



Mission Planning and Analysis Division
NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
MANNED SPACECRAFT CENTER
HOUSTON, TEXAS 77058

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R. A. LAPSON

REPLY TO
ATTN OF: FM73 (72-62)

April 3, 1972

MEMORANDUM

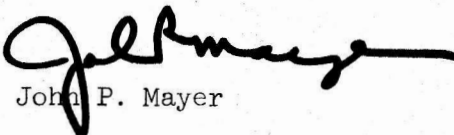
TO: FS/Chief, Flight Support Division

FROM: FM/Chief, Mission Planning and Analysis Division

SUBJECT: Final Prelaunch Erasable Load Parameters for Apollo 16
Flight Programs Using April 16, 1972, Launch Data

The enclosure to this memorandum defines the final Apollo 16 prelaunch erasable load parameters for Colossus 3 and Luminary LE that have been assigned to the Mission Planning and Analysis Division (MPAD). As requested by the Flight Support Division (FSD), MPAD is to supply on computer cards to the Guidance Program Section (GPS) the preliminary and final engineering values for the specified erasable quantities. The Guidance and Dynamics Branch (GDB) of the MPAD coordinated with the appropriate MPAD branches for the definition of those parameters and ensured that the data was supplied in the correct format and on schedule.

Tables II and III present a listing of these erasable quantities just as they appear on the computer cards. It should be noted that in some cases the listing gives both the engineering and octal values; however, the octal value is supplied to FSD only for the purpose of comparison. The name, description, and units for each erasable quantity are given in table I. Certain of the erasable parameters are launch date dependent variables which are given in tables II and III are valid for an April 16, 1972, launch date. Any questions or comments concerning the erasable load should be directed to Aldo Bordano or to Harry Miles, ext. 2855.


John P. Mayer

Enclosure

Distribution:
See list attached

FM73:HJMiles:bla:3/29/72:2855

CMC name	Description	Engineering Units	AGC Units
NO. PASS	Number of passes of P24 before landmark coordinate update	unitless	unitless
RTED1	First coefficient defining high speed V- γ target line polynomial	unitless	unitless
DVTHRESH	Threshold value of ΔV which must be sensed in two second period or SPS thrust failure is indicated	fps	m/cs
HORIZALT	Horizon altitude	km	m
ALTVAR	COAS angular error variance	(rad) ²	(rad) ²
EMDOT	SPS mass flow rate	lb/sec	kg/cs
UNITW ¹	Polar axis unit vector in reference coordinates	unitless	unitless
EIMPLSEC	SPS impulse acquired from a one second burn	lb-sec	kg-m/cs
EFIMPO1	SPS minimum impulse constant equal to the slope of the minimum impulse curve from 0 to 1 second. Used to estimate burn time when burn time is less than 1.0 second	lb	kg-m/cs/sec
EFIMPL6	SPS thrust used to estimate burn time when the burn is between 1 and 6 sec	lb	kg-m/cs/sec
E3J22R2M	Product of J22 lunar potential coeff., μ^{moon} , radius of the moon squared, and 3.0	$\text{m}^5/(\text{cs})^2$	$\text{m}^5/(\text{cs})^2$
E32C31RM	Product of C(3, 1) lunar potential coeff., μ^{moon} , radius of the moon cubed, and 1.5	$\text{m}^6/(\text{cs})^2$	$\text{m}^6/(\text{cs})^2$
WRENDPOS	W-matrix initialization for rendezvous navigation	ft	m
WRENDVEL	W-matrix initialization for rendezvous navigation	fps	m/cs
RMAX	Rendezvous navigation position update threshold limit	ft	m
VMAX	Rendezvous navigation velocity update threshold limit	fps	m/cs
WORBPOS	W-matrix initialization for orbital navigation	ft	m

Table I

CMC name	Description	Engineering Units	AGC Units
WORVEL	W-matrix initialization for orbital navigation	ft/sec	m/cs
S22WSUBL	W-matrix initialization for orbital navigation	ft	m
RPVAR	Primary body radius error variance	m ²	m ²
504LM ¹	Lunar libration correction vector in moon-fixed coordinates	rad	rad
EMSALT	EMS initialization altitude	ft	m
RLS ⁴	Lunar landing site radius vector	ft	m
TIMEMO ¹	Time at center of range over which lunar position polynomial is valid	hrs	cs
VECOEM ¹	Lunar position polynomial coefficients	$\frac{C_0 - C_9}{(m/cs^n)}$	$\frac{C_0 - C_9}{(m/cs^n)}$
RESO ¹	Position vector of sun at TIMEMO	ft	m
VESO ¹	Velocity vector of sun at TIMEMO	ft/sec	m/cs
OMEGAES ¹	Angular velocity of solar position vector at TIMEMO	deg/sec	rev/cs
DTF	Value subtracted from SIVB time-to-go in P15 to account for thrust decay	sec	cs
HAMDELH	Gives required altitude difference in P31	n.mi.	m
WRDIME	Used in MINKEY to check if sufficient time has elapsed since the previous W-matrix initialization	sec	cs
MINBLKTM	Used in MINKEY to check whether the time since last mark is excessive	sec	cs
TBEFCOMP	Used in MINKEY to check if sufficient time remains before the final targeting computation so that W-matrix reinitialization can take place	sec	cs
BRNBLKTM	Used in MINKEY to determine the age of W-matrix following next maneuver	sec	cs

