

Massachusetts Institute of Technology
C. S. Draper Laboratory
Cambridge, Massachusetts

LUMINARY Memo #160

To: Distribution
From: David Moore
Date: 16 July 1970
Subject: Descent Trajectory Monitor Package

Summary:

The LMS at KSC has asked for a comparison of the Descent Trajectory Monitor Chart between that supplied by MPAD (LM Timeline Book, 2/16/70) and that supplied by MIT all digital simulation data (Luminous 1D-level 4-Lunar Landing). The package is included in the memo and has been sent to KSC for verification on the LMS.

Purpose:

Ascertain reason for apparent difference in VI, \dot{H} , and H during descent as monitored on LMS/KSC and compared with MPAD/LM Timeline data.

Test:

Obtain nominal all-digital simulation, plot VI, HDOT, H with VI, \dot{H} , H from MPAD/LM Timeline vs. Time in Powered Descent. Ascertain curve relationship of each curve (if curves coincide, the data corresponds; if curves parallel, the data corresponds). The curves are on Figures 1, 2 and 3.

Results:

The curves are seen to be essentially parallel in that a general shape pattern is followed. The maximum error evident from the plots is:

VI (Figure 1) - 3%
H (Figure 2) - 2%
HDOT (Fig. 3) - 10%

The greatest difference in the HDOT curves is 10%; however it is noticed that the general curve shape of each curve is approximately the same. The curves converge at throttle down and at higate and otherwise follow a pattern of essentially the same shape. The VI and H curves differ only slightly.

Data was also available from LM TIMELINE BOOK (3/16/7) and included in the plots of Figures 4, 5, and 6. These curves are similar to those of Figures 1, 2 and 3 in that a general shape pattern is also followed.

For both sets of LM TIMELINE data, the curves are of the same family; in that, if $f(V)$, $f(HDOT)$, $f(H)$ are the trajectory curves as plotted; f_1 is MIT data, f_2 is MPAD data: Then

$$f_2(V) = f_1(V) + C_V$$

$$f_2(HDOT) = f_1(HDOT) + C_{HDOT}$$

$$f_2(H) = f_1(H) + C_H$$

where each C is a constant by which the curves of Figures 1-6 differ.

This means that a percentage error plot should tell that the curves do not converge; i.e., the C in the equations does not vary along the trajectory. If the C does vary, however, the curves will converge and the percentage error will be small at the points of convergence. For example in Fig. 1, $C_V \approx 200$ so that the percentage error at 4500 fps is less than that at 1000 fps. The C in that case does not vary as the curves plainly show no convergence. On the other hand, Figure 3 implies a varying C_{HDOT} along the curve which is evidenced by the convergence of the curves at $T=390$ (throttle-down) and $T=480$ (HIGATE).

Conclusion:

Readily apparent from the plots is the fact that the initial state of all sets of data closely agree. However, it is not so apparent what trajectory information was given to the simulated AGC so that the curves in Figures 1-6 resulted. Possible differences include:

- A. Terrain model difference
- B. Trajectory target parameter difference.

The main conclusion is that KSC/LMS should use the MIT/DL all-digital simulation initialization included in this memo to ascertain possibility of

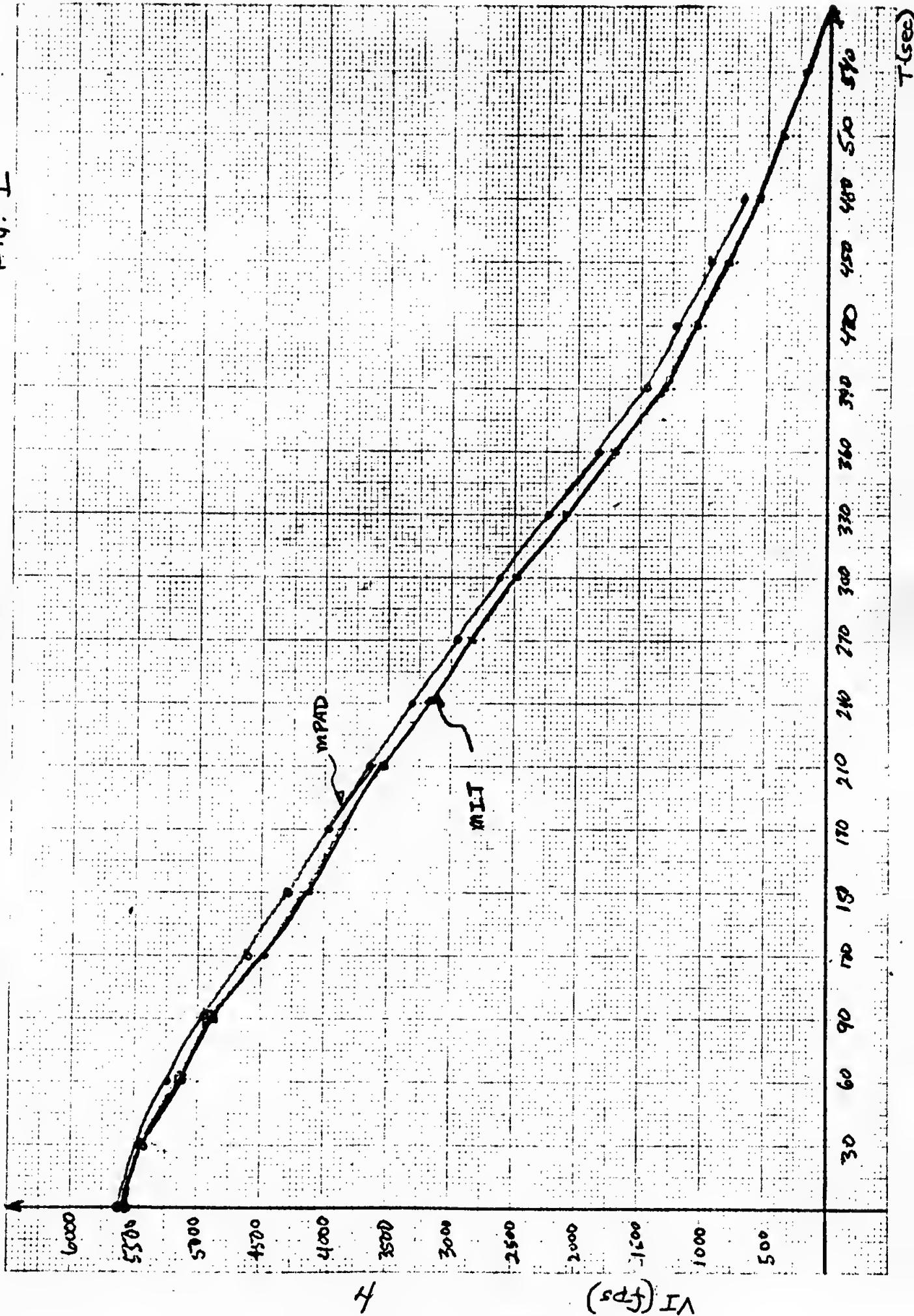
agreement between simulations. Failure of agreement between LMS/KSC and MIT then would open questions as to consistency of simulations. Agreement between the simulators would imply inconsistent initialization and communication of the correct data would be encouraged.

