

Date 22 August 1969

*check with
mission support plan*

*David
Benson
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S. ...*

MIT/IL SOFTWARE DEVELOPMENT PLAN
FOR
LUMINARY 1B LGC PROGRAM
(FINAL ISSUE)

George W. Cherry *8/26/69*
Luminary Project Manager's Signature Date

Fred H. Martin *8/26/69*
Director Mission Development Approval Date

This plan consists of 45 pages.

PREFACE

The Luminary 1B LGC Program is being prepared to support a manned Lunar or Earth Orbital LM flight of a Block II G&N System.

I
I

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SECTION I

PROGRAM DEVELOPMENT STATUS

Item No.	Description	Note	Predicted Date	Original Target	Actual Date	Required Dates	Scheduled Rev. Date	Resp.
1	Level 2/3 test plan Preliminary				6/18/69			
2	Level 2/3 testing Final		7/29/69	7/29/69	8/5/69			
	Start		6/5/69	6/5/69	6/5/69			
	Completion		8/12/69	7/15/69	8/12/69			
3	Level 4 test plan							
	PRELIMINARY		7/8/69	7/8/69	7/8/69			
	FINAL		7/11/69	7/11/69	7/11/69			
4	Level 4 testing							
	Start		7/16/69	7/16/69	7/14/69			
	Completion		8/12/69	8/12/69	8/12/69			
5	GSOP Publication dates							
	Section 1 (Change pages)							
	2 (Change pages)		8/28/69	8/1/69				
	3 (Change pages)		8/28/69	8/12/69				
	4 (Complete publication)		8/19/69	8/1/69	8/22/69			
	5 (Complete publication)		8/12/69	8/1/69	8/15/69			
6	MIT Documentation							
	Luminary Flowchart Completion		10/15/69	10/15/69				

Released on Rev 116
to Rope Manufacture

No changes necessary

SECTION II

GUIDANCE SOFTWARE OPERATIONAL PLAN STATUS

PCR	DESCRIPTION	Originator/ Date	MIT IMPACT	NASA/SCB DIRECTION	PCS	Approved Schedule Impact and Remarks
773.2	Fix constants for Planetary Inertial Subroutine.	Reber 5/13/69	No Slip	Implement and provide detailed evaluation.		
776.2	Improved R2 model timing.	Engel 5/15/69	No Slip	Implement and provide detailed evaluation.		
779	Leave Track Enable set when R29 is terminated.	Volante 5/20/69	No Slip	Implement and provide detailed evaluation.		
780	Provide pure RR Range, Range rate and time lag during P20, P22, and P25.	Cherry 5/19/69	Storage - 20 words	Implement and provide detailed evaluation.		
781.2	PIPA Bias Compensation scale.	Edmonds 5/19/69	Storage - 5 words	Implement and provide detailed evaluation.		
791.2	Do not allow a PROCEED Response to V21, V22, V23.	Copps 5/23/69	No Slip	Implement and provide detailed evaluation.		
797.2	Cause Key Release to Blank DSKY if no display waiting.	Copps 5/23/69		Disapproved		Not to be implemented because of difficult caused on KSC testin
798.2	Reset GLOKFAIL in Roo.	Copps 5/23/69	No Slip	Implement and provide detailed evaluation.		

PCR	DESCRIPTION	Originator/ Date	MIT IMPACT	NASA/SCB DIRECTION	PCS	Approved Schedule Impact and Remarks
801.2	Make BAILOUT a- larms start with 3XXXX and POODOO alarms start with 2XXXX.	Copps 5/23/69	No Slip	Implement and pro- vide detailed evalu- ation.		
802.2	Save alarm data af- ter "Error Reset".	Copps 5/23/69	No Slip	Implement and pro- vide detailed evalu- ation.		
805	Don't allow V66 on the surface.	Kernan 5/28/69	Storage - 4 words	Implement and pro- vide detailed evalu- ation.		
806.2	Allow N07 to address output channels.	Kernan 5/28/69		Disapproved		Not being incorporated because of schedule impact.
807.2	Add present time to P21.	Kernan 5/28/69	No Slip	Implement and pro- vide detailed evalu- ation.		
812.2	Resetting and Setting of the External ΔV flag.	Copps 6/2/69	No Slip	Implement and pro- vide detailed evalu- ation.		
814 (Rev)	Reduce keystrokes required to check and approve LR data.	Hamilton 6/2/69	No Slip	Implement and pro- vide detailed evalu- ation.		
816	Modify R03 to permit astronaut setting of 1° deadband.	Cherry 6/4/69	No Slip	Implement and pro- vide detailed evalu- ation.		
817	Eliminate Undesir- able LR position a- larms from R12.	Cherry 6/4/69	No Slip	Implement and pro- vide detailed evalu- ation.		Fixes Anomaly LNY 80

PCR	DESCRIPTION	Originator/ Date	MIT IMPACT	NASA/SCB DIRECTION	PCS	Approved Schedule Impact and Remarks
318	Permit rejection of Individual Measurement incorporations in P20.	Cherry 6/4/69	No Slip	Implement and provide detailed evaluation.		
320	Eliminate lighting of ALT light when low scale discrete is absent.	Cherry 6/4/69	No Slip	Implement and provide detailed evaluation.		
323	Delete P31 from Luminary 1B.	Greene 6/11/69	Storage - Saves 60 Words	Implement and provide detailed evaluation.		
325.2	Display option 3 in P52/P54.	Copps 6/13/69	No Slip	Implement and provide detailed evaluation.		
326.2	Reverse P76 Display.	Copps 6/13/69	No Slip	Implement and provide detailed evaluation.		
308.2	Limit range display in V83.	Kernan 5/28/69		Disapproved		
377	Affixed DUMPCNT	FSB 5/13/69	No Slip	Implement and provide detailed evaluation.		
379	Variable Insertion Computation	FCD/FDB 6/6/69	Storage - 80 words	Implement and provide detailed eval.		
327	Add ZDOTD to ascent descent downlist	Schulenburg 6/18/69	No Slip	Implement and provide detailed eval.		
331.2	Lambert Overflow protection	Robertson 6/19/69	No Slip	Implement and provide detailed eval.		
330	Supplementary ASTEER modification	Robertson 6/19/69	Storage - Save 4 words	Implement and provide detailed eval.		