Table I (continued)

CMC name	Description	Engineering Units	AGC Units
MAXWTIME	Used in MINKEY to determine if W-matrix age will be excessive after next maneuver	sec	cs
FINCMPIM	Used in MINKEY to allow for length of time of final targeting computation	sec	cs
INTVAR	Coasting integration error variance	m ²	m ²
LAUNCHAZ ¹	Desired azimuth of the stable member X-axis east of true north	deg	rev
WMIDPOS	W-matrix initialization for cislunar midcourse navigation	ft	m
WMIDVEL	W-matrix initialization for cislunar midcourse navigation	fps	m/cs
RVAR	VHF range error variance	perc err	perc err
RVARMIN	VHF minimum range error variance	ft ²	m ²
IADPAD	Reference L/D ratio	unitless	unitless
LODPAD	Final phase reference L/D ratio	unitless	unitless
ALFAPAD	Hypersonic value of angle of attack α for CM in trimmed flight	deg	rev
P37RANGE	Nominal inertial range from 300,000 ft altitude to 25,000 ft altitude	n.mi.	"rev"
ETDECAV	Value subtracted from time-to-go which compensates for SPS thrust tailoff	sec	cs
PACTOFF	Pitch angle to CSM c.g. at insertion	deg	edu puls
YACTOFF	Yaw angle to CSM c.g. at insertion	deg	edu puls
LEMMASS	LM mass (at transposition and docking)	lb	kg
CSMMASS	CSM mass (at transposition and docking)	lb	kg
POLYNUM + 0	Boost monitor pitch polynomial degree information	unitless	unitless
POLYNUM + 1	Boost monitor pitch polynomial coefficients, $A_0 - A_6$	$A_0 - A_6$ (deg/sec ⁿ)	$A_0 - A_6$ (rev/cs ⁿ)

Table I (continued)

CMC name	Description	Engineering Units	AGC Units
SATRLRT	Boost roll rate	deg/sec	rev/cs
RPSTART	Time from lift-off to start boost polynomial and booster roll	sec	cs
POLYSTOP	Time increment after RPSTART to stop boost polynomial and hold display of attitude error constant	sec	cs
HORISLP	Slope of function used for horizon altitude computation	unitless	unitless
LAT (SPL)	Entry target data for boost aborts, latitude	deg	rev
LNG (SPL)	Entry target data for boost aborts, longitude	deg	rev

Table I (continued)

LGC name	Description	Engineering Units	AGC Units
MASS	Total vehicle mass (CSM/LM separation)	lb	kg
LEMMASS	LM mass (at CSM/LM separation)	lb	kg
CSMMASS	CSM mass (at CSM/LM separation)	lb	kg
E3J22R2M	Product of J22 lunar potential coefficient, μ_{moon} , radius of the moon squared, and 3.0	$m^5/(cs)^2$	$m^5/(cs)^2$
E32C31RM	Product of C(3,1) lunar potential coefficient, μ_{moon} , radius of the moon cubed, and 1.5	$m^6/(cs)^2$	$m^6/(cs)^2$
TOOFEW	Minimum number of consecutive ROD calculations required before bypass the issuing of program alarm in P66	unitless	unitless
X789	Vector containing the best estimate of bias necessary to offset the rendezvous radar pointing error	rad	rad
-AYO ¹	True to mean pole rotation about the -Y-axis	rad	rad
AXO ¹	True to mean pole rotation about the +X-axis	rad	rad
REFSMMAT	Nominal descent REFSMMAT	unitless	unitless
RANGEVAR	Variance expected in measured range	unitless	unitless
RATEVAR	Variance expected in measured range rate	unitless	unitless
RVARMIN	Minimum rendezvous radar range error variance	(ft) ²	m ²
VVARMIN	Minimum rendezvous radar range rate error variance	ft ² /sec ²	(m/cs) ²
WRENDPOS	1st-3rd diagonal components of W-matrix	ft	m
WRENDVEL	4th-6th diagonal components of W-matrix	fps	m/cs
WSHAFT	7th diagonal component of W-matrix	millirad	rad

Table I (continued)

name	Description	Engineering Units	AGC Units
WTRUN	8th diagonal component of W-matrix	millirad	rad
RMAX	Maximum value of rendezvous position update allowed without astronaut approval	ft	m
VMAX	Maximum value of rendezvous velocity update allowed without astronaut approval	fps	m/cs
WSURFPOS	Initialization of position portion of W-matrix for P22(LM on lunar surface tracking CSM)	ft	m
WSURFVEL	Initialization of velocity portion of W-matrix for P22(LM on lunar surface tracking CSM)	fps	m/cs
SHAFTVAR	Variance associated with the measured value of the rendezvous radar shaft angle	(millirad) ²	(rad) ²
TRUNVAR	Variance associated with the measured value of the rendezvous radar trunnion angle	(millirad) ²	(rad) ²
504LM ¹	Lunar libration correction vector in moon-fixed coordinates	rad	rad
RLS ⁴	Lunar landing site radius vector	ft	m
TLAND	Nominal time of lunar landing	hr(g.e.t.)	cs(g.e.t.)
VELBIAS	Velocity portion of LR reasonability test	fps	m/cs
RBRFG	High-gate position aimpoint vector in the descent guidance coordinate system	ft	m
RAPFG	Low-gate position aimpoint vector in the descent guidance coordinate system	ft	m
VBRFG	High-gate velocity aimpoint vector expressed in the descent guidance coordinate system	fps	m/cs
VAPFG	Low-gate velocity aimpoint vector expressed in the descent guidance coordinate system	fps	m/cs

