

(05+40)

PROCEDURES FOR HI GAIN PWR UP

BACKUP ALIGN NAV STARS
NO. 14 (CANOPUS)
NO. 12 (RIGEL)
BACKUP ALIGN CHECK NAV STAR
NO. 15 (SIRIUS)

P00

LMP CB HI GAIN ANT 225
FLT RUS-CLOSED 225
GRP 2-CLOSED 2
HI GAIN ANT TRACK-MAN 2
HI GAIN ANT SERVO ELEC-PRIM 2
HI GAIN ANT BEAM-WIDE 2
HI GAIN ANT PWR-POWER 2

(04+15)

PROCEDURES FOR GDC ALIGN TO IMU
(ADDENDUM 3)

F 06 51

(RHO,GAMMA,BLANK)
LMP RECORD HI GAIN ANT
GMBL COORDINATES
CMP PRO

(04+20)

PROCEDURES FOR STAR/EARTH
HORIZON SIGHTINGS - 5 SETS
OF 3 MARKS EACH
(ADDENDUM 4) (P23)

LMP HI GAIN ANT POSITION 2
SET ANT PITCH TO RHO 2
SET ANT YAW TO GAMMA 2
S-BAND ANT OMNI-HI GAIN 3
VERIFY HI GAIN ANT S-BAND
ANT IND GREATER THAN 2
HALF SCALE 2
HI GAIN ANT TRACK-AUTO 2

RECOMMENDED NAV STARS
NO. 14 (CANOPUS) (FAR HOR)
NO. 15 (SIRIUS) (FAR HOR)

9

FINAL MISSION C-PRIME TRANSLUNAR,LUNAR PARKING ORBIT,AND TRANSEARTH TIMELINE -12 NOV 1968

HI GAIN ANT BEAM-AS REQUIRED
PERFORM HI GAIN ANT CHECKOUT

(07+30)

PROCEDURES FOR GDC DRIFT CHECK
(ADDENDUM 7)

(07+40)

PROCEDURES FOR IMU REALIGN TO
REFSMAT FOR DRIFT CHECK
(ADDENDUM 2) (P52)

RACKUP ALIGN NAV STARS

NO. 14 (CANOPUS)
NO. 12 (RIGEL)
RACKUP ALIGN CHECK NAV STAR
NO. 15 (SIRIUS)

(07+55)

PROCEDURES FOR GDC ALIGN TO I-U
(ADDENDUM 3)

(08+00)

PROCEDURES FOR GND UPLINK OF CSM
STATE VECTOR AND P30 TARGET LOAD
(ADDENDUM 4) (P27)

FOLLOWING GND UPLINK

P00 LEB WHEN COMP LT-OUT (NO INTEG)
KEY V96E
KEY V03E

F 16 54 (R, KDOT, THETA)
COPY R AND RDOT
PRO

F 04 06 (TRACK AXIS OPTION)
LOAD 00002 IN R2
PRO

KEY V89E

F 06 18 (COMPUTER R,P,Y)
COPY R,P,Y
KEY V30E

KEY V37E00E

P00

KEY V47E (TRANSFER CSM SV
FROM LM SLOTS TO CSM SLOTS)

(08+20)

PROCEDURES FOR EXT DV TARGETING
(ADDENDUM 9) (P30)

PROPULSION SYSTEM SELECTION
USE RCS FOR DV GREATER THAN
1FPS AND LESS THAN OR EQUAL
TO 5FPS
USE SPS FOR DV GREATER THAN
5FPS

TRIM DVS TO ZERO

(08+25)

PROCEDURES FOR SM RCS PROPULSION
MONITOR CHECK
(ADDENDUM 10)

(08+30)

PROCEDURES FOR SPS THRUST SETUP
(ADDENDUM 11) (P40)

(09*00) FIRST TRANS LUNAR MIDCOURSE
 (17*05) PROCEDURES FOR TERMINATING PTC
 (ADDENDUM 14)

(09*05) PROCEDURES FOR SM RCS AND SPS
 PROPULSION MONITOR CHECK
 (ADDENDUM 10)
 (17*10) PROCEDURES FOR IMU REALIGN
 TO REFSMMAT
 (ADDENDUM 2) (P52)

(09*10) PROCEDURES FOR STAR/EARTH
 LANDMARK SIGHTINGS - 3 SFTS
 OF 3 MARKS EACH
 (ADDENDUM 4) (P23)
 BACKUP ALIGN NAV STARS
 NO. 14(CANOPUS)
 NO. 12(RIGEL)
 BACKUP ALIGN CHECK NAV STAR
 NO. 15(SIRIUS)

(09*50) RECOMMENDED NAV STARS
 NO. 15(SIRIUS)
 NO. 15(SIRIUS)
 NO. 16(PROCYON)
 RECOMMENDED LANDMARK
 NO. 5(ALTERNATES-7,10,20,32,
 40,41,56,69,70,100,
 115,127,128)
 (17*25) PROCEDURES FOR GOC ALIGN TO IMU
 (ADDENDUM 3)
 (17*30) PROCEDURES FOR STAR/EARTH
 HORIZON SIGHTINGS - 5 SETS
 OF 3 MARKS EACH
 (ADDENDUM 4) (P23)
 RECOMMENDED NAV STARS
 NO. 22(REGULUS)(FAR HOR)
 NO. 22(REGULUS)(FAR HOR)
 NO. 16(PROCYON)(FAR HOR)
 NO. 16(PROCYON)(FAR HOR)
 NO. 16(PROCYON)(FAR HOR)

(09*55) PROCEDURES FOR ESTABLISHING PTC
 (ADDENDUM 13)
 (12*00) LMP RECEIVE BLOCK DATA PAD
 FROM GND
 (12*20) PROCEDURES FOR LOI POSITION
 DETERMINATION

(18+25)	(ADUENDUM 5) PROCEDURES FOR ESTABLISHING PTC (ADUENDUM 13)	(P21)	NO. 16 (PROCYON) (FAR HOR) NO. 22 (REGULUS) (FAR HOR) NO. 2A (SPICA) (NEAR HOR)
(25+10)	LMP RECEIVE BLOCK DATA PAD FROM GND	(26+50)	PROCEDURES FOR LOI POSITION DETERMINATION (ADDFNDUM 5) (P21)
(25+55)	PROCEDURES FOR TERMINATING PTC (ADUENDUM 14)	(27+00)	PROCEDURES FOR GND UPLINK OF CSM STATE VECTOR AND P30 TARGET LOAD (ADDFNDUM 8) (P27)
(26+00)	PROCEDURES FOR IMU REALIGN TO REFSMMAT (ADUENDUM 2)		FOLLOWING GND UPLINK
	RACKUP ALIGN NAV STARS NO. 14 (CANOPUS) NO. 12 (RIGEL) RACKUP ALIGN CHECK NAV STAR NO. 15 (SIRIUS)		P00 LER WHEN COMP LT-OUT (NO INTEG) KEY V96E KEY V83E F 16 54 (R, ROOT, THETA) COPY R AND RDOY PR0
(26+15)	PROCEDURES FOR GDC ALIGN TO I.U (ADUENDUM 3)		KEY V89E F 04 06 (TRACK AXIS OPTION) LOAD 00002 IN R2 PR0
(26+20)	PROCEDURES FOR STAR/EARTH HORIZON SIGHTINGS - 3 SETS OF 3 MARKS EACH (ADUENDUM 4)		F 06 1A (COMPUTED R,P,Y) COPY R,P,Y KEY V34E KEY V37E00E KEY V47E (TRANSFER CSM SV FROM LM SLOTS TO CSM SLOTS)
	RECOMMENDED NAV STARS		

(27+30)

PROCEDURES FOR EXT DV TARGETING
(ADDENDUM 9) (P30)

PROPULSION SYSTEM SELECTION
USE RCS FOR DV GREATER THAN
IFPS AND LESS THAN OR EQUAL
TO SFPS (A DV GREATER THAN
SFPS REQUIRES SCHEDULING THE
MIDCOURSE AT THE EARLIER
POSSIBLE TIME. HENCE THE SPS
SHOULD NOT BE REQUIRED FOR
THIS MIDCOURSE CORRECTION)

TRIM DVS TO ZERO

(29+00)

(27+35)

PROCEDURES FOR SM RCS PROPULSION
MONITOR CHECK
(ADDENDUM 10)

(29+05)

(27+40)

PROCEDURES FOR RCS THRUST SETUP
(ADDENDUM 12) (P41)

(31+55)

(28+00)

SECOND TRANSLUNAR MIDCOURSE

(34+00)

(28+05)

PROCEDURES FOR SM RCS AND SPS
PROPULSION MONITOR CHECKS
(ADDENDUM 10)

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Original Document

(28+20)

PROCEDURES FOR STAR/EARTH
HORIZON SIGHTINGS - 4 SETS
OF 3 MARKS EACH
(ADDENDUM 4) (P23)

RECOMMENDED NAV STARS
NO. 16 (PROCYON) (FAR HOR)
NO. 22 (REGULUS) (FAR HOR)
NO. 21 (ALPHARD) (FAR HOR)
NO. 26 (SPICA) (NEAR HOR)
(NO. 21 (ALPHARD) HAS
MAG 2.2 AND MAY NOT BE
VISIBLE. IF NOT SELECT
NO. 22 (REGULUS))

PROCEDURES FOR LOI POSITION
DETERMINATION
(ADDENDUM 5) (P21)

PROCEDURES FOR ESTABLISHING PIC
(ADDENDUM 13)

PROCEDURES FOR TERMINATING PTC
(ADDENDUM 14)

PROCEDURES FOR IMU REALIGN
TO REFSMMAT
(ADDENDUM 2) (P52)

BACKUP ALIGN STARS
NO. 14 (CANOPUS)
NO. 12 (RIGEL)

(34+15)	BACKUP ALIGN CHECK NAV STAR NO. 15(SIRIUS)	PROCEDURES FOR TERMINATING PTC (ADDENDUM 14)
(34+20)	PROCEDURES FOR GNC ALIGN TO I.U (ADDENDUM 3)	PROCEDURES FOR IMU REALIGN TO REFSMAY (ADDENDUM 2) (P52)
(34+20)	PROCEDURES FOR STAR/EARTH HORIZON SIGHTINGS - 3 SETS OF 3 MARKS EACH (ADDENDUM 4) (P21)	BACKUP ALIGN STARS NO. 14(CANOPUS) NO. 12(RIGEL) BACKUP ALIGN CHECK NAV STAR NO. 15(SIRIUS)
(34+50)	RECOMMENDED NAV STARS NO. 16(PROCYON) NO. 22(REGULUS) NO. 26(SPICA)	PROCEDURES FOR GNC ALIGN TO IMU (ADDENDUM 3)
(34+50)	PROCEDURES FOR LOI POSITION DETERMINATION (ADDENDUM 5) (P21)	PROCEDURES FOR STAR/LUNAR HORIZON SIGHTINGS - 5 SETS OF 3 MARKS EACH (ADDENDUM 4) (P23)
(34+55)	LMP RECEIVE BLOCK DATA PAD FROM GND	RECOMMENDED NAV STARS NO. 33(ANTARES) (NEAR HOR) NO. 33(ANTARES) (NEAR HOR) NO. 37(NUNKI) (NEAR HOR) NO. 45(FOMALHAUT) (FAR HOR) NO. 42(PEACOCK) (FAP HOR)
(35+00)	PROCEDURES FOR ESTABLISHING PTC (ADDENDUM 13)	
(44+00)	LMP RECEIVE BLOCK DATA PAD FROM GND	PROCEDURES FOR LOI POSITION DETERMINATION (ADDENDUM 5) (P21)
(44+35)		

(46+00)

PROCEDURES FOR GND UPLINK OF CSM
STATE VECTOR AND P30 TARGET LOAD
(ADDENDUM 8) (P27)

FOLLOWING GND UPLINK

P00 LEH WHEN COMP LT-OUT(NO INTEG)

KEY V96E

KEY V83E

F 16 54 (R,RDOT,THETA)

COPY R AND RDOT
PRO

KEY V89E

F 04 05 (TRACK AXIS OPTION)

LOAD 00002 IN R2

PRO

F 06 10 (COMPUTED R,P,Y)

COPY R,P,Y

KEY V34E

P00

KEY V37E00E

KEY V47E (TRANSFER CSM SV
FROM LM SLOTS TO CSM SLOTS)

(46+30)

PROCEDURES FOR EXT DV TARGETING
(ADDENDUM 9) (P30)

PROPULSION SYSTEM SELECTION

USE RCS FOR DV GREATER THAN
1FPS AND LESS THAN OR EQUAL
TO 5FPS (A DV GREATER THAN

5FPS REQUIRES SCHEDULING THE
MIDCOURSE AT THE EARLIEST
POSSIBLE TIME,HENCE THE SPS
SHOULD NOT BE REQUIRED FOR
THIS MIDCOURSE CORRECTION)

TRIM OVS TO ZERO

(46+35)

PROCEDURES FOR SM RCS PROPULSION
MONITOR CHECK
(ADDENDUM 10)

(46+40)

PROCEDURES FOR RCS THRUST SETUP
(ADDENDUM 12) (P41)

(47+00)

THIRD TRANSLUNAR MIDCOURSE

(47+05)

PROCEDURES FOR SM RCS AND SPS
PROPULSION MONITOR CHECKS
(ADDENDUM 10)

(47+25)

PROCEDURES FOR STAR/EARTH
HORIZON SIGHTINGS - 3 SETS
OF 3 MARKS EACH
(ADDENDUM 4) (P23)

RECOMMENDED NAV STARS

NO. 16 (PROCYON) (FAR HOR)

NO. 22 (REGULUS) (FAR HOR)

NO. 26 (SPICA) (NEAR HOR)

(47+55) PROCEDURES FOR LOI POSITION DETERMINATION (ADDENDUM 4) (P21) (P23)
 OF 3 MARKS EACH (ADDENDUM 4)

(48+00) RECOMMENDED NAV STARS
 NO. 37(NUNKI) (NEAR HOR)
 NO. 37(NUNKI) (NEAR HOR)
 NO. 33(ANTARES)(NEAR HOR)
 NO. 33(ANTARES)(NEAR HOR)
 NO. 33(ANTARES)(NEAR HOR)

(51+00) PROCEDURES FOR ESTABLISHING PTC (ADDENDUM 13)

(51+20) LMP RECEIVE BLOCK DATA PAD FROM GND (P21)
 PROCEDURES FOR LOI POSITION DETERMINATION (ADDENDUM 5) (P21)

(52+05) PROCEDURES FOR TERMINATING PTC (ADDENDUM 14)

(52+10) PROCEDURES FOR IMU REALIGN TO REFSMMAT (ADDENDUM 2) (P52)
 BACKUP ALIGN NAV STARS
 NO. 14(CANOPUS)
 NO. 12(RIGEL)
 BACKUP ALIGN CHECK NAV STAR
 NO. 15(SIRIUS)

(52+25) PROCEDURES FOR GDC ALIGN TO I.U.U (ADDENDUM 3)
 LMP RECEIVE GND BLOCK DATA PAD

(52+30) PROCEDURES FOR STAR/LUNAR HORIZON SIGHTINGS - 5 SETS (ADDENDUM 2) (P52)
 PROCEDURES FOR IMU ALIGN TO UPLINKED PREFERRED REFSMMAT (ADDENDUM 2) (P52)