PCR	DESCRIPTION	Originator/ Date	MIT IMPACT	NASA/SCB DIRECTION	PCS	Approved Schedule Impact and Remarks
832.2	Define Actual Restrictions on Running R05.	Cherry 7/7/69	No Slip	Implement and provide detailed eval.		
838	Prevent RCS Jet on Lunar Surface	Cherry 7/14/69	Storage - ≈ 0	Implement and provide detailed eval.		Fixes Anomaly LNY 8:
839	R12 and LR Repositioning Routine Improvements	Cherry 7/11/69	Storage - Saves \approx 20 words	Implement and provide detailed eval.		
840	Reduce Oscillation in P64/65	Cherry 7/14/69	No Slip	Implement and provide detailed eval.		Fixes Anomaly LNY 79
841	PGNCS Derived Vehicle Attitude Rates on FDAI Error Needle	Cherry 7/14/69	Storage - 13 words	Implement and provide detailed eval.		
842	Mod. of Criteria used to determine DAP phase plan parabola intercept	R.Goss 7/8/69	Storage - 1 word	Oral Approval 8/1/69		Fixes Anomaly LNY78
844	Deletion of P38/78 & P39/79	G.Cherry 7/16/69	No Slip	Implement and provide detailed eval.		Approved as part of PCR 279.
845	Do not turn on R29 during P70/71	G.Cherry 7/16/69	Storage - Save 2 words	Implement and provide detailed eval.		Approved as part of PCR 279.
280.2	Constants for Planetary Inertial Orientation (1970)	MPAD 6/25/69		Disapproved.		Superceded by PCR 773.2.

PCR	DESCRIPTION	Originator/ Date	MIT IMPACT	NASA/SCB DIRECTION	PCS	Approved Schedule Impact and Remarks
281	Constants for LOSSEM	MPAD 6/25/69		Disapproved.		Superceded by PCR 773.2.
846	More accurate DELTAT Tailoff for P70	G.Cherry 7/22/69		Pending		MSC to provide data
847	Eliminate Possible Lockout of Pitch- over from P12, P70, P71	G. Cherry 7/22/69	Save 6 words	Oral approval 7/22/69		
848	Prevent RR ECDU's from stealing LGC Memory Cycles	G.Cherry 7/22/69	Storage - \approx 8 words	Implement and pro- vide detailed eval.		
284	VGTIG's on C/A Downlist	C.D. Sykes 7/31/69	No Slip	Implement and pro- vide detailed eval.		
853	Restrict V35 to P00	B. McCoy 7/31/69	Storage - Saves 10 words	Oral Approval 8/1/69		Approved for Colossu on PCN 748.
854	Provide a Flexible Method for Crew to Modify RLS	G. Cherry 8/4/69		Approved by C. Kraft 7/31/69		
855	Begin Reading LR Velocity as soon as Velocity Data Good Appears	G. Cherry 8/4/69		Oral Approval 8/1/69		

PCR	DESCRIPTION	Originator/ Date	MIT IMPACT	NASA/SCB DIRECTION	PCS	Approved Schedule Impact and Remarks
856	Change 1502/1206 from POODOO's to BAILOUTS	B. McCoy 8/5/69		Oral Approval 8/6/69		

PCR	DESCRIPTION	Section 1			Section 2			Section 3			Section 4			Section 5		
		Req	Pred	Act	Req	Pred	Act	Req	Pred	Act	Req	Pred	Act	Req	Pred	Act
773.2	Fix constant for Planetary Inertial Subroutine.															8/5
776.2	Improved R2 model timing.													8/1	8/1	8/5
779	Leave Track Enable set when R29 is terminated.										7/7	8/1	8/12			
780	Provide pure RR Range, Range rate and time lag during P20, P22, and P25.										7/7	8/1	8/12	8/1	8/1	8/5
791.2	Do not allow a PROCEED Response to V21, V22, V23.										7/18	8/1	8/12			
798.2	Reset GLOKFAIL in Roo.										7/18	8/1	8/12			
801.2	Make BAILOUT alarms start with 3XXXX and POODOO alarms start with 2XXXX.										7/18	8/1	8/12			
802.2	Save alarm data after "Error Reset".					8/22	8/21				7/18	8/1	8/12			
805	Don't allow V66 on the surface.										7/18	8/1	8/12			

PCR	DESCRIPTION	Section 1			Section 2			Section 3			Section 4			Section 5		
		Req	Pred	Act	Req	Pred	Act	Req	Pred	Act	Req	Pred	Act	Req	Pred	Act
807.2	Add present time to P21.										7/18	8/1	8/12			
812.2	Resetting and Setting of the External ΔV flag.										7/18	8/1	8/12			
814 (Rev)	Reduce keystrokes required to check and approve LR data.										7/7	8/1	8/12			
816	Modify R03 to permit astronaut setting of 1 dead-band.					8/22	8/21		8/15	8/21	7/18	8/1	8/12			
817	Eliminate Undesirable LR position alarms from R12.										7/7	8/1	8/12	8/1	8/1	8/5
818	Permit rejection of Individual Measurement incorporations in P20.										7/7	8/1	8/12	8/1	8/1	8/5
820	Eliminate lighting of ALT light when low scale discrete is absent.										7/7	8/1	8/12	8/1	8/1	8/5
823	Delete P31 from Luminary 1B.					8/22	8/21				7/18	8/1	8/12	8/1	8/1	8/5
825.2	Display option 3 in P52/P54.										8/1	8/1	8/12			
826.2	Reverse P76 Display.										8/1	8/1	8/12			
277	Affixed DUMPCNT					8/22	8/21									
279	Variable insertion computation					8/22	8/21							8/1	8/1	8/5