VII VS TIME IN POWERED DESCENT

— = MIT ALL-DIGITAL

- - = MPD LM TIMELINE (2/16/79)

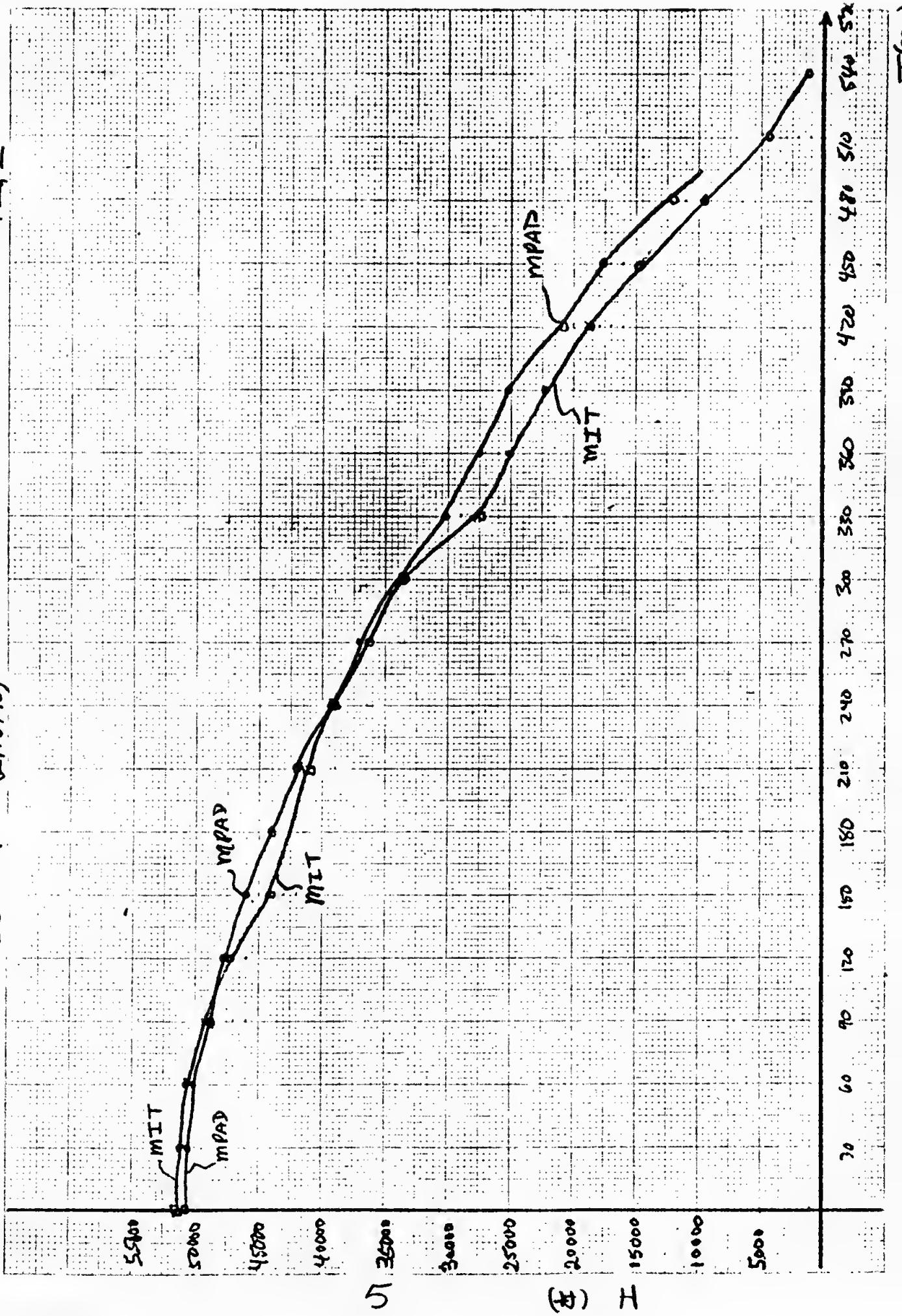
FIG. 8



H vs TIME IN POWERED DESCENT

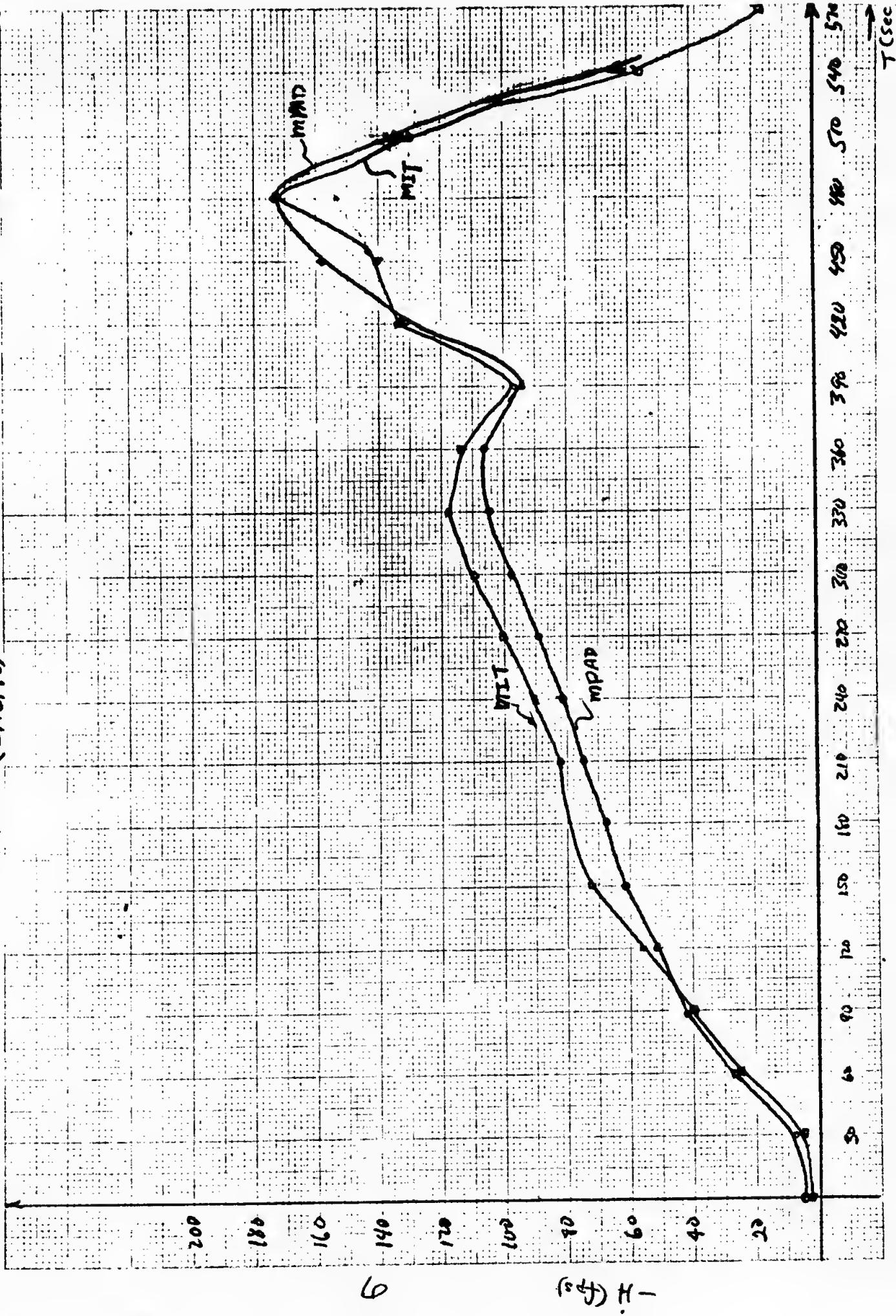
— MIT (2/16/70)

Fig 2



HOOT VS TIME IN POWERED DESCENT
- MPA (2/16/70)