(52+25) PROCEDURES FOR GND UPLINK OF CSM STATE VECTOR,P30 TARGET LOAD,AND PREFERREC LUNAR ORBIT REFSMMAT (ADDENDUM 8) (P27)

BACKUP ALIGN NAV STARS
NO. 14 (CANOPUS)
NO. 12 (RIGEL)
BACKUP ALIGN CHECK NAV STARS
NO. 15 (SIRIUS)

(60+25)

PROCEDURES FOR GDC ALIGN TO IMU
(ADDENDUM 3)

(60+30)

PROCEDURES FOR EXT DV TARGETING
(ADDENDUM 9) (P30)

PROPULSION SYSTEM SELECTION
USE RCS FOR DV LESS THAN OR
EQUAL TO SFPS (A DV GREATER
THAN SFPS WILL BE SCHEDULED
AT THE EARLIEST POSSIBLE TIME
AND HENCE THIS MICCOURSE
SHOULD NOT REQUIRE THE SPS)

TRIM DVS TO ZERO

(60+35)

PROCEDURES FOR SM RCS PROPULSION
MONITOR CHECK
(ADDENDUM 10)

(60+40)

PROCEDURES FOR RCS THRUST SETUP
(ADDENDUM 12) (P41)

(61+00)

LMP RECEIVE LO11 PAD FROM GND

FOURTH TRANSLUNAR MICCOURSE

(61+05)

PROCEDURES FOR SM RCS AND SPS
PROPULSION MONITOR CHECKS
(ADDENDUM 10)

(61+10)

PROCEDURES FOR LO11 POSITION
DETERMINATION
(ADDENDUM 5) (P21)

(61+15)

PROCEDURES FOR ESTABLISHING PTC
(ADDENDUM 13)

(65+55)

PROCEDURES FOR TERMINATING PTC
(ADDENDUM 14)

(66+00)

PROCEDURES FOR IMU REALIGN
TO REFSMAT
(ADDENDUM 2) (P52)
BACKUP ALIGN NAV STARS
NO. 14 (CANOPUS)
NO. 12 (RIGEL)
BACKUP ALIGN CHECK NAV STAR
NO. 15 (SIRIUS)

(66+15)

PROCEDURES FOR GDC ALIGN TO IMU
(ADDENDUM 3)

FOR STAR CHECK
(ROLL GMBL WILL RE THAT
WHICH PROVIDES COMM)

CDR FDUAI SCALE-5/1

FDAI SELECT-1/2
MAN ATT(3)-RATE CMD
LIMIT CYCLE-OFF
ATT DB-MAX
RATE-LOW
SC CONT-CMC
CMC MODE-AUTO
RMAG MODE(3)-RATE2
RHC-ARMED

LEB KEY V49E

F 06 22 (FINAL GMBL ANGLES)
LOAD BUJRN ATT STAR
CHECK GMBL ANGLES
KEY V62E

PRO

F 50 18 (COMMANDED K,P,Y)

PRO

06 18 (COMMANDED R,P,Y)
CDR MONITOR ATT MANEUVER

FOR GMBL LOCK

F 50 18 (COMMANDED R,P,Y)
VERIFY R,P,Y AGREE WITH
PAD R,P,Y WITHIN 5 DEG

NULL FDAI NEEDLES WITH RH
LER PRO

G/N PWR-OPTICS-ON

OPT ZERO-ZERO(15SEC)
OPT TELTRUN-SLAVE TO SXT
RETICLE BRT TW-ADJUST

OPT ZERO-OFF
OPT MODE-CMC
KEY V41NG1E

F 21 92 (BLANK,BLANK,BLANK)
LOAD PAD SHAFT ANGLE(+XXX.XX)
IN R1 AND PAD TRUNNION ANGLE
(+XX.XXX) IN R2

41 88 (SHAFT,TRUNNION,BLANK)
MONITOR OPT DRIVE TO PAD
VALUES

VERIFY PAD STAR IN SXT

OPT MODE-MAN

DRIVE TRUN LESS THAN 5 DEG

OPT ZERO-ZERO

RETICLE BRT TW-MIN ARTNESS

(66+30)

PROCEDURES FOR ESTABLISHING PTC
(ADDENDUM 13)

(67+30)

PROCEDURES FOR TERMINATING PTC
(ADDENDUM 14)

(67+35)

PROCEDURES FOR GDC DRIFT CHECK
(ADDENDUM 7)

(67+40)

PROCEDURES FOR IMU REALIGN TO
REFSMAT FOR DRIFT CHECK
(ADDENDUM 2) (P52)

HACKUP ALIGN NAV STARS

NO. 14 (CANOPUS)

NO. 12 (RIGEL)

RACKUP ALIGN CHECK NAV ST.R
NO. 15(SIRIUS)
PROCEDURES FOR SPS THRUST SETUP
(ADDENDUM 11) (P40)

(67+55) MSFN LOS
(68+57)

PROCEDURES FOR GDC ALIGN TO IMU
(ADDENDUM 3)

(69+07)

FIRST LUNAR ORBIT INJECTION

(68+00)

CDR RECEIVE GO/NO-GO FOR LOI

PROCEDURES FOR GND UPLINK OF ASM
STATE VECTOR AND P30 TARGET LOAD
(ADDENDUM 8) (P27)

*** BEGIN REV 1 ***
(69+09)

LMP INITIATE PRE-LOI SYSTEMS CHECK

(68+20)

PROCEDURES FOR LOI POSITION
DETERMINATION
(ADDENDUM 5) (P21)

LMP TLM INPUTS PCM-LAR
TAPE RCDR-FWD

(69+20)

PROCEDURES FOR SM RCS AND SPS
PROPULSION MONITOR CHECKS
(ADDENDUM 10)

(68+25)

PROCEDURES FOR EXT DV TARGETING
(ADDENDUM 9) (P30)

PROCEDURES FOR GDC ALIGN TO IMU
(ADDENDUM 3)

PROPULSION SYSTEM SELECTION
USE SPS AND DO NOT TRIM D/S

PROCEDURES FOR ORDEAL SETUP
(ADDENDUM 18)

(68+30)

PROCEDURES FOR SM RCS PROPULSION
MONITOR CHECKS
(ADDENDUM 10)

(69+20)

CDR ATT DR-MAX
RATE-LOW
BMAG MODE(3)-ATT1/RATE2
SC CONT-SCS
MAN ATT(R)-ACC CMD

(68+35)

MAN ATT(P,Y)-RATE CMD
MANEUVER SC IN ROLL 180 DEG
MAN ATT(3)-RATE CMD

LMP SELECT S BD ANT OMNI-A OR B
DVRU-ON
SH HANGING-OFF

MSFN AOS
(69+30)

PROCEDURES FOR HI GAIN ACQUISITION
OF MSFN
(ADDENDUM 19)

LMP RECEIVE BLOCK DATA FOR
TEI FROM MSFN

SUNDOWN
(70+14)

PROCEDURES FOR IMU REALIGN
TO REFSMMAT
(ADDENDUM 2) (P52)

RACKUP ALIGN NAV STARS
NO. 14(CANOPUS)
NO. 17(RIGEL)
RACKUP ALIGN CHECK NAV STAR
NO. 15(SIRIUS)

(70+30)

PROCEDURES FOR GDC ALIGN TO I-U
(ADDENDUM 3)

PROCEDURES FOR ORDEAL

VERIFICATION
(ADDENDUM 18)

LMP UP TLM-CMD RESET
UP TLM-NORMAL

MSFN LOS
(70+55)

CDR MAN ATT(P)-ACC CMD
MANEUVER SC TO 315 DEG PITCH
AND 180 DEG ROLL W.R.T.
LOCAL HORIZONTAL
MAN ATT(P)-MIN IMP
INITIATE ORBITAL PITCH RATE

SUNUP
(71+00)

(71+05)

CMP CONTROL POINT LMK
FAMILIARIZATION
IDENTIFY IDENTIFICATION POINTS
AND CONTROL POINTS AT
RESPECTIVE TIMES OF CLOSEST
APPROACH

*** REGIN REV 2 ***
(71+21)

(71+30)

***PREPARE FOR TV PICTURES**

CDR MAN ATT(Y)-ACCEL CMD
YAW SC RIGHT TO 45 DEG YAW
315 DEG PITCH AND 180 DEG
ROLL W.R.T. LOCAL HORIZONTAL
MAN ATT(Y)-RATE CMD
MAINTAIN ORBITAL PITCH RATE
FOR TV PICTURES

LMP SELECT S HD ANT OMNI-4 OR B
NVBU-ON
SR RANGING-OFF

MSFN AOS
(71+38)

PROCEDURES FOR HI GAIN ACQUISITION
OF MSFN
(ADDENDUM 19)

(71+40)

PROCEDURES FOR GND UPLINK OF FSM
STATE VECTOR AND P30 TARGET LOAD
(ADDENDUM 8)

(71+50)

CDR MAN ATT(Y)-ACCEL CMD
YAW SC LEFT TO 0 DEG YAW
315 DEG PITCH AND 180 DEG
ROLL W.R.T. LOCAL HORIZONTAL
MAN ATT(Y)-RATE CMD
MAINTAIN ORBITAL PITCH RATE

(72+00)

CMP PSEUDO LANDING SITE
FAMILIARIZATION
IDENTIFY IDENTIFICATION POINTS

AND PSEUDO LANDING SITE AT
RESPECTIVE TIMES OF CLOSEST
APPROACH

CDR STOP PITCH RATE BY SWITCHING
MAN ATT(3)-RATE CMD

(72+15)

CDR RECEIVE GO/NO-GO FOR LOI2

SUNDOWN
(72+22)

LMP RECEIVE BLOCK DATA FOR
TEI FROM MSFN

(72+30)

PROCEDURES FOR IMU REALIGN
TO REFSMMAT
(ADDENDUM 2) (P52)

BACKUP ALIGN NAV STARS
NO. 14(CANOPUS)
NO. 12(RIGEL)
BACKUP ALIGN CHECK NAV STAR
NO. 15(SIRIUS)

(72+45)

PROCEDURES FOR GDC ALIGN TO IMU
(ADDENDUM 3)

(72+50)

PROCEDURES FOR EXT DV TARGETING
(ADDENDUM 9) (P30)

PROPULSION SYSTEM SELECTION
USE SPS/DO NOT TRIM DV

(ADDENDUM 10)
PROCEDURES FOR GDC ALIGN TO IMU
(ADDENDUM 3)

(72+55)
PROCEDURES FOR SM RCS PROPULSION
MONITOR CHECK
(ADDENDUM 10)

PROCEDURES FOR ORDEAL SETUP
(ADDENDUM 18)

LMP UP TLM-CMU RESET
UP TLM-NORMAL

CDR ATT DB-MAX
RATE-LOW

MSFN LOS
(73+03)
PROCEDURES FOR SPS THRUST SETUP
(ADDENDUM 11)

BAG MODE(3)-ATT 1/RATE 2
SC CONT-SCS
MAN ATT(R,P)-ACCEL CMD
MAN ATT(Y)-RATE CMD
MANEUVER SC TO 307 DEG PITCH
AND 190 DEG ROLL W.R.T.
LOCAL HORIZONTAL

SUNUP
(73+09)
(73+31)
SECOND LUNAR ORBIT INJECTION
LMP TLM INPUTS PCM-LBR
TAPE RCDR-FWD

LMP SELECT S BD ANT OMNI-A OR B
OVBU-ON
SB RANGING-OFF

*** BEGIN REV 3 ***
(73+31)

MSFN AOS
(73+47)

PROCEDURES FOR HI GAIN ACQUISITION
OF MSFN
(ADDENDUM 19)

(73+35)
PROCEDURES FOR SM RCS AND SPS
PROPULSION MONITOR CHECKS

CDR WHEN PSEUDO LANDING SITE IS

ACQUIRED IN COAS-INCREASE
THE PITCH RATE TO KEEP X-AXIS
CAMERAS POINTED AT IT
TERMINATE PITCH RATE WHEN
LANDMARK IS NO LONGER VISIBLE
CDR MAN ATT(P)-ACCEL CMD
PITCH SC DOWN TO 0 DEG PITCH
AND 180 DEG ROLL W.R.T.
LOCAL HORIZONTAL
MAN ATT(P)-RATE CMD

SUNDOWN
(74+22)

LMP RECEIVE BLOCK DATA FOR
TEI FROM MSFN

(74+30)

PROCEDURES FOR IMU REALIGN
TO REFSMMAT
(ADDENDUM 2)

(P52)

BACKUP ALIGN NAV STARS
NO 14(CANOPUS)
NO 12(RIGEL)
BACKUP ALIGN CHECK NAV STAR
NO 15(SIRIUS)

(74+45)

PROCEDURES FOR GDC ALIGN TO IMU
(ADDENDUM 3)

(74+50)

PROCEDURES FOR ORDEAL
VERIFICATION
(ADDENDUM 18)

(74+55)

LMP BEGIN PHOTOGRAPHIC PREPARATION
FOR VERTICAL STEREO
NAVIGATION PHOTOGRAPHY

LMP UP TLM-CMD RESET
UP TLM-NORMAL

MSFN LOS
(74+59)

CDR MAN ATT(P)-ACCEL CMD
PITCH SC UP TO 270 DEG PITCH
AND 180 DEG ROLL W.R.T.
LOCAL HORIZONTAL
MAN ATT(P)-MIN IMP
INITIATE ORBITAL PITCH RATE

SUNUP
(75+09)

(75+13)

LMP KEY V06N65

(75+15)

CDR LMP WHEN X-AXIS CROSSES TERMINATOR
KEY ENTER/START CAMERA FOR
VERTICAL STEREO NAVIGATION
PHOTOGRAPHY

06 65 (GET)

LMP RECORD GET
CDR KEY N20E

06 20 (K,P,Y)

LMP RECORD PRESENT GMBL ANGLES

LMP MONITOR PHOTOGRAPHY

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FINAL MISSION C-PRIME TRANSLUNAR LUNAR PARKING ORBIT AND TRANSEARTH TIMELINE -12 NOV 1968

CDR MONITOR ORBITAL PITCH RATE

*** BEGIN REV 4 ***
(75+30)

(75+40) CDR MAN ATT(R)-ACCEL CMD
AT THE SUBSOLAR POINT-
ROLL SC 180 DEG TO 0 DEG ROLL
AND 270 DEG PITCH W.R.T.
LOCAL HORIZONTAL
MAN ATT(R)-RATE CMD
REINITIATE ORBITAL PITCH RATE

LMP SELECT S BD ANT OMNI-C OR U
DVRU-ON
SR RANGING-OFF

MSFN AOS
(75+46)

(75+55) CDR KEY V06N65
CDR,LMP KEY ENTER/STOP CAMERA
06 65 (GET)
LMP RECORD GET
CDR KEY N20E
06 20 (R,P,Y)
LMP RECORD PRESENT GMR L ANGLES
CDR STOP PITCH RATE

(76+00)-(TCA OF 70 DEG LONGITUDE WILL
BE UPDATED REALTIME)

CDR MAN ATT(P)-A CEL CMD
PITCH UP TO 5 DEG PITCH
AND 0 DEG ROLL W.R.T.
LOCAL HORIZONTAL
MAN ATT(P)-MIN IMP
INITIATE ORBITAL PITCH RATE
LMP ENTER LEB

(76+05)

***BEGIN LANDING SITE LIGHTING
EVALUATION**

LEB KEY V37E22E

P22

F 06 45 (BLANK,BLANK,MAX MGA)