PCR	DESCRIPTION	Section 1			Section 2			Section 3			Section 4			Section 5		
		Req	Pred	Act	Req	Pred	Act	Req	Pred	Act	Req	Pred	Act	Req	Pred	Act
827	Add ZDOTD to ascent/ Descent downlist				8/22	8/21										
831.2	Lambert Overflow Protection													8/1	8/1	8/5
830	Supplementary ASTEER modification													8/1	8/1	8/5
832.2	Remove Restriction of Running R05 only in POO										8/1	8/1	8/12			
838	Prevent RCS JET firings on Lunar Surface										8/1	8/1	8/12			
839	R12 and LR reposition routine Improvements				8/22	8/21					8/1	8/1	8/12	8/1	8/1	8/5
840	Reduce Attitude Oscilla- tions in P64/65										8/1	8/1	8/12			
841	PGNCS Derived Vehicle Attitude Rate onFDAI Error Needles				8/22	8/21		8/15	8/21	8/1	8/1	8/12				
844	Deletion of P38/78 and P39/79				8/22	8/21					8/1	8/1	8/12	8/1	8/1	8/5
845	Do not turn on R29 during P70/71										8/1	8/1	8/12			
847	Eliminate Possible Lock- out of Pitch-over from P12, P70/71										8/1	8/1	8/12	8/1	8/1	8/5

LUMINARY IB GSOP CHANGES AVAILABLE FOR REVIEW

DATE 8/22/69

PCR	DESCRIPTION	SECTION 1			SECTION 2			SECTION 3			SECTION 4			SECTION 5		
		Req	Pred	Act	Req	Pred	Act	Req	Pred	Act	Req	Pred	Act	Req	Pred	Act
842	Mod. of Criteria used to determine DAP phase plane parabola intercepts.							8/15	8/21							
848	Prevent RR ECDU's from stealing LGC memory cycles.	NO CHANGE TO GSOP														
284	VGTIG's on C/A Downlist.				8/22	8/21										
853	Restrict V35 to POO.											9/5				
854	Provide a Flexible Method for Crew to Modify RLS.											9/5				
855	Begin Reading LR Velocity Data Good Appears.						9/5					9/5			9/5	
856	Change 1502/1206 from POODOO's to BAILOUTS											9/5				

SECTION III

LEVEL 2 AND 3 TESTING

LUMINARY 1B LEVEL 3 TESTS

DATE 8/22/69

ANOM/ PCR/ ACB	TEST DESCRIPTION	RESPONSIBLE ENGINEER
	<u>LANDING</u>	
LN Y 58	Attempt V78 during Hybrid Landing Simulation. Observe operator error light.	D. Moore
LN Y 64	Cause engine fail during 10% and observe no throttle up at ZOOMTIME.	P. Adler
LN Y 72	Hybrid Simulation, place LR in POS 1 after HIGATE note 523 alarm, V32 response, note one 511 alarm instead of one every two secs.	D. Moore
LN Y 73	Do recycle on N61 in Hybrid Test - observe reappearance of N61.	P. Adler
LN Y 76	Hybrid Landing Simulation; notice 1 sec. flashing frequency of N60.	D. Moore
LN Y 88	Force restart during position job and note R12 is still running and LR repositions to desired position.	D. Moore
PCR 814 (REV)	Key V57E, obtain V06N68 flash; when acceptable delta H (R3) appears, proceed, obtain V50N18; key proceed to return to previous (P63, P64, P65, P66 & P67) displays.	D. Moore
PCR 817	Hybrid Landing Simulation has capability of making the 511, 522 and 523 alarm situation (possible in 1A) in 1B, 522 not noticed. Place LR in POS 2 (or repositioning) before HIGATE and reposition after HIGATE. When 523 occurs, key V34E; observe no 511 alarm and that R12 is turned off at V34E to 523 alarm.	D. Moore
PCR 820	Hybrid Landing Simulation; remove LO scale discrete; note no ALT light.	D. Moore
PCR 838	<ol style="list-style-type: none"> 1. Hybrid Simulation; at P68 check MINIMP flag to see bit setting. 2. At TIG in various programs check MINIMP flag to see bit reset. 3. Hybrid Simulation; notice attitude error needles zeroing when proceed to N60. 	D. Moore

LUMINARY IB LEVEL 3 TESTS

DATE 8/22/69

ANOM/ PCR/ ACB	TEST DESCRIPTION	RESPONSIBLE ENGINEER
PCR 839	Hybrid Simulation; note LR using POS 2 transformation even if POS 2 discrete has not appeared after HIGATE; from Hybrid edit, checking of LPOS2FLG (FLGWRD 11).	D. Moore
PCR 840	In P63 observe deadband not equal 0.3° deadband. Check bits 4 & 5 (DAPBOOLS) for 0.3° and 1° setting. Observe 0.3° deadband in P64.	D. Moore
PCR 848	Visual Inspection	D. Eyles
PCR 854	Visual Inspection	D. Eyles
PCR 855	Observe that Velocity Data Bad Light on DSKY comes on when average G is on.	B. McCoy
<u>ABORTS and ASCENTS</u>		
LNY 62	Test consists of obtaining basic trace of a portion of a P70 simulation which explicitly shows the DAP deadband being set to one degree (Digital Simulation).	W. Bernikowich
PCR 779	P12 run with POO selected at insertion. Verify from simulator output that track enable is still set.	P. Volante
PCR 279	<p>This test consists of four (4) separate tests, the PCR changed P70 and P71 so that the horizontal velocity at injection was continuously computed every two (2) seconds. All four runs will be edited to show relevant computations.</p> <ol style="list-style-type: none"> P70 run from early in descent (Digital Simulation) P70 run from the middle of descent (Digital Simulation) P71 run from the middle of descent (Digital Simulation) Test of P70 followed by a P71 (Hybrid Simulation) 	C. Schulenberg

