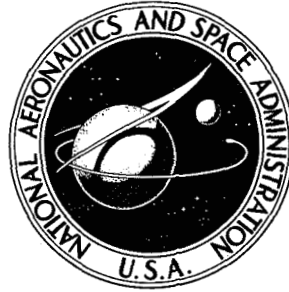


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GENENG — A PROGRAM FOR CALCULATING  
DESIGN AND OFF-DESIGN PERFORMANCE  
FOR TURBOJET AND TURBOFAN ENGINES

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16. Abstract A computer program titled GENENG is described. The program uses component performance maps to enable the user to do analytical steady-state engine cycle calculations. Through a scaling procedure, each of the component maps can be used to represent a family of maps (different design values of pressure ratios, efficiency, weight flow, etc.). Either convergent or convergent-divergent nozzles may be used. Included is a complete FORTRAN IV listing of the program. Sample results and input explanations are shown for one-spool and two-spool turbojets and two-spool separate- and mixed-flow turbopfans operating at design and off-design conditions. The computer program is available from the authors.			
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# GENENG - A PROGRAM FOR CALCULATING DESIGN AND OFF-DESIGN PERFORMANCE FOR TURBOJET AND TURBOFAN ENGINES

by Robert W. Koenig and Laurence H. Fishbach

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## SUMMARY

A digital computer program titled GENENG is described. The original version of the computer program is titled SMOTE (SiMulation Of Turbofan Engine) and was developed by the Turbine Engine Division of the Air Force Aero Propulsion Laboratory, Wright-Patterson Air Force Base, Ohio. SMOTE is capable of calculating only turbofan design and off-design performance using specific component performance maps. GENENG calculates steady-state design and off-design turbofan and one- and two-spool turbojet engine performance. Discussed in this report are changes to SMOTE which enable the user to do jet engine calculations of a general nature, thereby allowing the mission analyst a greater freedom in selection of engine design characteristics. Typical of these changes is an automatic redesign of fan and compressor pressure ratios when the static pressures do not balance at the mix point for mixed-flow turbofans. A convergent-divergent nozzle performance map addition is shown. Other changes that permit greater flexibility for generalized cycle studies for GENENG are described.

Included as an appendix to the report is a complete FORTRAN IV listing of GENENG. Sample results and input explanations are included for two-spool turbojet, one-spool turbojet, and two-spool turbofan engines operating at design and off-design conditions.

## INTRODUCTION

For preliminary as well as in-depth studies it is often necessary to study a broad range of engines operating at both design and off-design conditions in order to find an efficient airframe/engine combination. The spectrum of flight conditions through which an engine must operate will strongly affect the optimum design parameter for that engine.

The SMOTE code (SiMulation Of Turbofan Engines), discussed in references 1 and 2,

provided a computer program having off-design-point calculation capability for either existing engines or theoretical ones - a major advance. Theoretical engines are simulated by scaling component performances from existing engines to the design conditions of the theoretical engines.

GENENG (GENeralized ENGine), a computer code derived from SMOTE, was written to improve the versatility of SMOTE. Among the changes made are as follows:

- (1) One- and two-spool turbojets can be calculated, as well as turbofans.
- (2) Afterburner performance maps can be used.
- (3) Nozzle performance maps can be used.
- (4) Fan and compressor pressure ratios are automatically redesigned for mixed-flow turbofans if the static pressures at the mix point do not match.
- (5) Duct combustor pressure losses are calculated.
- (6) A new method of entering data into the program is used.

A derivative program from GENENG, called GENENG II, is reported in a companion report to this one (ref. 3). GENENG II calculates performance of two- or three-spool front- or aft-fan turbofan engines with as many as three nozzles (or airstreams).

These programs have proven to be very versatile, and minor changes to them can greatly increase the number of engine configurations that can be studied. As an example, GENENG was used to study nuclear-powered turbofan engines. This was accomplished by adding a heat-exchanger subroutine as a substitute for the combustor subroutine to simulate the use of nuclear power to raise the temperature of the air entering the first turbine.

In this report are included illustrative examples of the use of GENENG to study various one- and two-spool turbojets and separate- and mixed-flow turbofans.

GENENG is available from the authors upon request. This FORTRAN IV program can be used by computing centers having an IBM 7094 Mod 2 computer. With modifications, the program can be used on all machines that have a FORTRAN compiler.

## THERMODYNAMIC ANALYSIS OF ENGINE TYPES

All thermodynamic properties of air and gas are calculated by considering variable specific heats and no dissociation. The air and gas property tables of reference 4 were curve fit and are used herein.

The following discussion presents the thermodynamic analysis of the engine cycles that can be studied using GENENG.

## Two-Spool Afterburning Turbofan

The basic engine, a two-spool turbofan, is shown in figure 1. All other engine types are treated as variations of this basic engine. Free-stream conditions exist at station 1 and are determined by using the U. S. Standard Atmosphere Table of 1962 (ref. 5). The conditions at station 2 are determined by flight speed and inlet recovery.

GENENG compressor maps work with corrected values of airflow. At the entrance to the fan, the corrected airflow  $WA_{F, c}$  is

$$WA_{F, c} = \frac{WA_F \sqrt{T_2/518.668}}{P_2/P_{SLS}} \quad (1)$$

where  $P_2$  and  $P_{SLS}$  are in atmospheres and  $P_{SLS}$  equals 1.0. All symbols are defined in appendix C. Some symbols are formed as the combination of other symbols; thus WA is airflow, F is for fan, and c when following a compound symbol means corrected. Station numbers are defined on the appropriate figure.

All the fan air  $WA_F$  is compressed by the fan giving rise to conditions at station 21. The power required to do this is

$$\text{Fan power} = WA_F \times (H_{21} - H_2) \quad (2)$$

Some fan air may be lost to the cycle as fan bleed  $Bl_F$ , which is expressed as a fraction of the fan airflow

$$Bl_F = PC_{Bl, F} \times WA_F \quad (3)$$

The corrected airflow into the core compressor is

$$WA_{C, c} = \frac{WA_C \sqrt{T_{21}/518.668}}{P_{21}/1.0} \quad (4)$$

The remaining air goes through the fan duct where some leakage from the core air may also enter (see eq. (11)).

$$WA_D = WA_F - Bl_F - WA_C + Bl_{DU} \quad (5)$$

The air which may be heated by a duct burner to a temperature  $T_{24}$  undergoes a pressure drop

$$P_{25} = P_{24} \times \left[ 1 - \left( \frac{\Delta P}{P} \right)_{\text{DUCT}} \right] \quad (6)$$

The air would have been heated by the addition of fuel, which can be expressed as a fuel-air ratio so that

$$WG_{24} = WA_{23} \times \left[ 1 + (f/a)_{23} \right] \quad (7)$$

The gas is then either expanded through a nozzle (station 29) to produce thrust or is mixed with the core air as shown in figure 2 (mixed-flow turbofan). The bypass ratio of the engine is defined by

$$\text{BYPASS} = \frac{WA_D}{WA_C} \quad (8)$$

The air continuing into the core is compressed to conditions at station 3. The power required is

$$\text{Compressor power} = WA_C \times (H_3 - H_{21}) = WA_3 \times (H_3 - H_{21}) \quad (9)$$

Some core bleed air  $Bl_C$  may be used for turbine cooling. Some of the air is put back into the cycle into each of the two turbines, and some is lost to the cycle as overboard bleed or leakage into the fan duct.

$$Bl_C = PC_{Bl, C} \times WA_3 \quad (10)$$

$$Bl_{DU} = PC_{Bl, DU} \times Bl_C \quad (11)$$

$$Bl_{OB} = PC_{Bl, OB} \times Bl_C \quad (12)$$

$$Bl_{HP} = PC_{Bl, HP} \times Bl_C \quad (13)$$

$$Bl_{LP} = PC_{Bl, LP} \times Bl_C \quad (14)$$

Since  $Bl_{DU} + Bl_{OB} + Bl_{HP} + Bl_{LP} = Bl_C$ , the sum of  $PC_{Bl, DU} + PC_{Bl, OB} + PC_{Bl, HP} + PC_{Bl, LP}$  must be equal to 1.0.

The remaining air is

$$WA_4 = WA_3 - Bl_C \quad (15)$$

and is heated to a turbine inlet temperature  $T_4$  while undergoing a combustor pressure drop  $(\Delta P/P)_{\text{COMB}}$ . The fuel required to do this is expressed as a fuel-air ratio  $(f/a)_4$  so that the weight of the gas entering the first (high pressure) turbine  $WG_4$  can be expressed as

$$WG_4 = WA_4 \times \left[ 1 + (f/a)_4 \right] \quad (16)$$

This gas is then expanded through the turbine to conditions at station 5. The enthalpy at station 5 is first calculated by making a power balance, since this turbine drives the compressor and supplies any work extracted (HPEXT). By using equation (9),

$$WG_4 \times (H_4 - H_5) = WA_3 \times (H_3 - H_{21}) + \text{HPEXT} \quad (17)$$

In addition, the physical speeds must match

$$N_{\text{HP, TURBINE}} = N_{\text{COMP}} \quad (18)$$

If high-pressure-turbine bleed air  $Bl_{\text{HP}}$  is added back into the cycle at this point,  $H_5$  must be readjusted.

$$H_5 = \frac{(Bl_{\text{HP}} \times H_3) + WG_4 H_5}{WG_4 + Bl_{\text{HP}}} = \frac{(Bl_{\text{HP}} \times H_3) + WG_4 H_5}{WG_5} \quad (19)$$

Similarly,

$$WG_5 \times (H_5 - H_{55}) = WA_{\text{F}} \times (H_{21} - H_2) \quad (20)$$

$$N_{\text{LP, TURBINE}} = N_{\text{FAN}} \quad (21)$$

$$H_{55} = \frac{(Bl_{\text{LP}} \times H_3) + WG_5 H_{55}}{WG_5 + Bl_{\text{LP}}} = \frac{(Bl_{\text{LP}} \times H_3) + WG_5 H_{55}}{WG_{55}} \quad (22)$$

For non-mixed-flow turbofans, the gas flow at station 6,  $WG_6$ , is identical with that at station 55,  $WG_{55}$ . For mixed-flow turbofans, the air in the fan duct is added.

$$WG_6 = WG_{55} + WA_{\text{D}} \quad (23)$$

Mixed-flow turbofans additionally require that the static pressures at station 25 and at station 55 (fig. 2) match.

$$PS_{55} = PS_{25} \quad (24)$$

The gas flow  $WG_6$  then may be heated by an afterburner to a gas temperature  $T_7$  and may undergo a pressure drop.

$$P_7 = P_6 \left[ 1 - (\Delta P/P)_{\text{AFTERBURNER}} \right] \quad (25)$$

And the gas flow rate would be increased by any fuel burned.

$$WG_7 = WG_{55} + WFA$$

The gas is then expanded through the nozzle (station 9) to produce the remainder of the engine thrust.

### Two-Spool Turbojet

The two-spool turbojet is equivalent to a two-spool turbofan with a BYPASS of zero. This engine is shown in figure 3. In calculating this type of engine, there is no fan duct and the air entering the inner compressor is the same as the air entering the inlet less any bleed.

$$WA_C = WA_F - Bl_F \quad (26)$$

The thermodynamic calculations proceed identically to the previous case, the two-spool turbofan case, except that any equations referring to the fan duct are eliminated.

### One-Spool Turbojet

The one-spool turbojet is shown in figure 4. As can be seen, to simulate this engine the inner compressor and its driving turbine are eliminated. That is, stations 21 and 3 become identical and stations 4 and 5 become identical.

The only calculation changes required therefore are (1) to eliminate any thermodynamic equations relating to the fan duct and the inner spool of the two-spool turbofan engine and (2) to add the horsepower extracted to the power requirements of the outer turbine.

## BALANCING TECHNIQUE

An off-design engine cycle calculation requires satisfying various matching con-

straints (rotational speeds, airflows, compressor and turbine work functions, and nozzle flow functions) at each specified operating condition. GENENG internally searches for compressor and turbine operating points that will satisfy the constraints. It does this by generating differential errors caused by small changes in the independent variables. The program then uses a matrix that is loaded with the differential errors to solve for the zero-error condition. This procedure is known as the Newton-Raphson iteration technique.

For the two-spool turbofan or turbojet engines a solution for a set of six simultaneous linear algebraic equations is obtained; for the one-spool turbojet a set of three simultaneous linear equations is solved. The six independent variables selected are

ZF            Ratio of pressure ratios of outer compressor (fan) along a speed line,  

$$ZF = \frac{(\text{Pressure ratio along speed line}) - (\text{Low pressure ratio on speed line})}{(\text{High pressure ratio on speed line}) - (\text{Low pressure ratio on speed line})}$$

PCNF        Percent fan speed or turbine inlet temperature  
 or T4

ZC            Ratio of pressure ratios of inner compressor along speed line; calculated  
 same as ZF

PCNC        Percent compressor speed or turbine inlet temperature  
 or T4

TFFHP       Inner (high pressure) turbine flow function,  $WG_4 \sqrt{T_4/P_4}$

TFFLP       Outer (low pressure) turbine flow function,  $WG_5 \sqrt{T_5/P_5}$

ZC, PCNC, and TFFHP are not used for the one-spool turbojet.

The program initially selects new (perturbed) values for the variables, based on the design values. It is then possible to proceed through the entire engine cycle, where six (or three) errors are generated. The initial values of the six (or three) variables and six (or three) errors are base values.

As per reference 1, the partial differential equations for  $E = f(v)$  are

$$dE_i = \sum_{j=1}^{j_{\max}} \frac{\partial E_{ij}}{\partial V_j} dV_j \quad (27)$$

for  $i$  going from 1 to  $j_{\max}$  where  $j_{\max}$  is 6 for two-spool engines or 3 for one-spool turbojets;  $E$  is an error;  $V$  is a variable; and  $\partial E_{ij}$  is the change in  $E_i$  caused by a change in  $V_j$ .