Table I (continued)

LGC name	Description	Engineering Units	AGC Units
ABRFG	High-gate acceleration aimpoint vector expressed in the descent guidance coordinate system	ft/(sec) ²	m/(cs) ²
AAPFG	Low-gate acceleration aimpoint vector expressed in the descent guidance coordinate system	ft/(sec) ²	m/(cs) ²
VBRFG* ²	Z component of VBRFG multiplied by 18	fps	m/cs
VAPFG* ²	Z component of VAPFG multiplied by 18	fps	m/cs
ABRFG*	Z component of ABRFG multiplied by 6	ft/(sec) ²	m/(cs) ²
AAPFG*	Z component of AAPFG multiplied by 6	ft/(sec) ²	m/(cs) ²
JBRFG* ³	High-gate jerk aimpoint, Z component only	ft/(sec) ³	m/(cs) ³
JAPFG* ³	Low-gate jerk aimpoint, Z component only	ft/(sec) ³	m/(cs) ³
GAINBRAK	Gain constant used in the orientation of the descent guidance system in P63	unitless	unitless
GAINAPPR	Gain constant used in the orientation of the descent guidance coordinate system in P64	unitless	unitless
TCGFBRK	Latest time guidance frame is erected in braking phase	sec	cs
TCGIBRAK	Earliest time guidance frame is erected in braking phase	sec	cs
TCGFAPPR	Latest time guidance frame is erected in approach phase	sec	cs
TCGLAPPR	Earliest time guidance frame is erected in approach phase	sec	cs
VIGN	Speed desired at ignition, relative to the rotating moon	fps	m/cs
RIGNX	X component of desired position relative to the landing site. Expressed in the descent guidance coordinate system.	ft	m
RIGNZ	Z component of desired position relative to the landing site (desired cross range component is zero)	ft	m

Table I (continued)

LGC name	Description	Engineering Units	AGC Units
KIGNX/B4	Landing site vertical error scale factor used in the ignition-time test quantity	unitless	unitless
KIGNY/B8	Cross-range error scale factor	ft/ft ²	m ⁻¹
KIGNV/B4	Speed error scale factor used in the ignition-time test quantity	sec	cs
LOWCRIT	Upper limit on the variable throttle region in a situation of decreasing thrust commands	lbf	DPS throttle pulses
HIGHCRIT	Upper limit on the variable throttle region in a situation of increasing thrust commands	lbf	DPS throttle pulses
TAUHZ	Time constant for nulling horizontal velocity in P66 auto	sec	cs
QHZ	Feedback constant for previous acceleration level	unitless	unitless
AHZLIM	Maximum horizontal acceleration in auto P66	ft/sec ²	m/cs ²
2LATE466	Maximum time after start of servicer cycle for which 2nd ROD calculation will be started	sec	cs
DELQFIX	LR data reasonableness test parameter	ft	m
LRVMAX	Maximum limit for velocity calculations that are allowed to be updated by the LR	fps	m/cs
LRVF	Value at which velocity update coefficients are changed	fps	m/cs
LRWVZ	Weighting factor for LR Z-axis velocity update	unitless	unitless
LRWVY	Weighting factor for LR Y-axis velocity update	unitless	unitless
LRWVX	Weighting factor for LR X-axis velocity update	unitless	unitless
LRWVFZ	LR velocity weighting function	unitless	unitless

Table I (continued)

LGC name	Description	Engineering Units	AGC Units
LRWVFY	Lr velocity weighting function	unitless	unitless
LRWVFX	Lr velocity weighting function	unitless	unitless
LRWVFF	Weighting factor used for P66	unitless	unitless
ABSCO-ABSC4	Range to I th segment of lunar terrain	ft	m
SLOPEO-SLOPE4	Slope of I th segment of lunar terrain model	rad	rad
RODSALE	Click scale factor for rate of descent	fps	m/cs
TAUROD	Time constant for rate of descent	sec	cs
LAG/TAU	Lag time divided by TAUROD (P66)	unitless	unitless
MINFORCE	Minimum thrust P66 will command	lbf	kg-m/(cs) ²
MAXFORCE	Maximum thrust P66 will command	lbf	kg-m/(cs) ²
J1PARM	Parameter used in the computation of insertion velocity for aborts if the LM to CSM phase angle existing at the time of abort is less than THETCRIT	ft	m
K1PARM	Same definition as J1PARM	ft/rad	m/rev
J2PARM	Parameter used in the computation of insertion velocity for aborts if the LM to CSM phase angle existing at the time of abort is greater or equal to THETCRIT	ft	m
K2PARM	Same definition as J2PARM	ft/rad	m/rev
THETCRIT	Central angle switching criteria	deg	rev
RAMIN	The minimum apolune radius allowed for aborts from powered descent	ft	m
YLIM	Maximum cross range distance in aborts	n.mi.	m
ABTRDOT	Desired radial velocity for aborts	fps	m/cs

Table I (continued)