LEB PRO

F 05 70 (LANDMARK CODE)

LOAD 10001 IN R2

LEB OPT ZERO-OFF

OPT MODE-MAN

POSITION TRUNNION TO 10 DEG

LEB IDENTIFY IDENTIFICATION POINTS
AND PSEUDO LANDING SITE AT
THEIR RESPECTIVE ACQUISITION
TIMES

EVALIATE THE LIGHTING BETWEEN
THE LANDING SITE AND THE
TERMINATOR

LEB UPON ACQUISITION OF THE

TERMINATOR-

OPT MODE-CMC

LEB PRO

06 92 (SHAFT,TRUN,BLANK)

EVALIATE AUTO OPTICS ACQUI-
TION OF THE PSEUDO LANDING

SITE
CDR ROLL TO AVOID SHAFT AXIS
WITHIN 10 DEG OF LANDING SITE
LEB WHEN LANDING SITE NO LONGER
VISIBLE-
OPT MODE-MAN
EVALUATE EARTHSHINE TRACKING
FEASIBILITY
LEB POSITION TRUNNION TO 5 DEG
OPT ZERO-ZERO
OPT MODE-MAN
LEB KEY V37E00E

P00

(76+20)

CDR MAN ATT(P)-RATE CMD
MAN ATT(R)-ACCEL CMD
ROLL SC 180 DEG TO ABOUT
0 DEG PITCH AND 180 DEG
ROLL W.R.T. LOCAL HORIZONTAL
MAN ATT(R)-RATE CMD

SUNDOWN
(76+21)

PROCEDURES FOR HI GAIN ACQUISITION
OF MSFN
(ADDENDUM 19)

(76+25)

PROCEDURES FOR GND UPLINK OF ASM
STATE VECTOR
(ADDENDUM 8)

CMP KEY V47E

LMP RECEIVE BLOCK DATA FOR
TEI FROM MSFN

(74+30)

PROCEDURES FOR IMU REALIGN
TO REFSMMAT
(ADDENDUM 2) (P52)

BACKUP ALIGN NAV STARS
NO 14(CANOPUS)
NO 12(RIGEL)
BACKUP ALIGN CHECK NAV STAR
NO 15(SIRIUS)

(76+45)

PROCEDURES FOR GNC ALIGN TO IMU
(ADDENDUM 3)

(76+50)

PROCEDURES FOR ORDEAL VERIFICATION
(ADDENDUM 18)

LMP UP TLM-CMD RESET
UP TLM-NORMAL

MSFN LOS
(76+58)

CDR MAN ATT(R,P)-ACCEL CMD
MANUEVER SC TO 5 DEG PITCH
AND 0 DEG ROLL W.R.T.
LOCAL HORIZONTAL
MAN ATT(R)-RATE CMD
MAN ATT(P)-MIN IMP
INITIATE ORBITAL PITCH RATE
FOR LANDMARK SIGHTINGS

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FINAL MISSION C-PRIME TRANSLUNAR-LINAR PARKING ORBIT AND TRANSEARTH TIMELINE -12 NOV 1968

SUNUP (77+07)		SUNDOWN	PROCEDURES FOR HI GAIN ACQUISITION OF MSFN (ADDENDUM 19)
(77+25)	PROCEDURES FOR TRACKING THE SECOND CONTROL POINT LANDMARK (NO AUTO OPTICS) (ADDENDUM 16)	(7A+25)	PROCEDURES FOR GND UPLINK OF CSM STATE VECTOR (ADDENDUM 8)
*** BFGIN REV 5 *** (77+29)			CMP KEY V47E
	LMP SELECT S RD ANT OMNI-C OR D DVRU-ON SR RANGING-OFF	(7A+30)	LMP RECEIVE BLOCK DATA FOR TEI FROM MSFN
MSFN AOS (77+44)			PROCEDURES FOR IMU REALIGN TO REFSMAT (ADDENDUM 12) (P52)
(78+05)	PROCEDURES FOR TRACKING THE PSEUDO LANDING SITE (ADDENDUM 15)		HACKUP ALIGN NAV STARS NO 14 (CANOPUS) NO 12 (RIGEL) HACKUP ALIGN CHECK NAV STAR NO 15 (SIRIUS)
(78+15)	CNR MAN ATT(R)-ACCFL CMD ROLL SC 180 DEG TO ABOUT 300 DFG PITCH AND 180 DEG ROLL W.R.T. LOCAL HORIZO. TAI MAN ATT(R,P)-RATE CMD	(7A+45)	PROCEDURES FOR GDC ALIGN TO IMU (ADDENDUM 3)
(78+20)		(7A+50)	PROCEDURES FOR ORDEAL VERIFICATION (ADDENDUM 18)
			LMP UP TLM-CMD RESET

UP TLM-NORMAL

MSFN LOS
(79+55)

(80+00)

PROCEDURES FOR TRACKING THE
PSEUDO LANDING SITE
(ADDENDUM 15) (P22)

CDR MAN ATT(R,P)-ACCEL CMD
MANEUVER SC TO 5 DEG PITCH
AND 0 DEG ROLL W.R.T.
LOCAL HORIZONTAL
MAN ATT(R)-RATE CMD
MAN ATT(P)-MIN IMP
INITIATE ORBITAL PITCH RATE
FOR LANDMARK SIGHTINGS

(80+15)

CDR MAN ATT(R)-ACCEL CMD
ROLL SC 180 DEG TO ABOUT
300 DEG PITCH AND 180 DEG
ROLL W.R.T. LOCAL HORIZONTAL
MAN ATT(R,P)-RATE CMD

SUNUP
(80+18)

SUNUP
(79+06)

PROCEDURES FOR HI GAIN ACQUISITION
OF MSFN
(ADDENDUM 19)

(79+25)

PROCEDURES FOR TRACKING THE
SECOND CONTROL POINT LANDMARK
(WITH AUTO OPTICS) (P22)
(ADDENDUM 17)

(80+25)

PROCEDURES FOR GND UPLINK OF GSM
STATE VECTOR
(ADDENDUM 8)

*** BGIN REV 6 ***
(79+28)

CMP KEY V47E

LMP RECEIVE BLOCK DATA FOR
TEI FROM MSFN

LMP SELECT S HD ANT OMNI-C OR D
DVBU-ON
SB RANGING-OFF

(80+30)

PROCEDURES FOR IMU REALIGN
TO REFSMAT
(ADDENDUM 12) (P52)

MSFN AOS
(79+42)

BACKUP ALIGN NAV STARS
NO 14(CANOPUS)

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FINAL MISSION C-PRIME TRANS-LUNAR-LINAR PARKING ORBIT AND TRANSEARTH TIMELINE - 12 NOV 1968

(81*25) PROCEDURES FOR TRACKING THE
 SECOND CONTROL POINT LANDMARK
 (WITH AUTO OPTICS) (P22)
 (ADDENDUM 17)

(80*45) NO 12 (RIGEL)
 BACKUP ALIGN CHECK NAV STAR
 NO 15 (SIRIUS)

(80*50) PROCEDURES FOR GDC ALIGN TO I-U
 (ADDENDUM 3)

(80*50) PROCEDURES FOR ORDEAL VERIFICATION
 (ADDENDUM 18)

LMP UP TLM-CMU RESET
 I/P TLM-NORMAL

MSFN LOS
 (80*54)

CDR MAN ATT(R,P)-ACCEL CMD
 MANEUVER SC TO 5 DEG PITCH
 AND 0 DEG ROLL W.R.T.
 LOCAL HORIZONTAL
 MAN ATT(R)-RATE CMD
 MAN ATT(P)-MIN IMP
 INITIATE ORBITAL PITCH RATE
 FOR LANDMARK SIGHTINGS

SUNUP
 (81*04)

(81*10) PROCEDURES FOR TRACKING THE
 FIRST CONTROL POINT LANDMARK
 (NO AUTO OPTICS)
 (ADDENDUM 16) (P22)

(81*25) PROCEDURES FOR TRACKING THE
 SECOND CONTROL POINT LANDMARK
 (WITH AUTO OPTICS)
 (ADDENDUM 17) (P22)

*** REGIN REV 7 ***
 (81*27)

LMP SELECT S BD ANT OMNI-C OR D
 DVBU-ON
 SB RANGING-OFF

MSFN AOS
 (81*40)

(82*00) PROCEDURES FOR TRACKING THE
 PSEUDO LANDING SITE
 (ADDENDUM 15) (P22)

(82*15) CDR MAN ATT(R)-ACCEL CMD
 ROLL SC 180 DEG TO ABOUT
 300 DEG PITCH AND 180 DEG
 ROLL W.R.T. LOCAL HORIZONTAL
 MAN ATT(R,P)-RATE CMD

SUNDOWN
(82+17)

PROCEDURES FOR HI GAIN ACQUISITION
OF MSFN
(ADDENDUM 19)

(82+25)

PROCEDURES FOR GND UPLINK OF ASM
STATF VECTOR
(ADDENDUM 8)

CMP KEY V47E

LMP RECEIVE BLOCK DATA FOR
TEI FROM MSFN

(82+30)

PROCEDURES FOR IMU REALIGN
TO REFMMAT
(ADDENDUM 12) (P55)

BACKUP ALIGN NAV STARS
NO 14 (CANOPUS)
NO 12 (RIGEL)
BACKUP ALIGN CHECK NAV STAR
NO 15 (SIRIUS)

(82+45)

PROCEDURES FOR GDC ALIGN TO I.U
(ADDENDUM 3)

(82+50)

PROCEDURES FOR ORDEAL VERIFICATION
(ADDENDUM 18)

LMP UP TLM-CMD RESET
UP TLM-NORMAL

MSFN LOS
(82+52)

CDR MAN ATT(R,P)-ACCEL CMD
MANEUVER SC TO 5 DEG PITCH
AND 0 DEG ROLL W.R.T.
LOCAL HORIZONTAL
MAN ATT(R)-RATE CMD
MAN ATT(P)-MIN IMP
INITIATE ORBITAL PITCH RATE
FOR LANDMARK SIGHTINGS

SUNUP
(83+03)

PROCEDURES FOR TRACKING THE
FIRST CONTROL POINT LANDMARK
(WITH AUTO OPTICS)
(ADDENDUM 17) (P22)

(83+20)

PROCEDURES FOR TRACKING THE
SECOND CONTROL POINT LANDMARK
(WITH AUTO OPTICS)
(ADDENDUM 17) (P22)

*** BEGIN REV 8 ***
(83+26)

LMP SELECT S BD ANY OMNI-C OR D

30

FINAL MISSION C-PRIME TRANSLUNAR LUNAR PARKING ORBIT, AND TRANSEARTH TIMELINE -12 NOV 1968

OVRU=ON
SR RANGING=OFF

MSFN AOS
(83+39)

PROCEDURES FOR TRACKING THE
THIRD CONTROL POINT LANDMARK
(WITH AUTO OPTICS)
(ADDENDUM 17) (P22)

(84+00)

PROCEDURES FOR TRACKING THE
PSEUDO LANDING SITE
(ADDENDUM 15) (P22)

(84+10)

CDR MAN ATT(R)-ACCFL CMD
ROLL SC 180 DEG TO ABOUT
300 DEG PITCH AND 180 DEG
ROLL W.R.T. LOCAL HORIZONTAL
MAN ATT(R,P)-RATE CMD

(84+15)
SUNDOWN

PROCEDURES FOR HI GAIN ACQUISITION
OF MSFN
(ADDENDUM 19)

(84+20)

PROCEDURES FOR GND (UPLINK OF ASM
STATE VECTOR
(ADDENDUM 8)

CMP KEY V*7E

LMP RECEIVE BLOCK DATA FOR
TEI FROM MSFN

(84+25)

PROCEDURES FOR IMU REALIGN
TO REFSMMAT
(ADDENDUM 12) (P52)

BACKUP ALIGN NAV STARS
NO 14 (CANOPUS)
NO 12 (RIGEL)
BACKUP ALIGN CHECK NAV STAR
NO 15 (SIRIUS)

(84+40)

PROCEDURES FOR GDC ALIGN TO IMU
(ADDENDUM 3)

PROCEDURES FOR ORDEAL VERIFICATION
(ADDENDUM 18)

LMP BEGIN PHOTOGRAPHIC PREPARATION
FOR DARKSIDE AND SOLAR CORONA
PHOTOGRAPHY

CDR MAN ATT(P)-ACCEL CMD
PITCH SC UP TO 0 DEG PITCH
AND 180 DEG ROLL W.R.T.
LOCAL HORIZONTAL
MAN ATT(P)-MIN IMP
INITIATE ORBITAL PITCH RATE

(84+45)

LMP PERFORM DARKSIDE AND SOLAR
CORONA PHOTOGRAPHY

LMP UP TLM-CMD RESET
UP TLM-NORMAL

*** BEGIN REV 9 ***
(85*25)

MSFN LOS
(84*51)

(85*00)

LMP BEGIN PHOTOGRAPHIC PREPARATION
FOR CONVERGENT STEREO
NAVIGATION PHOTOGRAPHY
CDR MAN ATT(P)-ACFL CMD
PITCH SC UP TO 290 DEG PITCH
AND 180 DEG ROLL W.R.T.
LOCAL HORIZONTAL
MAN ATT(P)-MIN IMP
INITIATE ORBITAL PITCH RATE

(85*35)

CDR MAN ATT(P)-ACCEL CMD
PITCH SC UP 40 DEG TO 250 DEG
PITCH AND 180 DEG ROLL W.R.T.
LOCAL HORIZONTAL AT THE
SUBSOLAR POINT
MAN ATT(P)-MIN IMP
REINITIATE ORBITAL PITCH RATE
(ROLL AS REQUIRED TO SHADE
WINDOWS)

SUNUP
(85*02)

LMP SELECT S BD ANT OMNI-C OR D
DVBU-ON
SB RANGING-OFF

(85*05)

LMP KEY V06N65

MSFN AOS
(85*37)

CDR LMP WHEN X-AXIS CROSSES TERMINATOR
KEY ENTER/START CAMERA FOR
CONVERGENT STEREO NAVIGATION
PHOTOGRAPHY

06 65 (GET)

LMP RECORD GET

CDR KEY N20E

06 20 (R,P,Y)

LMP RECORD PRESENT GMBL ANGLES

LMP MONITOR PHOTOGRAPHY

CDR MONITOR ORBITAL PITCH RATE

SUNDOWN

CDR KEY V06N65
CDR LMP KEY ENTER/STOP CAMERA

06 65 (GET)

LMP RECORD GET

CDR KEY N20E

06 20 (R,P,Y)

LMP RECORD PRESENT GMBL ANGLES

CDR STOP PITCH RATE

(86+14)

CDR MAN ATT(P)-ACCEL CMD
PITCH SC DOWN 60 DEG TO APOUT
310 DEG PITCH AND 180 DEG
ROLL W.R.T. LOCAL HORIZONTAL
(ROLL 180 DEG IF REQUIRE)
MAN ATT(R,P)-RATE CMD

PROCEDURES FOR HI GAIN ACQUISITION
OF MSFN
(ADDENDUM 19)

(86+20)

PROCEDURES FOR GND (PLINK OF PSM
STATE VECTOR
(ADDENDUM 8)

CMP KEY V47E

LMP RECEIVE BLOCK DATA FOR
TEI FROM MSFN

(86+25)

PROCEDURES FOR IMU REALIGN
TO REFSMMAT
(ADDENDUM 12) (P5?)