The assumption of a small change in the variable results in the following approxima-



tions (B refers to a base value):

$$dE = E - EB \quad (28)$$

$$dV = V - VB \quad (29)$$

$$\frac{\partial E}{\partial V} = \frac{E}{V} \quad (30)$$

With these approximations and the knowledge that E should equal zero for the balanced engine, the set of partial differential equations (eq. (27)) reduces to

$$E_i - EB_i = \sum_{j=1}^{j_{\max}} \frac{\Delta E_{ij}}{\Delta V_j} dV_j = -EB_i \quad (31)$$

for i going from 1 to  $j_{\max}$ .

Thus the calculations made with the perturbed variables are used to compute  $\Delta E/\Delta V$ , and equation (31) is solved for  $dV_j$ . The variables V are then given new values from

$$V_j = V_j B + dV_j \quad (32)$$

If the engine cycle calculations were linear functions, the engine would balance (errors within some allowable limits) with the new values of the variables. However, this is not the case, and it is usually necessary to repeat the process of changing each variable by a small amount for each pass. A change in each error because of the small change in the variable is calculated for each pass, where the new values become base values. This process occurs several times before a balance is obtained.

A subroutine (MATRIX) to determine the solution of a matrix is used to solve the set of differential equations. After each pass through the engines, a matrix array is loaded with the appropriate values; after a number of passes equal to  $1 + j_{\max}$  (base value plus perturbation on each of the independent variables), the matrix subroutine is used to solve the matrix. This solution (E within some allowable limit) yields the correct values of the independent variable and satisfies all the component matching constraints.

The most-often-used independent variable and the differential errors for four types of engines that can be run on GENENG are listed in table I.

## SUBROUTINES AND INPUTS FOR ENGINE PERFORMANCE CALCULATIONS

Two forms of data are supplied to GENENG. Some data, such as all the constants and component map data, are in the form of a BLOCK DATA subprogram. The varying data are supplied at execution time by the use of input data cards.

The FORTRAN listings of GENENG are presented in appendix A. The function and description of the subroutines follows in the next section.

### GENENG Subroutine Functions and Descriptions

A flow chart of the computer program with the subroutines is shown in figure 5. The functions of the subroutines are listed here and the purpose of each described.

GENENG	Dummy main program to initiate the calculations and cause the input of the controlled output variables. Because of the looping between subroutines, control is never transferred back to this routine.
ENGBAL	Main routine. Controls all engine balancing loops; checks tolerances and number of loops and loads matrix; calls input.
GUESS	Determines initial values of independent variable (see table I) at each point.
MATRIX	Solves error matrix.
PUTIN	Calls input subroutine package. Controls loop on static pressures for mixed-flow turbofan.
ZERO	Zeros nearly all of common and certain controls.
COINLT	Determines ram recovery and performs inlet calculations.
ATMOS	1962 U. S. Standard Atmosphere Table.
RAM	Calculates ram recovery defined by MIL-E-5008B specifications.
RAM2	Calculates special cases of input ram recovery as a function of flight Mach number.
COFAN	Uses BLOCK DATA to perform outer-compressor (fan) calculations.
COCOMP	Uses BLOCK DATA to perform inner-compressor calculations (two spools only).

COCOMB Uses BLOCK DATA to perform combustor calculations. May use either T4 or WFB as the main parameter.

COHPTB Uses BLOCK DATA to perform inner-turbine calculations (two spools only).

COLPTB Uses BLOCK DATA to perform outer-turbine calculations.

CODUCT Performs duct and duct-burning calculations for turbofans. May use either T24 or WFD as main parameters.

COMIX Performs gas mixing calculations if in mixed-flow mode. At design points it calculates areas either from an input static pressure PS55 or from an input Mach number AM55 if PS55 = 0. At off-design points it calculates static pressures and Mach numbers from the design areas. Calculates ERR (5). Rescales pressure ratios for mixed-flow turbofans to match duct and core static pressures just prior to mixing. COMIX also calculates afterburner entrance area A6 as a function of afterburner entrance Mach number AM6.

COAFBN Performs afterburning calculations. May use either T7 or WFA as the main parameters.

FRTOSD Dummy routine to transfer values from common FRONT to common SIDE.

FASTBK Dummy routine to transfer values from common SIDE to common BACK.

COMNOZ Controls the main nozzle.

ERROR Controls all printouts if an error occurs. Prints names of subroutine where error occurred and also prints the values of all variables in the main commons.

SYG Controls printing from UNITO8. Throughout the program and particularly in ENGBAL, certain messages, variables, and matrix values are written on UNITO8 as an aid in determining why an error occurred or why a point did not balance. These values are printed out if subroutine ERROR is called and IDUMP is greater than zero, or after a good point if IDUMP = 2.

PERF Calculates performance after the engine is balanced.

OUTPUT Prints output except for controlled output. Prints the main commons after the design point.

CONOUT Controls and prints the controlled output variables.

THCOMP Performs isentropic calculations for compressors.

PROCOM Calculates thermodynamic gas properties for either air or a fuel-air mixture based on JP-4 using curve fits of the tables of reference 4.

SEARCH General table lookup and interpolation routine to obtain data from the BLOCK DATA subroutines.

MAPBAC Used when calculations result in values not on the turbine maps. Changes the map value and an independent variable (PCNF, PCNC, or T4) in an attempt to rectify the situation.

CONVRG Performs nozzle calculations for a convergent nozzle.

CONDIV Performs nozzle calculations for a convergent-divergent (C-D) nozzle.

THTURB Performs isentropic calculations for turbines.

THERMO Provides thermodynamic conditions using PROCOM.

AFQUIR General quadratic interpolation routine.

PARABO Parabolic curve-fit routine.

BLKFAN Performance data for outer compressor (fan) map (BLOCK DATA).

BLKCMP Performance data for inner compressor map (BLOCK DATA; two-spool engines).

CMBDAT BLOCK DATA for combustor.

HPTDAT Performance data for inner turbine map (BLOCK DATA; two-spool engines).

LPTDAT Performance data for outer turbine map (BLOCK DATA).

ETAAB Generalized afterburner performance BLOCK DATA as a function of fuel-air ratio with correction factors for off-design afterburner entrance pressure and Mach number.

FRATIO Convergent-divergent nozzle velocity coefficient (BLOCK DATA input as a function of nozzle pressure ratio and area expansion ratio).

INPUT Package of Huff input subroutines. (The Huff Input Routine is a very versatile input mechanism further detailed in appendix B.)

## Entering the Data

The Huff Input Routine, used to enter input data into the program at execution time, is discussed in appendix B. Appendix C presents the individual symbols internal to the program, station numbers, etc., from which compound names such as WAFCD (WA + F + C + DS) are formed. Table II and appendix C present the names of the variables, the values of which are supplied on data cards.

Choice of component maps - scaling laws. - Many of the engines that are studied using GENENG are theoretical. Therefore, actual component maps for these engines will be nonexistent. The program, however, does require component maps to do off-design calculations. To alleviate this problem, GENENG uses scaling laws to change data from one component map into a new component map. Hopefully, a component map can be found which could be expected to perform in a similar manner to the actual map for the engine type being studied. In fact, several maps that the authors have obtained are identified as to the range of pressure ratio and the engine component design type for which they are valid (i. e., pressure ratio range, 4 to 8; subsonic compressor or inner compressor). Thus, for example, a high-bypass-ratio, subsonic-flight-speed, low-pressure-ratio fan map for a CF6 engine would not properly simulate a low-bypass-ratio, high-pressure-ratio, supersonic multistage fan.

The scaling equations used for the compressor maps are

$$PR = \frac{PR_{\text{design}} - 1}{PR_{\text{map, design}} - 1} (PR_{\text{map}} - 1) + 1$$

$$WA = \frac{WA_{\text{design}}}{WA_{\text{map, design}}} \times WA_{\text{map}}$$

$$ETA = \frac{ETA_{\text{design}}}{ETA_{\text{map, design}}} \times ETA_{\text{map}}$$

Similar equations are used for combustor and turbine map scaling. These equations are found in the appropriate subroutines. In the output are printed the correction factors used in scaling the maps. The closer these values are to 1.0 (especially pressure ratio, a primary characteristic of a given compressor map), the more reasonable are the simulated maps of the engine. Conversely, however, not being close to 1.0 does not necessarily mean that the simulation is poor since many maps have been shown to be typical over quite a large range of variables.

BLOCK DATA input. - The two compressor performance maps are entered into the

code as the BLOCK DATA subprograms BLKFAN and BLKCMP. The subprograms supplied by the authors with the code and shown in appendix A are not to be taken as realistic maps. These maps are only of an illustrative nature and are the ones used to run the sample calculations.

Using subprogram BLKFAN as an example (the first nine cards of which are printed here) and referring to a typical compressor map (fig. 6), the data are programmed as follows:

```

$IBFTC BLKFAN DECK
C THIS IS A GENERALIZED FAN MAP FOR UNREALISTIC SUPERSONIC ENGINE 1
  BLOCK DATA 2
  COMMON / FAN/CN(15),PR(15,15),WAC(15,15),ETA(15,15),N,NP(15) 3
  DATA N,NP/10,6,3*7,5*10,8,5*0/ 4
  DATA CN/0.3,0.4,0.5,0.6,0.7,0.8,0.9,1.0,1.1,1.2,5*0./ 5
  DATA (PR( 1,J),WAC( 1,J),ETA( 1,J),J=1, 6)/ 6
  1 1.03000, 243.600, 0.75592, 1.01200, 229.800, 0.76120, 7
  2 1.02800, 199.800, 0.76648, 1.03840, 166.800, 0.75592, 8
  3 1.04480, 133.200, 0.72512, 1.04800, 86.400, 0.64152/ 9

```

Card 1 reminds the reader that these maps are fictitious. Card 2 identifies subprogram as BLOCK DATA. Card 3 identifies common block FAN into which data are to be stored and dimensions the program variables. Card 4 indicates that there are 10 speed lines (N) and the number of points NF on each line - six on the lowest speed, seven on the next three lines, etc. Card 5 assigns the value of speed to each of the 10 lines (low to high). Cards 6 to 9 along speed line CN=0.3 set pressure ratio PR, corrected airflow WAC, and efficiency ETA in sets of three going from low pressure (PR = 1.0) to the surge line (PR = 1.048). Note there are six sets of three values (NP(1) = 6). The rest of the cards (appendix A) set the values for each speed line.

The combustor map is also a BLOCK DATA subprogram (CMBDAT). It is a plot of temperature rise across the combustor against efficiency for constant input pressure. Entry to the map is through temperature rise and input pressure with efficiency being output. The cards in the subprogram CMBDAT are reproduced here; a typical combustor map is shown in figure 7.

```

$IBFTC CMBDAT DECK
  BLOCK DATA 1
  COMMON / COMB/PSI(15),DELT(15,15),ETA(15,15),N,NP(15) 2
  DATA N,NP / 15,15*15 / 3
  DATA PSI/4.9116,9.8232,14.735,19.646,24.558,29.470,34.381, 4
  139.293,44.207,73.674,100.,200.,300.,400.,500./ 5
  DATA DELT/15*200.,15*300.,15*400.,15*500.,15*600.,15*700.,15*800., 6
  115*900.,15*1000.,15*1100.,15*1200.,15*1300.,15*1400.,15*1500., 7
  215*1600./ 8
  DATA ETA/ 9
  1.600,.726,.777,.806,.826,.843,.855,.865,7*.870, 10
  2.758,.825,.858,.875,.888,.898,.906,.912,.914,6*.915, 11
  3.868,.893,.911,.925,.935,.942,.947,.951,7*.953, 12
  4.925,.936,.946,.955,.963,.969,.974,.977,.978,6*.979, 13
  5.960,.966,.972,.977,.982,.985,.990,.992,.993,6*.995, 14
  6.988,.991,.992,.994,.995,.997,.998,8*.999, 15
  78*1.03,7*.999,120*1.00/ 15
  END 17

```

Card 1 identifies subprogram as BLOCK DATA. Card 2 identifies common block COMB into which data are to be stored and the dimension of each variable. Card 3 indicates that there are 15 lines of constant PSI(P3) by the value of N, and that there are 15 values of DELT(DT) and ETA(ETAB) along each line of constant PSI(P3). Cards 4 and 5 assign values to each of the P3 lines from low to high pressure. Cards 6 to 8 assign values of  $\Delta T$  to each of the P3 lines starting at low  $\Delta T$ . The lowest value of  $\Delta T$  on each of the P3 lines is given starting with the lowest value of  $\Delta T$  on the lowest value of P3. Next comes the second lowest value of  $\Delta T$  on each P3, etc. Again this map is unrealistic, being used for illustrative purposes only. Cards 9 to 16 assign the value of  $\eta_B$  in a one-to-one correspondence with the  $\Delta T$  values just assigned. The order is the same. The combustor pressure loss is input as a design value (i. e., DPCODS = 0.05) for a combustor type being considered. During off-design operation, the pressure loss is related to the design value by the following equation:  $DPCOM = DPCODS \times (WA3C/WA3CDS)$ . This equation is found in the COCOMB routine.

Also entered as BLOCK DATA subprograms are the turbine maps (HPTDAT and LPTDAT). To illustrate the entering of turbine data, LPTDAT is used. A typical turbine map is shown in figure 8; the data are programmed as follows:

```

SIBFTC LPTDAT DECK
BLOCK DATA
COMMON / LTURB/TFF(15),CN(15,15),DH(15,15),ETA(15,15),N,NP(15)
DATA N,NP/11,9*15,12,9,4*0/
DATA TFF / 88.470, 102.795, 116.835, 129.330, 141.045,
1 145.725, 150.000, 153.345, 156.405, 159.780, 163.170,4*0./
DATA (CN( 1,J),DH( 1,J),ETA( 1,J),J=1,15)/
1 0.3582, 0.0018, 0.7120, 0.5336, 0.0026, 0.7300,
2 0.7365, 0.0035, 0.7472, 0.9754, 0.0044, 0.7300,
3 1.2146, 0.0051, 0.7140, 1.4173, 0.0056, 0.7000,
4 1.5201, 0.0059, 0.6850, 1.7673, 0.0061, 0.6730,
5 2.0247, 0.0062, 0.6452, 2.2827, 0.0061, 0.6200,
6 2.4565, 0.0057, 0.6000, 2.6137, 0.0053, 0.5750,
7 2.3166, 0.0044, 0.5310, 2.9456, 0.0035, 0.5000,
8 3.3138, 0.0001, 0.3850/

```

Card 1 identifies the subprogram as BLOCK DATA. Card 2 identifies the common block into which data are to be loaded and dimensions the program variables. Card 3 indicates the number of constant turbine flow function lines TFF as 11(N) and the number of points on each line from low to high TFF. Cards 4 and 5 set values of TFF from low to high. Cards 6 to 14 set values of corrected speed CN, work function DH, and efficiency ETA along TFF(1) starting from low CN(0.3682) and ending at high CN(3.3138). The rest of the cards set values along higher TFF lines.