LGC name	Description	Engineering Units	AGC Units
COSTHET1	Cosine 90°. Cosine of cone 1 angle for aborts. Used to determine direction of attitude rotation to preferred abort attitude	unitless	unitless
COSTHET2	Cosine 30°. Cosine of cone 2 angle for aborts. Used to determine direction of attitude rotation to preferred abort attitude	unitless	unitless
HIASCENT	Upper bound on the mass of the ascent stage	lb	kg
ROLLTIME	Time to drive the DPS gimbal around the R-axis, starting at the hard stop, to position it prior to PDI	sec	cs
PITTIME	Time to drive the DPS gimbal around the Q-axis, starting at the hard stop, to position it prior to PDI	sec	cs
IGNAOSQ	Initial DAP estimate of offset acceleration about the Q-axis	deg/(sec) ²	rev/(sec) ²
IGNAOSR	Initial DAP estimate of offset acceleration about the R-axis	deg/(sec) ²	rev/(sec) ²
LRHMAX	Maximum limit for altitude calculations that are allowed to be updated by the LR	ft	m
LRWH	Weighting factor for incorporation of LR altitude measurements into the LM state vector during braking phase	unitless	unitless
ZOOMTIME	Time after ignition to raise the DPS to full throttle position	sec	cs
TENDBRAK	Controls time of transition from P63 to P64	sec	cs
TENDAPPR	Controls time of transition from P64 to P66	sec	cs
DELTFAP	Time increment added to t_{go} when switching from P63 to P64	sec	cs
LEADTIME	Time increment specifying how far guidance is projected forward in P63 and P64	sec	cs

Table I (continued)

LGC name	Description	Engineering Units	AGC Units
RPCRTIME	Time from the end of the present descent guidance phase at which the IR may be repositioned to position 2	sec	cs
RPCRTQSW	Required X component of the X-body axis in platform coordinates at the time of IR reposition to position 2	unitless	unitless
TNEWA	Cycle period at which Lambert solution updates A-steer target parameters	cs	cs
LRWHL	Weighting factor for incorporation of IR altitude measurements into the IM state vector during approach phase	unitless	unitless

- 1 These values are valid for a launch date of April 16, 1972.
- 2 To achieve the proper octal value in the LGC, this number should be scaled B13.
- 3 To achieve the proper octal value in the LGC, this number should be scaled B-21.
- 4 Descartes landing site used for April 16, 1972 launch date.

FINAL E-LOAD ARTEMIS 72 APOLLO 16

	MNEMONIC	ADDRESS	ENGINEERING VALUE	AGC VALUE	WORD 3	WORD 2	WORD 1
BAC	NO.PASS + 0	0737 2		1.6383000000E+04			37777
DWH	RTEDI + 0	1342 1	1.6602637000E+00			06510	07025
HJM	DVTHRESH + 0	1344 1	2.0000000000E+00	6.0960000000E-03			
BAC	HORIZALT	1345 2		+2.8000000000E+04		00000	33260
BAC	ALTVAR	1347 2		+1.5216800000E+05			37723
HJM	EMDOT + 0	1350 1	6.5978420000E+01	2.9927308190E-01			
BAC	UNITW + 0	1711 2		.5428865552E-04		00000	34355
BAC	UNITW + 2	1713 2		.2460554242E-04		00000	14715
BAC	UNITW + 4	1715 2		.9999999963E+00		37777	37777
HJM	EIMPISEC + 0	1763 1	1.9286838000E+04	8.5792129720E+02			
HJM	EFIMPO1 + 0	1764 1	2.4195938000E+04	1.0762889446E+03			
HJM	EFIMP16 + 0	1765 1	2.0779000000E+04	9.2429614663E+02			
BAC	E3J22R2M	1766 2		+9.204790479E+16			12160
BAC	E32C31RM	1767 2		+1.312892560E+23			03363
BAC	WRENDPOS	2000 1	+1.000000000E+04	+3.048000000E+03			00137
BAC	WRENDVEL	2001 1	+1.000000000E+01	+3.048000000E-02			00763

FINAL E-LOAD ARTEMIS 72 APOLLO 16

MNEMONIC	ADDRESS	ENGINEERING VALUE	AGC VALUE	WORD 3	WORD 2	WORD 1
DAC	RMAX	2002 1 +2.000000000E+03	+6.096000000E+02			00023
DAC	VMAX	2003 1 +2.000000000E+00	+6.096000000E-03			00001
DAC	WORBPOS	2004 2	+0.000000000E+00			00000
DAC	WORBVEL	2005 2	+0.000000000E+00			00000
DAC	S22WSUBL	2006 2	+1.000000000E+04			00471
DAC	RPVAR	2007 2	+4.000000000E+06		00364	04400
DAC	504LM + 0	2011 2	-4.820525646E-05		77777	75361
DAC	504LM + 2	2013 2	.6953775883E-03		00013	14450
DAC	504LM + 4	2015 2	.2514459193E-03		00004	03651
DWH	EHSALT + 0	2017 1 2.9000010000E+05			00002	26244
DAC	RLS	2025 2	+1.653875942E+06		00311	34350
DAC	RLS +02	2027 2	+4.591698947E+05		00070	01504
DAC	RLS +04	2031 2	-2.718648303E+05		77736	72015
DAC	YIMEMO + 0	2033 2	.2572555765E+10	00011	22530	12765
DAC	VECOEM + 0	2036 2	-3.106865846E-64		77771	56461
DAC	VECOEM + 2	2040 2	.3882263800E-64		00010	05524