LUMINARY 1B LEVEL 3 TESTS

DATE 8/22/69

ANOM/ PCR/ ACB	TEST DESCRIPTION	RESPONSIBLE ENGINEER
PCR 827	Test by obtaining a downlink edit of test no. 3.17a (Digital Simulation).	C. Schulenberg
PCR 845	Visual Inspection	L. Berman
PCR 847	Visual Inspection	L. Berman
<u>NAVIGATION</u>		
LNY 61	P20 run with R31; force restart during shaft and trunnion incorporations. Navigations edit output and trace to verify correct incorporation.	P. Volante
LNY 66	Anomaly closed with PCR 780.	J. Stoppelman
LNY 71	Test run with ITRACE of predesignate area to demonstrate that XR1, used by orbital integration, was undisturbed.	V. Dunbar
LNY 81	Move antenna out of limits so that a reposition will be called for. Then do V41N72 during reposition and observe a designate to proper angles after reposition.	P. Volante
LNY 87	P22 run; check simulator output to verify that no display comes up after radar lock-on.	P. Volante
PCR 780	P20 - call P20 A. After PRO to V50N72, key V16N78 1. Observe 3 samples of data prior to computer activity light on solid, which indicates R22 processing.	B. McCoy

LUMINARY 1B LEVEL 3 TESTS

DATE 8/22/69

ANOM/ PCR/ ACB	TEST DESCRIPTION	RESPONSIBLE ENGINEER
PCR 780 (cont'd.)	<p>2. Observe 7 samples of data after computer activity light out, which indicates R22 finished processing.</p> <p>B. After PRO to V50N18 in P25</p> <ol style="list-style-type: none"> 1. Observe no tracker fail lamp without radar lock on. 2. Observe samples every seven seconds with radar lock on. <p>C. Assure that R04 still works.</p>	B. McCoy
PCR 818	P20 run with RMAX, VMAX set to zero; observe V32, V33 and V34 responses made to V06N49 display.	P. Volante
PCR 826.2	Earth centered state vectors to be established for the 2 vehicles and P76 will be called. A time of ignition will be entered in response to the V06N33 display and a delta \bar{V} will be entered in response to the V06N84.	T. Crocker
LNY 91	.In P20 - lock on manually, then switch to LGC mode - observe that both CDU fail discrete and CDU zero discrete is present and no 515 alarm	P. Volante
<u>POWERED FLIGHT</u>		
LNY 63	Cause engine fail during 10% and observe no throttle up at ZOOMTIME.	P. Adler
PCR 830	Do Lambert ITRACE to verify counter was set to 20.	P. Adler
PCN 831.2	Twenty-nine (29) bench tests of the Lambert Routine will be run, and the results will be compared by means of a Lambert MAC edit against the MAC Lambert program in KWK conics which has been the standard for several years. Twenty-eight (28) of the bench tests will cover a wide range of transfer angles and eccentricities and initial and terminal radius vectors. The one other case will be the	W. Robertson

LUMINARY IB LEVEL 3 TESTS

DATE 8/22/69

ANOM/ PCR/ ACB	TEST DESCRIPTION	RESPONSIBLE ENGINEER
PCR 284	<p>precise input of the only known case ever to have caused the problem for which PCN 831.2 is the fix.</p> <p>Load VGTIG with 3XXXX - select P00 - observe octal down list has 3XXXX in sight down list.</p>	B. McCoy
LNY 68	<p><u>ALIGNMENTS</u></p> <p>Test to be done in System Test Lab and the Hybrid Simulator. The test requires that IMU compensation be pulsing gyros when N22 or N93 are answered. The test involves loading the GCOMP registers with larger values on the N22 or N93 display, waiting for IMU compensation to start, then answering the display with a proceed.</p>	D. Millard
LNY 74	<p>Two digital tests to be run. P57 AT-2 to REFSMMAT use a star and the sun P57 AT-3 to Landing Site using the sun</p>	D. Millard
PCR 825.2	<p>Key V37E52E and observe 3 appear in Register 3 of first display.</p>	D. Millard
LNY 69	<p><u>DAP</u></p> <p>Patch in correction on rollback of powered ascent run in which anomaly was observed to verify that anomaly has been repaired without degrading performance.</p>	

ANOM/ PCR/ ACB	TEST DESCRIPTION	RESPONSIBLE ENGINEER
PCR 816	<p>In ascent coasting flight with 5° DB specified:</p> <ol style="list-style-type: none"> 1. Initiate limit cycle with pulse from RHC in manual rate command/attitude hold mode 2. Observe limit cycle amplitude to verify selection of 5° DB by program 3. Specify 1° DB via R03 (V48) and observe subsequent limit cycle amplitude to verify selection of 1° DB by program 4. Specify 0.3° DB via R03 and observe subsequent limit cycle amplitude to verify selection of 0.3° DB by program 	G. Kalan
PCR 841	<p>In ascent coasting flight:</p> <ol style="list-style-type: none"> 1. Initiate 1°/sec rates about the P, Q, and R axes using RHC in manual rate command/attitude hold mode. 2. Select DAP derived rate display via V60 and verify rate display coding by observing attitude error needle input. 3. Terminate manual rate commands to allow normal attitude hold mode limit cycling. 4. Select V61 and observe attitude error needle inputs to verify that mode I display works properly. 5. Select V62 and observe attitude error needle inputs to verify that mode II display work properly. 	G. Kalan
PCR 842	Patch in correction on rollback of powered ascent run in which anomaly LNY 78 was observed to verify that anomaly has been repaired without degrading performance.	G. Kalan
PCR 812.2	<p><u>TARGETTING</u></p> <p>Observe setting of XDELFLG after N81 display but before N42 display.</p>	P. White