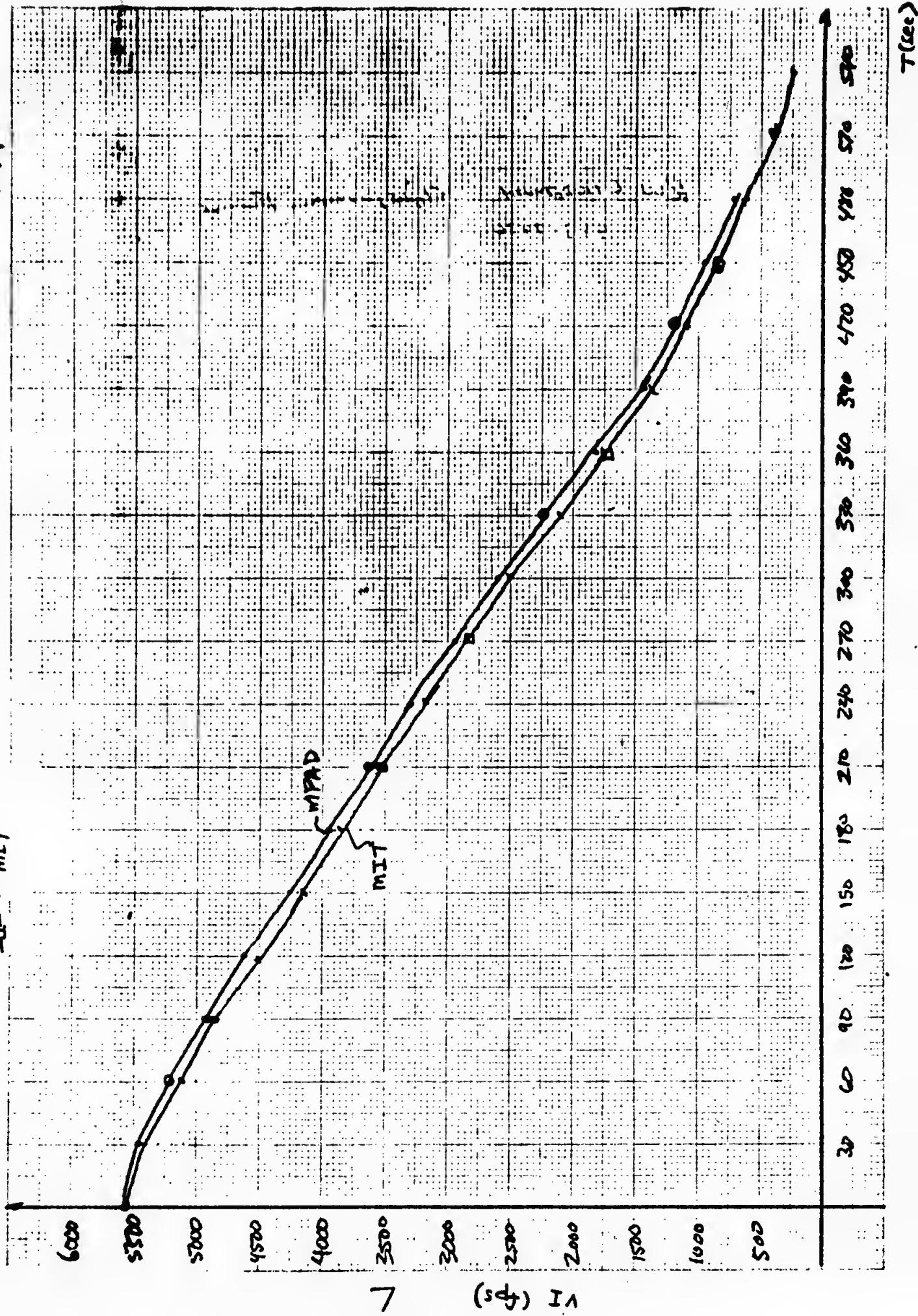
FIG. 3



V_I VS TIME IN REVERSE DESCENT

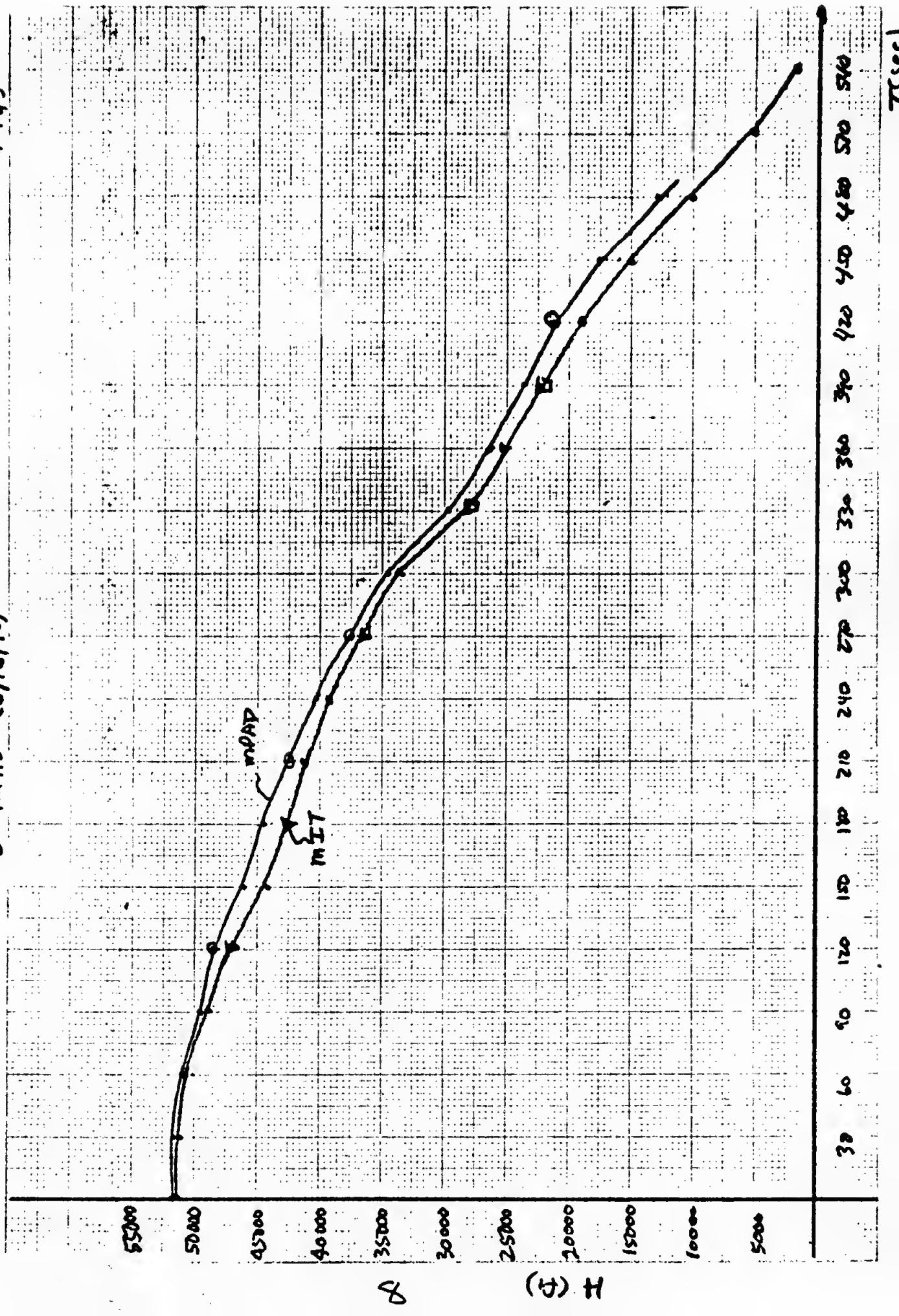
— M₂T — M₂T

Fig 4



H vs TIME IN DESCENT

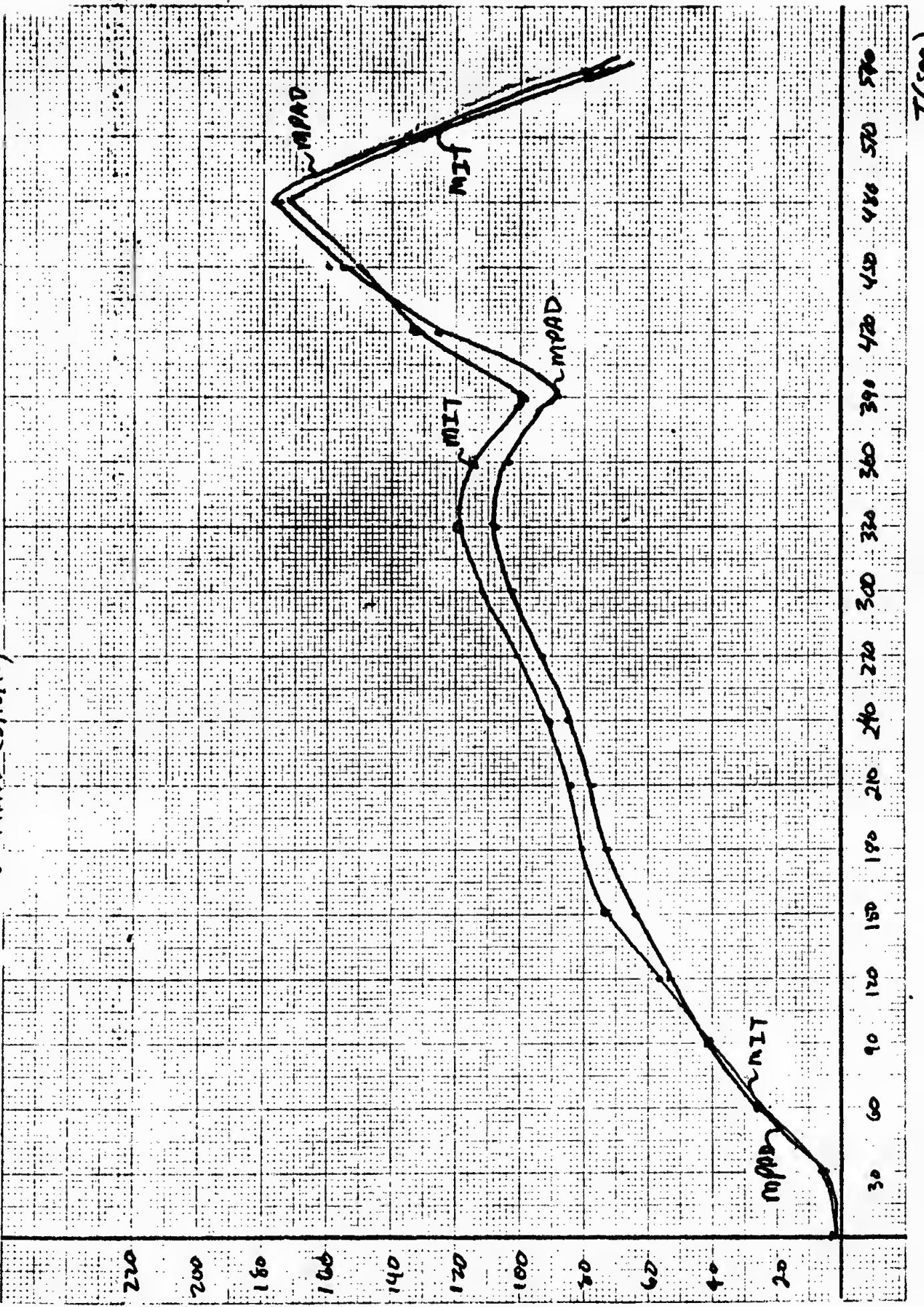