BACKUP ALIGN NAV STARS
NO 14(CANOPUS)
NO 12(RIGEL)
BACKUP ALIGN CHECK NAV STAR
NO 15(SIRIUS)

(86+40)

PROCEDURES FOR GDC ALIGN TO I.U

(ADDENDUM 3)

PROCEDURES FOR IDEAL VERIFICATION
(ADDENDUM 1)

LMP UP TLM-CMD) RESET
UP TLM-NORMAL

MSFN LOS
(86+49)

CDR MAN ATT(P)-ACCEL CMD
PITCH SC UP TO 270 DEG PITCH
AND 180 DEG ROLL W.R.T.
LOCAL HORIZONTAL
MAN ATT(P)-MIN IMP
INITIATE ORBITAL PITCH RATE
FOR LUNAR OBSERVATION

SUN/UP
(87+00)

*** REGIN REV 10 ***
(87+24)

LMP SELECT S BD ANT OMNI-A OR B
DVRU-ON
SB RANGING-OFF

MSFN AOS
(87+35)

CDR MAN ATT(P)-RATE CMD

33

FINAL MISSION C-PRIME TRANSLUNAR LUNAR PARKING ORBIT AND TRANSEARTH TIMELINE -12 NOV 1968

PROCEDURES FOR HI GAIN ACQUISITION
OF MSFN
(ADDENDUM 19)

(87+40)

PROCEDURES FOR GND UPLINK OF -SM
STATE VECTOR AND P30 TARGET LOAD
(ADDENDUM 8)

LMP RECEIVE BLOCK DATA FOR
TEI FROM MSFN

(87+55)

PROCEDURES FOR TEI BURN ATT
CHECK EXCEPT ROLL

CDR FDAI SCALE-5/1
FDAI SELECT-1/2
MAN ATT(3)-RATE CMD
LIMIT CYCLE-OFF
ATT DB-MAX
RATE-LOW
SC CONT-CMC
CMC MODE-AUTO
RMAG MODE(3)-RATE2
RMC-ARMED

LEB KEY V+9E

F 06 22 (FINAL GMBL ANGLES)
LOAD BURN ATT STAR
CHECK GMBL ANGLES
KEY V62E
PRO

F 50 18 (COMMANDED R,P,Y)
PRO

06 18 (COMMANDED R,P,Y)
CDR MONITOR ATT MANEUVER
FOR GMBL LOCK
F 50 18 (COMMANDED R,P,Y)
VERIFY R,P,Y AGREE WITH
PAD R,P,Y WITHIN 5 DEG
NULL FDAI NEEDLES WITH RMC
LEB PRO

G/N PWR,OPTICS-ON
OPT ZERO-ZERO(15SEC)
OPT TELTRUN-SLAVE TO SXT
RETICLE BRT TW-ADJUST
OPT ZERO-OFF
OPT MODE-CMC
KEY V+IN9LE
F 21 92 (BLANK,BLANK,BLANK)
LOAD PAD SHAFT ANGLE(+XXX.XX)
IN R1 AND PAD TRUNNION ANGLE
(+XX.XXX) IN R2

41 88 (SHAFT,TRUNNION,BLANK)
MONITOR OPT DRIVE TO PAD
VALUES

VERIFY PAD STAR IN SXT
OPT MODE-MAN
DRIVE TRUN LESS THAN 5 DEG
OPT ZERO-ZERO
RETICLE BRT TW-MIN BRTNESS

(8A+05)

PROCEDURES FOR GDC DRIFT CHECK
(ADDENDUM 7)

(8A+10)

CDR RECEIVE GO/NO-GO FOR TEI

<p>(88+13) SUNDOWN</p> <p>(88+15) PROCEDURES FOR IMU REALIGN TO REFSMAT FOR DRIFT CHECK (ADDENDUM 2)</p> <p>(88+30) PROCEDURES FOR GDC ALIGN TO I.U (ADDENDUM 3)</p> <p>(88+35) PROCEDURES FOR EXT DV TARGETING (ADDENDUM 9)</p> <p>(88+40) PROCEDURES FOR SM RCS PROPULSION MONITOR CHECK (ADDENDUM 10)</p>	<p>LMP INITIATE PRE-TEI SYSTEMS CHECK</p> <p>PROCEDURES FOR IMU REALIGN TO REFSMAT FOR DRIFT CHECK (P52)</p> <p>BACKUP ALIGN NAV STARS NO. 14 (CANOPUS) NO. 12 (RIGEL) BACKUP ALIGN CHECK NAV STARS NO. 15 (SIRIUS)</p> <p>PROCEDURES FOR GDC ALIGN TO I.U (ADDENDUM 3)</p> <p>PROCEDURES FOR EXT DV TARGETING (P30)</p> <p>***** PROPULSION SYSTEM SELECTION USE SPS AND TRIM DVS TO 2FPS *****</p> <p>PROCEDURES FOR SM RCS PROPULSION MONITOR CHECK (ADDENDUM 10)</p> <p>LMP UP TLM CMU-RESET THEN NOR..</p>	<p>MSFN LOS (8A+48)</p> <p>(8A+45) PROCEDURES FOR SPS THRUST SETUP (ADDENDUM 11)</p> <p>SUNIIP (8A+59)</p> <p>(89+15) TRANSEARTH INJECTION</p> <p>(89+20) PROCEDURES FOR SM RCS AND SPS PROPULSION MONITOR CHECKS (ADDENDUM 10)</p> <p>P00 CDR MAN ATT(P)-ACC CMD RMC-ARMED MANEUVER SC IN PITCH(UP) TO MONITOR MOON IN CMD WINDOW MAN ATT(3)-RATE CMD</p> <p>MSFN AOS (8C+20)</p> <p>(90+00) PROCEDURES FOR IMU REALIGN TO REFSMAT (ADDENDUM 2)</p>	<p>PROCEDURES FOR HIGH GAIN ANT ACQUISITION (ADDENDUM 19)</p> <p>PROCEDURES FOR IMU REALIGN TO REFSMAT (ADDENDUM 2) (P52)</p>
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(90+15) BACKUP ALIGN NAV STARS
 NO. 14(CANOPIUS)
 NO. 12(RIGEL)
 BACKUP ALIGN CHECK NAV STAR
 NO. 15(SIRIUS)

(90+10) PROCEDURES FOR TERMINATING PTC
 (ADDENDUM 14)

(90+15) PROCEDURES FOR GDC ALIGN TO IMU
 (ADDENDUM 3)

(90+30) PROCEDURES FOR STAR/LUNAR
 HORIZON SIGHTINGS - 8 SETS
 OF 3 MARKS EACH
 (ADDENDUM 4)

(91+50) RECOMMENDED NAV STARS
 NO. 1(ALPHERATZ) (NEAR HOR)
 NO. 1(ALPHERATZ) (NEAR HOR)
 NO. 10(MIRFAK) (FAR HOR)
 NO. 10(MIRFAK) (FAR HOR)
 NO. 11(ALDEBARAN) (FAR HOR)
 NO. 11(ALDEBARAN) (FAR HOR)
 NO. 2(DIPHDA) (NEAR HOR)
 NO. 2(DIPHDA) (NEAR HOR)

(91+55) PROCEDURES FOR ESTABLISHING PTC
 (ADDENDUM 13)

(100+05) PROCEDURES FOR RETURN TO
 EARTH TARGETING
 (ADDENDUM 6)

(100+25) PROCEDURES FOR GDC ALIGN TO IMU
 (ADDENDUM 3)

(100+30) PROCEDURES FOR STAR/LUNAR
 HORIZON SIGHTINGS - 3 SETS
 OF 3 MARKS EACH
 (ADDENDUM 4)

(101+20) RECOMMENDED NAV STARS
 NO. 1(ALPHERATZ) (NEAR HOR)
 NO. 11(ALDEBARAN) (FAR HOR)
 NO. 2(DIPHDA) (NEAR HOR)

(101+20) PROCEDURES FOR STAR/EARTH
 HORIZON SIGHTINGS - 6 SETS
 OF 3 MARKS EACH
 (ADDENDUM 4)

(100+05) RECOMMENDED NAV STARS

(P52) BACKUP ALIGN NAV STARS
 NO. 14(CANOPIUS)
 NO. 12(RIGEL)
 BACKUP ALIGN CHECK NAV STAR
 NO. 15(SIRIUS)

(P23) PROCEDURES FOR GDC ALIGN TO IMU
 (ADDENDUM 3)

(P23) RECOMMENDED NAV STARS
 NO. 1(ALPHERATZ) (NEAR HOR)
 NO. 11(ALDEBARAN) (FAR HOR)
 NO. 2(DIPHDA) (NEAR HOR)

(P37) PROCEDURES FOR ESTABLISHING PTC
 (ADDENDUM 13)

(P37) PROCEDURES FOR RETURN TO
 EARTH TARGETING
 (ADDENDUM 6)

NO. 22 (REGULUS) (FAR HOR)
 NO. 22 (REGULUS) (FAR HOR)
 NO. 26 (SPICA) (NEAR HOR)
 NO. 26 (SPICA) (NEAR HOR)
 NO. 31 (ARCTURIUS) (NEAR HOR)
 NO. 31 (ARCTURIUS) (NEAR HOR)

(102+20) PROCEDURES FOR RETURN TO
 EARTH TARGETING
 (ADDENDUM 6) (P37)

(102+50) PROCEDURES FOR GND UPLINK OF CSM
 STATE VECTOR, P30 TARGET LOAD, AND
 PREFERRED ENTRY REFSMMAT
 (ADDENDUM 9)

P00 FOLLOWING GND UPLINK
 LEB WHEN COMP LT=0JIT (NO INTEG)
 KEY V96E
 KEY V83E
 F 16 54 (R, RDOT, THETA)
 COPY R AND RDOT
 PRO
 KEY V89E
 F 04 06 (TRACK AXIS OPTION)
 LOAD 00002 IN R2
 PRO
 F 06 18 (COMPUTED R,P,Y)
 COPY R,P,Y
 KEY V34E
 P00 KEY V37E00E

KEY V47E (TRANSFER CSM SV
 FROM LM SLOTS TO CSM SLOTS)

(103+10) PROCEDURES FOR IMU ALIGN TO
 UPLINKED PREFERRED REFSMMAT (P52)
 (ADDENDUM 2)

BACKUP ALIGN NAV STARS
 NO. 14 (CANOPUS)
 NO. 12 (RIGEL)
 BACKUP ALIGN CHECK NAV STAR
 NO. 15 (SIRIUS)

(103+25) PROCEDURES FOR GDC ALIGN TO IMU

(103+30) PROCEDURES FOR EXT DV TARGETING
 (ADDENDUM 9) (P30)

 PROPLUSION SYSTEM SELECTION
 USE RCS FOR DV LESS THAN OR
 EQUAL TO 12FPS
 USE SPS FOR DV GREATER
 THAN 12FPS
 TRIM DVS TO ZERO

(103+35) PROCEDURES FOR SM RCS PROPLUSION
 MONITOR CHECK
 (ADDENDUM 10)

(103+40) PROCEDURES FOR RCS THRUST SETUP (P41)
(ADDENDUM 12)

(104+00) FIRST TRANSEARTH MIDCOURSE

(104+05) PROCEDURES FOR SM RCS AND SPS
PROPULSION MONITOR CHECKS
(ADDENDUM 10)

(105+15) PROCEDURES FOR STAR/EARTH
HORIZON SIGHTINGS - 5 SETS
OF 3 MARKS EACH
(ADDENDUM 4)
RECOMMENDED NAV STARS (P23)
NO. 22 (REGULUS) (FAR HOR)
NO. 22 (REGULUS) (FAR HOR)
NO. 26 (SPICA) (NEAR HOR)
NO. 31 (ARCTURIUS) (NEAR HOR)
NO. 31 (ARCTURIUS) (NEAR HOR)

(106+15) PROCEDURES FOR STAR/LUNAR
HORIZON SIGHTINGS - 3 SETS
OF 3 MARKS EACH
(ADDENDUM 4)
RECOMMENDED NAV STARS (P23)
NO. 1 (ALPHERATZ) (NEAR HOR)
NO. 11 (ALDERARAN) (FAR HOR)
NO. 2 (DIPHA) (NEAR HOR)

(106+45) PROCEDURES FOR RETURN TO
EARTH TARGETING
(ADDENDUM 6) (P37)

(107+40) PROCEDURES FOR IMU REALIGN
TO REFSMMAT
(ADDENDUM 2) (P52)
BACKUP ALIGN NAV STARS
NO. 14 (CANOPUS)
NO. 12 (RIGEL)
BACKUP ALIGN CHECK NAV STAR
NO. 15 (SIRIUS)

(107+55) PROCEDURES FOR GDC ALIGN TO IMU
(ADDENDUM 3)

(108+00) PROCEDURES FOR STAR/EARTH
HORIZON SIGHTINGS - 5 SETS
OF 3 MARKS EACH
(ADDENDUM 4) (P23)
RECOMMENDED NAV STARS
NO. 22 (REGULUS) (FAR HOR)
NO. 22 (REGULUS) (FAR HOR)
NO. 26 (SPICA) (NEAR HOR)
NO. 31 (ARCTURIUS) (NEAR HOR)
NO. 31 (ARCTURIUS) (NEAR HOR)

(108+50) PROCEDURES FOR ESTABLISHING PTC
(ADDENDUM 13)

(108+55)	PROCEDURES FOR RETURN TO EARTH TARGETING (ADDENDUM 6)	(P37)	(120+30)	PROCEDURES FOR RETURN TO EARTH TARGETING (ADDENDUM 6)	(P37)
(119+35)	PROCEDURES FOR TERMINATING PTC (ADDENDUM 14)		(121+00)	PROCEDURES FOR GND UPLINK OF CSM STATE VECTOR AND P30 TARGET LOAD (ADDENDUM 8)	
(119+40)	PROCEDURES FOR IMU REALIGN TO REFSMMAT (ADDENDUM 2)	(P52)	P00	FOLLOWING GND UPLINK LEB WHEN COMP LT-OUT(NO INTEG) KEY V96E KEY V83E F 16 54 (R,RDOT,THETA) COPY R AND RDOT PRO	
	BACKUP ALIGN NAV STARS NO. 14 (CANOPUS) NO. 12 (RIGEL) BACKUP ALIGN CHECK NAV STAR NO. 15 (SIRIUS)		F 04 06	(TRACK AXIS OPTION) LOAD 00002 IN R2 PRO	
(119+55)	PROCEDURES FOR GDC ALIGN TO I.U (ADDENDUM 3)		F 06 18	(COMPUTED R,P,Y) COPY R,P,Y KEY V34E	
(120+00)	PROCEDURES FOR STAR/EARTH HORIZON SIGHTINGS - 3 SETS OF 3 MARKS EACH (ADDENDUM 4)	(P21)	P00	KEY V37E00E KEY V47E (TRANSFER CSM SV FROM LM SLOTS TO CSM SLOTS)	
	RECOMMENDED NAV STARS NO. 22 (REGULUS) (FAR HOR) NO. 26 (SPICA) (NEAR HOR) NO. 31 (ARCTURIUS) (NEAR HOR)		(121+30)	PROCEDURES FOR EXT DV TARGETING (ADDENDUM 9)	(P30)
				***** PROPULSION SYSTEM SELECTION	*****

