

(2)

M. HAMILTON

T/DL PROGRAM CHANGE ROUTING SLIP

PCR/PCN # SL003

ANOMALY # _____

ADR # _____

- COLOSSUS 3
- COLOSSUS _____
- SKYLARK _____
- LUMINARY 1E
- LUMINARY _____

- MIT Approved PCN
- MIT Approved ADR
- NASA Approved PCR
- NASA Approved PCN
- NASA Approved Software Anomaly
- MIT Approved Software Anomaly

A. Coding

- Begin coding immediately

Carl Beals

ACTION: _____

Program Supervisor's Approval: Margaret Hamilton

- Do not code until new GSOP material has been approved by the MIT Mission Design Review Board (MDRB) and distributed.

B. GSOP Preparation

- Prepare GSOP revisions for MDRB consideration

J. KLAWENIK / E. OLSSON

ACTION: _____

- Technical Committee Meeting not required.

- Technical Committee Meeting(s) held on _____
Attendees: _____

C. KSC Testing and Checkout

- Review for possible impact on KSC testing and checkout

ACTION: _____

D. Other Programs Affected

- Review for corresponding changes in _____

ACTION: _____

Special Instructions

Project Manager Stephen L. Coffey
 Date 1/28/71

APOLLO SPACECRAFT SOFTWARE CONFIGURATION CONTROL BOARD
PROGRAM CHANGE REQUEST

NUMBER (Completed by FSB)

SL003

1.0 COMPLETED BY ORIGINATOR

1.1 ORIGINATOR R. O. Nobles	DATE 2/25/70	1.2 ORGANIZATION MPAD	APPROVAL RO. Nobles	DATE 2/25/70
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1.3 EFFECTIVITY AAP	1.4 TITLE OF CHANGE Improved Short Burn Logic
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1.5 REASON(S) FOR CHANGE
Make the CSM short burn logic less sensitive to the dispersion in the performance of the SPS engine.

1.6 DESCRIPTION OF CHANGE
See attached flow chart

2.0 SOFTWARE CONTROL BOARD OR FLIGHT SOFTWARE BRANCH
DECISION FOR VISIBILITY IMPACT ESTIMATE BY MIT

2.1 <input type="checkbox"/> APPROVED <input type="checkbox"/> DISAPPROVED	2.2 REMARKS:
2.3 SOFTWARE CONTROL BOARD OR FLIGHT SOFTWARE BRANCH SIGN OFF	
DATE	

3.0 MIT VISIBILITY IMPACT EVALUATION:

3.1 SCHEDULE IMPACT	3.2 IMPACT OF PROVIDING DETAILED EVALUATION
3.3 STORAGE IMPACT	3.4 REMARKS:
3.5 MIT COORDINATOR	
DATE	

4.0 SOFTWARE CONTROL BOARD ACTION

4.1 <input checked="" type="checkbox"/> IMPLEMENT AND PROVIDE DETAILED CHANGE EVAL. <input type="checkbox"/> PROVIDE DETAILED CHANGE EVALUATION <input type="checkbox"/> DIS-APPROVED	4.2 REMARKS
4.3 SOFTWARE CONTROL BOARD SIGN OFF <i>[Signature]</i>	
DATE 3/4/70	