LUMINARY 1B LEVEL 3 TESTS

DATE 8/22/69

ANOM/ PCR/ ACB	TEST DESCRIPTION	RESPONSIBLE ENGINEER
PCR 823	Check to see opp error when P31 is killed.	P. Adler
PCR 844	Hybrid Simulation; attempt entry to P38, P39, P78 and P79 by V37E38E, V37E39E, V37E78E and V37E79E; resulting in reception of operator error light signifying non-existence of said programs.	D. Moore
<u>MISCELLANEOUS</u>		
LNY 67	Visual Inspection	P. Rye
PCR 773.2	Visual Inspection	J. Stoppelman
PCR 776.2	Start simulation in orbit with precision integration on. Run for about 160 seconds with no other computer activity. Using R2 Model, edit, observe and evaluate the integration process, and compare the results with the output of identical tests done on Colossus. Repeat this sequence using the six cases, both lunar and earth orbit, used in the Colossus tests.	M. Albert
PCR 781.2	Test to be conducted in System Tests lab to verify change of maximum PIPA Bias from 2.28 cm/sec ² to 9 cm/sec ² using an erasable program.	E. Grace
PCR 791.2	Observe PRO rejected when verb lights contain V21, V22 & V23 and accepted when verb lights contain V20 or V24.	J. Vella

LUMINARY IB LEVEL 3 TESTS

DATE 8/22/69

ANOM/ PCR/ ACB	TEST DESCRIPTION	RESPONSIBLE ENGINEER
PCR 798.2	Verify by a trace that if GLOKFAIL is set before V37 it is reset after V37 is selected.	E. Hughes
PCR 801.2	Visual Inspection	P. Rye
PCR 802.2	With APSFLG set, select P40 four (4) times; do an error reset and select P40 four (4) times again in order to get alarm 1706. Verify by a trace that FAILREG +2 always contains the most recent alarm and bit 15 is not longer set for the 4th alarm.	E. Hughes
PCR 805	Verify the operator error when V66 is selected with SURFFLAG set.	E. Hughes
PCR 807.2	Two (2) tests will be run. In each test, P21 is called to compute values for earth-centered state vectors and the "this vehicle" option is employed. In test "a" the initial (present) time is accepted and in test "b" the initial (present) time is rejected in favor of one requested by a V25 response to the V06N34 display. In both tests a second computation will be requested via recycle on the V06N43, to provide a second set of data for the earlier time plus 10 minutes.	T. Crocker
PCR 277	Digital test of V74 with trace.	E. Denniston
PCR 832.2	Call R05 during P63 and obtain antenna angles display; reset REFSMMAT flag and show that RDS is still callable.	T. Crocker
PCR 853	Attempt V35 during all possible major mode - observe operator error.	B. McCoy
LNY 89	Assure that LM State Vector time tag is "large". Select P00 set present time of LGC to zero; call P27 with V71 - wait until computer activity light goes on. Verify via V16N38 that state vector is not being integrated backward.	B. McCoy
PCR 856	Visual Inspection	B. McCoy
ACB 43	Tested during Level 4 Runs	D. Eyles/ L. Berman

LUMINARY IB LEVEL 3 TESTS

DATE 8/22/69

ANOM/
PCR/
ACB

TEST DESCRIPTION

RESPONSIBLE
ENGINEER

ACB 44	Tested during Level 4 Runs	C. Schulenberg
ACB L1	Tested during Level 4 Runs	D. Moore
ACB L2	Tested during Level 4 Runs	D. Eyles
ACB L3	Make the Special test of erasable for R3 in Noun 49 unshared.	P. Volante/ J. Stoppelman

LUMINARY 1B PROGRAM DEVELOPMENT STATUS

DATE 8/22/69

PCR/ Anom/ ACBs	DESCRIPTION	LEVEL 2 AND LEVEL 3 TESTING									RESPONSIBLE ENGINEER	
		START			COMPLETE			RESULTS EVALUATED				
		REQ	PRED	ACT	REQ	PRED	ACT	REQ	PRED	ACT		
	<u>Landing</u>											
LNy 58	It is possible to enter a program which uses the RR or LR while R12 is reading the LR.	7/2	7/2	7/15	7/15	7/11	7/18	7/18	7/11			
LNy 64	Restart occurring in P63.	6/24	6/24	7/15	7/1	7/1	7/3	7/3	7/3			
LNy 72	Program alarm light every two seconds after V32E to 523 alarm.	7/2	7/3	7/15	8/12	8/12	7/18	8/12	8/12			
LNy 73	Bug in recycle option to N61 in P63.	6/24	6/24	7/15	7/1	7/1	7/3	7/15	7/15			
LNy 76	N60 Made to be HCALC1 (ALTRATE)	6/24	6/24	7/15	7/7	7/7	7/18	7/7	7/7			
LNy 88	Landing Radar Repositioning routine is not restart protected.	7/28	7/28	7/31	7/31	7/31	7/31	7/31	7/31			
814 (REV)	Reduce keystrokes required to check and approve LR data.	7/1	7/10	7/10	7/15	7/15	7/14	7/18	7/18	7/14		
817	Eliminate undesirable LR position alarms from R12.	7/7	7/9	7/15	7/20	7/14	7/23	7/23	7/14			
820	Eliminate lighting of ALT light when low scale discrete is absent.	7/2	7/9	7/15	7/15	7/14	7/18	7/18	7/14			
838	Prevent RCS jet firings on lunar surface.	7/10	7/10	7/14	7/14	7/14	7/23	7/23	7/23			
839	R12 and LR Repositioning Routine Improvements.	7/28	7/28	7/31	8/12	8/12	7/31	8/12	8/12			