FINAL E-LOAD ARTEMIS 72 APOLLO 16

MNEMONIC	ADDRESS	ENGINEERING VALUE	AGC VALUE	WORD 3	WORD 2	WORD 1
VECOEM	+ 4 2042 2		.1641410656E-64		00003	16503
VECOEM	+ 6 2044 2		.6325714477E-56		00023	33241
VECOEM	+ 8 2046 2		-.1426626252E-57		77777	61527
VECOEM	+10 2050 2		.4608531752E-57		00001	16222
VECOEM	+12 2052 2		-.2267778214E-48		77765	54467
VECOEM	+14 2054 2		-.7438416423E-48		77735	46620
VECOEM	+16 2056 2		-.3828722397E-48		77756	43000
VECOEM	+18 2060 2		-.1410582067E-41		77777	40425
VECOEM	+20 2062 2		.7412106920E-40		00063	24751
VECOEM	+22 2064 2		.3644418223E-40		00031	14567
VECOEM	+24 2066 2		.7483079488E-32		00115	26534
VECOEM	+26 2070 2		-.1965742175E-31		77463	73546
VECOEM	+28 2072 2		-.9021137439E-32		77642	52154
VECOEM	+30 2074 2		-.5938236055E-23		76151	43216
VECOEM	+32 2076 2		.6002446572E-24		00134	34211
VECOEM	+34 2100 2		-.2198476457E-24		77735	77273

FINAL E-LOAD ARTEMIS 72 APOLLO 16

	MNEMONIC	ADDRESS	ENGINEERING VALUE	AGC VALUE	WORD 3	WORD 2	WORD 1
BAC	VECOEM	+36 2102	2	.2447903777E-15		01064	16257
BAC	VECOEM	+38 2104	2	.9020309289E-15		04037	36105
BAC	VECOEM	+40 2106	2	.4619401380E-15		02051	05125
BAC	VECOEM	+42 2110	2	.1194130768E-06		10007	00042
BAC	VECOEM	+44 2112	2	-.4158987110E-07		75152	77347
BAC	VECOEM	+46 2114	2	-.1011791007E-07		77244	53172
BAC	VECOEM	+48 2116	2	-.3532147765E+01		74357	61224
BAC	VECOEM	+50 2120	2	-.8139928222E+01		67670	53324
BAC	VECOEM	+52 2122	2	-.4267223120E+01		73567	45641
BAC	VECOEM	+54 2124	2	-.3764565440E+09		72307	73523
BAC	VECOEM	+56 2126	2	.1276269280E+09		01715	26726
BAC	VECOEM	+58 2130	2	.3062934400E+08		00351	25674
BAC	RES0	+ 0 2132	2	.1252304896E+12		16450	12150
BAC	RES0	+ 2 2134	2	.7652284723E+11		10721	03577
BAC	RES0	+ 4 2136	2	.3318243328E+11		03671	32360
BAC	VESO	+ 0 2140	2	-.1640953331E+03		65574	76301

FINAL E-LOAD ARTEMIS 72 APOLLO 16

	MNEMONIC	ADDRESS	ENGINEERING VALUE	AGC VALUE	WORD 3	WORD 2	WORD 1
BK	VESO	+ 2 2142 2		.2260438023E+03		16101	14665
BK	VESO	+ 4 2144 2		.9801039886E+02		06100	12514
BK	OMEGAES	+ 0 2146 2		.3131201409E-09		00530	10737
L.G.	DTF	+ 0 2354 1	3.0000000000E-01	3.0000000000E+01			
AGS	HANDELH	+ 0 2367 1	1.5000000000E+01	2.7780000000E+04			
BK	WRD TIME	2371 2		+2.4000000000E+05			00017
BK	MINBLKTM	2372 2		+3.2880000000E+04			00002
BK	TBEFCOMP	2373 2		+8.2200000000E+04			00005
BK	BRNBLKTM	2374 2		+8.2200000000E+04			00005
BK	MAXWTIME	2375 2		+3.6168000000E+05			00026
BK	FINCMPTM	2376 2		+4.9320000000E+04			00003
BK	INTVAR	2377 2		+1.9600000000E+02			00142
L.G.	LAUNCHAZ	2633 1	7.2034439090E+01	+2.000956643E-01			
BK	WMIDPOS	3000 2	+3.0000000000E+04	+9.1440000000E+03			00436
BK	WMIDVEL	3001 2	+3.0000000000E+01	+9.1440000000E-02			02732
BK	RVAR	3002 2	+0.0000000000E+00	+0.0000000000E+00		00000	00000