USE RCS FOR DV LESS THAN 0.4
EQUAL TO 12FPS
USE SPS FOR DV GREATER
THAN 12FPS

TRIM OVS TO ZERO

(121+35) PROCEDURES FOR SM RCS PROPULSION
MONITOR CHECKS
(ADDENDUM 10)

(121+40) PROCEDURES FOR RCS THRUST SETUP
(ADDENDUM 12) (P41)

(122+00) SECOND TRANSEARTH MIDCOURSE

(122+05) PROCEDURES FOR SM RCS AND SPS
PROPULSION MONITOR CHECKS
(ADDENDUM 10)

(122+35) PROCEDURES FOR STAR/LUNAR
HORIZON SIGHTINGS - 3 SETS
OF 3 MARKS EACH
(ADDENDUM 4) (P23)

RECOMMENDED NAV STARS
NO. 2 (DIPHDA) (NEAR HOR)
NO. 2 (DIPHDA) (NEAR HOR)
NO. 1 (ALPHERATZ) (NEAR HOR)

(123+35) PROCEDURES FOR STAR/EARTH
HORIZON SIGHTINGS - 5 SETS
OF 3 MARKS EACH
(ADDENDUM 4) (P23)
RECOMMENDED NAV STARS
NO. 22 (REGULUS) (FAR HOR)
NO. 22 (REGULUS) (FAR HOR)
NO. 26 (SPICA) (NEAR HOR)
NO. 31 (ARCTURUS) (NEAR HOR)
NO. 31 (ARCTURUS) (NEAR HOR)

(124+25) PROCEDURES FOR ESTABLISHING PTC
(ADDENDUM 13)

(124+30) PROCEDURES FOR RETURN TO
EARTH TARGETING
(ADDENDUM 6) (P37)

(129+35) PROCEDURES FOR TERMINATING PTC
ADDENDUM 14)

(129+40) PROCEDURES FOR IMU REALIGN
TO REFSMMAT
(ADDENDUM 2) (P52)

BACKUP ALIGN NAV STARS
NO. 14 (CANOPUS)
NO. 12 (RIGEL)
BACKUP ALIGN CHECK NAV STARS
NO. 15 (SIRIUS)

(129+55) PROCEDURES FOR GNC ALIGN TO LUNAR (P21)
 (ADDENDUM 3)

(EI-17HRS)

(130+00) PROCEDURES FOR STAR/LUNAR HORIZON SIGHTINGS - 2 SETS OF 3 MARKS EACH (ADDENDUM 4)

RECOMMENDED NAV STARS
 NO. 2 (DIPHA) (NEAR HOR)
 NO. 2 (DIPHA) (NEAR HOR)

(FI-16HRS)

(130+50) PROCEDURES FOR STAR/EARTH HORIZON SIGHTINGS - 6 SETS OF 3 MARKS EACH (ADDENDUM 4)

RECOMMENDED NAV STARS
 NO. 22 (REGULUS)
 NO. 22 (REGULUS)
 NO. 26 (SPICA)
 NO. 24 (SPICA)
 NO. 31 (ARCTURUS)
 NO. 31 (ARCTURUS)

(131+50) PROCEDURES FOR ESTABLISHING PTC (ADDENDUM 13)

(131+55) PROCEDURES FOR RETURN TO EARTH TARGETING (ADDENDUM 5) (P37)

(EI-15HRS)

(132+05) PROCEDURES FOR GND UPLINK OF CSM STATE VECTOR (UPLINK INTO LM SLOTS ONLY) (ADDENDUM 8) (P27)

FOLLOWING GND UPLINK

P00 LEB WHEN COMP I.T-OUT(NN INTEG)
 KEY V96E
 KEY V83E
 F 16 54 (R,RDOT,THETA)
 COPY R AND RDOT
 PRO
 KEY V89E
 F 04 06 (TRACK AXIS OPTION)
 LOAD 00002 IN R2
 PRO
 F 06 1A (COMPUTED R,P,Y)
 COPY R,P,Y
 KEY V34E

P00 KEY V37E00E

(140+25) PROCEDURES FOR TERMINATING PTC (ADDENDUM 14)

(140+30)

PROCEDURES FOR IMU ALIGN TO REFSMMAT (ADDENDUM 2) (P52)
 BACKUP ALIGN NAV STARS NO. 14 (CANOPUS)
 NO. 12 (RIGEL)
 BACKUP ALIGN CHECK NAV STARS NO. 15 (SIRIUS) (140+45)
 PROCEDURES FOR GDC ALIGN TO IMU (ADDENDUM 3) (140+50)
 PROCEDURES FOR ESTABLISHING PTC (**ONLY IF COMM AVAILABLE**) (ADDENDUM 13)

 THE FOLLOWING SIGHTING PERIODS REQUIRED ONLY FOR A COMM FAILURE (EI-6HRS)
 PROCEDURES FOR STAR/EARTH HORIZON SIGHTINGS - 3 SETS OF 3 MARKS EACH (COMM FAILURE ONLY) (ADDENDUM 4) (140+50) (P23)
 RECOMMENDED NAV STARS NO. 22 (REGULUS) (FAR HOR)
 NO. 26 (SPICA) (NEAR HOR)
 NO. 31 (ARCTURUS) (NEAR HOR) (141+20)
 PROCEDURES FOR RETURN TO EARTH TARGETING (ADDENDUM 6) (P37)
 PROCEDURES FOR IMU REALIGN TO REFSMMAT (ADDENDUM 2) (P52)
 BACKUP ALIGN NAV STARS NO. 14 (CANOPUS)
 NO. 12 (RIGEL)
 BACKUP ALIGN CHECK NAV STARS NO. 15 (SIRIUS) (142+00)
 PROCEDURES FOR GDC ALIGN TO IMU (ADDENDUM 3) (142+15)
 PROCEDURES FOR STAR/EARTH HORIZON SIGHTINGS - 2 SETS OF 3 MARKS EACH (COMM FAILURE ONLY) (ADDENDUM 4) (P23)
 RECOMMENDED NAV STARS NO. 22 (REGULUS) (FAR HOR)
 NO. 26 (SPICA) (NEAR HOR) (142+40)

PROCEDURES FOR RETURN TO
EARTH TARGETING
(ADDENDUM 6)

(P37)

(143+25)

PROCEDURES FOR TERMINATING PTC
(*ONLY IF COMM AVAILABLE*)
(ADDENDUM 14)

(143+30)

PROCEDURES FOR IMU REALIGN
TO REFSMMAT
(ADDENDUM 2)

(P52)

BACKUP ALIGN NAV STARS
NO. 14(CANOPUS)
NO. 12(RIGEL)
BACKUP ALIGN CHECK NAV STAR
NO. 15(SIRIUS)

(143+45)

PROCEDURES FOR GDC ALIGN TO IMU
(ADDENDUM 3)

(143+50)

PROCEDURES FOR GND UPLINK OF SSM
STATE VECTOR AND P30 TARGET LOAD
(ADDENDUM 8)

(144+20)

PROCEDURES FOR EXT DV TARGETING
(ADDENDUM 9)

(P30)

PROPULSION SYSTEM SELECTION
USE RCS FOR DV LESS THAN OR
EQUAL TO 12FPS
USE SPS FOR DV GREATER
THAN 12FPS

TRIM DV TO ZERO

(144+25)

PROCEDURES FOR SM RCS PROPULSION
MONITOR CHECKS
(ADDENDUM 10)

(144+30)

PROCEDURES FOR RCS THRUST SETUP
(ADDENDUM 12)

(P41)

(144+50)

THIRD TRANSEARTH MIDCOURSE

(144+55)

PROCEDURES FOR SM RCS AND SPS
PROPULSION MONITOR CHECKS
(ADDENDUM 10)

THE FOLLOWING SIGHTING PERIOD
REQUIRED ONLY FOR A COMM FAILURE

(144+05)

PROCEDURES FOR STAR/EARTH
HORIZON SIGHTINGS - 1 SET
OF 3 MARKS
(COMM FAILURE ONLY)

(P23)

RECOMMENDED NAV STAR
NO. 33 (ANTARES) (NEAR HOR)

(146.35)

CM/SM SEPARATION

(146.50)

ENTRY INTERFACE

Section 3.2.1

SUMMARY OF ADDENDUMS

ADDENDUM 1
ASSUMED PRE-SEP SC CONFIGURATION

ADDENDUM 2
IMU ALIGN (OR REALIGN) (P52)

ADDENDUM 3
GDC ALIGN TO IMU

ADDENDUM 4
STAR/EARTH (OR LUNAR) HORIZON
(OR LANDMARK) SIGHTINGS (P23)

ADDENDUM 5
GROUND TRACK DETERMINATION (P21)

ADDENDUM 6
RETURN TO EARTH TARGETING (P37)

ADDENDUM 7
GDC DRIFT CHECK

ADDENDUM 8
GND UPLINK (P27)

ADDENDUM 9
EXTERNAL OV TARGETING (P30)

ADDENDUM 10
SM RCS AND SPS PROPUSSION CHECKS

ADDENDUM 11
SPS THRUST SETUP (P40)

ADDENDUM 12
RCS THRUST SETUP (P41)

ADDENDUM 13
ESTABLISH PTC

ADDENDUM 14
TERMINATE PTC

ADDENDUM 15
ORBITAL NAVIGATION PROCEDURES (P22)
(PSEUDO LANDING SITE TRACKING)

ADDENDUM 16
ORBITAL NAVIGATION PROCEDURES (P22)
(LANDMARK TRACKING-NO AUTO OPTICS)

ADDENDUM 17
ORBITAL NAVIGATION PROCEDURES (P22)
(LANDMARK TRACKING-WITH AUTO OPT)

ADDENDUM 18
ORDEAL INITIALIZATION/VERIFICATION

ADDENDUM 19
HIGH GAIN ANT ACQUISITION

Section 3.2.2

 ADDENDUM 1
 ASSUMED PRE-SEP SC CONFIGURATION

D	P	C	P
I	L	R	A
S	A	F	N
P	W		E
			L

PROCEDURES

CNR**ASSUMED SWITCH POSITIONS**

CMC ATT-IMU	1
FDAI SCALE-5/1	1
FDAI SELECT-1/2	1
FDAI SOURCE-CMC	1
ATT SET-GDC	1
MAN ATT(3)-RATE CMD	1
LIMIT CYCLE-OFF	1
ATT DB-MAX	1
RATE-LOW	1
THC LOCKED	
RHC LOCKED	
TRANS CONT PWR-OFF	1
RHC PWR NORM(BOTH)-AC/DC	1
RHC PWR DIRECT(ROTH)-OFF	1
SC CONT-CMC	1
CMC MODE-FREE	1
RMAG MODE(3)-RATE 2	1
SPS THRUST DIRECT-NORMAL	1
DV THRUST(A,R)-OFF	1
SCS TVC(2)-AUTO	1

SPS GMBL MTR(4)-OFF	1
DV CR-CSM	1
ELS LOGIC-OFF	1
ELS AUTO-MAN	1
CM RCS LOGIC-OFF	1
CM PRPLNT DUMP-OFF	1
CM PRPLNT PURG-OFF(DOWN)	1
IMU CAGE-OFF	1
FMS ROLL-OFF	1
.05 G-OFF	1
LV/SPS IND(2)-PC,GPI	1
TVC GMBL DRIVE(BOTH)-AUTO	1
FCSM(A,R)-RESET/OVERRIDE	1
FMS FUNCTION-OFF	1
EMS MODE-STBY	1
COAS PWR-ON	15
UTIL PWR-ON	15
PL BCN LT-OFF(CENTER)	15
PL DYE MARKER-OFF(DOWN)	15
PL VENT-OFF	15
U/P TLM(CM,IU)-BLOCK	2
RCS TRNFR-SM	2
PANEL 8 CB CLOSED EXCEPT	8
ELS BAT A,BAT B(2)-OPEN	8
SECS ARM(BOTH)-OPEN	8
SECS LOGIC(BOTH)-OPEN	8
PL VENT FLT/PL-OPEN	8
FLOAT BAG(3)-OPEN	8
RCS LOGIC(2)-OPEN	8
AUTO RCS SEL(16)-OFF	8
FLOAT BAG(ALL)-VENT	8
SECS LOGIC(BOTH)-OFF	8
SECS PYRO ARM(2)-SAFE	8

122
122
122
122
122

OPT SPEED=HI
OPT COUPLING=DIRECT
OPT TELTRUN=SLAVE TO SXT
COND LAMPS=ON
UP TLM(LEB)=ACCEPT

ASSUMED NAP CONFIGURATION

KEY V48E

F 04 46 (DAP CONFIGURATION)

LOAD
11103
11111
B

PRO

F 06 47 (CSM AND LM WT)

LOAD PAD DATA
+XXXXX.
+00000.
B

PRC

F 06 48 (SPS GMBL TRIM)

LOAD PAD DATA
+XXX.XX
+XXX.XX
B

PRO

KEY V46E

ASSUMED PROGRAM CONFIGURATION

KEY V37E47E

6

47

FDS PWR-OFF 7
TVC SERVO PWR(?)=OFF 7
FDAI/GPI PWR=BOTH 7
LOGIC PWR 2/3=ON(UP) 7
SCS ELEC PWR=GDC/ECA 7
SIG CONDR/DR BTAS(2)=AC1 7
RMAG PWR(BOTH)=ON 7

LMP**ASSUMED SWITCH POSITIONS**

G/N PWR=AC1 5
MN BUS TIE(2)=OFF 5
RATT CHGR=AC1 5
NONESS BUS=MNB 5
PANEL 5 G/N CB(10)=CLOSED 5

PANEL 229 CB CLOSED EXCEPT 229
MN REL PYRO(2)=OPEN 229
S-RAND ANT OMNI= 3
SELECT REQD OMNI ANT 3
S-RAND NORM MODE VOICF=VOICE 3
S-RAND NORM MODE PCM=PCM 3
S-BAND NORM MODE RNG=RANGING 3
S-BAND AUX TAPE=OFF 3
S-BAND AUX TV=OFF 3
UP TLM DATA=DATA 3
HIP TLM CMD=NORM 3

LEB**ASSUMED SWITCH POSITIONS**

G/N PWR=OPTICS=ON 100
G/N PWR IMU=ON 100
OPT MODE=MAN 122
OPT ZERO=ZERO 122

5

FINAL MISSION C-PRIME PROCEDURES DATED=12 NOV 196A

F 16 83 (DVX,DVY,UVZ)
 V32 KEYED FOLLOWING TLI
 CUTOFF

 ADDENDUM 2
 IMU ALIGN (OR REALIGN) (P52)

DISPLAYS
 PROGRAMS
 C R E W PROCEDURES
 PANEL

***PROCEDURES FOR IMU REALIGN**

P00 LMP G/N PWR=ACI
 LEB G/N PWR OPTICS=ON
 OPT ZERO=ZERO (15 SEC)
 OPT ZERO=OFF
 OPT MODE=MAN
 COND LAMPS=ON
 RETICLE BRT TW=ADJUST
 CDR ATT DR=MIN
 RATE=LOW
 LIMIT CYCLE=OFF
 HMAG MODE(3)=ATT1/RATE2
 MAN ATT(3)=RATE CMD
 SC CONT=SCS

P52 LEB KEY V37E52E
 F 04 06 (ALIGN OPTION CODE)
 LOAD 00001 IN R2 FOR ALIGN TO
 UPLINKED PREFERRED REFSMMAT
 AND GO TO F 06 22 DISPLAY
 OR LOAD 00003 FOR REALIGN TO