In many cases, turbine maps for high-performance engines operate at a choked condition (constant TFF). Thus, a turbine map to be represented could possibly have no lines representing constant TFF for a significant portion of the map. For complete map representation, lines of constant TFF may be estimated on the map up to the limit load-

ing line by inputting slight changes for the values of TFF (example: one line for TFF is 62.105, the next may be input as equal to 62.108). This will eliminate computational difficulties which would arise if constant values for TFF lines were input.

Afterburner performance has been programmed in a generalized form in subroutine COAFBN. The afterburner performance map included in the program is shown in figure 9(a). The performance map shows afterburner combustion efficiency ratio as a function of fuel-air ratio. The value of afterburner combustion efficiency correction factor  $\Delta ETAA$  during off-design operation is shown against design afterburner inlet Mach number ratio (fig. 9(b)) and design afterburner inlet total-pressure ratio (fig. 9(c)). Other correction factors or performance maps could be added as desired. The afterburner efficiency, fuel-air ratio, inlet total pressure, and Mach number are generalized external to the program.

A specific afterburner performance is generalized by dividing the specific off-design values by the design values, as shown here. The symbols shown are the symbols used in the ABETTA subroutine, where the generalized and specific values are input. The generalized afterburner values are obtained as follows:

$$\text{Efficiency (ETABRT)} = \frac{ETAA}{ETAADS}$$

$$\text{Fuel-air ratio (FART)} = \frac{FART}{FARTDS}$$

$$\text{Entrance total pressure (P6T)} = \frac{P6}{P6DS}$$

$$\text{Entrance Mach number (EM6T)} = \frac{AM6}{AM6DS}$$

However, the correction factor for efficiency  $\Delta ETAA$  is not a generalized value. The value of  $\Delta ETAA$  is an efficiency degradation. These degradations are input in ABETTA as discrete points on a curve. The values of  $\Delta ETAA$  are input as functions of the following:

- (1) The change in efficiency as a function of EM6T is input as DELM6 (which is really  $\Delta ETAA = f(AM6)$ ).
- (2) The change in efficiency as a function of P6T is input as DELP6 (which is really  $\Delta ETAA = f(P6)$ ).

At execution time for the design point, values of afterburner combustion efficiency ETAADS, exit total temperature T7DS, and entrance Mach number AM6DS are input. Design fuel-air ratio and entrance pressure ratio are calculated from the input values



and the other design engine characteristics.

The afterburner pressure loss is input as a design value (DPAFDS) during the design case. During afterburning the pressure loss due to combustion is calculated and the regular pressure drop is included for a total afterburner pressure loss. The equations for off-design and combustion pressure losses are contained in the COAFBN subroutines.

To achieve a reasonable accuracy in cycle calculations when using any generalized component map, the usage of the map should be limited within a certain range of the original design values and configuration changes. Therefore if, for example, an afterburner has a design task that differs significantly from an example used, a new generalized performance map should be used in order to simulate the component more accurately.

SMOTE, the original code, uses a single-point input for nozzle velocity coefficients (CVMNOZ and CVDNOZ) when calculating engine performance. GENENG, however, uses a convergent-divergent nozzle velocity coefficient which is input in map form in the FRATIO subroutine. The velocity coefficient is input as a function of nozzle total-pressure ratio ( $P_8/P_1$  or  $P_{28}/P_1$ ). The data for the nozzle are programmed as shown below for a performance map similar to that shown in figure 10.

```

$IBFTC FRATIO DECK
SUBROUTINE FRASHO (PRATIO,ARATIO,CF)
DIMENSION PR(30),AR(10),CFR(30,10),XX(3),YY(3),ZZ(2)
DATA PR/1.0,1.25,1.5,1.75,2.0,2.25,2.5,2.75,3.0,3.25,3.5,3.75,4.0,
14.25,4.5,4.75,5.0,5.25,5.5,5.75,6.0,6.25,6.5,7.0,7.5,8.0,9.0,11.0,13.0,15.0,17.0,19.0,21.0,23.0/
DATA AR/ 1.0,1.05,1.092,1.228,1.318,1.423,1.8,1.9,1.97,2.15/
DATA (CFR(1,1),I=1,180)/
1.971,.975,.978,.980,.981,.980,.9784,.9765,.9742,.9715,
2.9688,.9658,.9632,.960,.957,.9538,.9506,.9445,.938,.932,.927,.918,
3.910,.897,.8855,.875,.867,.859,.8515,.844,
4.963,.966,.970,.9743,.978,.9805,.982,.9805,.9786,.9762,.974,.9718,
5.9693,.967,.9645,.962,.9595,.954,.949,.944,.9395,.9308,.9238,
6.9117,.901,.892,.884,.8762,.869,.862,
7.956,.9505,.9638,.967,.970,.9731,.9754,.9775,.9795,.981,.980,.978,
8.9761,.9742,.9723,.9702,.968,.9637,.9594,.9554,.9517,.945,.9387,
9.9276,.9177,.9082,.8998,.891,.883,.8752,
$.947,.949,.9517,.954,.957,.9601,.9632,.9668,.971,.9737,.976,.9777,
$.978,.9772,.976,.9752,.974,.970,.9655,.9637,.9603,.955,.950,.9403,
$.9317,.9232,.915,.907,.8996,.892,
$.949,.9502,.9508,.9473,.9438,.941,.939,.940,.950,.956,.9605,.964,
$.967,.968,.9692,.970,.9703,.9698,.9596,.9684,.966,.960,.9444,
$.9438,.9347,.9265,.9193,.9128,.9067,.901,
$.948,.9485,.9487,.9484,.947,.9435,.936,.9275,.9185,.9155,.9155,
$.929,.941,.9455,.950,.953,.9558,.9591,.9608,.9609,.9601,.957,
$.9531,.9448,.9362,.9279,.9188,.910,.901,.8921/

```

Card 1 identifies the subroutine. Card 2 identifies the subroutine from its calling name used in the subprogram in addition to the variables that are desired to be transferred. Card 3 dimensions the program variables. Cards 4 and 5 set the value of PR( $P_9/P_1$  or  $P_{29}/P_1$ ) from low to high. Card 6 sets the values of AR( $A_9/A_8$  or  $A_{29}/A_{28}$ ) from low to high. Card 7 fills the first  $30 \times 6$  array of CFR(CVMNOZ or CVDNOZ) and is followed by another data statement which fills the remaining  $30 \times 4$  values. Cards 8 to 24, etc., are the first values of CFR being loaded for constant values of pressure ratio for varying

ratios going from low to high.

After the data input, the equations required to interpolate between points of the input values are included in the FRATIO subroutine.

Usually, a turbojet or a mixed-flow turbofan has a single exhaust nozzle; therefore the velocity coefficient is determined as a function of  $A9/A8$  and  $P9/P1$ . A separate-flow turbofan is considered to have two separate nozzles; thus in addition to using  $A9/A8$  and  $P9/P1$ , it will use the duct  $A29/A28$  and  $P29/P28$  to determine a velocity coefficient to correct each stream separately. When desired, however, a different nozzle map (i. e., convergent nozzle) or single-point inputs can be readily incorporated.

Inputs required at execution time. - Basically, what must be supplied are a list of the desired output variables, design values of any component existing in the engine (compressors, combustors, turbines, etc.), and engine operational controls. The variables that are to be output are selected by the first section of data cards. Any variables that are in one of the main commons (DESIGN, FRONT, SIDE, BACK, or DUMMYS) may be selected for output by punching, in columns 1 to 6, the name of the variable as it appears in the common. Up to 150 variables (25 lines of six variables) may be chosen for a particular run. During the output phase, the name of the variable is printed out, with its value printed immediately below the name.

Another feature of the controlled output is the ability to change the name of the variable to be output; for example, it may be desired to change a station designation to one more common to a particular programmer. In this case, the variable name would be punched in columns 1 to 6 as previously described, but, in addition, the described name would be punched in columns 13 to 18. Special symbols, such as /, may be used in the new name. The last card of the controlled output must be a card with "THEEND" punched in columns 1 to 6.

Table II summarizes the design inputs for the four basic engines discussed in this report.

The following control variables should always be supplied at the design point. The value used is dependent on how the user wants the engine to operate. The symbols and their purpose are listed in subroutine PUTIN but are shown here for the reader's convenience. The superscripts (1) to (4) on the symbols have the following meanings: (1) means "automatically returned to zero after each point is calculated, must be re-input if option is again desired"; (2) means "option can be used for design or off-design", whereas the other two modes can only be used at off-design; (3) means "these input values remain as input unless changed by a new input"; (4) means "before using these options which are not equal to zero, a setup case must be run where all the components are first matched, and then the identical case may be repeated exercising these options."

IDES = 1 <sup>(1)</sup>	For calculating design points.
(2)MODE = 0 <sup>(3)</sup>	Specify T4.
MODE = 1 <sup>(3)</sup>	Specify PCNC.
(2)MODE = 2 <sup>(3)</sup>	Specify WFB.
MODE = 3 <sup>(3)</sup>	Specify PCNF.
INIT = 0 <sup>(3)</sup>	Initializes point.
INIT = 1 <sup>(3)</sup>	Will not initialize point.
IDUMP = 0 <sup>(3)</sup>	Will not dump looping write-outs.
IDUMP = 1 <sup>(3)</sup>	Will dump looping write-outs if error occurs.
IDUMP = 2 <sup>(3)</sup>	Will dump looping write-outs after every point.
IAMTP = 0 <sup>(3)</sup>	Will use input AM and military specification ETAR.
IAMTP = 1 <sup>(3)</sup>	Will use input AM and input ETAR.
IAMTP = 2 <sup>(3)</sup>	Will use input T2 to determine $T1 = 518.668 + T2$ and standard P1. (T2 value needs to be input at every point or an error will occur whenever used.)
IAMTP = 3 <sup>(3)</sup>	Will use inputs P2 and standard T1.
IAMTP = 4 <sup>(3)</sup>	Will use input T2 and input P2.
IAMTP = 5 <sup>(3)</sup>	Will use specific schedule of ETAR located in subroutine RAMTWO.
IGASM <sup>X</sup> = -1 <sup>(3)</sup>	Separate flow, input A6.
IGASM <sup>X</sup> = 0 <sup>(3)</sup>	Separate flow, A6 = A55.
IGASM <sup>X</sup> = 1 <sup>(3)</sup>	Will mix fan duct and main streams, A6 = A25 + A55.
IGASM <sup>X</sup> = 2 <sup>(3)</sup>	Will mix fan duct and main streams, input A6.
IDBURN = 1 <sup>(1)</sup>	For duct burning (fan stream only), input T24. <sup>(4)</sup>
IDBURN = 2 <sup>(1)</sup>	For duct burning (fan stream only), input WFD. <sup>(4)</sup>

IAFTBN = 1 <sup>(1)</sup>	For afterburning (main stream or mixed stream of fan and main stream), input T7. <sup>(4)</sup>
IAFTBN = 2 <sup>(1)</sup>	For afterburning (main stream or mixed stream of fan and main stream), input WFA. <sup>(4)</sup>
IDCD = 1 <sup>(3)</sup>	Fan duct nozzle will be convergent-divergent.
IMCD = 1 <sup>(3)</sup>	Main nozzle will be convergent-divergent.
NOZFLT = 1 <sup>(3)</sup>	For floating main nozzle exit area. <sup>(4)</sup>
NOZFLT = 2 <sup>(3)</sup>	For floating fan duct nozzle exit area. <sup>(4)</sup>
NOZFLT = 3 <sup>(3)</sup>	For floating fan duct and main nozzle exit area. <sup>(4)</sup>
ITRYS = N	Number of passes through engine before quitting.
TOLALL = X	Tolerance which the errors must satisfy before engine is matched.

The following are other input variables for which some value, depending on the engine design, should be input at the discretion of the user:

DELFG, DELFN, DELSFC	Normally input as 1.0 unless correction is desired.
A6, AM55, AM23, AM6, HPEXT, AM, ALTP	See appendix C, Input Symbols.
PCBLF, PCBLC, PCBLDU, PCBLOB	Value for bleed out of the cycle; decimal equivalent of percent compressor flow.
PCBLHP, PCBLLP	Value of total bleed returned to turbines for cycle; fractional equivalent of flow. The sum of these variables plus PCBLDU and PCBLOB should equal 1.

Inputs required for additional options to basic cycles. - To run duct-burning (available only in fan stream duct) cases load ETAD, T24 or WFDDS, and DPDUDS. To run afterburning (mixed-flow fan or unmixed fan - available only for core and fan stream) cases, load T7DS or WFADS, ETAADS, and DPAFDS. Afterburner operation is the same as in reference 2 with the exception of a generalized afterburner performance map addition. For changing the generalized map to a specific map for a specific engine design, the preceding design values are needed at the design point.

## Means of Specifying Mode of Engine Operations

Shown in the section SAMPLE CALCULATIONS (pp. 21 to 46) are the methods of specifying off-design operation points. The most common one and that used exclusively herein is to select a Mach number, altitude, and turbine inlet temperature other than design values. There are, however, several other possibilities which the user may employ. For example, changing the following controls:

MODE = 0    Specify a new turbine inlet temperature T4.

MODE = 1    Specify a compressor rotational speed PCNC.

MODE = 2    Specify a fuel flow rate WFB.

MODE = 3    Specify a fan rotational speed PCNF.