5.0 MIT DETAILED PROGRAM CHANGE EVALUATION

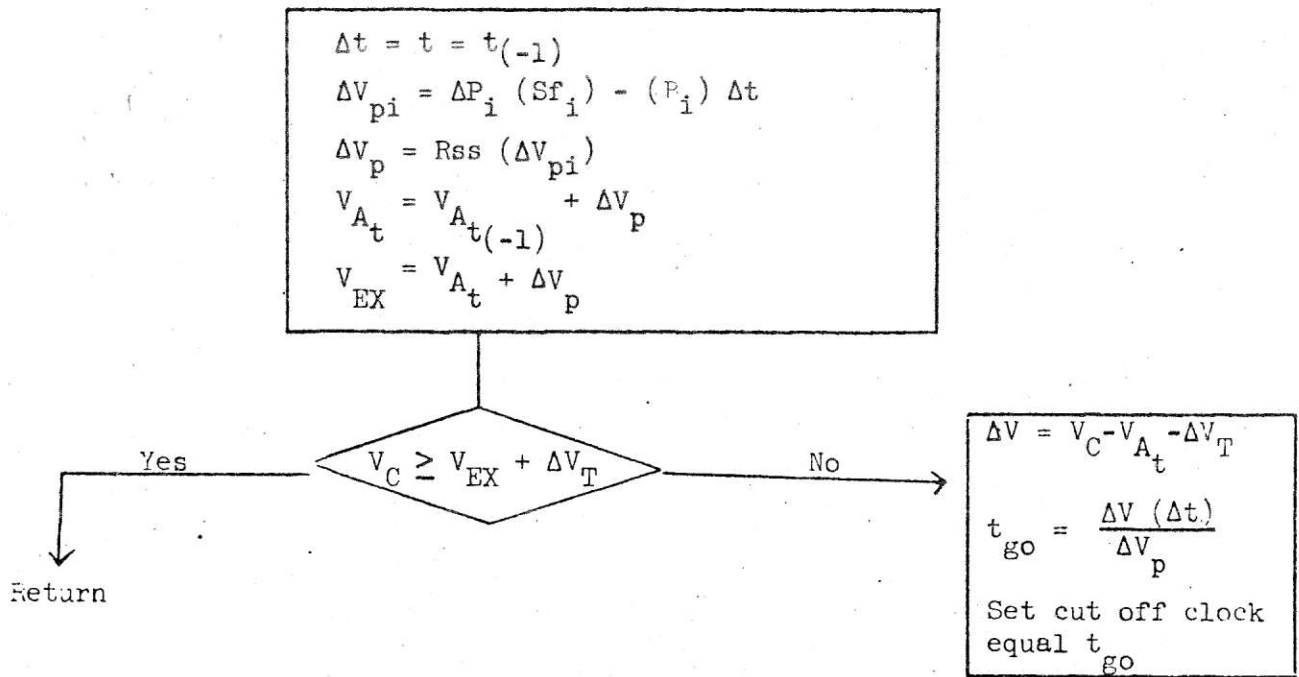
5.1 MIT COORDINATOR	5.2 MIT EVALUATION
DATE	

6.0 SOFTWARE CONTROL BOARD DECISION ON MIT
DETAILED PROGRAM CHANGE EVALUATION

6.1 <input type="checkbox"/> START OR CONTINUE IMPLEMENTATION <input type="checkbox"/> DISAPPROVED OR STOP IMPLEMENTATION	6.2 REMARKS:
6.3 SOFTWARE CONTROL BOARD SIGN OFF	
DATE	

Figure 1

Proposed SPS Short Burn Logic



where:

- $\Delta P_i =$ X, Y, Z accelerometer pulses
- $P_i, Sf_i =$ X, Y, Z accelerometer bias and scale factor
- $V_C =$ targeted velocity, Rss (external ΔV targets)
- $V_{A_t} =$ thrust velocity magnitude
- $V_{EX} =$ thrust velocity extrapolated ahead
- $\Delta t =$ computer computational cycle (0.1 sec)
- $\Delta V_T =$ expected tail off velocity (preburn calculation)

$$\Delta V_T = \frac{F}{M - \dot{M} T_{go}} (\Delta T)$$

- $\Delta T =$ single bore tail off constant
- $M =$ vehicle mass at burn initiation
- $\dot{M} =$ SPS mass flow rate
- $F =$ SPS thrust
- $T_{go} =$ SPS burn duration (preburn calculation)

$$T_{go} = \frac{V_C (M)}{F + \frac{V_C (\dot{M})}{2}}$$