PCR/ Anom/ ACBs	DESCRIPTION	LEVEL 2 AND LEVEL 3 TESTING									RESPONSIBLE ENGINEER
		START			COMPLETE			RESULTS EVAL			
		Req	Pred	Act	Req	Pred	Act	Req	Pred	Act	
840	Reduce Oscillation in P64/P65.		7/10	7/10	7/10	7/10	7/14	7/14	7/14	7/14.	
848	Prevent RR ECDU's from stealing LGC Memory Cycles.		7/28	7/28	7/31	7/31	8/7	7/31	7/31	8/7	
854	Provide a Flexible Method for crew to Modify RLS.		8/5	8/5	8/12	8/7	8/7	8/12	8/7	8/7	
855	Begin Reading LR Velocity as soon as Velocity Data Good Appears.		8/5	8/6	8/12	8/6	8/6	8/12	8/6	8/6	
LN 90	If an engine fail occurs between TIG and throttle up, the flashing V97N63 display will be overwritten by the static V06N63 display.		8/5	8/5	8/12	8/7	8/7	8/12	8/7	8/7	

LUMINARY IB PROGRAM DEVELOPMENT STATUS

DATE 8/22/69

PCR/ Anom/ ACBs	DESCRIPTION	LEVEL 2 AND LEVEL 3 TESTING									RESPONSIBLE ENGINEER
		START		COMPLETE		RESULTS EVALUATED					
		REQ	PRED ACT	REQ	PRED ACT	REQ	PRED	ACT			
	<u>Navigation</u>										
LNy 61	If a restart occurs during the transition or shaft incorporations of P20 Rendezvous Navigation, an incorrect State Vector Update can occur.	6/5	6/5	7/15	6/20	6/20	7/3	7/10	7/14		
LNy 66	R04 Erasable Conflict w/R65.	FIXED BY PCR 780									
LNy 71	In P22 Lunar Surface Navigation the RR predesignate routine stores into an erasable used by orbital integration.	6/5	6/5	7/15	6/20	6/20	7/3	7/3	7/14		
LNy 81	Eliminate erasables shared between Radar Tasks and jobs.	7/28	7/31	7/31	7/31	7/31	7/31	7/31	7/31		
LNy 87	In P22 the flashing V50N72 is displayed following radar lock on in R21.	7/10	7/25	7/15	7/31	8/8	7/18	7/31	8/8		
780	Provide pure RR Range, Range rate and time lag during P20, P22, and P25.	7/7	7/7	7/15	7/15	7/10	7/18	7/18	7/10		
818	Permit rejection of Individual Measurement incorporations in P20.	7/7	7/10	7/10	7/15	7/20	7/15	7/23	7/23	7/15	
826.2	Reverse P76 Display.	7/10	7/16	7/15	7/15	7/16	7/15	7/15	7/16		
LNy 91	CDU Fail program alarm is given in P20.	8/5	8/5	8/12	8/6	8/6	8/12	8/6	8/6		

LUMINARY 1B PROGRAM DEVELOPMENT STATUS

DATE 8/22/69

PCR/ Anom/ ACBs	DESCRIPTION	LEVEL 2 AND LEVEL 3 TESTING									
		START			COMPLETE			RESULTS EVALUATED			RESPONSIBLE ENGINEER
		REQ	PRED	ACT	REQ	PRED	ACT	REQ	PRED	ACT	
816	Modify R03 to permit astronaut setting of 1°-deadband.		7/1	7/1	7/15	7/15	7/15	7/15	7/15	7/15	
841	PGNCS Derived Vehicle Attitude Rates on FDAI Error Needles.		7/28	7/31	7/31	7/31	7/31	7/31	7/31	7/31	
842	Modification of Criteria used to determine DAP phase plane parabola intercept.		8/7	8/7	8/12	8/8	8/8	8/12	8/8	8/8	
	<u>Targetting</u>										
812.2	Resetting and Setting of the External ΔV flag.	7/1	7/10	7/14	7/15	7/15	7/14	7/18	7/18	7/14	
823	Delete P31 from Luminary 1B.		6/24	6/24	7/15	7/1	7/1	7/3	7/3	7/1	
844	Deletion of P38/78 and P39/79.		6/24	6/24	7/15	7/1	7/1	7/3	7/3	7/1	
	<u>Miscellaneous</u>										
LNy 67	Make ATTSTALL 1210 into BAILOUT instead of POODOO.		6/25	7/7	7/15	7/2	7/8	7/5	7/5	7/8	
773.2	Fix constants for Planetary Inertial Subroutine.		8/5	8/5	8/12	8/5	8/5	8/12	8/5	8/5	
776.2	Improved R2 Model timing.		6/25	5/28	7/15	7/15	6/7	7/18	7/18	6/7	
781.2	PIPA Bias Compensation Scale.	7/1	7/9	7/16	7/15	7/28	7/28	7/15	7/31	7/31	
791.2	Do not allow a PROCEED Response to V21, V22, V23..		6/25	6/25	7/15	7/28	7/28	7/15	7/31	7/28	

LUMINARY 1B PROGRAM DEVELOPMENT STATUS

DATE 8/22/69

PCR/ Anom/ ACBs	DESCRIPTION	LEVEL 2 AND LEVEL 3 TESTING									RESPONSIBLE ENGINEER
		START			COMPLETE			RESULTS EVALUATED			
		REQ	PRED	ACT	REQ	PRED	ACT	REQ	PRED	ACT	
798.2	Reset GLOKFAIL in ROO.	6/25	6/25	7/15	7/10	7/3	7/13	7/13	7/3		
801.2	Make BAILOUT alarms start with 3XXXX and POODOO alarms start with 2XXXX.	6/25	6/25	7/15	7/1	7/1	7/3	7/3	7/1		
802.2	Save alarm data after "Error Reset".	6/25	6/25	7/15	7/10	7/3	7/13	7/13	7/14		
805	Don't allow V66 on the surface.	6/25	6/25	7/15	7/10	7/3	7/13	7/13	7/17		
807.2	Add present time to P21.	6/25	6/25	7/15	7/10	7/1	7/13	7/13	7/14		
277	Affixed DUMPCNT.	6/24	6/24	7/15	7/10	7/23	7/15	7/15	7/23		
832.2	Define Actual Restrictions on Running R05.	7/28	7/28	7/31	7/31	7/31	7/31	7/31	7/31		
853	Restrict V35 to POO.	8/5	8/5	8/12	8/5	8/5	8/12	8/5	8/5		
---*	Provide CH13STAL Coding.	7/10	7/10	8/12	8/12	8/11	8/12	8/12	8/11		
856	Change 1502/1206 from POODOO's to BAILOUTS.	8/5	8/10	8/12	8/11	8/11	8/12	8/11	8/11		
LNy-89	State Vector was integrated backwards while in POO, P27.	8/5	8/5	8/12	8/5	8/5	8/12	8/5	8/5		
ACB 43	Change the Displays at the end of descent and ascent guidance to be "non-R" type displays.	LEVEL 4 TESTED									
ACB 44	Change the call to ZATTEROR in the ascent guidance to a call to the STOPRATE routine.	LEVEL 4 TESTED									