If the engine has all its nozzles fixed, an input such as turbine inlet temperature, fuel flow, or speed will set the thrust level. But other means of changing engine operation can be accomplished by varying such nozzle thrust areas as

A8    Main nozzle thrust area

A28   Fan nozzle thrust area

For example, an off-design condition may exist where, in an attempt to satisfy continuity of mass flow (one of the component matching requirements), the fan operating point may lie outside the limits of the data map that was input for the component map. A fan nozzle thrust area change could be used to return the fan operating condition on the map such that a match would occur. This would indicate a possibility exists that variable fan nozzle would be required on this engine for operation at the desired condition. The area is changed by inputting (example:  $A28 = 1.2 \times A28$ ). Since the design areas are not known prior to running the design point, the Huff Input Routine provides the versatility in which A28 is increased by 20 percent, as was shown. It should be noted that any area changed remains changed until it is recalculated by a new design case or altered by a new input. The preceding example and statements would also apply if changes were made instead to A8.

The nozzle exit area (such as A9 or A29) may then be fully expanded (if A8 and A28 are sonic) after component matching is achieved by using the control variables NOZFLT = 1, 2, or 3. The significance of these values was explained in the previous section.

## SAMPLE CALCULATIONS

Shown in this section are complete listings of sample calculations for a single-spool turbojet, a two-spool turbojet, a separate-flow turbofan, and a mixed-flow turbofan. Detailed instructions for operating the program are disclosed through means of these sample calculations plus comments in the program listing (appendix A) and the PUTIN routine (pp. 50 to 52). The input and output data associated with one complete design and off-design computer run are shown.

### Single-Spool Turbojet

Design point. - This case is for design-point operation of a single-spool turbojet engine and is represented by three groups of data printout: (1) card images of input data, (2) results, and (3) the common variables printed out only at the design point.

For this point, the component maps were loaded as BLOCK DATA and scaled to maps for the desired engine design. The common variables appearing in the printout are defined by the relative locations of the common blocks of DESIGN, FRONT, SIDE, and BACK. The code for the common variables shown on pages 23 and 24 is as follows:

```

ZF,   PCNF,      ZC,   PCNC,  T4,      MODE

COMMON /DESIGN/
1PCNFGJ,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC ,
2ZPDS ,PCNFDS,PRFDS ,ETAFDS,WAFDS ,PRFCF ,ETAFCF,WAFCF ,
3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,
4T4DS ,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCCCF,ETABCF,
5TFHPDS,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS ,
6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLPFC,ETLPCF,DHLPFC,T2LDS ,
7T24DS ,WFDDS ,DTDUDS,ETAADS,WA23DS,DPDUDS,DTDUCF,ETADCF,
8T7DS ,WFADS ,DTAFDS,ETAADS,WG6CDS,DPAFDS,DTAFCF,ETAACF,
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
$PS55 ,AM55 ,CVDNOZ,CVMNOZ,A8SAV ,A9SAV ,A28SAV,A29SAV

COMMON / FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLDB ,
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,
7CNHP ,ETATHP,DHTCHP,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,
8CNLP ,ETATLP,DHTCLP,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU,PCBLDB,PCBLHP,PCBLP

COMMON / SIDE/
1XP1 ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3 ,DUMS1 ,DUMS2 ,
2XT21 ,XP21 ,XH21 ,XS21 ,T23 ,P23 ,H23 ,S23 ,
3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 ,
4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 ,
5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS,DUMS3 ,
6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29

COMMON / BACK/
1XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,
2XWFB ,XWG55 ,XFAR55,XWFD ,XWG24 ,XFAR24,XXP1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC

```

It should be noted that the program uses fixed effective areas calculated at the design point. The design case which calculates the area uses a convergent nozzle. Thus, as seen on the listing (pp. 23 and 24),  $A_8 = A_9$ . Before an afterburning case or before the nozzle is allowed to float (become fully expanded to a convergent-divergent nozzle), the engine components must first be matched in a setup run. Then, the next case may be run with IAFITBN or NOZFLT control cards unequal to zero and may be used as shown in the off-design cases 4 and 5 or 6 and 7. Either a convergent or a convergent-divergent subroutine may be used, depending on the input controls. If afterburning has been selected, the nozzle area is allowed to float to satisfy continuity. However, the areas are returned to their original design values after the afterburning or nozzle float points are completed.

Because the first example is a single-spool turbojet, values for the inner compressor and turbine, as well as duct parameters, are either zero or 1. The actual or corrected values of compressor exit flow (WAC or WACC, respectively) are printed out prior to any bleed extraction. The value of WA3 is the combustor entrance airflow after bleed extraction.

The first 11 lines of the following example show the storage locations reserved for the input data. The locations are relative to the common block locations shown on page 21. Only locations for which values of data will be input are required. Following the end of the storage locations are the input data which are typical for running a single-spool turbojet engine.

```

$DATA(1) #1  INPUT DATA CARDS
$TABLE(.REAL,22=TOLALL,34=DELFG,35=DELFN,36=DELSFC,37=ZFDS,38=PCNFDS,39=PRFDS,
40=ETAFDS,41=WAFDS,45=ZCDS,46=PCNCDS,47=PRCDS,48=ETACDS,49=WACDS,53=T4DS,54=
WFBDS,55=DTCCDS,56=ETABDS,58=DPCDS,61=TFHPDS,62=CNHPDS,63=ETHPDS,69=TFLPDS,
70=CNLPDS,71=ETLPDS,82=DPDUDS,85=T7DS,88=ETAADS,90=DPAFDS,95=A6,97=A8,101=PS55,
102=AM55,103=CVDNOZ,104=CVMNOZ,113=T2,114=P2,125=T4,144=WAFCD,152=WACCD,
172=HPEXT,173=AM,174=ALTP,175=ETAR,177=PCNF,179=PCNC,183=PCBLF,184=PCBLC,
185=PCBLDU,186=PCBLOB,187=PCBLHP,188=PCBLLP,205=T24,225=ETAD,257=T7,270=WFA,
273=ETAA,279=AM6,313=AM23,99=A28,180=WFB,222=WFD,.INTEGER,2=IDES,5=MODE
,7=IDUMP,8=IAMTP,9=IGASM,10=IDBURN,11=IAFITBN,12=IDCD,13=IMCD,16=NOZFLT,17=
ITRYS,314=ISPOOL,.LOGICAL,319=FAN) # END OF TABLE GIVING NAMES AND LOCATIONS

#0 INPJT DATA TO RUN ONE SPOOL TURBOJET
FAN=.FALSE.,ISP00L=1,ZFDS=.83333333,PRFDS=8,WAFCD=633,ETAFDS=.829,PCNFDS=100,
PCBLF=0,PCBLC=.097,PCBLHP=0,PCBLLP=1,PCBLOB=0,T4DS=3260,DPCDS=.056,ETABDS=.985,
ETLPDS=.903,TFLPDS=130.0,CNLPDS=2.300,HPEXT=0,AM55=.238,AM6=.238,ETAADS=.90,
T7DS=3560,DPAFDS=.048,DELFG=1,DELFN=1,DELSFC=1,TOLALL=.005,ITRYS=210,IDCD=0,
IMCD=0,IDES=1,IAMTP=0,MODE=0,IDUMP=0,AM=0,ALTP=0,

```

The following is an example of the normal design-case printout. The fan, compressor, combustor, high-pressure- and low-pressure-turbine correction factors are printed out, as are the various designated design areas. Following these are the internal engine thermodynamic characteristics and thrust and specific fuel consumption.

\*1 DESIGN POINT WITH NORMAL PRINTOUT WHEN IDES=1 (PRINTED OUT FOR DESIGN  
 \* POINT ONLY) - - NEXT PAGE HAS THE CLOSE FORMAT (SEE PAGE OF KEY TO VARIABLES)

FAN DESIGN PRFCF= 0.1750000E+02 ETACF= 0.9420454E+00 WAFCF= 0.1055000E+01 T2DS= 0.5186682E+03  
 COMPRESSOR DESIGN PRCCF= 0.1000000E+01 ETACCF= 0.1000000E+01 WACCF= 0.1000000E+01 T21DS= 0.1019159E+04  
 COMBUSTER DESIGN WA3CDS= 0.15521097E+03 ETABCF= 0.9850000E+00 DTCOCF= 0.1000000E+01  
 L.P. TURBINE DESIGN CNLPCF= 0.13132174E+01 TFLPCF= 0.42574495E+00 ETLPCF= 0.10235344E+01 DHLPCF= 0.23187937E+01  
 TURBINE AREA DESIGN A55= 0.20933135E+02 AM55= 0.23823719E+00  
 AFTERBURNER DESIGN WG6CDS= 0.89939300E+04  
 NOZZLE DESIGN A8= 0.85851488E+01 AM8= 0.1000000E+01 A9= 0.85851488E+01 AM9= 0.1000000E+01  
 OUTPUT AM= 0. ALTP= 0. T4= 3260.00 ETAR= 1.0000  
 PCNF CNF ZF PRF WAF WAF  
 0.100000E+03 0.100000E+01 0.833333E+00 0.800000E+01 0.633000E+03 0.633000E+03  
 PCNC CNC ZC PRC WACC WAC  
 0.100000E+03 0.100000E+01 0. 0.100000E+01 0.110915E+03 0.633000E+03  
 T2 P2 T21 P21 T3 P3  
 0.518668E+03 0.100000E+01 0.101916E+04 0.800000E+01 0.101916E+04 0.800000E+01  
 PCBLF BLF PCBLC BLC PCBLOB BLOB  
 0. 0. 0.970000E-01 0.614010E+02 0. 0.  
 PCBLHP BLHP PCBLLP BLLP T4 P4  
 0. 0. 0.100000E+01 0.614010E+02 0.326000E+04 0.755200E+01  
 WA3 WFB WG4 FAR4 ETAB DPCOM  
 0.571599E+03 0.219359E+02 0.593535E+03 0.383764E-01 0.985000E+00 0.560000E-01  
 TFFHP CNHP DHTCHP DHTC T5 P5  
 0. 0. 0. 0. 0.326000E+04 0.755200E+01  
 TFFLP CNLP DHTCLP DHTF T55 P55  
 0.305347E+03 0.175142E+01 0.171750E-01 0.129830E+03 0.269721E+04 0.378187E+01  
 PCBLOU RLDU T24 P24 T25 P25  
 0. 0. 0. 0. 0.101916E+04 0.800000E+01  
 WAD WFD WG24 FAR24 ETAD DPDU  
 0. 0. 0.633000E+03 0. 0. 0.  
 ETAF ETAC ETATHP ETATLP AM55 AM25  
 0.829000E+00 0.100000E+01 0.100000E+01 0.903000E+00 0.238237E+00 0.  
 T6 P6 PS6 AM5 V6 WG6  
 0.269721E+04 0.378187E+01 0.364695E+01 0.238237E+00 0.579433E+03 0.654936E+03  
 T7 WFA WG7 FAR7 ETAA DPAFT  
 0.269721E+04 0. 0.654936E+03 0.346539E-01 0. 0.480000E-01  
 PS8 AMB V8 PS9 AM9 V9  
 0.198204E+01 0.100000E+01 0.228834E+04 0.198204E+01 0.100000E+01 0.228834E+04  
 PS28 AM28 V28 PS29 AM29 V29  
 0. 0. 0. 0. 0. 0.  
 BYPASS HPEXT WFT WGT VA FRD  
 0. 0. 0.219359E+02 0.654936E+03 0. 0.  
 CVMNOZ VJM FVNOZ VJD FGM FGP  
 0.967632E+00 0.221427E+04 0. 0. 0.450738E+05 0.178418E+05  
 MAIN SCNIC CONVERGENT NOZZLE FG= 62915.60 FN= 52915.60 SFC= 1.25516  
 DUCT SHOCK OUTSIDE C-D NOZZLE  
 CONVERGED AFTER 1 LOOPS

The following shows the common variables which can be read, if desired, by referring to the common designation (p. 21).

COMMON	0.833333E+00	0.100000E+03	-0.000000E-19	-0.000000E-19	0.	0.100000E+03	0.326000E+04	0
0.100000E+03	0.	0.326000E+04	0.100000E+03	0.	0.100000E+01	0.100000E+01	0.100000E+01	
0.833333E+00	0.100000E+03	0.800000E+01	0.829000E+00	0.633000E+03	0.175000E+02	0.942045E+00	0.105500E+01	
0.326000E+04	0.	0.224084E+04	0.985000E+00	0.633000E+03	0.100000E+01	0.100000E+01	0.100000E+01	
0.	0.	0.	0.100000E+01	0.155211E+03	0.568000E-01	0.100000E+01	0.985000E+00	
0.130000E+03	0.230000E+01	0.903000E+00	0.425745E+00	0.131322E+01	0.102353E+01	0.231879E+01	0.101916E+04	
0.	0.	0.	0.	0.	0.	0.	0.	
0.356000E+04	0.	0.	0.900000E+00	0.899393E+04	0.480000E-01	0.	0.	
0.209331E+02	0.	0.209331E+02	0.209331E+02	0.858515E+01	0.858515E+01	0.	0.	
0.364695E+01	0.238237E+00	0.	0.967632E+00	0.	0.	0.	0.	
0.518668E+03	0.100000E+01	0.123918E+03	0.159103E+01	0.518668E+03	0.100000E+01	0.123918E+03	0.159103E+01	
0.101916E+04	0.800000E+01	0.245654E+03	0.161236E+01	0.101916E+04	0.800000E+01	0.245654E+03	0.161236E+01	
0.326000E+04	0.755200E+01	0.916119E+03	0.196079E+01	0.326000E+04	0.755200E+01	0.916119E+03	0.196079E+01	
0.269721E+04	0.378187E+01	0.735604E+03	0.194649E+01	0.	0.614010E+02	0.	0.	
0.100000E+01	0.800000E+01	0.829000E+00	0.633000E+03	0.633000E+03	0.571599E+03	0.593535E+03	0.383764E-01	
0.100000E+01	0.100000E+01	0.	0.	0.633000E+03	0.985000E+00	0.560000E-01	0.	
0.100000E+01	0.100000E+01	0.	0.	0.	0.593535E+03	0.383764E-01	0.111699E+04	
0.230000E+01	0.903000E+00	0.398252E-01	0.129830E+03	0.614010E+02	0.654936E+03	0.	0.	
0.	0.	0.100000E+01	0.833333E+00	0.100000E+03	0.	0.100000E+03	0.219359E+02	
0.	0.130000E+03	0.	0.970000E-01	0.	0.	0.	0.100000E+01	