FINAL E-LOAD ARTEMIS 72 APOLLO 16

	MNEMONIC	ADDRESS	ENGINEERING VALUE	AGC VALUE	WORD 3	WORD 2	WORD 1	
	RVARMIN	3004	2	-4.000000000E+04	-3.716121600E+03	77777	77777	42757
DWH	LADPAD	+ 0	3007	1	2.700000000E-01			10510
DWH	LODPAD	+ 0	3010	1	2.070000000E-01			06477
DWH	ALFAPAD	+ 0	3011	1	-1.851000000E+01			74552
DWH	P37RANGE	+ 0	3012	1	1.086533000E+03			01470
HJM	ETDECAY	+ 0	3013	1	6.555000000E-01	6.555000000E+01		
HJM	PACTOFF	+ 0	3023	1	-.527000000E-00	-.2221285657E+02		
HJM	YACTOFF	+ 0	3024	1	1.898000000E+00	8.000000000E+01		
HJM	LEMMASS	+ 0	3072	1	3.620830000E+04	1.6423808773E+04		
HJM	CSMMASS	+ 0	3073	1	6.688760000E+04	3.0339705308E+04		
HJM	POLYNUM	+ 0	3261	1	5.000000000E+00	5.000000000E+00		
HJM	POLYNUM	+01	3262	1	-.9520749700E-01	-.2644652695E-03		
HJM	POLYNUM	+03	3264	1	1.9666296000E-01	5.4628600000E-06		
HJM	POLYNUM	+05	3266	1	9.5150055000E-03	2.6430570840E-09		
HJM	POLYNUM	+07	3270	1	-.1449852400E-03	-.4027367778E-12		
HJM	POLYNUM	+09	3272	1	1.9137262000E-06	5.3159061120E-17		

FINAL E-LOAD ARTEMIS 72 APOLLO 16

MNEMONIC	ADDRESS	ENGINEERING VALUE	AGC VALUE	WORD 3	WORD 2	WORD 1
HJM	POLYNUM +11 3274 1	-.1512142600E-07	-.4200396111E-21			
HJM	POLYNUM +13 3276 1	4.3469835000E-11	1.2074954170E-25			
HJM	SATRLRT + 0 3300 1	1.0000000000E+00	2.7777777778E-05			
HJM	RPSTART + 0 3302 1	1.1850000000E+01	1.1850000000E+03			
HJM	POLYSTOP + 0 3303 1	-.1470000000E+03	-.1470000000E+05			
<i>WFC</i>	HORISLP 3375 2		+0.0000000000E+00		00000	00000
<i>WFC</i>	LAT(SPL) + 0 3400 1	2.6500000000E+01	7.3611111120E-02			
<i>WFC</i>	LNG(SPL) + 0 3402 1	-.1700000000E+02	-.4722222223E-01			

FINAL E-LOAD LUMINARY 210 APOLLO 16

	MNEMONIC		ADDRESS		ENGINEERING VALUE	AGC VALUE	WORD 3	WORD 2	WORD 1
HJM	MASS	+ 0	1243	1	3.6685200000E+04	1.6640126972E+04			
HJM	LRWHI		1315	1	3.5000000000E-01				13146
HJM	LEMMASS	+ 0	1326	1	3.6685200000E+04	1.6640126972E+04			
HJM	CSHMASS	+ 0	1327	1	3.9354100000E+04	1.7850719665E+04			
HJM	E3J22R2M		1347	2		+9.204790479E+16			12160
HJM	E32C3IRM		1350	2		+1.312892560E+23			03363
HJM	TOOFEN		1354	1	3.0000000000E+00				00003
HJM	X789		1700	2		0.0000000000E+00		00000	00000
HJM	X789	+ 2	1702	2		0.0000000000E+00		00000	00000
HJM	X789	+ 4	1704	2		0.0000000000E+00		00000	00000
HJM	-AYO	+ 0	1711	2		.5428865552E-04		00000	34355
HJM	AXO	+ 0	1713	2		.2460554242E-04		00000	14715
HJM	REFSHMAY		1731	1	6.7000809000E-01			12560	26464
HJM	REFSHMAY	+02	1733	1	-5.998490400E-01			66316	41130
HJM	REFSHMAY	+04	1735	1	-4.373934700E-01			71001	51020
HJM	REFSHMAY	+06	1737	1	1.2951155000E-01			02044	36532

FINAL E-LOAD LUMINARY 210 APOLLO 16

MNEMONIC	ADDRESS	ENGINEERING VALUE	AGC VALUE	WORD 3	WORD 2	WORD 1
<i>DATA</i> REFSMMAT	+08 1741 1	-4.856443700E-01			70165	63173
<i>DATA</i> REFSMMAT	+10 1743 1	8.6450928000E-01			15652	01727
<i>DATA</i> REFSMMAT	+12 1745 1	-7.309691600E-01			64233	74643
<i>DATA</i> REFSMMAT	+14 1747 1	-6.358692600E-01			65646	76540
<i>DATA</i> REFSMMAT	+16 1751 1	-2.476981900E-01			74022	73317
<i>DATA</i> RANGEVAR	1766 2		+1.111111111E-05		01351	24734
<i>DATA</i> RATEVAR	1770 2		+1.877777778E-05		02354	04761
<i>DATA</i> RVARMIN	1772 2		+6.600000000E+01			00410
<i>DATA</i> VVARMIN	1773 2		+1.744500000E-06			00165
<i>DATA</i> WRENDPOS	2000 2		+3.048000000E+03			05750
<i>DATA</i> WRENDVEL	2001 2		+0.304800000E-01			00763
<i>DATA</i> WSHAFT	2002 2		+0.015000000E+00			17270
<i>DATA</i> WTRUN	2003 2		+0.015000000E+00			17270
<i>DATA</i> RMAX	2004 2		+6.096000000E+02			00023
<i>DATA</i> VMAX	2005 2		+0.006096000E+00			00001
<i>DATA</i> WSURFPOS	2006 1	+0.000000000E+00	+0.000000000E+00			00000