-□- MIT
-○- MPAD (3/16/70)



HDOT vs T. in POWERED DESCENT

DATA (3/16/70)

FIG 6.



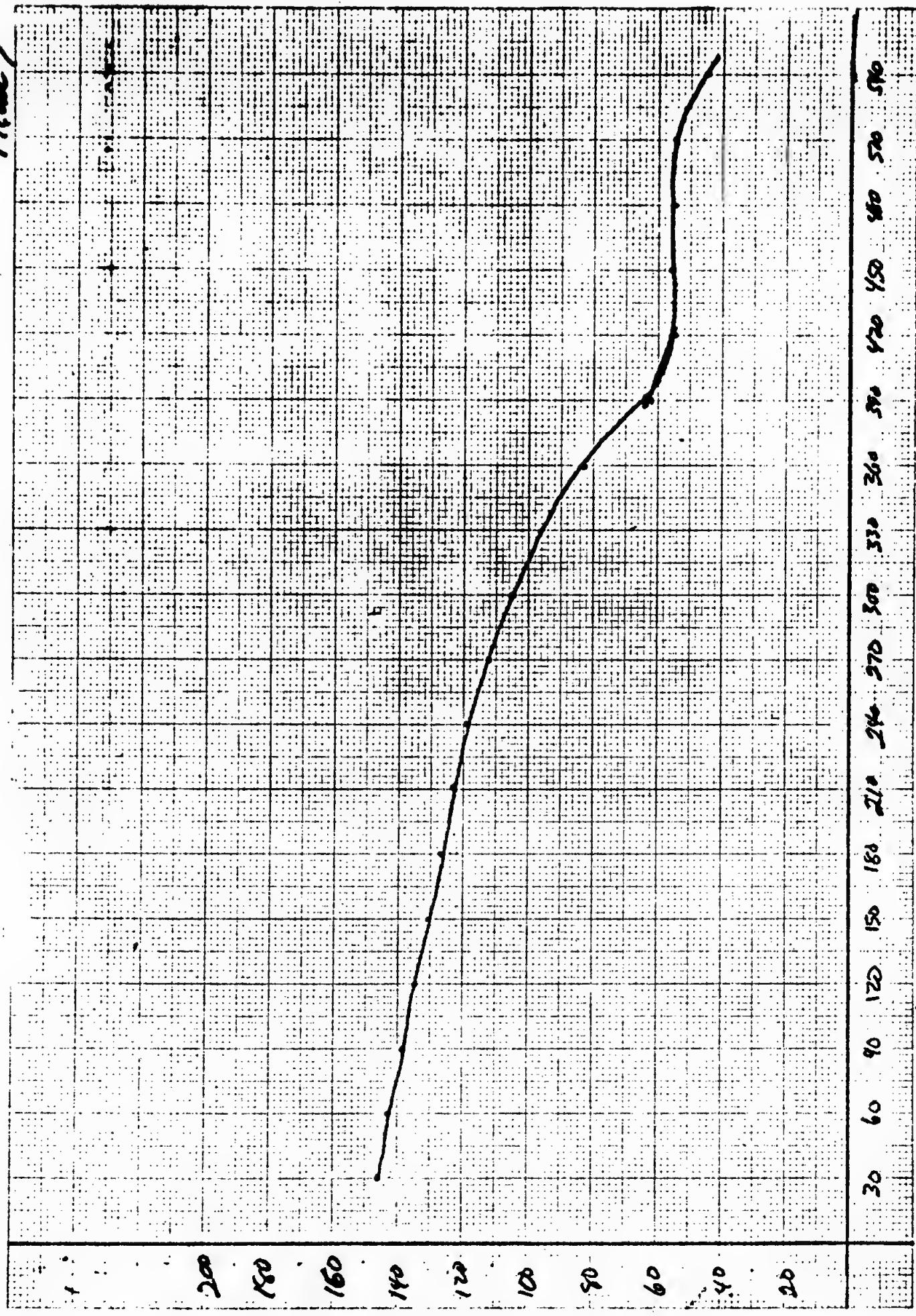
(54) H -

T(s)

7. THRUST VS TIME IN POWERED DESCENT

).