G.1C000E+01	0.63300E+03	0.63300E+03	0.	0.	0.245654E+03	0.	0.
C.101916E+04	C.800000E+01	0.245654E+03	0.161236E+01	0.	0.	0.	0.
0.	0.	0.	0.	0.101916E+04	0.800000E+01	0.245654E+03	0.161236E+01
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.63300E+03	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.269721E+04	0.378187E+01	0.735604E+03	0.194649E+01	0.101916E+04	0.800000E+01	0.245654E+03	0.161236E+01
U.219359E+02	0.654936E+03	0.346539E-01	0.	0.633000E+03	0.	0.100000E+01	0.
C.269721E+04	0.378187E+01	0.735604E+03	0.194649E+01	0.269721E+04	0.360034E+01	0.735601E+03	0.194987E+01
0.269721E+04	0.360034E+01	0.735601E+03	0.194987E+01	0.269721E+04	0.360034E+01	0.735601E+03	0.194987E+01
0.654936E+03	0.	0.654936E+03	0.346539E-01	0.	0.480000E-01	0.579433E+03	0.
0.364675E+01	0.579433E+03	0.238237E+00	0.267322E+04	0.345885E+01	0.610475E+03	0.251101E+00	0.
0.235513E+04	0.198204E+01	0.228834E+04	0.100000E+01	0.235513E+04	0.198204E+01	0.228834E+04	0.100000E+01
0.	0.	0.	0.	0.221427E+04	0.450738E+05	0.	C.178418E+05
0.450738E+05	0.178418E+05	0.219359E+02	0.654936E+03	0.346539E-01	0.629156E+05	0.629156E+05	C.125516E+01

**Off-design.** - In the following cases, only the input data that are changed are printed out as input. The other data (normal design-case printout, p. 23) are not affected. The common variables are not repeated because IDES was set to zero internally in the program. That printout is eliminated and the design printout converts to the normal off-design printout format.

Case 1: The engine is operating at a reduced turbine inlet temperature T4 at a sea-level-static condition. The only input change required is T4.

```

SD(1),T4=2660, #1 PART POWER ENGINE OPERATION -- OFF-DESIGN
* RESULTS NORMAL OFF-DESIGN OUTPUT
OUTPUT          AM= 0.          ALTP= 0.          T4= 2660.00          ETAR= 1.0000
PCNF            CNF            ZF            PRF            WAF            WAF
0.843207E+02   0.843207E+00   0.766751E+00   0.584906E+01   0.525128E+03   0.525128E+03
PCNC            CNC            ZC            PRC            WACC            WAC
0.949398E+02   0.100000E+01   0.100000E+01   0.100000E+01   0.119482E+03   0.525128E+03
T2             P2             T21            P21            T3             P3
0.518666E+03   0.100000E+01   0.918624E+03   0.584906E+01   0.918615E+03   0.584906E+01
PCBLF          BLF            PCBLC          BLC            PCBLOB         BLOB
0.            0.            0.970000E-01   0.509374E+02   0.            0.
PCBLHP        BLHP          PCBLLP         SLLP            T4             P4
0.            0.            0.100000E+01   0.509374E+02   0.266000E+04   0.549621E+01
WA3           WFB           WG4            FAR4            ETAB           DPCOM
0.474191E+03   0.133727E+02   0.487563E+03   0.282011E-01   0.985000E+00   0.603254E-01
TFFHP         CNHP          DHTCHP        DHTC            T5             P5
0.            0.            0.            0.            0.266000E+04   0.549621E+01
TFFLP         CNLP          DHTCLP        DHTF            T55            P55
0.311322E+03   0.163491E+01   0.169092E-01   0.104294E+03   0.219476E+04   0.278698E+01
PCBLDU        BLDU          T24            P24            T25            P25
0.            0.            0.            0.            0.918624E+03   0.584906E+01
WAD           WFD           WG24           FAR24           ETAD           DPDU
0.            0.            0.525128E+03   0.            0.            0.
ETAF          ETAC          ETATHP         ETATLP          AM55           AM25
0.843568E+00   0.100000E+01   0.100000E+01   0.907545E+00   0.238077E+00   0.
T6            P6            PS6            AM6             V6             WG6
0.219476E+04   0.278698E+01   0.269024E+01   0.238077E+00   0.525868E+03   0.538501E+03
T7            WFA           WG7            FAR7            ETAA           DPAPT
0.219476E+04   0.            0.538501E+03   0.254656E-01   0.            0.483102E-C1
PSB           AM8           VB             PS9            AM9            V9
0.145228E+01   0.100000E+01   0.207131E+04   0.145228E+01   0.100000E+01   0.207131E+04
PS28          AM28          V28            PS29           AM29           V29
0.            0.            0.            0.            0.            0.
RYPASS        HPEXT        WFT            NGT             VA             FRD
0.            0.            0.133727E+02   0.538501E+03   0.            0.
CVMNOZ        VJM           CVDNOZ         VJD             FGM            FGP
0.977273E+00   0.202425E+04   0.            0.            0.338801E+05   0.821708E+04
MAIN SCNIC CONVERGENT NOZZLE          FG= 42097.19          FN= 42097.19          SFC= 1.14308
DUCT SHOCK OUTSIDE C-D NOZZLE
CONVERGED AFTER 14 LOOPS

```

Case 2: The engine is operating at design turbine inlet temperature at sea-level-static condition. The engine components must always be matched at the desired turbine inlet temperature prior to running the engine at afterburning conditions.

SD(1), T4=3260., \* SET UP FOR AFTERBURNER OPERATION

OUTPUT	AM= 0.	ALTP= 0.	T4= 3260.00		ETAR= 1.0000
	PCNF	CNF	ZF	PRF	WAF
0.100000E+03	0.100000E+01	0.833333E+00	0.800000E+01	0.633000E+03	0.633000E+03
	PCNC	CNC	ZC	PRC	WAC
0.100000E+03	0.100000E+01	0.	0.100000E+01	0.110915E+03	0.633000E+03
	T2	P2	T21	P21	T3
0.518668E+03	0.100000E+01	0.101916E+04	0.800000E+01	0.101916E+04	0.800000E+01
	PCBLF	BLF	PCBLC	BLC	PCBL0B
0.	0.	0.970000E-01	0.614010E+02	0.	0.
	PCBLHP	BLHP	PCBLLP	BLLP	T4
0.	0.	0.100000E+01	0.614010E+02	0.326000E+04	0.755200E+01
	WA3	WFB	WG4	FAR4	ETAB
0.571599E+03	0.219359E+02	0.593535E+03	0.383764E-01	0.985000E+00	0.560000E-01
	TFFFHP	CNHP	DHTCHP	DHTC	T5
0.	0.	0.	0.	0.326000E+04	0.755200E+01
	TFFFLP	CNLP	DHTCLP	DHTF	T55
0.305347E+03	0.175142E+01	0.171750E-01	0.129830E+03	0.269721E+04	0.378187E+01
	PCBLDU	BLDU	T24	P24	T25
0.	0.	0.	0.	0.101916E+04	0.800000E+01
	WAD	WFD	WG24	FAR24	ETAD
0.	0.	0.633000E+03	0.	0.	0.
	ETAF	ETAC	ETATHP	ETATLP	AM55
0.829000E+00	0.100000E+01	0.100000E+01	0.903000E+00	0.238190E+00	0.
	T6	P6	P56	AM6	WG6
0.269721E+04	0.378187E+01	0.365225E+01	0.238190E+00	0.579408E+03	0.654936E+03
	T7	WFA	WG7	FAR7	ETAA
0.269721E+04	0.	0.654936E+03	0.345539E-01	0.	0.480000E-01
	PS8	AMB	V8	PS9	AM9
0.198204E+01	0.100000E+01	0.228834E+04	0.198204E+01	0.100000E+01	0.228834E+04
	PS29	AM28	V28	PS29	AM29
0.	0.	0.	0.	0.	0.
	BYPASS	HPEXT	WFT	WGT	VA
0.	0.	0.219359E+02	0.654936E+03	0.	0.
	CVMNOZ	VJM	CVDNOZ	VJD	FGM
0.967632E+00	0.221427E+04	0.	0.	0.450738E+05	0.178418E+05

MAIN SCNIC CONVERGENT NOZZLE  
 DUCT SHOCK OUTSIDE C-D NOZZLE  
 CONVERGED AFTER 1 LOOPS

FG= 62915.60

FN= 62915.60

SFC= 1.25516

Case 3: The engine is operating at maximum design afterburner temperature at sea-level-static condition. Once the engine components are matched at the desired turbine inlet temperature, the engine performance for any afterburner temperature that is greater than the turbine exit temperature may be calculated.

A new value of A8 is computed when afterburning to account for the temperature, pressure, and mass flow changes. A9 is allowed to float to provide for complete expansion to PS9 = PS0 at station 9. The new values of A8 and A9 are printed at the top of the normal off-design output page. The value of IAFTBN is set to zero after each case; therefore, IAFTBN = 1 must be used for each afterburning case.

SD(1), T4=3260, T7=3560, IAFTBN=1, \* AFTERBURNER OPERATION AT DESIGN TEMP  
 \* RESULTS NORMAL OFF-DESIGN OUTPUT

NOZZLE DESIGN	A8= 0.10260896E+02	AM8= 0.10000000E+01	A9= 0.12334022E+02	AM9= 0.15117356E+01	
OUTPUT	AM= 0.	ALTP= 0.	T4= 3260.00	T7= 3560.00	ETAR= 1.0000
	PCNF	CNF	ZF	PRF	WAF
0.100000E+03	0.100000E+01	0.833333E+00	0.800000E+01	0.633000E+03	0.633000E+03
	PCNC	CNC	ZC	PRC	WAC
0.100000E+03	0.100000E+01	0.	0.100000E+01	0.110915E+03	0.633000E+03
	T2	P2	T21	P21	T3
0.518668E+03	0.100000E+01	0.101916E+04	0.800000E+01	0.101916E+04	0.800000E+01
	PCBLF	BLF	PCBLC	BLC	PCBL0B
0.	0.	0.970000E-01	0.614010E+02	0.	0.
	PCBLHP	BLHP	PCBLLP	BLLP	T4
0.	0.	0.100000E+01	0.614010E+02	0.326000E+04	0.755200E+01
	WA3	WFB	WG4	FAR4	ETAB
0.571599E+03	0.219359E+02	0.593535E+03	0.383764E-01	0.985000E+00	0.560000E-01
	TFFFHP	CNHP	DHTCHP	DHTC	T5
0.	0.	0.	0.	0.326000E+04	0.755200E+01
	TFFFLP	CNLP	DHTCLP	DHTF	T55
0.305347E+03	0.175142E+01	0.171750E-01	0.129830E+03	0.269721E+04	0.378187E+01
	PCBLDU	BLDU	T24	P24	T25
0.	0.	0.	0.	0.101916E+04	0.800000E+01
	WAD	WFD	WG24	FAR24	ETAD
0.	0.	0.633000E+03	0.	0.	0.

ETAF	ETAC	ETATHP	ETATLP	AM55	AM25
0.829000E+00	0.100000E+01	0.100000E+01	0.903000E+00	0.238190E+00	0.
T6	P6	PS6	AM6	V6	WG6
0.269721E+04	0.378187E+01	0.365225E+01	0.238190E+00	0.579408E+03	0.654936E+03
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.356000E+04	0.106971E+02	0.665633E+03	0.515530E-01	0.900029E+00	0.480000E-01
PS8	AM8	V8	PS9	AM9	V9
0.196671E+01	0.100000E+01	0.261777E+04	0.100000E+01	0.151174E+01	0.369243E+04
PS28	AM28	V28	PS29	AM29	V29
0.	0.	0.	0.	0.	0.
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.	0.	0.326330E+02	0.665633E+03	0.	0.
CVMNDZ	VJM	CVDNOZ	VJD	FGM	FGP
0.977004E+00	0.360752E+04	0.	0.	0.746342E+05	0.

MAIN SHOCK OUTSIDE C-D NOZZLE      FG= 74634.20      FN= 74634.20      SFC= 1.57406  
 DUCT SHOCK OUTSIDE C-D NOZZLE

CONVERGED AFTER 1 LOOPS

Case 4: The engine is operating at a subsonic Mach number (0.7) at 25 000 feet altitude with a turbine inlet temperature less than design. The performance shown is for A8 = A9 (i.e., the engine is using a convergent nozzle). If desired, a convergent-divergent nozzle could have been simulated by using this case as a setup and having an additional case with NOZFLT = 1 or IMCD = 1 for complete expansion to occur.