REFSMAT AND GO TO
 F 50 25 DISPLAY
 PRO
 F 06 22 (R.P.Y AT PROPOSED ALIGNMENT)
 VERIFY NO GMAL LOCK AT
 PROPOSED ALIGNMENT
 PRO/MONITOR COARSE ALIGN
 F 50 25 (00015, PERFORM STAR ACQ)
 RMC-ARMED
 PRO
 IF TWO STARS NOT AVAILABLE, A
 F 05 09 (CODE 405) ALARM
 WILL OCCUR. COR MANUALLY
 MANEUVER SC TO ACQUIRE BACKUP
 ALIGN NAV STARS IN SCT FAV.
 CMP KEY V32E, RETURN TO
 F 50 25 (00015) DISPLAY AND PRO
 IF TWO STARS AVAILABLE AT
 THIS ALTITUDE
 F 01 70 (STAR CODE)
 CHECK FIRST STAR CODE
 OR LOAD THIRD STAR CODE
 IF USING ALIGN CHECK
 OPT MODE-CMC
 PRO
 06 92 (SHAFT, TRUN, BLANK)
 MONITOR OPTICS DRIVE TO
 STAR ONE
 (IF VERIFYING SXT DRIVE TO
 THIRD STAR, KEY V37E ONE)
 IDENTIFY STAR ONE
 OPT MODE-MAN
 F 51 88 (PLEASE MARK)
 MARK ON STAR UNF WITH SXT
 F 50 25 (00016, TERMINATE MARK SEQ)
 PRO

F 01 71 (MARKED STAR CODE)
 PRO
 F 01 70 (STAR CODE)
 CHECK SECOND STAR CODE
 OPT MODE-CMC
 PRO
 06 92 (SHAFT, TRUN, BLANK)
 MONITOR OPTICS DRIVE TO
 STAR TWO
 IDENTIFY STAR TWO
 OPT MODE-MAN
 F 51 88 (PLEASE MARK)
 MARK ON STAR TWO WITH SXT
 F 50 25 (00016, TERMINATE MARK SEQ)
 PRO
 F 01 71 (MARKED STAR CODE)
 PRO
 F 06 05 (ANGLE DIFF)
 IF GREATER THAN .05 DEG, KEY
 V32F, GO TO F 50 25 (00014)
 AND REPEAT ALIGNMENT
 IF .05 DEG OR LESS, PRO
 F 06 93 (GYRO TORQ ANGLES)
 LMF RECORD GYRO TORQ ANGLES
 VOICF TO GND
 LEB PRO IF GYRO TORQ ANGLES OK
 F 50 25 (00014, PERFORM FINE ALIGN)
 IF REPEAT ALIGNMENT REQUIRED,
 PRO, RETURN TO F 50 25 (00015),
 VERIFY STAR ID AND REPEAT
 MARK PROCESS ON TWO STARS
 IF ONLY ALIGN CHECK REQUIRED,
 PRO, RETURN TO F 50 25 (00015),
 KEY ENTER, GO TO F 01 70,
 AND VERIFY SXT DRIVES TO
 THIRD STAR

 ADDENDUM 3
 GDC ALIGN TO IMU

(CALLED AFTER THIRD STAR SHW)
 OPT MODE=MAN
 DRIVE TRUN TO LESS THAN 5 DEG
 OPT ZERO=ZERO
 RETICLE BRT TW-MIN BRTNES

DISPLAY PROCEDURES
 PROGRAM
 C R E W
 P A N E L

 PROCEDURES FOR GDC ALIGN TO IMU

P00 COR ATT DB-MIN
 RATE-LOW
 BMAG MODE(3)-ATT1/RATE2
 MAN ATT(3)-RATE CMD
 SC CONT-SCS
 FDAI SELECT=1
 FDAI SOURCE=ATT SET
 ATT SET=IMU
 KEY V16N20E
 16 20 (CURRENT R,P,Y)
 NULL ATT ERROR NEEDLES
 ON FDAI 1 WITH
 ATT SET THUMBWHEELS
 (AVOID FALSE NULL RY
 COMPARING ATT SET IND
 WITH N20)
 ATT SET=GDC
 DEPRESS GDC ALIGN BUTTON
 FDAI SELECT 1/2
 ATT DR-MAX

 ADDENDUM 4
 STAR/EARTH(OR LUNAR) HOR (P23)
 (OR LMK SIGHTINGS)

D I S P
 P L A Y S
 P R A C
 E U G R
 T E W
 P R O C E D U R E S

P
A
N
E
L

PROCEDURES FOR CISELUNAR NAV

P00 LMP G/N PWR=AC1
 TLM INPUTS PCM=LOW
 UP TLM CMD=RESET THEN NORM
 TAPE RCUR PCM=PCM/ANLG
 TAPE RCDR RCD=RCD
 TAPE RCDR FWD=FWD
 TAPE MOTION TB=BP
 LEB G/N PWR,OPTICS=ON(OPTICS PWR
 MUST BE ON AT LEAST
 30 MIN PRIOR TO TAKING MARKS)
 OPT ZERO=ZERO(15SEC)
 OPT TELTRUN=SLAVE TO SXT
 COND LAMPS=ON
 OPT MODE=MAN
 RETCL BRT TW=ADJUST
 CDR SC CONT=CMC
 CMC MODE=AUTO
 BMAG MODE(3)=RATE2
 MAN ATT(3)=RATE CMD

P23 F 05 70 KEY V37E23L
 (MEASUREMENT IDENTIFICATION)
 LOAD
 000XX STAR ID
 XXXXX LMK ID
 00XX0 HOR ID
 FOR HOR MARKS.R2=00000
 AND LOAD HOR DATA IN R3
 FOR LMK MARKS.R3=00000
 AND LOAD LMK DATA IN R2
 (ASSUMED ALL KNOWN LMK'S)
 PRO
 F 50 25 (00202,PERFORM AUTO MANEUVER)
 KEY ENTER
 F 59 88 (PERFORM CALIBRATION MARK)
 CALIBRATION OF OPTICS TRUNNING
 REQUIRED AT THE BEGINNING
 OF EACH SIGHTING PERIOD.
 IF OPTICS CALIBRATION ALREADY
 PERFORMED,
 OPT MODE=CMC
 OPT ZERO=OFF
 KEY ENTER AND GO TO
 06 92 DISPLAY
 IF OPTICS CALIBRATION REQUIRED
 OPT MODE=MAN
 RMC 2=ARMED.
 MANEUVER SC TO ACQUIRE A STAP
 WITH THE S.T LMK LOS
 CDR CMC MODE=FREE
 LER FIX STAR IN SXT LMK LOS FIELD
 OF VIEW WITH MIN IMP CONTPGI
 OPT ZERO=OFF
 OPT COUPLING=RSLV
 OPT SPEED=LOW

15

F 06 87 SUPERIMPOSE SXT STAR LOS
 ON SXT LMK LOS WITH THE
 OPTICS HAND CONTROLLER
 MARK WHEN THE TWO LINES OF
 SIGHT ARE SUPERIMPOSED
 (BLANK,TRUN ANGLE BIAS,BLANK)
 COPY TRUN ANGLE BIAS
 KEY V32E

F 59 88 (PERFORM CALIBRATION MARK)
 SUPERIMPOSE SXT STAR LOS
 ON THE SXT LMK LOS
 MARK WHEN THE TWO LINES OF
 SIGHT ARE SUPERIMPOSED
 (BLANK,TRUN ANGLE BIAS,BLANK)
 COPY TRUN ANGLE BIAS
 IF THE TWO TRUN ANGLE BIAS
 MEASUREMENTS ARE NOT WITHIN
 .003 DEG,
 KEY V32E AND REPEAT THE
 CALIBRATION PROCESS
 IF ANY TWO TRUN ANGLE BIAS
 MEASUREMENTS ARE
 WITHIN .003 DEG,
 PRO(INCORPORATE TRUN BIAS)
 PLEASE MARK)

F 51 88 KEY V94E (PERFORM LMK(HOR)
 ACQUISITION)
 CDR SC CONT=CMC
 CMC MODE=AUTO
 LER PRO
 06 18 (COMMANDED R,P,Y ANGLES)
 CDR MONITOR AUTOMATIC MANEUVER
 FOR GIMBAL LOCK
 LER MONITOR AUTO MANEUVER,VERIFY
 LMK/HOR IN SXT FIELD OF VIEW

16

F 50 18 (COMMANDED R.P.Y ANGLES)
 OPT MODE-CMC
 KEY ENTER

06 92 (COMMANDED OPTICS ANGLES)
 MONITOR OPTICS DRIVE
 TO SELECTED STAR
 OPT MODE-MAN

F 51 88 (PLEASE MARK ON STAR-LMK/HOR)
 CDR CMC MODE-FREE
 LFR MANEUVER SC WITH MIN TMP
 CONTROLLER TO POSITION
 LMK(HOR) IN SXT AT
 SIX STELLAR POINT
 FIX LMK(HOR) IN SXT
 OPT COUPLING-RSLV
 OPT SPEED-LO
 MANEUVER OPTICS TO SUPERIMPOSE
 STAR ON LMK/HOR
 MARK ON STAR-LMK/HOR

F 50 25 (00016, TERMINATE MARK SEQ)
 PRO

F 05 71 (MEASUREMENT IDENTIFICATION)
 VERIFY MARKED DATA
 PRO

F 06 49 (DR,DV,BLANK)
 **HOLD AT THIS DISPLAY FOR
 20 SECONDS**
 KEY VOINOLE 362E
 CCOPY TRUN (OCTAL) IN R1
 **HOLD AT THIS DISPLAY FOR
 20 SECONDS**
 KEY RELEASE
 PRO IF DR AND DV ACCEPTABLE
 OR IF NOT ACCEPTABLE,
 KEY V32E, RETURN TO F 05 70

F 37 88

DISPLAY, AND REPEAT MARK
 TAKE 2 ADDITIONAL MARKS ON
 EACH STAR/LMK(HOR) SET
 KEY 23E, RETURN TO
 F 05 70 DISPLAY
 AND REPEAT MARK PROCESS
 WHEN SELECTING A NEW
 STAR/LMK(HOR) SET TO MARK ON,
 KFY 23E, RETURN TO THE
 F 05 70 DISPLAY PERFORM THE
 REQUIRED ATT MANEUVER, AND
 TAKE THREE MARKS ON THE SET
 WHEN TERMINATING THE NAV
 SIGHTING PERIOD
 KEY 00E
 OPT MODE-MAN
 DRIVE TRUN LESS THAN 5 DEG
 OPT ZERO-ZERO
 RETICLE BRT 7M-MIN ARTNESS
 CMC MODE-AUTO
 LMP IF TAPE RCDR OPERATING
 TAPE RCDR FWD-OFF(CTR)
 TAPE MOTION TR-GRAY
 (ASSUME GND PLAYBACK AT
 NEXT OPPORTUNITY)

P00

ADDENDUM 5
GROUND TRACK DETERMINATION (P21)

D I S P O S I T I O N
P L A C E M E N T
P R O C E D U R E S
P00
P21

```
CMP KEY V37E21E  
F 04 06 (00002,VEHICLE OPTION)  
LOAD 00001 IN R2  
PRO  
F 05 34 (T LAT AND LONG)  
LOAD NOMINAL TIME OF  
LOI1 RECEIVED FROM GND  
PRO  
F 06 43 (LAT,LONG,ALT)  
COPY LAT,LONG,AND ALT  
IF A 10MIN INCREMENT FROM THE  
LOADED TIME IS DESIRED,  
KEY V32E AND RETURN TO  
F 06 34 DISPLAY(TIME IS  
AUTOMATICALLY INCREMENTED)  
IF PROGRAM TERMINATION  
IS DESIRED,  
PRO  
F 37 RB KEY 00E  
P00
```

P00
P37

ADDENDUM 6
RETURN TO EARTH TARGETING (P37)

**PROCEDURES FOR RETURN TO
EARTH TARGETING**
CMP KEY V37E37E
F 06 33 (TIG OF MIDCOURSE)
LOAD TIG (GET) OF MIDCOURSE
PRO
F 06 60 (BLANK,VPRED,GAMMA EI)
LOAD ZERO IN R2(VPRED) AND
LOAD ZERO IN R3(GAMMA EI)
PRO
KEY V16N3BE,MONITOR ITERATION
KEY RELEASE WHEN KEY REL LT-ON
IF F 06 09 ALARM 612,605,OR
613 OCCUR,KEY V32E,GO TO
F 06 33 DISPLAY,AND ADJUST
INPUT PARAMETERS
F 06 61 (IMPACT LAT,IMPACT LONG,BLANK)
COPY LAT AND LONG AND IF NOT
ACCEPTABLE,KEY V32E,GO TO
F 06 33 DISPLAY,AND ADJUST
INPUT PARAMETERS
PRO IF ACCEPTABLE
F 06 39 (DELTA T TRANSFER)
COPY DELTA T TRANSFER (GET)
AND IF NOT ACCEPTABLE,KEY
V32E,GO TO F 06 33 DISPLAY,
AND ADJUST INPUT PARAMETERS
PRO IF ACCEPTABLE
F 06 60 (BLANK,VPRED,GAMMA EI)

COPY VPRED AND GAMMA FI AND
 IF NOT ACCEPTABLE, KEY V32F,
 GO TO F 06 33 DISPLAY, AND
 ADJUST INPUT PARAMETERS
 PRO IF ACCEPTABLE
 F 06 81 (DELTA VX, VY, VZ=LOCAL VERT)
 COPY DVS
 PRO
 IF F 05 09 ALARM 605 OR 613
 OCCUR, KEY V32F, GO TO F 04 37
 DISPLAY, AND ADJUST
 INPUT PARAMETERS
 F 06 61 (IMPACT LAT, IMPACT LONG, BLANK)
 COPY PRECISION IMPACT
 LAT AND LONG
 IF NOT ACCEPTABLE, KEY V32F,
 RETURN TO F 06 33 DISPLAY,
 AND ADJUST INPUT PARAMETERS
 PRO IF ACCEPTABLE
 F 06 39 (DELTA T TRANSFER)
 COPY PRECISION DELTA T
 TRANSFER
 IF NOT ACCEPTABLE, KEY V32F,
 GO TO F 06 33 DISPLAY, AND
 ADJUST INPUT PARAMETERS
 PRO IF ACCEPTABLE
 F 06 60 (BLANK, VPRED, GAMMA EI)
 COPY PRECISION VPRED
 AND GAMMA EI
 IF NOT ACCEPTABLE, KEY V32F,
 GO TO F 06 33 DISPLAY, AND
 ADJUST INPUT PARAMETERS
 PRO IF ACCEPTABLE
 F 06 81 (DELTA VX, VY, VZ=LOCAL VERT)
 COPY PRECISION DELTA V
 PRO

F 04 06 (00007, 00001, BLANK)
 VERIFY R2=00001 FOR SPS OR
 LOAD R2=00002 FOR RCS
 PRO
 F 06 33 (TIG OF MIDCOURSE)
 COPY TIG OF MIDCOURSE
 ADJUSTED FOR FINITE
 BURN TIME
 PRO
 F 16 45 (MKS, TFI, MGA)
 COPY MGA
 PRO
 F 37 BB KEY ONE
 P00