Figure 7



THECM³ (lb SEC²) = 10,500 lb Thrust

DESCENT TRAJECTORY MONITOR
PACKAGE

The following package contains

- A. DESCENT TRAJECTORY TIME CHART
- B. ERASABLE PADLOAD
- C. INITIALIZATION EDIT (ENV. vs. AGC)
- D. ASTRONAUT FILE
- E. EVENT LIST

DESCENT TRAJECTORY TIME CHART

T(from TIG)	V	H	H	MODE
0	5558.3	-2.9	51457	63
30	5478.9	-5.0	51387	
60	5193.9	-25.1	50905	
90	4900.2	-40.1	49920	
120	4496.0	-55.8	47989	
150	4183.4	-73.7	44084	
210	3532.0	-82.1	41743	
240	3191.7	-90.7	39150	
270	2840.3	-100.1	36627	
300	2476.1	-111.9	33826	
330	2099.7	-118.7	27838	
360	1711.2	-113.9	25100	
390	1308.8	-97.4	22254	
420	1059.5	-134.1	18915	
450	818.6	-160.5	14632	
480	575.7	-173.7	9777	
510	364.6	-135.6	4761	64
540	232.9	-57.6	1674	
570	132.2	-18.8	641	
600	56.2	-6.3	294	
630	6.7	-3.5	141	
660	0	-1.3	66	66
690	0	-1.3	20	66

DATE FEBRUARY 16, 1970

DESCENT TRAJECTORY THRU TD+3 MIN

RESET WATCH	MASTER ARM-ON	V1	(-HMAX) -HDOT	(AH)	DPS	SBD P/Y	
- 1:00	MASTER ARM-ON						
- :30	ENG ARM-DES						
- :07.5	ULLAGE						
- :05	PRO						
+ :00	PDI						
+ :02	(NO IGN) -						
	START PB - PUSH						
+ :05	DES ENG OVRD						
- :05	MASTER ARM-OFF						
+0:26	THROTTLE UP						
	$\sqrt{T/W} > 1.6$						
	V21N69						
	V57E - (+) LR HIGHER						
	THAN LGC PRO TO						
	PERMIT LR DATA						
	✓ ED BATTs						
	79 4:30 2962.9	89.5	81.7	39675	61 19/-18	PGNS MODE CONT- ATT HOLD	(63.8)
	79 4:00 3310.8	74.5	44151	72 15/-14			
	80 5:00 2603.5	97.4	42020	67			
	77 5:00 2603.5	89.5	(+17000)	56			
	77 5:30 2231.6	104.1	33795	51 23/-21			
	75 5:30 2231.6	104.1	(+15600)	45			
	75 5:30 2231.6	104.1	(+13500)				
	72 6:00 1846.9	105.4	27542	40 26/-24			
N68	223+00060 (DO NOT ENTR)	70 6:30 1449.7	93.8	(+11800)	35	X-PNTR-LO MULT BINGO FUEL DES QTY LT+1+34 TOUCHDOWN	(35.7)
	66 7:00 1209.8	(434.6)	(+ 9800)	25071			
	66 7:00 1209.8	134.1	21962	32 32/-28			
	64 7:30 969.5	(388.8)	(+ 7600)				
	60 8:00 726.5	(321.0)	(+ 6000)				
		173.3	12704	25 37/-31			
	SEQ CAMR - ON					ENG STOP - PUSH PRO	
						ABORT STAGE - PUSH MODE CONTROL (BOTH) - AUTO DES ENG CMD OVRD - OFF ENG START - PUSH MODE CONTROL (2) - AUTO	

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RECYCLE PARKER VALVE

	H	(H MAX) -H DOT)	DPS	TERR CORR	LPD
	7000	(228.2)	22	(TBD)	
	5000	(186.9)	20		
	4000	(163.2)	20		
	3000	(134.4)			
	2000	(110.9)			
	1000	(85.8)	19		
	500	(104.6)			
	400	(28.0)	15		
	300	(21.0)			
	200	(12.8)	12		
		(12.2)			
		9.7	12		

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PDI THRU

PDI THRU TD+3 MIN

	RESET WATCH	MASTER ARM-ON	(ΔH) -HDOT	(ΔH) H	DPS	SBD P/Y	
- :1:00							
- :30	ENG ARM-DES						
- :07.5	ULLAGE						
- :05	PRO						
+ :00	PDI	(NO IGN) -					
+ :02	START PB - PUSH						
+ :05	DES ENG OVRD -ON	104 1:00	5202.1	26.0	50906	93 2/-2	P64
+0:26	MASTER ARM-OFF THROTTLE UP ✓T/W > 1.6	99 1:30	4907.3	41.6	49880	88	P64 + 15 SEC: NO THROTTLE DN - ABORT
V2 IN69		90 2:30	4293.7	62.5	46707	77	523 ALARM
V57E - (+) LR HIGHER THAN LGC PRO TO PERMIT LR DATA	86 3:00	3974.7	70.2	44713	72 15/-14	V58	LDG ANT-HOVER
	83 3:30	3646.8	77.4	42498	66	RESET	2000 73.2 17
	81 4:00	3309.4	84.5	40071	61 19/-18	WAIT 2 SEC.	(63.8)
	79 4:30	2961.3	92.4	37143	56	PRO	1000 30.6 15
	77 5:00	2601.6	100.4	34544	50 22/-20	PGNS MODE CONT- ATT HOLD	500 13.5 12
	74 5:30	2229.1	107.7	(+17500)		P66	(35.7)
	72 6:00	1846.1	101.4	(+12500)	45	X-PNTR-LO MULT	400 (28.7)
	70 6:30	1449.4	85.4	(+15200)	39 27/-24	BINGO FUEL	400 12.6 12
	- 6:7	1200.5	(429.9)	(+10900)	34	DES TTY LT+1+34	(21.0)
			(328.1)	(+9500)	31 31/-27	TOUCHDOWN	300 11.5 11
			(328.1)	(+6000)			200 (12.2)
			(389.0)	(+7600)			9.6 11
N68	223+00060 (DO NOT ENTR)		(158.3)	17943	28	ENG STOP - PUSH	
			(328.1)	(+ 6000)	24 35/-30	PRO	
	SEQ CAMR - ON		(389.0)	(+7600)		MODE CONTROL (BOTH) - AUTO	
EVAL MAN CONT		65 7:30	962.3	179.7	13209	DES ENG CMD OVRD - OFF	
		62 8:00	722.1	179.7		ENG ARM - OFF	
						ENG START - PUSH	
						MODE CONTROL(2) - AUTO	
						413 + 1	

RECYCLE PARKER VALVE

INITIAL CONDITIONS EDIT

STATE VECTORS (METERS)											
TIME (SEC.)	X	Y	Z	POSITIV	VELAG.	XDOT	YDOT	ZDOT	VAG.	VELOCITY (METERS/SEC.)	
AGC = 372491.4799 -	11434.2	-	11734.7	0.0	-	314312.9	+ 1754527.5	-1269.606	+1040.857	+ 290.767	+1692.542
ENV = 372491.4799 -	1143498.0	-	1225726.4	-	514313.0	+ 1754527.5	-1269.606	+1040.857	+ 290.767	+1692.542	
DIFF=- 0.0000 -	0.0	-	0.0	+ 0.0	0.0	+ 1754527.5	-1269.606	+1040.857	- 0.000	+ 0.000	+ 0.000

SIGHTING (METERS)											
TIME (SEC.)	X	Y	Z	POSITIV	VELAG.	XDOT	YDOT	ZDOT	VAG.	VELOCITY (METERS/SEC.)	
AGC = 372491.4799 -	677652.4	-	1905957.4	-	615355.4	+ 1348627.6	-1507.542	+ 603.306	+ 105.199	+1627.277	
ENV = 372491.4799 -	677652.4	-	1905957.4	-	615355.4	+ 1348627.6	-1507.542	+ 603.306	+ 105.199	+1627.277	
DIFF=- 0.0000 +	0.0	-	0.0	- 0.1	0.1	+ 0.1	- 0.1	+ 0.000	+ 0.000	+ 0.000	+ 0.000

LINE-OF-SIGHT VELOCITY SIGHTING (METERS)											
TIME (SEC.)	X	Y	Z	POSITIV	VELAG.	XDOT	YDOT	ZDOT	VAG.	VELOCITY (METERS/SEC.)	
AGC = 372491.4799 +	445025.0	-	379250.0	-	103542.4	+ 0.211	- 251.035	- 477.551	- 185.568	+ 145.986	
ENV = 372491.4799 +	445025.0	-	379250.0	-	103542.3	+ 0.211	- 251.035	- 477.551	- 185.568	+ 145.986	
DIFF=- 0.0000 +	0.0	-	0.0	- 0.1	0.1	+ 0.1	- 0.1	+ 0.000	+ 0.000	+ 0.000	+ 0.000