FINAL E-LOAD LUMINARY Z10 APOLLO 16

MNEMONIC	ADDRESS	ENGINEERING VALUE	AGC VALUE	WORD 3	WORD 2	WORD 1
<i>ATC</i> WSURFVEL	2007 1	+0.000000000E+00	+0.000000000E+00			00000
<i>ATC</i> SHAFTVAR	2010 2		+1.000000000E-06			00103
<i>ATC</i> TRUNVAR	2011 2		+1.000000000E-06			00103
<i>ATC</i> 504LM + 0	2012 2		-.4820525646E-05		77777	75361
<i>ATC</i> 504LM + 2	2014 2		.6953775883E-03		00013	14450
<i>ATC</i> 504LM + 4	2016 2		.2514459193E-03		00004	03651
<i>ATC</i> RLS	2025 <u>2</u> 2020		+1.653875942E+06		00311	34350
<i>ATC</i> RLS +02	2027 <u>2</u> 2022		+4.591698947E+05		00070	01504
<i>ATC</i> RLS +04	2031 <u>2</u> 2024		-2.718648303E+05		77736	72015
<i>ATC</i> TLAND	2026 2		3.5560224000E+07		04172	15440
<i>ATC</i> VELBIAS	2400 1	2.500000000E+00			00001	36331
<i>ATC</i> RBRFGX	2402 1	-3.118358800E+03			77777	42227
<i>ATC</i> RAPFGX	2404 1	1.585000000E+02			00000	01405
<i>ATC</i> RBRFGZ	2406 1	-1.174144100E+04			77774	60122
<i>ATC</i> RAPFGZ	2410 1	-2.735540000E+01			77777	77572
<i>ATC</i> VBRFGX	2412 1	-1.964691600E+02			77766	55311

FINAL E-LOAD LUMINARY 210 APOLLO 16

	MNEMONIC	ADDRESS	ENGINEERING VALUE	AGC VALUE	WORD 3	WORD 2	WORD 1
<i>Jha</i>	VAPFGX	2414	1	-3.534760000E+00		77777	72367
<i>Jha</i>	VBRFGZ	2416	1	-1.667599700E+02		77767	73603
<i>Jha</i>	VAPFGZ	2420	1	2.49505000000E-02		00000	00024
<i>Jha</i>	ABRFGX	2422	1	-7.182481400E-01		77772	50265
<i>Jha</i>	AAPFGX	2424	1	7.71783000000E-02		00000	23567
<i>Jha</i>	ABRFGZ	2426	1	-8.302450300E+00		77675	65140
<i>Jha</i>	AAPFGZ	2430	1	-5.896270000E-01		77773	51173
<i>Jha</i>	VBRFG*	2432	1	-3.001680000E+03		77755	66351
<i>Jha</i>	VAPFG*	2434	1	4.49109000000E-01		00000	00055
<i>Jha</i>	ABRFG*	2436	1	-4.981470200E+01		77161	77106
<i>Jha</i>	AAPFG*	2440	1	-3.537762000E+00		77743	67344
<i>Jha</i>	JBRFG*	2442	1	-1.512365900E-02		77541	63456
<i>Jha</i>	JAPFG*	2444	1	4.31736000000E-02		00704	04643
<i>Jha</i>	GAINBRAK	2446	1	9.999999963E-01		37777	37777
<i>Jha</i>	GAINAPPR	2450	1	0.000000000E+00		00000	00000
<i>Jha</i>	TCGFBRAK	2452	1	3.000000000E+01			00567

FINAL E-LOAD LUMINARY 210 APOLO 16

MEMORIC	ADDRESS	ENGINEERING VALUE	AGC VALUE	WORD 3	WORD 2	WORD 1
JJA	TCGIBRAK	2453 1 9.0000000000E+02				25762
JJC	TCGFAPPR	2454 1 6.0000000000E+00				00113
JJK	TCGIAPPR	2455 1 2.0000000000E+02				04704
JJL	VIGN	2456 1 5.5434605000+<03			00416	12774
JJM	RIGNX	2460 1 -1.595487200E+05			77720	60230
JJN	RIGNZ	2462 1 -1.547623300E+06			77063	52706
JJO	RIGNX/B4	2464 1 -3.340000000E-01			77251	77371
JJP	RIGNY/B8	2466 1 -2.207000000E-07			76366	60601
JJQ	RIGNV/B4	2470 1 -4.980000000E+02			71727	57777
JJR	IOWCRIT	2472 1 5.9850000000E+03				04114
JJS	HIGHCRIT	2473 1 6.6150000000E+03				04454
JJT	TAUH7	2474 1 5.0000000000E+00				07640
JJU	OH7	2475 1 4.0000000000E-01				14632
JJV	AF7LIM	2476 1 1.9389800000E+00				00017
JJW	PLATE456	2477 1 1.5000000000E+00			00000	00226
JJX	DELQFIX	2503 1 1.0000000000E+02			00000	00750