LMP RECORD DRIFT DATA AND
VOICE TO GND
COR ATT DB=MAX
FDAI SEL=1/2
ATT SET=IMU

APPENDUM 7
GDC DRIFT CHECK

P
G R A Y S
T O G S
C R E W
P A N E L

PROCEDURES

PROCEDURES FOR GDC DRIFT TEST

P00 COR ATT DB=MIN
RATE=LOW
RMAG MODE(3)=ATT1/RATE2
MAN ATT(3)=RATE CMD
SC COM=SCS
KEY V06N20E
06 20 (R,P,Y)
FDAI SELECT=1
FDAI SOURCE=ATT SET
ATT SET=GDC
NULL ATT ERROR NEEDLES
ON FDAI 1 WITH
ATT SET THUMB WHEELS
(AVOID FALSE NULL BY
COMPARING ATT SET IND
WITH N20)
KEY ENTER WHEN FDAI ERROR
NEEDLES NULLEN
READ DSKY N20 AND ATT SET
VALUES AND CONFIRM DRIFT
LESS THAN 10 DEG PER HR

 ADDENDUM 8
 GND UPLINK
 ***** (P27) *****

D I S P L A Y S
 P L A N E L
 C R E W PROCEDURES
 G R E O T G S

***PROCEDURES FOR GND UPLINK**

P00 CMP KEY V37E00E
 CDR UP TLM(CM) (MDC)-ACCEPT
 P27 CMP MONITOR UPLINK ACT LT-ON
 MONITOR GND UPLINK
 P00 MONITOR UPLINK ACT LT-OFF
 LMP RECORD VOICE SV AND MANEUVER
 PAD DATA
 CDR UP TLM(CM) (MDC)-BLOCK

 ADDENDUM 9
 EXTERNAL DV TARGETING (P30)

D I S P L A Y S
 P L A N E L
 C R E W PROCEDURES
 G R E O T G S

***PROCEDURES FOR EXT DV TARGETING**

P30 LEB KEY V37E30E
 F 06 33 (GETI OF BURN)
 VERIFY GND GETI
 PRO
 F 06 81 (VG = LV)
 VERIFY GND DVX,DVY,DVZ
 PRO
 F 06 42 (MA,HP,VG)
 COPY VG/COMPARE WITH PAD DVY
 PRO
 F 16 45 (MKS,TFI,MGA)
 CDR SET MDC ET COUNT DWN
 CONFIRM MGA LESS THAN 45 DEG
 LEB PRO
 F 37 88
 P00 KEY 00E

 ADDENDUM 10
 SM RCS AND SPS PROPULSION CHECKS

OPERATIONAL PROCEDURES
 P A N E L
 C R E W

 PROC FOR SM RCS MONITOR CHECK
 SM RCS HE TB(8)-GRAY 2
 SM RCS PRIM PRPLNT TB(4)-GRAY 2
 SM RCS SEC PRPLNT TB(4)-GRAY 2
 RCS IND SEL-SM A,B,C,D 2
 CHECK GAUDES A,R,C,D
 SM RCS PKG TEMP
 IND-105-195 DEG
 SM RCS HE PRESS IND-RECORD
 SM RCS IND SW-HE TK TEMP
 SM RCS HE TK TEMP
 IND-RECORD
 SM RCS IND SW-PRPLNT QTY
 SM RCS PRPLNT QTY
 IND-RECORD
 SM RCS MANF PRESS
 IND-178-192PSIA
 WHEN SM RCS MANF PRESS 2
 IND-150PSIA
 RCS SEC FUEL PRESS
 (A,R,C,D)-OPEN

 PROC FOR SPS MONITORING CHECK

SPS PRPLNT TK TEMP IND
 *53 TO *75 DEG
 IF<55DEG,SPS LINE HTS-A
 IF>75DEG,SPS LINE HTS-OFF
 SPS PRESS IND SW-HE,N2A,N2B 3 3
 SPS HE/N2 PRESS IND 3
 SPS HE PRESS-3900PSIA MAX
 SPS N2A PRESS-2900PSIA MAX
 SPS N2B PRESS-2900PSIA MAX 3
 SPS PRESS IND SW-HE 3
 SPS FUEL AND OXID PRESS 3
 INDS-170-195PSIA
 (FUEL/OXID MAX DELTAP=20PSIA)
 SPS ENG INJ VLV IND(4)-CLOSE 3
 RECORD 3
 SPS OXID QTY IND
 SPS FUEL QTY IND
 SPS OXID UNBAL IND
 OXID FLOW VLV PRIM-PRIM 3
 SPS HE VLV(BOTH)-AUTO 3
 SPS HE VLV TB(BOTH)-BP 3

 ADDENDUM 11
 SPS THRUST SETUP (PAD)

D I S P L A Y S
 P R O C E D U R E S
 C R E W
 P A N E L
 A R E
 N E E
 L

***PROCEDURES FOR SPS THRUST**

P00 LEB KEY V37E00E
 CDR CONFIRM GND PIFA BIAS CHECK

***PROCEDURES FOR EMS TEST AND e/U**

CDR CR,EMS(2)/CLOSED
 SC CONT=CMC
 CMC MODE=FREE
 EMS MODE=STBY
 EMS FUNCTION=OFF
 EMS FUNCTION=DV
 EMS MODE=AUTO
 VERIFY DV LESS THAN
 3.27PS PER 10NSEC
 CMC MODE=AUTO
 EMS FUNCTION=OFF
 EMS MODE=STBY
 EMS FUNCTION=DV SET
 EMS MODE=AUTO
 LOAD DV = 1586.8

EMS FUNCTION=DV TEST
 MONITOR SPS LT-ON
 MONITOR DV COUNTDOWN
 MONITOR SPS LT-OFF
 CONFIRM DV = -20.8 +/- 20.7
 EMS MODE=STBY
 EMS FUNCTION=DV SET
 LOAD BURN VC
 EMS FUNCTION=DV

F 06 22 LEB KEY V49E
 (FINAL GMBL ANGLES)
 LOAD BURN R,P,Y GMBL ANGLES
 KEY V62E

CDR BMAG MODE(3)=RATE2
 RMC=ARMED

F 50 18 LEB PRO
 (COMMANDED R,P,Y)
 PRO

06 18 (COMMANDED R,P,Y)
 CDR MONITOR ATT MANEUVER
 FOR GMBL LOCK

F 50 18 (COMMANDED R,P,Y)
 VERIFY R,P,Y AGREE WITH BURN
 PAD R,P,Y WITHIN 5 DEG
 LEB KEY ENTER

PROC FOR SPS THRUST PROG

P40 LEB KEY V37E40E
 (COMMANDED R,P,Y)
 VERIFY R,P,Y AGREE WITH PAD
 VALUES WITHIN 5 DEG
 CDR ALIGN SC IN ROLL TO PAD VALUE
 LEB PRO
 06 18 (COMMANDED R,P,Y)

CDR MONITOR BURN ATT TRIM
 F 50 18 (COMMANDED R,P,Y)
 LEB G/N PWR OPTICS-ON
 OPT ZERO-ZERO(15SEC)
 OPT TELTRUN-SLAVE TO SXT
 RETICLE BRT TW-ADJUST
 OPT ZERO-OFF
 OPT MODE-CMC
 KEY V-INGIE
 F 21 92 (BLANK,BLANK,BLANK)
 LOAD PAD SHAFT ANGLE(XXX,XX)
 IN R1 AND PAD TRUNNION A,GLF
 (XX,XX) IN R2
 41 RB (SHAFT,TRUNNION,PLANK)
 MONITOR OPT URIVE TO PAD
 VALUES
 F 50 18 (COMMANDED R,P,Y)
 VERIFY PAD STAR IN SXT
 OPT MODE-MAN
 DRIVE TRUNN LESS THAN 5 DEG
 OPT ZERO-ZERO
 RETICLE BRT TW-MIN BRTNESS
 MOVE TO CENTER SEAT
 LMP MN BUS TIE(2)-ON
 PLUG MODE-NORM(AUX FOR TEI)
 SPS HE VLV TA (ROTH)-RP
 SPS HE VLV (ROTH)-AUTO
 CDR RHC PWR DIRECT(ROTH)-OFF
 SC CONT-CMC
 CMC MODE-AUTO
 SCS TVC (BOTH)-RATE CMD
 TVC GMRL DRIVE (2)-AUTO
 PERFORM MIVC CHECKS
 TVC SERVO PWR 1-AC 1/MNA
 TVC SERVO PWR 2-AC 2/MNB
 TRANS CONTR PWR-ON (UP)
 RHC PWR NORM 2-AC
 GMRL MTRS PITCH 1-START-ON
 GMRL MTRS YAW 1-START-ON
 THC-CLOCKWISE
 RHC-VERIFY NO MTVC
 GMRL MTRS PITCH 2-START ON
 GMRL MTRS YAW 2-START ON
 SFT GPI PTRIM TO PAD VALUE
 YTRIM TO PAD VALUE
 RHC-VERIFY MTVC
 THC-NEUTRAL
 RHC PWR NORM 2-AC/DC
 ALIGN SC IN ROLL TO PAD VALUE
 CMP PRO
 06 18 (COMMANDED R,P,Y)
 MONITOR ATT TRIM
 F 50 18 (COMMANDED R,P,Y)
 VERIFY R,P,Y AGREE WITH PAD
 VALUES WITHIN 5 DEG
 KEY ENTER
 F 50 25 (00204, GMRL DRIVE TEST)
 CDR RHC PWR DIRECT(ROTH)-MNA/MNB
 MAN ATT (3)-RATE CMD
 ATT DR-MIN
 RATE-HIGH
 HMAG MODE (3)-ATT1/RATE 2
 AUTO RCS SEL(16)-MNA
 ALIGN GNC TO IMU
 CDR FDAI SELECT 1
 FDAI SOURCE-ATT SET
 ATT SET-IMU
 CMP KEY V16N20E
 F 16 20 (R,P,Y)
 CDR NULL FDAI NEEDLES WITH

-15 ATT SET TW (AVOID FALSE
 NULL BY COMPARING ATT SET
 IND WITH N20)
 -5 ATT SET-GDC
 DEPRESS GDC ALIGN PB
 FDAI SELECT-1/2
 CMP KEY RELEASE
 F 50 25 (00204, GMBL DRIVE TEST)
 CMP PRO
 CDR MONITOR GMHL DRIVE
 SEQ AND TRIM
 06 40 (TFI, VG, DVM)
 VERIFY SPS GPI TRIM AGREES
 WITH PAD WITHIN 0.5 DEG
 -2MIN LMP NONESS BUS-MNA
 TAPE RCDR FWD-OFF(CTR)
 TLM INPUTS PCM-HIGH
 UP TLM CMD-RESET THEN NORM
 TAPE RCDR PCM-PCM/ANLG
 TAPE RCDR RCD-RCD
 TAPE RCDR FWD-FWD
 TAPE MOTION TB-RP
 FLT RCDR-RECORD
 (FOR LO11, LO12, AND TEI)
 CDR FDAI SCALE-5/5
 VERIFY SPS TH LT-OFF
 CB SPS P2-OPEN(TEI ONLY)
 CB SPS Y2-OPEN(TEI ONLY)
 DV THRUST(BOTH)-NORMAL
 THC-ARMED
 RMC-ARMED
 DSKY BLANKS
 06 40 (TFI, VG, DVM)
 CDR EMS MODE-AUTO
 -35SEC CMP CK DVM FOR HI PIPA BIAS
 -30
 -25

33

FINAL MISSION C-PRIME PROCEDURES DATED 12 NOV 1968

(LESS THAN 2FPS PER 5 SEC)
 CDR APPLY ULLAGE THRUST
 (IF RQUIRED)
 -5 F 99 40 (REQUEST FOR ENGINE ENABLE)
 CMP PRO
 0 06 40 (TFC, VG, DVM)
 CDR MONITOR SPS LT-ON/START
 MDC ET COUNT UP
 TERMINATE ULLAGE 1 SEC
 AFTER SPS IGNITION
 CONFIRM FDAI 1 RATES LESS THAN
 10 DEG/SEC
 MONITOR PC GAUGE 95-105 PSIA
 VERIFY VG/DECR AND DVM/INCR
 CMP MONITOR FDAI 2 RATES LESS THAN
 10 DEG/SEC
 LMP MONITOR VALVES OPEN AND
 SPS GAUGES
 CDR MONITOR SPS ENGINE CUTOFF
 MONITOR SPS LT-OFF
 F 16 40 (TFC, VG, DVM) (COPY VG AND DVM)
 CUTOFF+1SEC DV THRUST(BOTH)-OFF
 MONITOR PC GAUGE ZERO
 LMP MONITOR VALVES CLOSED
 SPS HE VLV TB(BOTH)-BP
 FLT RCDR-OFF(CTR)
 CDR CB SPS P2-CLOSED
 CB SPS Y2-CLOSED
 SPS GMBL MTRS(4)-ONF
 (SEQUENTIALLY-P2, Y2, P1, Y1)
 TVC SERVO PWR(BOTH)-OFF
 (SEQUENTIALLY-2, 1)
 CMP PRO
 F 16 85 (VG-BODY)
 CDR NULL VG RESIDUALS AS READ
 THC-LOCKED

34

51

 ADDENDUM 12
 RCS THRUST SETUP (P41)

P00
 CALL IDLING PROGRAM
 KEY 00E
 KEY V66E (TRANSFER POST BURN
 SV FROM CSM SLOTS
 TO LM SLOTS)
 LMP NONESS BUS-OFF
 TAPE RCDR FWD-OFF (CTR)
 TAPE MOTION TB-GRAY
 (ASSUME GND PLAYRACK AT
 NEXT OPPORTUNITY)
 MN BUS TIE(2)-OFF
 CHARGE BATTERIES
 CDR EMS MODE-STBY
 EMS FUNCTION-OFF
 FOAI SCALE-5/1
 ATT DB-MAX
 RATE-LOW
 TRANS CONT PWR-OFF
 RHC PWR DIRECT (BOTH)-OFF
 RMAG MODE (3)-RATE 2

P00
 P R A Y S
 G R A Y S
 E O Y S
 T G S

 PROCÉDURES FOR RCS THRUST
 LER KEY V37E00E
 CDR CONFIRM: GND PIPA BIAS CHECK
 PROCÉDURES FOR EMS SETUP

CDR SC CONT-CMC
 CMC MODE-FREE
 EMS MODE-STBY
 EMS FUNCTION-OV
 EMS MODE-AUTO
 VERIFY DV LESS THAN
 3.2FPS PER 100SEC
 CMC MODE-AUTO
 EMS FUNCTION-OFF
 EMS MODE-STBY
 EMS FUNCTION-OV SET
 LOAD BURN VC
 EMS FUNCTION-OV

LER KEY V49E

RHC-LOCKED
 CMP PRO
 F 37 88

P00

F 06 22 (FINAL GMBL ANGLES)
 LOAD BURN R,P,Y GMBL ANGLES
 KEY V62E
 CDR RMAG MODE(3)-RATE2
 RMC-ARMED
 LEB PRO
 F 50 18 (COMMANDED R,P,Y)
 PRN
 06 18 (COMMANDED R,P,Y)
 CDR MONITOR ATT MANFUVER
 FOR GMBL LOCK
 F 50 18 (COMMANDED R,P,Y)
 VERIFY R,P,Y AGREE WITH
 PAD R,P,Y WITHIN 5 DFG
 LEB KEY ENTER
 PROC FOR RCS THRUST PROG
 P41 LEB KEY V37E41E
 F 50 18 (COMMANDED R,P,Y)
 CDR ALIGN SC IN ROLL TO PAD VALUF
 LEB PRO
 06 18 (COMMANDED R,P,Y)
 CDR MONITOR BURN ATT TRIM
 F 50 18 (COMMANDED R,P,Y)
 VERIFY R,P,Y AGREE WITH PAD
 VALUES WITHIN 5 DEG
 LEB OPT MODE-MAN
 G/N PWR OPTICS-ON
 OPT ZERO-ZERO(15SEC)
 OPT TELTRUN-SLAVE TO SXT
 RETICLE BRT TM-ADJUST
 OPT ZERO-OFF
 OPT MODE-CMC
 KEY V41N91E
 F 21 92 (BLANK,BLANK,BLANK)