RELATIVE AND VERTICAL SIGHTING (METERS)											
TIME (SEC.)	X	Y	Z	POSITIV	VELAG.	XDOT	YDOT	ZDOT	VAG.	VELOCITY (METERS/SEC.)	
ALCNU LINE-OF-SIGHT (X), ORTHOGONAL TO (X) AND (Z)	-6.325	-0.249	-0.156	-	0.0	-	0.0	-	0.000	+ 0.000	
REFSMAT = -0.367	-0.332	+0.940	+0.309	-	0.0	-	0.0	-	0.000	+ 0.000	
T = -0.266	+0.909	+0.309	+0.309	-	0.0	-	0.0	-	0.000	+ 0.000	
S1 = -0.206	+0.993	+0.309	+0.309	-	0.0	-	0.0	-	0.000	+ 0.000	

173 NASA 2021112-141 NEW SIMULATION

MARSBOT 18001320 / 170UE-E-R

COMM

TIME 2 2D + 37249148
ITEM 2D + 37249148
TETCSM 2D + 37249148

+3.72491480 E+ 5

INITIAL CONDITIONS

(START
TIME)

RCVLEM 2D -1.14390826 E+ 68-27 - -1.14390826 E+ 4
RCVLEM +2 2D -1.22670645 E+ 68-27 R -1.22670645 E+ 5
RCVLEM +4 2D -5.14813045 E+ 58-27 LC0N -5.14813045 E+ 5

RRECTLEM 2D -1.14390826 E+ 6B-27 - -1.14390826 E+ 6
RRECTLEM +2 2D -1.22570645 E+ 6B-27 R -1.22570645 E+ 6
RRECTLEM +4 2D -5.14313045 E+ 5B-27 LO -5.14313045 E+ 5

VCVLEM 2D -1.26960674 E+ 1B-05 - -1.26960674 F+ 3
 VCVLEM +2 2D +1.08085729 E+ 1B-05 V +1.08085729 F+ 3
 VCVLEM +4 2D +2.90767590 E+ 0B-05 LCUN +2.90767590 E+ 2

VRECTLEM	2D -1.26960674 E+ 1B-05 -	-1.26960674 E+ 3
VRECTLEM +2	2D +1.08085729 E+ 1B-05 V	+1.08085729 E+ 3
VRECTLEM +4	2D +2.90767580 E+ 08-05 LO	+2.90767580 E+ 2

RCVCSM	2D	-6.77982641	E+	5B-27	-	-6.77982641	F+	5	
RCVCSM	+2	2D	-1.60595727	E+	6B-27	R	-1.60595727	E+	6
RCVCSM	+4	2D	-6.15355379	E+	5B-27	CCON	-6.15355379	E+	5

KRECTCSM 2D -6.77982641 E+ 5B-27 - -6.77982641 E+ 5
 RRRECTCSM +2 2D -1.60595727 E+ 6B-27 R -1.60595727 E+ 6
 KRECTCSM +4 2D -6.15355379 E+ 5B-27 CO -6.15355379 E+ 5

VCVCSM 2D -1.50764261 E+ 1B-05 - -1.50764261 E+ 3
 VCVCSM +2 2D +6.03306059 E+ 0B-05 V +6.03306059 E+ 2
 VCVCSM +4 2D +1.05199093 E+ 0B-05 CCON +1.05199093 E+ 2

VRECTCSM -2D -1.50764261 E+ 1B-05 - - - - -
VRECTCSM +2 2D +6.03306059 E+ 0B-05 V
VRECTCSM +4 2D +1.05199093 E+ 0B-05 C0

END H14PD1

— + + + + + + + + + + +

ELAGWRD3 EFT 10000

FLAGWRD8 OCT 00000
FLAGWRD10 OCT 00000

MASS 20+1.53618127

LEMMASS 1D +1.53618127 E+ 4B-16 M +1.53618127 E+ 4
CSMMASS + 0 1D -1.68393445 E+04B-16 3.712440000+004 LBS

10. The following table shows the number of hours worked by 1000 workers in a certain industry.

DSPTAB +110 OCT - 0

~~CDUZ~~ OCT 0

LGC IMU COMPENSATION PARAMETERS

PB1A5X 907 05503

2.2 CM/SEC

COMMON

RY AGG 173 NASA 2021112-141 NEW SIMULATION

MARSROT 19001320 / MOORE.D

PIIASY	OCT	72274	-2.2 CM/SEC
PIASZ	OCT	01217	0.5 CM/SEC
PIPASCFX	OCT	10142	5//MILLION
PIPASCFY	OCT	67635	-500 /MILLION
PIPASCFZ	OCT	37746	1950 /MILLION
NBDX	OCT	02371	10 MERU
NBDY	OCT	75406	-10 MERU
NBDZ	OCT	12737	44 MERU
ADIFAX	OCT	00606	15 MERU/G
ADIAY	OCT	01212	25 M RU/G
AUTAZ	OCT	13126	220 MERU/G
ADSRAZ	OCT	00606	15 MERU/G
ADSRAY	OCT	13126	220MERU/G
ADSRAZ	OCT	77171	-15 MERU/G
GCOMP +0	2DEC	0	
GCOMP +2	2DEC	0	
GCOMP +4	2DEC	0	
GCOMP SW + 0		10 00000	0.000000000+000

FRA MAURO TERRAIN PROFILE II - DATA BY KRIEGSMAN - ERASABLES BY KLUMPP

ABSC0	1D	72324
ABSC1	1D	-45720.0 B-18
ABSC2	1D	-11276.4 B-18
ABSC3	1D	-4572.00 B-18
ABSC4	1D	-1524.00 B-18
SLOPE0	1D	-.100000 B-06
SLOPE1	1D	-.011905 B-06
SLOPE2	1D	+.055555 B-06
SLOPE3	1D	-.040000 B-06
SLOPE4	1D	-.020000 B-06

ALIGNMENT OPTICAL TELESCOPE DETENT PADLOADS

AOTAZ	OCT	65252
AOTAZ +1	OCT	00000
AOTAZ +2	OCT	12525
AOTAZ +3	OCT	25252
AOTAZ +4	OCT	40000
AOTAZ +5	OCT	52525
AOTEL	OCT	10000
AOTEL +1	OCT	10000
AOTEL +2	OCT	10000
AOTEL +3	OCT	10000
AOTEL +4	OCT	10000
AOTEL +5	OCT	10000

RENDEZVOUS AND LUNAR SURFACE NAVIGATION INITIALIZATION

ENDPOS	OCT	05750
ENVEL	OCT	00763
IRFPOS	OCT	02764
WSURFVEL	OCT	00372
WSHAFT	OCT	17270
WTURN	OCT	17270
WMAX	OCT	0023

COMMON.E

Y AGC 173 NASA 2021112-141 NEW SIMULATION

MARSROT 19001320 / MOORE.D

VMAX	OCT	00001
SE_TVAR	DEC	E-6 B12
TRUNVAR	DEC	E-6 B12
RVARMIN	DEC	66.0 B-12
VVARMIN	DEC	.17445 E-5 B12
RANGEVAR	2DEC*	.11111111 E-4 B12*
RATEVAR	2DEC*	1.877777 E-5 B12*
X789	2DEC	0
X789 +2	2DEC	0
X789 +4	2DEC	0

PADLOADS FOR THE R2 LUNAR GRAVITATIONAL POTENTIAL MODEL

E3J22R2M	OCT	12160
E32C31RM	OCT	03363

MISCELLANEOUS INITIALIZATION

ATIGINC	2DEC	18000
PTIGINC	2DEC	18000
AGSK	OCT	03671
AGSK+1	OCT	21200
TNEWA + 0	10	20000
TNEWA + 1	10	00000

DKTRAP + 0	1D	-3.88888888 E-03B+03	-3.888888889-003
DK_EGAN + 0	1D	1.00000000 E+01B-14	1.000000000+001
DKKAOSN + 0	1D	6.00000000 E+01B-14	6.000000000+001
LMTRAP + 0	1D	-3.88888888 E-03B+03	-3.888888889-003
LMOMEGAN + 0	1D	0.00000000 E+00B-14	0.000000000+000
LMKADSN + 0	1D	6.00000000 E+01B-14	6.000000000+001
DKDB + 0	1D	2.56000000 E+02B-15	2.560000000+002