FINAL E-LOAD LUMINARY 210 APOLLO 16

MNEMONIC	ADDRESS	ENGINEERING VALUE	AGC VALUE	WORD 3	WORD 2	WORD 1
<i>Jha</i> LRVMAX	2511	1	2.5000000000E+03			01717
<i>Jha</i> LRVF	2512	1	2.0000000000E+02			00116
<i>Jha</i> LRWVZ	2513	1	3.0000000000E-01			11463
<i>Jha</i> LRWVY	2514	1	3.0000000000E-01			11463
<i>Jha</i> LRWVX	2515	1	3.0000000000E-01			11463
<i>Jha</i> LRWVFZ	2516	1	2.0000000000E-01			06315
<i>Jha</i> LRWVFY	2517	1	2.0000000000E-01			06315
<i>Jha</i> LRWVFX	2520	1	2.0000000000E-01			06315
<i>Jha</i> LRWVFF	2521	1	1.0000000000E-01			03146
<i>Jha</i> ABSCO	2522	1	-6.9200000000E+05			46200
<i>Jha</i> ABSC1	2523	1	-5.2400000000E+05			54401
<i>Jha</i> ABSC2	2524	1	-2.3400000000E+05			67225
<i>Jha</i> ABSC3	2525	1	-1.6400000000E+05			71713
<i>Jha</i> ABSC4	2526	1	-1.0000000000E+05			74216
<i>Jha</i> SLOPE0	2527	1	9.8214280000E-02			00031
<i>Jha</i> SLOPE1	2530	1	-5.1724100000E-03			77776

FINAL E-LOAD LUMINARY 210 APOLLO 16

MNEMONIC	ADDRESS	ENGINEERING VALUE	AGC VALUE	WORD 3	WORD 2	WORD 1
<i>JHA</i> SLOPE2	2531	1	-6.428571000E-02			77757
<i>JHA</i> SLOPE3	2532	1	1.562500000E-02			00004
<i>JHA</i> SLOPE4	2533	1	0.000000000E+00			00000
<i>JHA</i> RODSCALE	2534	1	1.000000000E+00			14370
<i>JHA</i> TAUR0D	2535	1	1.500000000E+00		11300	00000
<i>JHA</i> LAG/TAU	2537	1	2.333300000E-01		07356	34075
<i>JHA</i> MINFORCE	2541	1	9.800000000E+02		00001	27631
<i>JHA</i> MAXFORCE	2543	1	6.300000000E+03		00013	06551
<i>f.m.g.</i> JIPARM + 0	2545	1	6.051373000E+06	1.844458490E+06		
<i>f.m.g.</i> KIPARM + 0	2547	1	-6.055552000E+05	-1.159707775E+06		
<i>f.m.g.</i> JZPARM + 0	2551	1	6.0383801000E+06	1.8404982540E+06		
<i>f.m.g.</i> KZPARM + 0	2553	1	-3.169623600E+05	-6.070193324E+05		
<i>f.m.g.</i> THETCRIT + 0	2555	1	5.6304007300E+00	1.5640002029E-02		
<i>f.m.g.</i> RAMIN + 0	2557	1	5.8838246720E+06	1.7933897600E+06		
<i>JHA</i> YLIM	2561	1	8.200000000E+00		00016	32446
<i>JHA</i> ABYRDOT	2563	1	1.950000000E+01		00007	23346

FINAL E-LOAD LUMINARY 210 APOLLO 16

MNEMONIC	ADDRESS	ENGINEERING VALUE	AGC VALUE	WORD 3	WORD 2	WORD 1
<i>JHA</i> COSTHET1	2565	1	0.0000000000E+00		00000	00000
<i>JHA</i> COSTHET2	2567	1	8.6602540370E-01		06733	07535
HJM HIASCENT + 0	3000	1	1.0900000000E+04	4.9441568330E+03		
HJM ROLLTIME + 0	3001	1	3.2300000000E+01	3.2300000000E+03		
HJM PIYTIME + 0	3002	1	3.2410000000E+01	3.2410000000E+03		
HJM IGNAOSQ + 0	3012	1	5.4420000000E+00	1.5116666667E-02		
HJM IGNAOSR + 0	3013	1	9.4000000000E-02	2.6111111111E-04		
<i>JHA</i> LRHMAX	3420	1	5.0000000000E+04			35610
<i>JHA</i> LRWH	3421	1	3.5000000000E-01			13146
<i>JHA</i> ZOOMTIME	3422	1	2.6000000000E+01			05050
<i>JHA</i> TENDBRAK	3423	1	6.2000000000E+01			01407
<i>JHA</i> TENDAPPR	3424	1	1.2000000000E+01			00226
<i>JHA</i> DELTTFAP	3425	1	-7.0000000000E+01			76224
<i>JHA</i> LEADTIME	3426	1	-2.2000000000E+00			77743
<i>JHA</i> RPCRTIME	3427	1	6.2000000000E+01			01407
<i>JHA</i> RPCRTQSW	3430	1	-1.0000000000E+00			57777

FINAL E-LOAD LUMINARY 210 APOLLO 16

MNEMONIC	ADDRESS	ENGINEERING VALUE	AGC VALUE	WORD 3	WORD 2	WORD 1
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HJM INEWA	* 0 3431 1	1.3421772800E+08	1.3421772800E+08			
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LM STATE VECTOR AT LOI IGNITION

L.G. X		-2830280.9 FT				
L.G. Y		4728937.6 FT				
L.G. Z		3008921.6 FT				
L.G. DX		7697.7982 FPS				
L.G. DY		2452.4478 FPS				
L.G. DZ		636.01637 FPS				
L.G. T		74 HR 28 MIN 38.554 SEC G.E.T				