* Directed by G. Low NASA/MSC CCB

SECTION IV

LEVEL 4 TESTING

LUMINARY 1B LEVEL 4 TEST DESCRIPTION AND STATUS

DATE 8/22/69

TEST NO.	DESCRIPTION	RESPONSIBLE ENGINEER	START			COMPLETION			DOCUMENTATION		
			Req.	Pred.	Act.	Req.	Pred.	Act.	Req.	Pred.	Act.
4.1A	RENDEZVOUS: NOMINAL - LM ACTIVE (P00, P52, P20, P42, P30, P41, P33, P41, P34, P41, P35, P41)	P. White			7/14	8/12	8/5	8/10	8/12	8/11	8/12
4.1B	RENDEZVOUS: NOMINAL - CSM ACTIVE (P25, P52, P20, P72, P76, P73, P25 - R36, P76, P20, P73, P76, P74, P76, P75, P76, P75, P76)	P. White			7/14	8/12	8/5	8/10	8/12	8/11	8/12
4.2A	ASCENT & ABORT: Ascent (normal liftoff) (continuance from area 4) (P12, P20 < 400 min., P32)	L. Berman			7/14	8/12	8/5	8/11	8/12	8/11	8/12
4.2B	ASCENT & ABORT: Aborts from Descent 1. (P6x, P70(early), P20, P32) 2. (P6x, P71(late), P20, P32) 3. (P6x, P70, P71, P20)	L. Berman		7/31	7/31	8/12	8/12	8/12		8/12	8/12
4.2C	ASCENT & ABORT: Aborts from Surface. (Quick liftoff) (Continuance from area 3) (P68, P00, P12, P20)	L. Berman		7/31	7/31	8/12	8/12	8/12		8/12	8/12
4.3A	LANDINGS: DOI (P00, P20, P30, P40, P52, P40, P00)	D. Eyles			7/14	8/12	8/5	8/10	8/12	8/11	8/11
4.3B	LANDINGS: PDI (P00, P63, P64, P65, P66 at 120 ft. & redesignation, P68)	D. Eyles			7/14	8/12	7/31	8/10	8/12	8/11	8/12
4.3C	LANDINGS: PDI (P00, P63, P64, P66 at 500 ft., P67 at 125 ft., P68)	D. Eyles			7/14	8/12	7/31	8/10	8/12	8/11	8/12
4.3D	LANDINGS: PDI (P00, P63, P64, P66 at 500 ft., P67 at 500 ft.,	D. Eyles			7/14	8/12	7/31	8/10	8/12	8/11	8/12

LUMINARY 1B LEVEL 4 TEST DESCRIPTION AND STATUS

DATE 8/22/69

TEST NO.	DESCRIPTION	RESPONSIBLE ENGINEER	START			COMPLETION			DOCUMENTATION		
			Req.	Pred.	Act.	Req.	Pred.	Act.	Req.	Pred.	Act.
L4.4A	LUNAR SURFACE: (P68, P12(without liftoff), P57(AT-1 to Refsmmat), P57(AT-2 to Refsmmat), P57(AT-3 to Refsmmat), P12)	D. Millard		7/31	7/31	8/12	8/7	8/11	8/12	8/12	8/12
L4.4B	LUNAR SURFACE: (P68, P22(update mode), P57 (AT-3 to Refsmmat), P22(no update), P57(AT-3 to Landing Site at Liftoff), P12)	D. Millard		7/31	7/31	8/12	8/7	8/11	8/12	8/12	8/12
L4.5A	MISCELLANEOUS: Restart Test Programs in powered flight alignments and navigation.	B. McCoy		7/28	7/28	8/12	8/7	8/12	8/12	8/12	8/12
L4.5B	MISCELLANEOUS: Extended Verb usage during powered flight, alignments and navigation.	B. McCoy		7/28	7/28	8/12	8/7	8/12	8/12	8/12	8/12

SECTION V

STG TESTING

Number	Title	Engineer	Test Complete			Document Approved		
			Req.	Pred.	Act.	Req.	Pred.	Act.
1	Test of New PIPA Bias Compensation	Grace	8/12		7/23		8/8	8/12
2	Test of C13STA11	Sheridan	8/12	8/8	8/8		8/15	8/12
3	Test of New RR CDU Zero	Reedy	8/12	8/8	8/8		8/15	8/12

Test	Title	Engineer	Test Complete			Document Approved		
			Req.	Pred.	Act.	Req.	Pred.	Act.
ICP. 4	Polarity & Scaling Test for Accelerometers	Sheridan		9/12			9/26	
ALM1.1	Alarm Code Test	Sheridan		9/12			9/26	
RP5.1	Anti Max Limit & Remode	Reedy		9/5			9/19	
RP5.2	LR + RR data R04	Reedy		8/29			9/12	
RP5.3	P20 Options	Reedy		9/5			9/19	
RP5.4	Operations of Lunar Surface Prog. P22	Reedy		9/12			9/26	
RP5.5	R29 Test	Reedy		9/5			9/19	
RP5.6	R77 Test	Reedy		9/5			9/19	
IP.1	LGC/CMC Clock Synch. (R33) Including V55	St. Amand		9/5			9/19	
IP.2	AGS Initialization Prog.	St. Amand		9/5			9/19	
IOP.1	Nominal Turn On	Grace		8/29			9/12	
IOP.2	Turn On with Failures Test	Grace		8/29			9/12	
IOP.3	Gimbal Lock Protection Test	Grace		8/29			9/12	
IOP.4	Cage Test	Grace		8/29			9/12	
IOP.5	IMU Error Monitor Test	Grace		8/29			9/12	
AAP.1	P51-IMU Orientation Determination	St. Amand		9/5			9/19	
AAP.2	P-51 Options Case	St. Amand		9/5			9/19	

This schedule to be transferred to Mission H Plan.