\$D11),AM=.7,ALTP=25000,T4=2760, \* PART POWER OPERATION AT CRUISE CONDITION  
 \* RESULTS NORMAL OFF-DESIGN OUTPUT

OUTPUT	AM= 0.700	ALTP= 25000.	T4= 2760.00	ETAR= 1.0000
PCNF	CNF	ZF	PRF	WAF
0.900352E+02	0.944024E+00	0.806890E+00	0.714555E+01	0.594428E+03
PCNC	CNC	ZC	PRC	WACC
0.936879E+02	0.100000E+01	0.100000E+01	0.100000E+01	0.114549E+03
T2	P2	T21	P21	T3
0.471787E+03	0.514853E+00	0.894556E+03	0.367891E+01	0.894555E+03
PCBLF	BLF	PCBLC	BLC	PCBLOB
0.	0.	0.970000E-01	0.311261E+02	0.
PCDLHP	BLHP	PCBLLP	BLLP	T4
0.	0.	0.100000E+01	0.311261E+02	0.276000E+04
WA3	WFB	WG4	FAR4	ETAB
0.289761E+03	0.882030E+01	0.298582E+03	0.304399E-01	0.985000E+00
TFFHP	CNHP	DHTCHP	DHTC	T5
0.	0.	0.	0.	0.276000E+04
TFFLP	CNLP	DHTCLP	DHTF	T55
0.307945E+03	0.171379E+01	0.171341E-01	0.109754E+03	0.227177E+04
PCBLDU	BLDU	T24	P24	T25
0.	0.	0.	0.	0.894556E+03
WAD	WFD	WG24	FAR24	ETAD
0.	0.	0.320887E+03	0.	0.
ETAF	ETAC	ETATHP	ETATLP	AM55
0.836023E+00	0.100000E+01	0.100000E+01	0.905173E+00	0.238293E+00
T6	P6	PS6	AM6	V6
0.227177E+04	0.173692E+01	0.167666E+01	0.238293E+00	0.534812E+03
T7	WFA	WG7	FAR7	ETAA
0.227177E+04	0.	0.329708E+03	0.274872E-01	0.
PS8	AM8	V8	PS9	AM9
0.906790E+00	0.100000E+01	0.210597E+04	0.905790E+00	0.100000E+01
PS28	AM28	V28	PS29	AM29
0.	0.	0.	0.	0.
BYPASS	HPEXT	WFT	WGT	VA
0.	0.	0.882030E+01	0.329708E+03	0.711594E+03
CVMNDZ	VJM	CVDNOZ	VJD	FGM
0.957531E+00	0.201653E+04	0.	0.	0.206646E+05

MAIN SCNIC CONVERGENT NOZZLE      FG= 30397.21      FN= 23300.14      SFC= 1.36278  
 DUCT SHOCK OUTSIDE C-D NOZZLE

CONVERGED AFTER 7 LOOPS

Case 5: The engine is operating at the same subsonic condition as in case 4, but at the design turbine inlet temperature. Afterburner operation may be added at this point. If a different turbine inlet temperature is desired, a new setup case, as in case 4, must be calculated for component matching purposes immediately prior to running the afterburner case.

SD(1),T4=3260,\* DESIGN T4 AT FLIGHT CONDITION  
 \* RESULTS NORMAL OFF-DESIGN OUTPUT

OUTPUT	AM= 0.700	ALTP= 25000.	T4= 3260.00	ETAR= 1.0000		
PCNF	CNF	ZF	PRF	WAF	WAF	
0.984254E+02	0.103200E+01	0.856827E+00	0.869637E+01	0.660746E+03	0.356688E+03	
PCNC	CNC	ZC	PRC	WACC	WACC	
0.976137E+02	0.100000E+01	0.100000E+01	0.100000E+01	0.109007E+03	0.356688E+03	
T2	P2	T21	P21	T3	P3	
0.471789E+03	0.514853E+00	0.971095E+03	0.447735E+01	0.971092E+03	0.447735E+01	
PCBLF	BLF	PCBLC	BLC	PCBLOB	BLOB	
0.	0.	0.970000E-01	0.345987E+02	0.	0.	
PCBLHP	BLHP	PCBLLP	BLLP	T4	P4	
0.	0.	0.100000E+01	0.345987E+02	0.326000E+04	0.423093E+01	
WA3	WFR	WG4	FAR4	ETAR	DPCDM	
0.322089E+03	0.125982E+02	0.334688E+03	0.391140E-01	0.985000E+00	0.590366E-01	
TFHHP	CNHP	DHTCHP	DHTC	T5	P5	
0.	0.	0.	0.	0.326000E+04	0.423093E+01	
TFHLP	CNLP	DHTCLP	DHTF	T55	P55	
0.307337E+03	0.172385E+01	0.171404E-01	0.128965E+03	0.269690E+04	0.213226E+01	
PCBLDU	BLDU	T24	P24	T25	P25	
0.	0.	0.	0.	0.971095E+03	0.447735E+01	
WAD	WFD	WG24	FAR24	ETAD	DPDUC	
0.	0.	0.356688E+03	0.	0.	0.	
ETAF	ETAC	ETATHP	ETATLP	AM55	AM25	
0.800114E+00	0.100000E+01	0.100000E+01	0.904580E+00	0.238236E+00	0.	
T6	P6	PS6	AM6	V6	WG6	
0.269690E+04	0.213226E+01	0.205917E+01	0.238236E+00	0.579402E+03	0.369286E+03	
T7	WFA	WG7	FAR7	ETAA	DPAFT	
0.269690E+04	0.	0.369286E+03	0.353199E-01	0.	0.480007E-01	
PS8	AMB	V8	PS9	AM9	V9	
0.111781E+01	0.100000E+01	0.228808E+04	0.111781E+01	0.100000E+01	0.228808E+04	
PS28	AM28	V28	PS29	AM29	V29	
0.	0.	0.	0.	0.	0.	
BYPASS	HPEXT	WFT	WGT	VA	FRD	
0.	0.	0.125982E+02	0.369286E+03	0.711594E+03	0.788888E+04	
CVMNOZ	VJM	CVNOZ	VJD	FGM	FGP	
0.944853E+00	0.216190E+04	0.	0.	0.248138E+05	0.135664E+05	
MAIN SCNIC CONVERGENT NOZZLE		FG= 38380.11	FN= 30491.23	SFC= 1.46743		
DUCT SHOCK OUTSIDE C-U NOZZLE						
CONVERGED AFTER 14 LOOPS						

Case 6: The engine is now operating at a supersonic flight Mach number (2.5) at an altitude of 50 000 feet. The engine components are matched with an engine nozzle throat area A8 equal to the sea-level-static design value. This case is a setup case for matching the components only.

SD(1),AM=2.5,ALTP=50000,T4=3260,\* SET UP FOR MATCHING ONLY

OUTPUT	AM= 2.500	ALTP= 50000.	T4= 3260.00	ETAR= 0.8703		
PCNF	CNF	ZF	PRF	WAF	WAF	
0.832186E+02	0.640467E+00	0.665972E+00	0.361453E+01	0.385510E+03	0.505199E+03	
PCNC	CNC	ZC	PRC	WACC	WACC	
0.114102E+03	0.100000E+01	0.100000E+01	0.100000E+01	0.131289E+03	0.505199E+03	
T2	P2	T21	P21	T3	P3	
0.875661E+03	0.170275E+01	0.132686E+04	0.615464E+01	0.132686E+04	0.615464E+01	
PCBLF	BLF	PCBLC	BLC	PCBLOB	BLOB	
0.	0.	0.970000E-01	0.490043E+02	0.	0.	
PCBLHP	BLHP	PCBLLP	BLLP	T4	P4	
0.	0.	0.100000E+01	0.490043E+02	0.326000E+04	0.574667E+01	
WA3	WFB	WG4	FAR4	ETAR	DPCDM	
0.454195E+03	0.153057E+02	0.471501E+03	0.335508E-01	0.985000E+00	0.662867E-01	
TFHHP	CNHP	DHTCHP	DHTC	T5	P5	
0.	0.	0.	0.	0.326000E+04	0.574667E+01	
TFHLP	CNLP	DHTCLP	DHTF	T55	P55	
0.318769E+03	0.145751E+01	0.160734E-01	0.121704E+03	0.274010E+04	0.302911E+01	
PCBLDU	BLDU	T24	P24	T25	P25	
0.	0.	0.	0.	0.132686E+04	0.615464E+01	
WAD	WFD	WG24	FAR24	ETAD	DPDUC	
0.	0.	0.505199E+03	0.	0.	0.	
ETAF	ETAC	ETATHP	ETATLP	AM55	AM25	
0.815437E+00	0.100000E+01	0.100000E+01	0.909909E+00	0.238017E+00	0.	
T6	P6	PS6	AM6	V6	WG6	
0.274010E+04	0.302911E+01	0.292530E+01	0.238010E+00	0.583946E+03	0.520505E+03	
T7	WFA	WG7	FAR7	ETAA	DPAFT	
0.274010E+04	0.	0.520505E+03	0.302964E-01	0.	0.480007E-01	
PS8	AMB	V8	PS9	AM9	V9	
0.158578E+01	0.100000E+01	0.230699E+04	0.158578E+01	0.100000E+01	0.230699E+04	
PS28	AM28	V28	PS29	AM29	V29	
0.	0.	0.	0.	0.	0.	
BYPASS	HPEXT	WFT	WGT	VA	FRD	
0.	0.	0.153057E+02	0.520505E+03	0.242123E+04	0.380184E+05	
CVMNOZ	VJM	CVNOZ	VJD	FGM	FGP	
0.835763E+00	0.192812E+04	0.	0.	0.311927E+05	0.267311E+05	
MAIN SCNIC CONVERGENT NOZZLE		FG= 57923.74	FN= 19905.36	SFC= 2.76813		
DUCT SHOCK OUTSIDE C-U NOZZLE						
CONVERGED AFTER 8 LOOPS						

Case 7: The performance data from this case reflect an expansion to the maximum attainable area ratio for A9/A8, thus simulating a convergent-divergent nozzle. Performance is calculated such that the nozzle static pressure PS9 is equal to the free-stream static pressure PS0. This is accomplished by setting NOZFLT equal to 1. The improvement in thrust and specific fuel consumption from case 6 should be noted.

Although engine performance is calculated as if complete expansion occurred (A9/A8 = 3.449), a performance penalty from the fully expanded case resulted because this particular nozzle has a physical expansion limit A9/A8 of 2.15. This means that the nozzle has a lower-than-optimum thrust coefficient, resulting in a performance penalty.

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SD(1),NOZFLT=1, * PERFORMANCE DATA WITH MAXIMUM ATTAINABLE NOZZLE RATIO
NOZZLE DESIGN          A8= 0.85821256E+01  AM8= 0.10000000E+01  A9= 0.29598886E+02  AM9= 0.27020654E+01
ARATIO= 3.449      OUT OF RANGE,USE DATA FOR 2.150
CUTPUT
AM= 2.500          ALTP= 50000.          T4= 3260.00          ETAR= 0.8703
PCNF              CNF              ZF              PRF              WAF              WAF
0.832186E+02     0.640467E+00     0.665972E+00     0.361453E+01     0.385510E+03     0.505199E+03
PCNC              CNC              ZC              PRC              WACC              WAC
0.114102E+03     0.100000E+01     0.100000E+01     0.100000E+01     0.131289E+03     0.505199E+03
T2               P2               T21             P21              T3               P3
0.875661E+03     0.170275E+01     0.132686E+04     0.615464E+01     0.132686E+04     0.615464E+01
PCBLF            BLF              PCBLC           BLC              PCBL0B          RLOB
0.               0.               0.970000E-01     0.490043E+02     0.               0.
PCBLHP          BLHP            PCBLLP          BLLP             T4               P4
0.               0.               0.100000E+01     0.490043E+02     0.326000E+04     0.574667E+C1
WA3             WFB             WG4             FAR4             ETAB             DPCOM
0.456195E+03     0.153057E+02     0.471501E+03     0.335508E-01     0.985000E+00     0.662867E-01
TFFHP          CNHP            DHTCHP          DHTC             T5               P5
0.               0.               0.               0.               0.326000E+04     0.574667E+C1
TFFLP          CNLP            DHTCLP          DHTF             T55              P55
0.313767E+03     0.145751E+01     0.160734E-01     0.121704E+03     0.274010E+04     0.302911E+01
PCHLDU         BLDU            T24             P24              T25              P25
0.               0.               0.               0.               0.132686E+04     0.615464E+01
WAD            WFD             WG24            FAR24            ETAD             DPDUC
0.               0.               0.505199E+03     0.               0.               0.
ETAF           ETAC            ETATHP          ETATLP           AM55             AM25
0.815437E+00     0.100000E+01     0.100000E+01     0.909909E+00     0.238010E+00     0.
T6             P6             PS6             AMS              V6              WG6
0.274010E+04     0.302911E+01     0.292530E+01     0.238010E+00     0.583946E+03     0.520505E+03
T7             WFA            WG7             FAR7             ETAA             DPAPT
0.274010E+04     0.               0.520505E+03     0.302964E-01     0.               0.480047E-01
PSH           AM8            V8             PS9             AM9             V9
0.158634E+01     0.100000E+01     0.230699E+04     0.114456E+00     0.270307E+01     0.462581E+04
PS28          AM28           V28            PS29            AM29            V29
0.               0.               0.               0.               0.               0.
BYPASS        HPEXT          WFT            WGT              VA              FRD
0.               0.               0.153057E+02     0.520505E+03     0.242123E+04     0.380184E+C5
CVMNOZ        VJM            CVDNOZ          VJD              FGM              FGP
0.958966E+00     0.443600E+04     0.               0.               0.717646E+05     0.
MAIN SHOCK JJT SIDE C-D NOZZLE          FG= 71764.62          FM= 33746.24          SFC= 1.62279
DUCT SHOCK OUTSIDE C-D NOZZLE
CONVERGED AFTER 1 LOOPS

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Case 8: The engine is operating in the supersonic flight condition with partial afterburning. This case illustrates that for an afterburning condition (IAFTBN = 1), the program automatically expands the nozzle so PS9 is equal to PS0. There is no need to input NOZFLT = 1 as in the previous case.

SD(1),T4=3260,T7=2900,IAFTBN=1,+ PARTIAL AFTERBURNING DURING SUPERSONIC OPER.