AUTOPILOT INITIALIZATION

ROLLTIME	OCT	05556
PITTIME	OCT	04431
POSTURKP	2DEC	0
POSTORKU	2DEC	0
POSTORKV	2DEC	0

ABORT INITIALIZATION

J1PARM + 0	2D	1.83969285 E+06B-23	6.035737700+006	FT
K1PARM + 0	2D	-6.12342049 E+05B-23	-3.197416800+005	FT/RAD
J2PARM + 0	2D	1.84096185 E+06B-23	6.039901100+006	FT
K2PARM + 0	10	73337	-6.247310200+005	FT/RAD
K2PARM + 1	10	47011		
THETCRIT + 0	2D	-4.77123789 E-02B+00	-1.717663642+001	DEG
RAMIN + 0	2D	1.79020009 E+06B-24	5.873359885+006	FT
YLTIM + 0	2D	1.51864000 E+04B-24	8.200000000+000	N.M.
ABDDOT + 0	2D	5.94360000 E-02B-07	1.950000000+001	FT/SEC
COSPHET1 + 0	2D	0.00000000 E+00B-02	0.000000000+000	
COSTHET2 + 0	2D	8.66025403 E-01B-02	8.660254037-001	

P12 INITIALIZATION

JARY 4CC 173 NASA 2021112-141 NEW SIMULATION MARSROT 19001320 / MOORE.D

GNAOSQ + 0	1D	1.93333333 E-02B+02	6.960000000+000	DEG/SEC2
GNAOSR + 0	1D	1.55555555 E-03B+02	5.600000000-001	DEG/SEC2
HIASCENT + 0	1D	4.94415683 E+03B-16	1.090000000+004	LBS

LANDING SITE DEPENDENT DATA

TLAND	2D	+ 37347617	TLAND	+ 3.73476171 E+ 5
RLS	2D	+ 1.65305549 E+ 6B-27		+ 1.65305549 E+ 6
RLS +2	2D	- 5.20705500 E+ 5B-27 R		- 5.20705500 E+ 5
RLS +4	2D	- 1.11122499 E+ 5B-27 LS		- 1.11122499 E+ 5
TEPHEM	1D	+ 3	T (EPOCH	+ 3.88197999 E+ 6
TEPHEM +1	2D	+ 32891632	0 SECS U.T.)	
AXO	2DEC*	3.962911018 E-5*		
-AYO	2DEC*	- 5.58111439 E-6*		

LUNAR LIBRATION

504LM	2D	- .000398466	-	- 3.98466794 E- 4
504LM +2	2D	- .000002989	L	- 2.98927218 E- 6
504LM +4	2D	- .000379924	LIBRATION	- 3.79924699 E- 4

GEOMETRY OF INERTIAL REFERENCE AND PLATFORM COORDINATES

REFSMMAT	2D	- .477840406	-	- 9.55680812 E- 1
REFSMMAT +2	2D	- .124686983	REFSMMAT	- 2.49373967 E- 1
REFSMMAT +4	2D	- .078241307	0	- 1.56482615 E- 1
REFSMMAT +6	2D	- .033691028	-	- 6.73820565 E- 2
INGREFSMMAT +10	2D	- .166068687	REFSMMAT	- 3.32137374 E- 1
INGREFSMMAT +12	2D	+ .470410571	3	+ 9.40821142 E- 1
INGREFSMMAT +14	2D	- .143295013	-	- 2.86590026 E- 1
INGREFSMMAT +16	2D	+ .454834417	REFSMMAT	+ 9.09668834 E- 1
INGREFSMMAT +20	2D	+ .150306992	6	+ 3.00613985 E- 1

IGNITION ALGORITHM PARAMETERS

VIGN	2D	+ 1.69022706 E+ 1B-10 VIGG	-	+ 1.69022706 E+ 3
RIGNX	2D	- 4.06516453 E+ 4B-24 RIGXG	-	- 4.06516453 E+ 4
RIGNZ	2D	- 4.40457183 E+ 5B-24 RIGZG	-	- 4.40457183 E+ 5
KIGNX/B4	2D	- .020587499	KX	+ 3.304999999 E- 1
KIGNY/B8	2D	- .126199802	KY	+ 1.92565616 E- 6
KIGNV/B4	2D	- .167033740	KV	+ 4.38000000 E+ 2

BRAKING AND APPROACH PHASE TARGETS

TCGIBRAK	1D	+ 9.00000000 E+ 4B-17 TCGIBRAK	-	+ 9.00000000 E+ 2
TCGF BRAK	1D	+ 3.00000000 E+ 3B-17 TCGF BRAK	-	+ 3.00000000 E+ 1
ENDBRAK	1D	+ 6.20000000 E+ 3B-17 TENDBRAK	-	+ 6.20000000 E+ 1
GAINBRAK	2D	+ 9.99999998 E- 1B+00 GAINBRAK	-	+ 9.9999998 E- 1
RBREFGX	2D	- 1.08571283 E+ 3B-24 -	-	- 1.08571283 E+ 3
RBREFGZ	2D	- 4.17750039 E+ 3B-24 OFG	-	- 4.17750039 E+ 3

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MARSROT 19001320 / MOORE

VBRFGX	2D -5.69680495 E- 1B-10 -	-5.69680495 E+ 1
VBRFGZ	2D -3.00954002 E- 1B-10 OFG	-3.00954002 E+ 1
ABRFGX	2D -1.37236046 E- 5B+04 -	-1.37236046 E- 1
ABRFGZ	2D -2.90020170 E- 4B+04 OFG	-2.90020170 E+ 0
VBRFG*	2D -6.77146506 E- 1B-10 18/8 VOFZG	-6.77146506 E+ 1
ABRFG*	2D -1.74012102 E- 3B+04 6 AOFZG	-1.74012102 E+ 1
JBRFG*	2D -3.59485996 E- 8B+18 8 JOFZG	-3.59485996 E- 2
DELTIFAP	1D -9.00000000 E+ 3B-17 TGOAUG	+9.00000000 E+ 1
TCGIAPPR	1D +2.00000000 E+ 4B-17 TCGIAPPR	+2.00000000 E+ 2
TCGFAPPR	1D +6.00000000 E+ 2B-17 TCGFAPPR	+6.00000000 E+ 0
TENDAPPR	1D +1.20000000 E+ 3B-17 TENDAPPR	+1.20000000 E+ 1
GAINAPPR	2D +0.00000000 E+ 0B+00 GAINAPPR	+0.00000000 E+ 0
RAPEGX	2D +2.52763019 E+ 1B-24 -	+2.52763019 E+ 1
RAPEGZ	2D -6.14492039 E+ 0B-24 1FG	-6.14492039 E+ 0
VAPEGX	2D -9.72311999 E- 4B-10 -	-9.72311999 E- 2
VAPEGZ	2D +9.51981839 E- 4B-10 1FG	+9.51981839 E- 2
AAPEGX	2D +9.13851359 E- 6B+04 -	+9.13851359 E- 2
AAPEGZ	2D -1.22422919 E- 5B+04 1FG	-1.22422919 E- 1
VAPFG*	2D +2.14195913 E- 3B-10 18/8 V1FZG	+2.14195913 E- 1
AAPFG*	2D -7.34537518 E- 5B+04 6 A1FZG	-7.34537518 E- 1
JAPFG*	2D +9.19164631 E- 8B+18 8 J1FZG	+9.19164631 E- 2
		J1FZG +1.14895578 E- 2