Test	Title	Engineer	Test Complete			Document Approved		
			Req.	Pred.	Act.	Req.	Pred.	Act.
AAP.3	P52-IMU Realign Prog ⁽²⁾ .	St. Amand		9/5			9/19	
AAP.4	P52-Options Case	St. Amand		9/12			9/26	
AAP.5	AOT Bias Calibration Test	St. Amand		9/12			9/26	
AAP.6	P57-Lunar Surface Align	St. Amand		9/12			9/26	
SEV.1	Verb 40 Zero IMU-CDU	Grace		8/29			9/12	
SEV.2	Verb 41 Coarse Align IMU	Grace		8/29			9/12	
SEV.3	Verb 42 IMU to Inertial Mode	Grace		8/29			9/12	
SEV.4	Verb 43 Load IMU Error Needles	Grace		8/29			9/12	
ICP.1	Free Fall Bias Comp. Test	Sheridan		9/12			9/26	
ICP.2	Thrusting & Comp. Test	Sheridan		9/12			9/26	
ICP.3	Gyro Drift & Acceler. Comp. Test	Sheridan		9/12			9/26	

This schedule to be transferred to Mission H Plan.

E-MEMORY PROGRAMS	Engineer	First Program Assemble (1)			Good Digital Sim (1)			Level V STG Test Doc			With K-Start Tape Good STG Lab (2) Level 5		
		Req	Pred	Act	Req	Pred	Act	Req ⁽⁵⁾	Pred	Act	Req	Pred	Act
IMU Performance Test STP 2	Grace	8/15	8/15	8/15	8/15	8/30		10/1	9/15		8/23	8/23	
IRIG SF STP 3	Grace	8/15	8/15	8/15	8/15	8/30		10/1	9/15		8/23	8/23	
Self Test STP 4	Estabrook	8/15	8/30		8/15	8/30		10/1	9/15		8/30	8/30	
AGS Align STP 5 Digit Lab	St. Amand	8/15	8/30		8/15	8/30		10/1	9/15		8/30	8/30	

- (1) With current version of LUMINARY
- (2) With version of LUMINARY released for manufacture.

This schedule to be transferred to Mission H Plan.

SECTION VI

LUMINARY FLOWCHART STATUS

ITEM NO.	DESCRIPTION	RESPONSIBILITY	FIRST RELEASE		
			Required	Predicted	Actual
P00	LGC Idling	Danforth		10/15/69	
P20	Rendezvous Navigation	Danforth		9/30/69	
P25	Preferred Tracking Attitude	Danforth		9/30/69	
P32/72	Co-elliptic Sequence Initiation (CSI)	Hubbard		9/15/69	
P38/78	Stable Orbit Rendezvous (SOR)	Hubbard		9/30/69	
P39/79	Stable Orbit Midcourse (SOM)	Hubbard		9/30/69	
P40	DPS Thrusting	Danforth		6/30/69	6/20/69
P42	APS Thrusting	Danforth		6/30/69	6/5/69
P47	Thrust Monitor	Danforth		8/29/69	
P51	IMU Orientation Determination	Sorant		10/15/69	
P52	IMU Align	Sorant		10/15/69	
R00	Final Automatic Request Terminate	Danforth		9/30/69	
P63 P67	Landing Programs	Danforth		9/30/69	
R03	DAP Data Load	Entes		10/15/69	
R04	RR/LR Self Test	Danforth		9/30/69	
R31	Rendezvous Parameter Display	Danforth		6/30/69	6/27/69
R40	DPS/APS Thrust Fail	Danforth		9/30/69	
R47	AGS Initialization	Danforth		7/30/69	8/8/69
R56	Terminate Tracking	Danforth		9/30/69	
P12	Powered Ascent	Sorant		9/15/69	
P70 P71	Aborts	Sorant		9/15/69	
This schedule to be transferred to Mission H Plan.					

SECTION VII

LUMINARY ANOMALY STATUS

ANSI

LUMINARY 1B

ANOMALY STATUS REPORT

DATE 8/22/69

MSC NO.	DESCRIPTION	Originator/ Date/ Control No.	RECOMMENDED DISPOSITION (Date Submitted to NASA)	NASA DIRECTION/ DATE	CLOSING ACTION TAKEN/ DATE
L-1B- 01	Selection of P22 before the CSM has entered Mode 2 Radar Coverage can result in an erroneous 530 Alarm.	MIT 8/19/69	Work-around in LUMINARY 1B; fix in LUMINARY 1C. 8/25/69		
L-1B- 02	R29 will not achieve RR lockon.	MIT 8/20/69	Work-around in LUMINARY 1B; fix in LUMINARY 1C. 8/25/69		
L-1B- 03	During powered ascent, for a period of two seconds, the quantity COEFFR, computed in the 1/ACCS program, can be incorrectly determined to an extremely small (essentially zero) value.	MIT 8/26/69	Work-around in LUMINARY 1B; fix in subsequent release. 8/26/69		

THIS SCHEDULE TO BE TRANSFERRED TO MISSION H PLAN.