NOZZLE DESIGN A8= 0.88924085E+01 AM8= 0.10000000E+01 A9= 0.30845876E+02 AM9= 0.26977559E+01

ARATIO= 3.459 OUT OF RANGE,USE DATA FOR 2.150

OUTPUT AM= 2.500 ALTP= 50000. T4= 3260.00 T7= 2900.00 ETAR= 0.8703

PCNF	CNF	ZF	PRF	WAF	WAF
0.832186E+02	0.640467E+00	0.665972E+00	0.361453E+01	0.385510E+03	0.505199E+03
PCNC	CNC	ZC	PRC	WACC	WAC
0.114102E+03	0.100000E+01	0.100000E+01	0.100000E+01	0.131289E+03	0.505199E+03
T2	P2	T21	P21	T3	P3
0.875661E+03	0.170275E+01	0.132686E+04	0.615464E+01	0.132686E+04	0.615464E+01
PCBLF	BLF	PCBLC	BLC	PCBL0B	BL0B
0.	0.	0.970000E-01	0.490043E+02	0.	0.
PCBLHP	BLHP	PCBLLP	BLLP	T4	P4
0.	0.	0.100000E+01	0.490043E+02	0.326000E+04	0.574667E+01
WA3	WFB	WG4	FAR4	ETAB	DPDOM
0.456195E+03	0.153057E+02	0.471501E+03	0.335508E-01	0.985000E+00	0.662867E-01
TFFHP	CNHP	DHTCHP	DHTC	T5	P5
0.	0.	0.	0.	0.326000E+04	0.574667E+01
TFFLP	CNLP	DHTCLP	DHTF	T55	P55
0.318769E+03	0.145751E+01	0.160734E-01	0.121704E+03	0.274010E+04	0.302911E+01
PCBLDU	BLDU	T24	P24	T25	P25
0.	0.	0.	0.	0.132686E+04	0.615464E+01
WAD	WFD	WG24	FAR24	ETAD	DPDUC
0.	0.	0.505199E+03	0.	0.	0.
ETAF	ETAC	ETATHP	ETATLP	AM55	AM25
0.815437E+00	0.100000E+01	0.100000E+01	0.909909E+00	0.238010E+00	0.
T6	P6	PS6	AM6	V6	WG6
0.274010E+04	0.302911E+01	0.292530E+01	0.238010E+00	0.583946E+03	0.520505E+03
TT	WFA	WG7	FAR7	ETAA	DPAFT
0.290000E+04	0.160194E+01	0.522107E+03	0.334673E-01	0.866827E+00	0.480047E-01
PS8	AM8	V8	PS9	AM9	V9
0.158508E+01	0.100000E+01	0.237121E+04	0.114456E+00	0.269776E+01	0.476814E+04
PS28	AM28	V28	PS29	AM29	V29
0.	0.	0.	0.	0.	0.
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.	0.	0.169076E+02	0.522107E+03	0.242123E+04	0.380184E+05
CVMNDZ	VJM	CVDNDZ	VJD	FGM	FGP
0.959036E+00	0.457282E+04	0.	0.	0.742059E+05	0.

MAIN SHOCK OUTSIDE C-D NOZZLE FG= 74209.85 FN= 36187.47 SFC= 1.6E2C1

DUCT SHOCK OUTSIDE C-D NOZZLE

CONVERGED AFTER 1 LOOPS

Case 9: The engine is operating in the supersonic condition at the maximum design afterburner temperature.

SD(1),T4=3260,T7=3560,IAFTBN=1,+ MAX AFTERBURNING AT SUPERSONIC OPERATION

NOZZLE DESIGN A8= 0.10150120E+02 AM8= 0.10000000E+01 A9= 0.36281996E+02 AM9= 0.26857718E+01

ARATIO= 3.575 OUT OF RANGE,USE DATA FOR 2.150

OUTPUT AM= 2.500 ALTP= 50000. T4= 3260.00 T7= 3560.00 ETAR= 0.8703

PCNF	CNF	ZF	PRF	WAF	WAF
0.832186E+02	0.640467E+00	0.665972E+00	0.361453E+01	0.385510E+03	0.505199E+03
PCNC	CNC	ZC	PRC	WACC	WAC
0.114102E+03	0.100000E+01	0.100000E+01	0.100000E+01	0.131289E+03	0.505199E+03
T2	P2	T21	P21	T3	P3
0.875661E+03	0.170275E+01	0.132686E+04	0.615464E+01	0.132686E+04	0.615464E+01
PCBLF	BLF	PCBLC	BLC	PCBL0B	BL0B
0.	0.	0.970000E-01	0.490043E+02	0.	0.
PCBLHP	BLHP	PCBLLP	BLLP	T4	P4
0.	0.	0.100000E+01	0.490043E+02	0.326000E+04	0.574667E+01
WA3	WFB	WG4	FAR4	ETAB	DPDOM
0.456195E+03	0.153057E+02	0.471501E+03	0.335508E-01	0.985000E+00	0.662867E-01
TFFHP	CNHP	DHTCHP	DHTC	T5	P5
0.	0.	0.	0.	0.326000E+04	0.574667E+01
TFFLP	CNLP	DHTCLP	DHTF	T55	P55
0.318769E+03	0.145751E+01	0.160734E-01	0.121704E+03	0.274010E+04	0.302911E+01
PCBLDU	BLDU	T24	P24	T25	P25
0.	0.	0.	0.	0.132686E+04	0.615464E+01
WAD	WFD	WG24	FAR24	ETAD	DPDUC
0.	0.	0.505199E+03	0.	0.	0.
ETAF	ETAC	ETATHP	ETATLP	AM55	AM25
0.815437E+00	0.100000E+01	0.100000E+01	0.909909E+00	0.238010E+00	0.
T6	P6	PS6	AM6	V6	WG6
0.274010E+04	0.302911E+01	0.292530E+01	0.238010E+00	0.583946E+03	0.520505E+03
TT	WFA	WG7	FAR7	ETAA	DPAFT
0.356000E+04	0.179442E+01	0.528449E+03	0.460213E-01	0.908862E+00	0.480047E-01
PS8	AM8	V8	PS9	AM9	V9
0.157511E+01	0.100000E+01	0.261903E+04	0.114456E+00	0.268577E+01	0.532344E+04
PS28	AM28	V28	PS29	AM29	V29
0.	0.	0.	0.	0.	0.
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.	0.	0.232499E+02	0.528449E+03	0.242123E+04	0.380184E+05
CVMNDZ	VJM	CVDNDZ	VJD	FGM	FGP
0.959380E+00	0.510720E+04	0.	0.	0.838843E+05	0.

MAIN SHOCK OUTSIDE C-D NOZZLE FG= 83884.25 FN= 45865.87 SFC= 1.82498

DUCT SHOCK OUTSIDE C-D NOZZLE

CONVERGED AFTER 1 LOOPS

Case 10: Ability to fix the mechanical shaft speed PCNF is a revision to reference 2 which was supplied by its author. This engine operation method is accomplished by setting the control MODE equal to 3. MODE = 3 allows the engine to run at some desired value of shaft speed while the primary stream area A8 is varied. The value of turbine inlet temperature necessary to properly match the components is calculated in the program.

The supersonic flight condition is used for illustrative purposes. Any condition desired could have been used. In case 9, the outer compressor shaft speed was considerably less than 100 percent when the design turbine inlet temperature T4 and area A8 were used. By using MODE = 3 instead of MODE = 0, the design T4 was exceeded when a 100-percent shaft speed was specified. A8 was increased by 1 percent. An attempt was made to run the design value of A8; but the T4 exceeded the program limit value of 4000° R and the components did not match. The limit value of a 4000° R temperature is used because the equations for the gas properties do not account for the dissociation of the products of combustion which would become significant at higher temperatures.

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SD(1),MODE=3,PCNF=100,A8=8.586*1.01,NOZFLT=0, * SETUP FOR 100 PERCENT COMP. SPEED
OUTPUT          AN= 2.500      ALTP= 50000.      T4= 3962.97      ETAR= 0.8703
                PCNF          CNF          ZF          PRF          WAF          WAF
0.100000E+03   0.769621E+00   0.736343E+00   0.502448E+01   0.473919E+03   0.621058E+03
                PCNC          CNC          ZC          PRC          WAC          WAC
0.119204E+03   0.100000E+01   0.100000E+01   0.100000E+01   0.121299E+03   0.621058E+03
                T2          P2          T21         P21         T3          P3
0.875661E+03   0.170275E+01   0.144819E+04   0.855543E+01   0.144819E+04   0.855543E+01
                PCBLF        BLF          PCBLC       BLC         PCBLOB      BLOB
0.          0.          0.970000E-01   0.602426E+02   0.          0.
                PCBLHP       BLHP        PCBLLP      BLLP        T4          P4
0.          0.          0.100000E+01   0.602426E+02   0.396297E+04   0.803147E+01
                WA3         WFB         WG4         FAR4        ETAB        DPCOM
0.560815E+03   0.259670E+02   0.586782E+03   0.463022E-01   0.985000E+00   0.612431E-01
                TFFHP       CNHP        DHTCHP      DHTC        T5          P5
0.          0.          0.          0.          0.396297E+04   0.803147E+01
                TFFLP       CNLP        DHTCLP      DHTF        T55         P55
0.312963E+03   0.158851E+01   0.167439E-01   0.153692E+03   0.332634E+04   0.412729E+01
                PC9LDU      BLDU        T24         P24         T25         P25
0.          0.          0.          0.          0.144819E+04   0.855543E+01
                WAD         WFD         WG24        FAR24       ETAD        DPDUC
0.          0.          0.621058E+03   0.          0.          0.
                ETAF        ETAC        ETATHP      ETATLP      AM55        AM25
0.840396E+00   0.100000E+01   0.100000E+01   0.908362E+00   0.240982E+00   0.
                T6         P6         PS6         AM6         V6          WG6
0.332634E+04   0.412729E+01   0.398387E+01   0.240982E+00   0.647423E+03   0.647025E+03
                T7         WFA        WG7         FAR7        ETAA        DPAFT
0.332634E+04   0.          0.647025E+03   0.418109E-01   0.          0.482537E-01
                PS8         AM8        V8          PS9         AM9         V9
0.217407E+01   0.100000E+01   0.253416E+04   0.217407E+01   0.100000E+01   0.253416E+04
                PS28        AM28       V28         PS29        AM29        V29
0.          0.          0.          0.          0.          0.
                BYPASS      HPEXT      WFT         WGT         VA          FRD
0.          0.          0.259670E+02   0.647025E+03   0.242123E+04   0.467372E+05
                CVMNOZ      VJM        CVDNOZ      VJD         FGM         FGP
0.801550E+00   0.203126E+04   0.          0.          0.408489E+05   0.377970E+05
                MAIN SCNIC CONVERGENT NOZZLE      FG= 78645.91      FN= 31908.72      SFC= 2.92964
                DUCT SHOCK OUTSIDE C-D NOZZLE
CONVERGED AFTER 7 LOOPS

```

Case 11: This case indicates that if MODE = 3 and the area is changed, T4 will change. The nozzle throat area has been opened to 120 percent of its design value (A8 = 8.586 × 1.20). The engine components match at a point where the calculated T4 is less than the design value and the compressor shaft speed is 100 percent of design. Area

variation A8 and compressor speed variations are the options that are available when MODE = 3. The engine components are matched in all setup cases with the engine operating with the throat area A8 equal to A9 as if the engine has a convergent nozzle. In order to return to a convergent nozzle, an input of the control variable NOZFLT = 0 is required when a new setup case is run because NOZFLT is not reset to zero internally. This case is a setup case and illustrates only one of the many options available for operation of the GENENG program.

```

$D(1),NOZFLT=1,* PERFORMANCE WITH BEST NOZZLE AREA RATIO
NOZZLE DESIGN      A8= 0.86801134E+01   AM8= 0.10000000E+01   A9= 0.37183272E+02   AM9= 0.28765206E+01
ARATIO= 4.284      OUT OF RANGE,USE DATA FOR 2.150
CUTPUT
AM= 2.500          ALTP= 50000.          T4= 3962.97          ETAR= 0.6703
PCNF 0.100000E+03  CNF 0.769620E+00  ZF 0.736343E+00  PRF 0.502448E+01  W AFC 0.473919E+03  WAF 0.621058E+03
PCNC 0.119204E+03  CNC 0.100000E+01  ZC 0.100000E+01  PRC 0.100000E+01  WACC 0.121299E+03  WAC 0.621058E+03
T2 0.875661E+03  P2 0.170275E+01  T21 0.144819E+04  P21 0.855543E+01  T3 0.144819E+04  P3 0.855543E+01
PCBLF 0.  BLF 0.  PCBLC 0.970000E-01  BLC 0.602426E+02  PCBLDB 0.  BLOB 0.
PCBLHP 0.  BLHP 0.  PCBLLP 0.100000E+01  BLLP 0.602426E+02  T4 0.  P4 0.
WA3 0.560815E+03  WFB 0.259670E+02  WG4 0.586782E+03  FAR4 0.463022E-01  ETAB 0.396297E+04  DPCOM 0.303147E+01
TFFHP 0.  CNHP 0.  DHTCHP 0.  DHTC 0.985000E+00  T5 0.612431E-01  P5 0.
TFFLP 0.  CNLP 0.  DHTCLP 0.  DHTF 0.396297E+04  T55 0.803147E+01  P55 0.
PCBLDU 0.312963E+03  BLDU 0.158851E+01  T24 0.167439E-01  P24 0.153692E+03  T25 0.332634E+04  P25 0.412729E+01
WAD 0.  WFD 0.  WG24 0.621058E+03  FAR24 0.  ETAD 0.855543E+01  OPDUC 0.
ETAF 0.840396E+00  ETAC 0.100000E+01  ETATHP 0.100000E+01  ETATLP 0.908362E+00  AM55 0.  AM25 0.
T6 0.332634E+04  P6 0.412729E+01  PS6 0.398386E+01  AM6 0.240982E+00  V6 0.  WG6 0.547025E+03
T7 0.332634E+04  WFA 0.  WG7 0.647025E+03  FAR7 0.418109E-01  ETAA 0.  DPAFT 0.482538E-01
PS8 0.217200E+01  AM8 0.100000E+01  V8 0.257416E+04  PS9 0.114456E+00  AM9 0.287652E+01  V9 0.529546E+04
PS28 0.  AM28 0.  V28 0.  PS29 0.  AM29 0.  V29 0.
BYPASS 0.  HPEXT 0.  WFT 0.  WGT 0.  VA 0.  FRD 0.
CVMNOZ 0.  VJM 0.  CVDNOZ 0.259670E+02  VJD 0.647025E+03  FGM 0.242123E+04  FGP 0.467372E+05
0.948016E+00 0.502018E+04 0. 0. 0. 0.100956E+06 0.
MAIN SHOCK OUTSIDE C-D NOZZLE FG=100956.46 FN= 54219.26 SFC= 1.72413
DUCT SHOCK OUTSIDE C-D NOZZLE
CONVERGED AFTER 1 LOOPS

```

Case 12: The final example case is the performance for the previous case (11) when the maximum attainable area expansion is used (PS9 = PS0).