LOAD PAD SHAFT ANGLF(XXX,XX)
 IN R1 AND PAD TRUNNION ANGLE
 (XX,XXX) IN R2
 41 88 (SHAFT,TRUNNION,BLANK)
 MONITOR OPT DRIVE TO PAD
 VALUES
 F 50 18 (COMMANDED R,P,Y)
 VERIFY PAD STAR IN SXT
 OPT MODE-MAN
 DRIVE TRUN LESS THAN 5 DEG
 OPT ZERO-ZERO
 RETICLE BRT TM-MIN RTNESS
 MOVE TO CENTER SEAT
 KEY ENTER
 06 85 (VG-RDNY)
 CDR ATT DR-MIN
 RATE-LOW
 RMAG MODE(3)-ATT1/RATE2
 AUTO RCS SEL(16)-MNA
 TRANS CONT PWR-ON(UP)
 ALIGN GDC TO IMU
 CDR FDAI SELECT 1
 FDAI SOURCE-ATT SET
 ATT SET-IMU
 CMP KEY V16N20E
 16 20 (R,P,Y)
 CDR NULL FDAI NEEDLES WITH
 ATT SET TW (AVOID FALSE
 NULL BY COMPARING ATT SET
 IND WITH N20)
 ATT SET-GDC
 DEPRESS GDC ALIGN PA
 FOAI SELECT-1/2
 CMP KEY RELEASE

EMS FUNCTION-OFF
 ATT NB-MAX
 BMAG MODE(3)-RATE 2
 TRANS CONT PWR-OFF

-2MIN LMP TLM INPUTS PCM-HIGH
 UP TLM CMD-RESET THEN NOR.
 TAPE RCDR PCM-PCM/ANLG
 TAPE RCDR RCD-RCD
 TAPE RCDR FWJ-FWD
 TAPE MOTION TB-RP
 DSKY BLANKS
 -35 SEC 16 AS (VG-BODY)
 -30 EMS MODE-AUTO
 -25 SEC CMP KEY N40E
 16 40 (TFI, VG, DVM)
 MONITOR DVM FOR HT PIPA BIAS
 (LESS THAN 2FPS PER 5SEC,
 KEY RELEASE
 CDR RMC-ARMED
 THC-ARMED
 -0 F 16 AS (VG-BODY)
 SET MUC ET COUNTING UP AND
 NULL VG AS REQUIRED
 THC-LOCKED
 RMC-LOCKED
 CMP PHO
 F 37 AB
 CALL IDLING PROGRAM
 P00 KEY 00F
 KEY V66E (TRANSFER POST BURN
 SV FROM C.M SLOTS
 TO LM SLOTS)
 LMP TAPE RCDR FWD-OFF(CTR)
 TAPE MOTION TB-GRAY
 (ASSUME GND PLAYBACK AT
 NEXT OPPORTUNITY)
 CDR EMS MODE-STBY

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FINAL MISSION C-PRIME PROCEDURES DATED 12 NOV 196A

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 APPENDUM 13
 ESTABLISH PTC

LMP RECEIVE PTC R,P,Y FROM GND
 OR USE PREVIOUS PTC ANGLES
 IF STILL VALID.COORDINATE PTC
 GMBL ANGLE UPDATES WITH GND

D I S P L A Y S
 P R O C E D U R E S
 P A N E L

CMP KEY V49E
 F 06 22 (FINAL GMBL ANGLES)
 LOAD PTC GMBL ANGLES
 KEY V62E
 RHC=ARMED
 PRO
 F 50 18 (COMMANDED R,P,Y)
 PRO
 06 18 (COMMANDED R,P,Y)
 CDR MONITOR ATT MANEUVER
 FOR GMBL LOCK
 F 50 18 (COMMANDED R,P,Y)
 VERIFY R,P,Y AGREE WITH PTC
 PAD R,P,Y WITHIN 5 DEG
 NULL FDAI NEEDLES WITH RHC
 CMP KEY ENTER
 CDR BMAG MODE(3)=ATT1/RATE2
 SC CONT=SCS
 AUTO RCS SEL ROLL(8)=MNA
 CONFIGURE AUTO RCS SEL FOR
 SINGLE JET PITCH AND YAW
 (PITCH=(*)A3,(*)C4)
 (YAW=(*)B3,(*)D4)
 LIMIT CYCLE=ON(UP)
 ATT DB=MAX
 RATE=HIGH
 BMAG MODE(R)=RATE2
 MAN ATT(P,Y)=RATE CMD
 MAN ATT(R)=ACC CMD
 MANEUVER SC IN ROLL ABOUT THE

P00 CDR ATT DB=MIN
 RATE=LOW
 MAN ATT(3)=RATE CMD
 BMAG MODE(3)=RATE2
 SC CONT=CMC
 CMC MODE=AUTO
 FDAI SCALE=5/1
 FDAI SELECT=1/2

CMP KEY V49E
 F 04 46 (DAP CONFIGURATION)
 LOAD OR VERIFY
 11112
 11111
 B
 PRO
 F 06 47 (CSM AND LM WT)
 PRO
 F 06 48 (SPS GMBL TRIM)
 PRO

BODY X AXIS TO SET UP A
 .2 DEG/SEC BODY ROLL RATE
 RHC-LOCKED
 LMP RECORD PTC START GET IN LOG
 CDR MAINTAIN PTC ATT TO +/- 2.0 DEG
 MAINTAIN BODY ROLL RATE TO
 .2 DEG/SEC +OR- .1 DEG/SEC
 DURING PERIODS OF PTC

LMP RECORD PTC CORRECTIONS IN LOG
 IF THE PTC ATTITUDE MUST BE
 INTERRUPTED AND A NEW ATT
 SELECTED, NULLING THE NEEDLES
 WILL RETURN SC TO PTC ATT

 ADDENDUM 14
 TERMINATE PTC

D I S P L A Y S
 P R A E W
 G R A E I
 P 0 0

C R E W P R O C E D U R E S
 P A N E L

CMP KEY V16N20E
 16 20 (CURRENT R.P.Y)
 CDR RHC-ARMED
 LIMIT CYCLE-OFF
 STOP PTC ROLL AT THE
 PREFERRED ATTITUDE FOR NAV
 SIGHTINGS OR AN IMU ALIGNMENT
 BY SWITCHING
 MAN ATT(3)-RATE CMD AT THE
 APPROPRIATE GMRL ANGLES
 BMAG MODE(3)-ATT1/RATE2
 RATE-LOW
 FDAI SELECT-1/2
 FDAI SCALE-5/1
 AUTO RCS SEL(16)-MNA
 LMP RECORD PTC STOP GET IN LOG

 APPENDUM 15
 ORBITAL NAVIGATION PROCEDURES (P22)
 (PSEUDO LANDING SITE TRACKING)

C I S P
 P L A R C
 G R A R F
 T G S F W
 P A N E L
 PROCEDURES

LEB KEY V37E22E

F 06 45 (BLANK, BLANK, MAX MGA)
 LEB PRO
 F 05 70 (LANDMARK CODE)
 LOAD 10001 IN R2
 LEB OPT ZERO-OFF
 OPT MODE-MAN
 POSITION TRUNNION TO 35 DEG
 OPT MODE-CMC

LEB PRO P00
 06 92 (SHAFT, TRUN, BLANK)
 MONITOR OPT DRIVE TO LANDING
 SITE

LEB IDENTIFY IDENTIFICATION POINTS
 IF VISABLE
 LEB, CDR WHEN LANDING SITE IDENTIFIED.
 START 0.3 DEG/SEC
 PITCH RATE

CDR ROLL TO AVOID SHAFT AXIS
 WITHIN 10 DEG OF LANDING

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FINAL MISSION C-PRIME PROCEDURES DATED 12 NOV 196A

SITE
 LEB OPT MODE-MAN
 F 51 88 (PLEASE MARK)
 MAKE 5 MARKS-20 SEC APART
 POSITION TRUNNION TO 5 DEG
 OPT ZERO-ZERO

CDR STOP PITCH RATE

LEB PRO
 F 05 71 (LANDING SITE DATA)
 VERIFY R2

LEB PRO
 F 06 49 (DELTA R, DELTA V, BLANK)
 RECORD VALUES

**HOLD AT THIS DISPLAY FOR
 ONE MINUTE**

LEB PRO
 F 05 89 (LAT, LONG/2, ALT)
 RECORD VALUES

**HOLD AT THIS DISPLAY FOR
 20 SECONDS**

LEB KEY V37E00E

67

46

 ADDENDUM 16
 ORBITAL NAVIGATION PROCEDURES (P2?)
 (LANDMARK TRACKING-NO AUTO OPTICS)

DISPLAY PROCEDURES
 P L C R E W
 G R A R E W
 E O Y E W
 T G S W
 LEB KEY V37E22E

F 06 45 (BLANK,BLANK,MAX MGA)
 LEB PRO
 F 05 70 (LANDMARK CODE)
 LOAD 2000 IN R2
 LEB OPT ZERO-OFF
 OPT MODE-MAN
 LFB PRO
 F 51 RR (PLEASE MARK)
 PLACE OPTICS AT 10 DEG
 TRUNNION ANGLE
 IDENTIFY IDENTIFICATION POINTS
 AT THEIR RESPECTIVE
 ACQUISITION TIMES
 LEB,CDR WHEN LANDMARK IDENTIFIED-
 START 0.3 DEG/SEC
 PITCH RATE
 CDR ROLL TO AVOID SHAFT AXIS
 WITHIN 10 DEG OF LANDMARK
 LEB MAKE 5 MARKS-20 SEC APART

POSITION TRUNNION TO 5 DEG
 OPT ZERO-ZERO
 CDR STOP PITCH RATE
 MAN ATT(P)-ACCEL CMD
 PITCH SC UP TO 5 DEG PITCH
 AND 0 DEG ROLL W.R.T.
 LOCAL HORIZONTAL
 MAN ATT(P)-MIN IMP
 INITIATE INITIAL PITCH RATE
 FOR NEXT LANDMARK SIGHTING

LEB PRO
 F 05 71 (LANDMARK DATA)
 VERIFY R2
 LEB PRO
 F 06 49 (DELTA R,DELTA V,BLANK)
 RECORD VALUES

**HOLD AT THIS DISPLAY FOR
 20 SECONDS**

LEB PRO
 F 06 89 (LAT,LONG/2,ALT)
 RECORD VALUES

**HOLD AT THIS DISPLAY FOR
 20 SECONDS**

LEB KEY V37E00E

 ADDENDUM 17
 ORBITAL NAVIGATION PROCEDURES (P22)
 (LANDMARK TRACKING-WITH AUTO OPT)

C
 I
 S
 P
 G R A C
 E O Y R F
 T G S W
 P R O C E D U R E S
 P 2 2
 L E B K E Y V 3 7 2 2 E

F 06 45 (BLANK,BLANK,MAX MGA)
 LEB PRO
 F 05 70 (LANDMARK CODE)
 LOAD 10000 IN R2
 LEB PRO
 F 06 89 (LAT, LONG/2, ALT)
 LOAD PREVIOUSLY COMPUTED
 LANDMARK COORDINATES
 LEB OPT ZERO-OFF
 OPT MODE=MAN
 POSITION TRUNNION TO 35 DEG
 OPT MODE=CMC
 PRO
 06 92 (SHAFT, TRUN, BLANK)
 MONITOR OPT DRIVE TO LANDMARK
 IDENTIFY IDENTIFICATION POINTS
 IF VISABLE
 LEB, COR WHEN LANDMARK IDENTIFIED-
 START 0.3 DEG/SEC

PITCH RATE
 CDR ROLL TO AVOID SHAFT AXIS
 WITHIN 10 DEG OF LANDMARK
 LEB OPT MODE=MAN
 F 51 88 (PLEASE MARK)
 MAKE 3 MARKS=20 SEC APART
 POSITION TRUNNION TO 5 DEG
 OPT ZERO=ZERO
 CDR STOP PITCH RATE
 MAN ATT(P)-ACCEL CMD
 PITCH SC UP TO 5 DEG PITCH
 AND 0 DEG ROLL W.R.T.
 LOCAL HORIZONTAL
 MAN ATT(P)-MIN IMP
 INITIATE ORBITAL PITCH RATE
 FOR NEXT LANDMARK SIGHTING
 LEB PRO
 F 05 71 (LANDMARK DATA)
 VERIFY R2
 LEB PRO
 F 06 89 (LAT, LONG/2, ALT)
 VERIFY LANDMARK COORDINATES
 LEB PRO
 F 06 49 (DELTA R, DELTA V, BLANK)
 RECORD VALUES
 **HOLD AT THIS DISPLAY FOR
 20 SECONDS**
 LEB PRO
 F 06 89 (LAT, LONG/2, ALT)
 RECORD VALUES
 **HOLD AT THIS DISPLAY FOR
 20 SECONDS**

LER KEY V37E00E

P00

ADDENDUM 18
ORDEAL INITIALIZATION/VERIFICATION

D I S P
P L A Y S
P R O G R A M
G R A Y S
E T

C R E W
P A N E L
P R O C E D U R E S

PROCEDURES FOR ORDEAL SETUP

CDR SET ORDEAL ON FDAI 1
FDAI/GPI PWR=BOTH
LOGIC 2/3 PWR=ON
CMC ATT=IMU
FDAI SELECT=1/2
FDAI 1=ORB RATE
EARTH/LUNAR=LUNAR
MODE=HOLD/FAST
KEY V82E
F 04 06 (00002,00001)
PR0
F 16 44 (HA,MP,TFF)
ALT SET=SET ALT TO HA,MP AVG
PR0
KEY V83E
F 16 54 (R,ROOT,THETA)
VERIFY 0 DEG YAW
SLEW FDAI 1 TO THETA
MODE=OPR/SLOW
SLEW FDAI PRECISELY

PRO

ADDENDUM 19
HIGH GAIN ANT ACQUISITION

P00

LMP HI GAIN ANT TRACK=MAN
HI GAIN ANT BEAM=WIDE

CMP KEY V64E

F 06 51 (RHO,GAMMA,BLANK)
LMP RECORD HI GAIN ANT

GMBL COORDINATES

CMP PRO

LMP HI GAIN ANT POSITION
SET ANT PITCH TO RHO
SET ANT YAW TO GAMMA
S=BAND ANT OMNI=HI GAIN
VERIFY HI GAIN S=BAND
ANT IND GREATER THAN
HALF SCALE

HI GAIN ANT TRACK=

AUTO OR REACO AS REQUIRED
HI GAIN ANT BEAM=AS REQUIRED

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- (16) MSC Memo 68-FM47-441; "Detailed Sighting Schedule for the Transearth Leg of the 21 December 1968 Launch Windows (Fast Turn)"; dated 13 November 1968 (Amended per 18 November Meeting)