P66 INITIALIZATION

RODSCALE + 0	1D 3.04800000 E-03B+07	1.000000000+000	FT/SEC
TAUROD + 0	2D 1.50000000 E+02B-09	1.500000000+000	SEC
LAG/TAU + 0	2D 4.13333000 E-01B+00	4.133330000-001	
MINFORCE + 0	2D 4.35925718 E-01B-12	9.800000000+002	LBF
MAXFORCE + 0	2D 2.80237961 E+00B-12	6.300000000+003	LBF
TAUHZ	1D 0.5 E+2 B-11		
QHZ	1D 0.4		
AHZLIM	1D .591 E-4 B+4		
HZROFF	2D +15.24 B-24		
TOOFFW	1D 0.0003		
2LATE466	2D 1.5 E+02 B-28		
ZOOMTIME + 0	1D 2.60000000 E+03B-14	2.600000000+001	SEC
LEADTIME	1D -2.19999999 E+ 2B-17 LEADTIME	+2.19999999 E+ 0	
LOWCRIT	1D + 2136 FLO	+2.66226063 F+ 4	
HIGHCRIT	1D + 2361 FHI	+2.94249859 E+ 4	
ELBIAS	DEC .001		
AZRIAS	DEC .001		
VELBIAS	ZDEC .02286 B-6		

PADLOADS FOR MOUN 69

DLAND + 0	2D 0.00000000 E+00B-24	0.000000000+000	FT
DLAND + Z	2D -0.00000000 E+00B-24	0.000000000+000	FT
DLAND - Z	2D 0.00000000 E+00B-24	0.000000000+000	FT

MARY ACC

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COMMON
MARSROT 19001320 / MOORE.D

LANDING RADAR PADLOADS

RPCRTIME + 0	1D	6.20000000 E+03B-17	6.200000000+001	SEC
RPCRTQSW + 0	1D	-1.00000000 E+00B-01	-1.000000000+000	
DELQFIX + 0	2D	15.2400000 E+01B-24	5.000000000+002	FT
LRWH + 0	1D	3.50000000 E-01B+00	3.500000000-001	
LRWHL	DFC	0.25		
LRHMAX + 0	1D	1.52400000 E+04B-14	5.000000000+004	FT
LRVMAX + 0	1D	6.09600000 E+00B-07	2.000000000+003	FT/SEC
LRVF + 0	1D	6.09600000 E-01B-07	2.000000000+002	FT/SEC
LRWVZ + 0	1D	3.00000000 E+01B+00	3.000000000-001	
LRWVY + 0	1D	3.00000000 E-01B+00	3.000000000-001	
LRWVX + 0	1D	3.00000000 E-01B+00	3.000000000-001	
LRWVFZ + 0	1D	2.00000000 E-01B+00	2.000000000-001	
LRWVFY + 0	1D	2.00000000 E-01B+00	2.000000000-001	
LRWVFX + 0	1D	2.00000000 E-01B+00	2.000000000-001	
LRWVFF + 0	1D	1.00000000 E-01B+00	1.000000000-001	

END HBPADS

+++++ INSERT LSR002 +++++

TGCSM	2DEC	0	
NUVCSM	2DEC	0	
NUVCSM +2	2DEC	0	
NUVCSM +4	2DEC	0	
XKEPCSM	2DEC	0	
ELTACSM	2DEC	0	
DELTACSM +2	2DEC	0	
DELTACSM +4	2DEC	0	
TCLEM	2DEC	0	
NUVLEM	2DEC	0	
NUVLEM +2	2DEC	0	
NUVLEM +4	2DEC	0	
XKEPLEM	2DEC	0	
DELTALEM	2DEC	0	
DELTALEM +2	2DEC	0	
DELTALEM +4	2DEC	0	
END	LSR002		

READACCS	ENVSUM	2		
PREREAD	ENVSUM	2		
RODCOMP	CFDUMP		1.1	
SLAPI	ENVMODE	0		
SLAPI	PIRIT DUTYCYLE			
SLAPI	CFDUMP		1.3	
SLAPI	DUMP	32	END-E7	1.4
R60INIT +2	DUMP	32	END-E7	1.5
READACCS +1	CFDUMP			1.1
ENDLL JCB	DUMP	A	E7,1777	1.2

CUX = 25032

CARDS READ BY ASTRONAUT INPUT-PROG 5 :

```

A PRINT OFF
A WAIT 12 V 37 E 00 F
A VERIFY WITHIN 20 MODE=00 THEN WAIT 2 V 48 E
A IF V 01 N 46 THEN WAIT 1 V 21 F 21112 F WAIT 3 PROCEED
A IF V 06 N 47 THEN WAIT 1 V 21 F +33380 E WAIT 1 PROCEED
A IF V 06 N 43 THEN WAIT 1 V 34 E WAIT 2
A W I LEMRADAR OFF
A V 37 E +3 F WAIT 2
A DISPLAY INERTIAL DATA ON . . . WAIT 1
A IF V 06 N 61 THEN WAIT 2 PROCEED WAIT 2
A IF V 50 N 25 R1 00014 THEN WAIT 2 ENTER WAIT 2
A IF V 50 N 18 THEN GUIDEMODE PRIMARY SCSCMODE AUTO WAIT 2 PROCEED
A IF V 50 N 18 THEN WAIT 2 ENTER
A IF V 99 N 62 THEN DCENGARM ON LEMTHRUT 10 WAIT 2 PROCEED
A WAIT 100 V 57 E WAIT 10 V 34 E
A WAIT 30 V 57 E WAIT 25 V 34 E WAIT 1 LEMRADAR OFF WAIT 50 V 57 E W
ATRYIT VERIFY WITHIN 120 NOABORT V 06 N 68 R3 UNEQ +99999 THEN GO TO LRON
A OTHERWISE V 34 E WAIT 10 V 57 E GO 10 TRYIT END
ALRON WAIT 5 PROCEED
A WAIT 15
A WAIT 25 V 57 E WAIT 6 PROCEED WAIT 10 PROCEED
ALROFF IF V 06 N 64 THEN W 2 PROCEED W 2
A LRHCRATE 99999 LEMHCTRL R -5 W .03 LRHCRATE 99999 LEMHCTRL R 0 W .5
A W 15
A LRHCRATE 99999 LEMHCTRL Q -5 W .03 LRHCRATE 99999 LEMHCTRL Q 0 W .5
A LRHCRATE 99999 LEMHCTRL Q -5 W .03 LRHCRATE 99999 LEMHCTRL Q 0 W .5
ACONT S 72 H
A W 5 V 37 E 68 F
A IF V 06 N 43 THEN WAIT 2 PROCEED WAIT 2
A WAIT 5 V 69 E

```

END OF ASTRONAUT INPUT CARDS.

```

LABEL TRYIT IS FUN # 2921BL .
LABEL LRUN IS FUN # 292200 .
LABEL LROFF IS FUN # 292220 .
LABEL CONT IS FUN # 292295 .

```

INPUTPROG INITIALIZATION CARDS:

	PRNGTRL	FLIGHT #	ALTT
000000000	00	5.04000000000	02 -1.40496414599 03 -3.058

VERSE ROUTINE IS USING DATA FOR THE PERIOD FROM JULY 1, 1970 THROUGH JULY 1971

EVENT LIST (AGC TIME)

TIG = 372791.5
THROTTLE-UP = 372817.5
THROTTLE-DOWN = 373195.
HIGATE = 373298
REDES (-AZ) = 373304.5
REDES (+EL) = 373321.0
REDES (+EL) = 373322.4
LOGATE = 373450.7
TOUCHDOWN = 373496

ADDENDUM to LUMINARY Memo #160

Summary:

A further study of the Descent Trajectory Monitor Package of Luminary Memo #160 led to an all-digital simulation at MIT/DL using a smooth moon in the environment and the LGC; i.e. no terrain model used. This data was compared with the LM Timeline chart of 3/16/70 and the result was found to be that the LM Timeline data corresponds much better with the MIT/DL data without terrain than MIT/DL data with terrain. The H and VI plots are such that the general curve shape is followed nearly coincidental. The HDOT curve follows a general shape pattern nearly parallel with a constant difference of about 8 fps until throttle-recovery, at which time the curves for LM Timeline Data and MIT/DL continue nearly coincidental to the end of the run sample period.

Suggestion:

Communication of Apollo 14 (Luminary 1D) trajectory parameters, using the latest known Fra Mauro terrain parameters, to all simulators so that the Descent Trajectory Monitor chart for a final Apollo 14 LM Timeline Book can be verified from independent observers at MSC, KSC, and MIT/DL.