```

$D(1),MODE=3,PCNF=100,A8=8.586*1.20,NOZFLT=0,* SET UP FOR 100 PERCENT
OUTPUT
AM= 2.500          ALTP= 50000.          T4= 3157.72          ETAR= 0.8703
PCNF 0.100000E+03  CNF 0.769620E+00  ZF 0.613211E+00  PRF 0.435150E+01  W AFC 0.488534E+03  WAF 0.640210E+03
PCNC 0.117487E+03  CNC 0.100000E+01  ZC 0.100000E+01  PRC 0.100000E+01  WACC 0.142298E+03  WAC 0.640210E+03
T2 0.875661E+03  P2 0.170275E+01  T21 0.140676E+04  P21 0.740952E+01  T3 0.140676E+04  P3 0.740952E+01
PCBLF 0.  BLF 0.  PCBLC 0.970000E-01  BLC 0.621004E+02  PCBLDB 0.  BLOB 0.
PCBLHP 0.  BLHP 0.  PCBLLP 0.100000E+01  BLLP 0.621004E+02  T4 0.  P4 0.
WA3 0.578110E+03  WFB 0.174823E+02  WG4 0.595992E+03  FAR4 0.302405E-01  ETAB 0.315772E+04  DPCOM 0.687718E+01
TFFHP 0.  CNHP 0.  DHTCHP 0.  DHTC 0.985000E+00  T5 0.718451E-01  P5 0.
TFFLP 0.331151E+03  CNLP 0.177956E+01  DHTCLP 0.196409E-01  DHTF 0.315772E+04  T55 0.687718E+01  P55 0.311714E+01
0.331151E+03 0.177956E+01 0.196409E-01 0.144425E+03 0.258009E+04 0.311714E+01

```



PCBLDU	BLDU	T24	P24	T25	P25
0.	0.	0.	0.	0.140676E+04	0.740952E+01
WAD	WFD	WG24	FAR24	ETAD	DPDUC
0.	0.	0.640210E+03	0.	0.	0.
ETAF	ETAC	ETATHP	ETATLP	AM55	AM25
0.810314E+00	0.100000E+01	0.100000E+01	0.917729E+00	0.287188E+00	0.
T6	P6	PS6	AM6	V6	WG6
0.258009E+04	0.311714E+01	0.296110E+01	0.287188E+00	0.683798E+03	0.657693E+03
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.258009E+04	0.	0.657693E+03	0.273072E-01	0.	0.571973E-01
PS8	AM8	V8	PS9	AM9	V9
0.161452E+01	0.100000E+01	0.224079E+04	0.161452E+01	0.100000E+01	0.224079E+04
PS28	AM28	V28	PS29	AM29	V29
0.	0.	0.	0.	0.	0.
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.	0.	0.174823E+02	0.657693E+03	0.242123E+04	0.481785E+05
CVMNOZ	VJM	CVDNOZ	VJD	FGM	FGP
0.833962E+00	0.186873E+04	0.	0.	0.382001E+05	0.327071E+05

MAIN SONIC CONVERGENT NOZZLE      FG= 70907.18      FN= 22728.66      SFC= 2.76903  
 DUCT SHOCK OUTSIDE C-D NOZZLE  
 CONVERGED AFTER 8 LUOPS

## Two-Spool Turbojet

Design point. - An example is given for design-point operation of a two-spool turbojet. For the design point the data are printed out in three groups: (1) card images of the input data that were changed from the original input data, (2) results, and (3) common variables at the design point. The card images show two cards having \$D(1), IDES = 1, imprinted. These two cards in sequence recall the original design inputs. This method can be used as many times as desired in any computer run to design a new engine. A typical use for this may be a study of several engines where design-point performance only is sufficient. Thus, only changes in engine design characteristics need be input and a new engine is run. Or, as in this case, the engine was changed from a single-spool turbojet to a two-spool turbojet by changing ISPOOL = 1 to ISPOOL = 2 and by inputting the design pressure ratio and efficiency of the inner compressor and the design efficiency speed and flow function of the inner turbine. Also a change to the front compressor pressure ratio was made to have an overall pressure ratio of 8:1. The MODE was reset to zero (MODE = 0) from the previous cases of the single-spool turbojet engine.

```

SD(1),IDES=1, * FIRST OF TWO IDES=1 CARDS NECESSARY TO CHANGE DESIGN POINT
SD(1),IDES=1, * SECOND OF TWO IDES=1 CARDS NECESSARY TO CHANGE DESIGN POINT
ISPOOL=2,* TWO SPOOL TURBOJET
PRFDS=2,PRCDS=4,ETACDS=.8>,PCNCDS=100,ZCDS=.81433225, * OVERALL P RATIO STILL 8
TFHPDS=50.00,CNHPS=2.000,ETHPDS=.90,MODE=0, * ADDITIONAL INPUT WAS REQUIRED TO
* RUN TWO SPOOL TURBOJETS, MODE = 0 RESETS TO USE OF T4 RATHER THAN SPEED

```

FAN DESIGN	PRFCF= 0.2500000E+01	ETAFCF= 0.9420454E+00	WAFCF= 0.1055000E+01	T2DS= 0.51866820E+03
COMPRESSOR DESIGN	PRCCF= 0.42857143E+00	ETACCF= 0.98837209E+00	WACCF= 0.35584073E+01	T21DS= 0.65562200E+03
COMBUSTER DESIGN	WA3CDS= 0.15548823E+03	ETABCF= 0.98500000E+00	DTCCCF= 0.10000000E+01	
H.P. TURBINE DESIGN	CNHPCF= 0.11419282E+01	TFHPCF= 0.16375691E+00	ETHPCF= 0.10000000E+01	DHHPCF= 0.48959067E+00
L.P. TURBINE DESIGN	CNLPCF= 0.12507844E+01	TFLPCF= 0.27070545E+00	ETLPCF= 0.10235344E+01	DHLPCF= 0.68989028E+00
TURBINE AREA DESIGN	A55= 0.21028949E+02	AM55= 0.23823737E+00		
AFTERBURNER DESIGN	WG6CDS= 0.90353905E+04			
NOZZLE DESIGN	A8= 0.86244577E+01	AM8= 0.10000000E+01	A9= 0.86244577E+01	AM9= 0.10000000E+01

```

OUTPUT          AM= 0.      ALTP= 0.      T4= 3260.00      ETAR= 1.0000
PCNF           CNF       ZF          PRF          WAF          WAF
0.100000E+03  0.100000E+01  0.833333E+00  0.200000E+01  0.633000E+03  0.633000E+03
PCNC           CNC       ZC          PRC          WACC          WACC
0.100000E+03  0.100000E+01  0.814332E+00  0.400000E+01  0.355841E+03  0.633000E+03
T2            T21      T21         P21          T3           T3
0.518666E+03  0.100000E+01  0.655622E+03  0.200000E+01  0.102280E+04  0.800000E+01
PCHLF         BLF      PCBL        BLC          PCBL0B       BLOB
0.           0.       0.970000E-01  0.614010E+02  0.           0.
PCBLHP        BLHP     PCBLP       BLLP         T4           P4
0.           0.       0.100000E+01  0.614010E+02  0.326000E+04  0.755200E+C1
WA3           WFB      WG4         FAR4         ETAR         DPCOM
0.571594E+03  0.219038E+02  0.593503E+03  0.383203E-01  0.985000E+00  0.560000E-01
TFHP         CNHP     DHTCHP      DHTC         T5           P5
0.305331E+03  0.175142E+01  0.600000E-01  0.957639E+02  0.295739E+04  0.457332E+01
TFFLP        CNLP     DHTCLP      DHTF         T55          P55
0.480227E+03  0.183885E+01  0.171750E-01  0.350417E+02  0.269456E+04  0.376248E+01
PCBL0U       BLDU     T24         P24          T25          P25
0.           0.       0.           0.           0.655622E+03  0.200000E+01
WAD           WFD      WG24        FAR24        ETAD         DPDOC
0.           0.       0.633000E+03  0.           0.           0.
ETAF          ETAC     ETATHP      ETATLP       AM55         AM25
0.829000E+00  0.850000E+00  0.900000E+00  0.903000E+00  0.238237E+00  0.
T6           P6       PS6         AM5          V6           WG6
0.269456E+04  0.376248E+01  0.362824E+01  0.238237E+00  0.579167E+03  0.654904E+03
T7           WFA      WG7         FAR7         E7AA        DPAFT
0.269456E+04  0.           0.654904E+03  0.345032E-01  0.           0.480000E-01
PS8          AM8      PS8         AM9          V9           P9
0.197183E+01  0.100000E+01  0.228725E+04  0.197183E+01  0.100000E+01  0.228725E+04
PS28         AM28     PS29        AM29         V29          P29
0.           0.       0.           0.           0.           0.
BYPASS        HPEXT    WFT         HGT          VA           FRD
0.           0.       0.219038E+02  0.654904E+03  0.           0.
CVMNDZ        VJM      CVDNDZ      VJD          FGM          FGP
0.967850E+00  0.221372E+04  0.           0.           0.450603E+05  0.177371E+05
MAIN SCNIC CONVERGENT NOZZLE      FG= 62797.45      FN= 62797.45      SFC= 1.25568
DUCT SHOCK OUTSIDE C-D NOZZLE

```

CONVERGED AFTER 1 LOOPS

```

COMMON 0.833333E+00 0.100000E+03 -0.000000E-19 -0.000000E-19 0.814332E+00 0.100000E+03 0.326000E+04 0
0.100000E+03 0.100000E+03 0.326000E+04 0.100000E+03 0. 0.100000E+01 0.100000E+01 0.100000E+01
0.833333E+00 0.100000E+03 0.200000E+01 0.829000E+00 0.633000E+03 0.250000E+01 0.942045E+00 0.105500E+01
0.814332E+00 0.100000E+03 0.400000E+01 0.850000E+00 0.633000E+03 0.428571E+00 0.988372E+00 0.355841E+01
0.326000E+04 0. 0.223720E+04 0.985000E+00 0.155488E+03 0.560000E-01 0.100000E+01 0.985000E+00
0.500000E+02 0.200000E+01 0.900000E+00 0.163757E+00 0.114193E+01 0.100000E+01 0.4489591E+00 0.518666E+03
0.130000E+03 0.230000E+01 0.903000E+00 0.270705E+00 0.125078E+01 0.102353E+01 0.689890E+00 0.655622E+03
0. 0. 0. 0. 0. 0. 0. 0.
0.356000E+04 0. 0. 0.900000E+00 0.903539E+04 0.480000E-01 0. 0. 0.
0.210249E+02 0. 0.210289E+02 0.210289E+02 0.862446E+01 0.862446E+01 0. 0. 0.
0.362824E+01 0.238237E+00 0. 0.967850E+00 0. 0. 0. 0.
0.518670E+03 0.100000E+01 0.123918E+03 0.159103E+01 0.518668E+03 0.100000E+01 0.123918E+03 0.159103E+01
0.655622E+03 0.200000E+01 0.156774E+03 0.159970E+01 0.102280E+04 0.800000E+01 0.246562E+03 0.161325E+01
0.326000E+04 0.755200E+01 0.916050E+03 0.196075E+01 0.295739E+04 0.457332E+01 0.820286E+03 0.196436E+01
0.269456E+04 0.376248E+01 0.734740E+03 0.194651E+01 0. 0.614010E+02 0. 0.
0.100000E+01 0.200000E+01 0.829000E+00 0.633000E+03 0.633000E+03 0.571599E+03 0.593503E+03 0.383203E-01
0.100000E+01 0.400000E+01 0.850000E+00 0.355841E+03 0.633000E+03 0.985000E+00 0.560000E-01 0.
0.700000E+01 0.900000E+00 0.293754E-01 0.957639E+02 0. 0.593503E+03 0.383203E-01 0.116999E+04
0.230000E+01 0.903000E+01 0.118488E-01 0.350417E+02 0.614010E+02 0.654904E+03 0.346032E-01 0.
0. 0. 0. 0.833333E+00 0. 0.100000E+03 0.814332E+00 0. 0. 0.219038E+02 0.100000E+03
0.500000E+02 0.130000E+03 0. 0.970000E-01 0. 0. 0. 0.
0.100000E+01 0.633000E+03 0.633000E+03 0. 0. 0.246562E+03 0. 0.
0.655622E+03 0.200000E+01 0.156774E+03 0.159970E+01 0. 0. 0. 0.
0. 0. 0. 0. 0. 0.655622E+03 0.200000E+01 0.156774E+03 0.159970E+01
0. 0. 0. 0. 0. 0. 0. 0.
0. 0. 0.633000E+03 0. 0. 0. 0. 0. 0.
0.269456E+04 0.376248E+01 0.734740E+03 0.194651E+01 0.655622E+03 0.200000E+01 0.156774E+03 0.159970E+01
0.219038E+02 0.654904E+03 0.346032E-01 0. 0.633000E+03 0. 0.100000E+01 0.
0.269456E+04 0.376248E+01 0.734740E+03 0.194651E+01 0.269456E+04 0.358188E+01 0.734737E+03 0.194989E+01
0.269456E+04 0.358188E+01 0.734737E+03 0.194989E+01 0.269456E+04 0.358188E+01 0.734737E+03 0.194989E+01
0.654904E+03 0. 0.654904E+03 0.346032E-01 0. 0.480000E-01 0.579167E+03 0.
0.362824E+01 0.579167E+03 0.238237E+00 0.267059E+04 0.344111E+01 0.610194E+03 0.251101E+00 0.
0.235272E+04 0.197183E+01 0.228725E+04 0.100000E+01 0.235272E+04 0.197183E+01 0.228725E+04 0.100000E+01
0. 0. 0. 0. 0.221372E+04 0.450603E+05 0. 0.177371E+05 0.
0.450603E+05 0.177371E+05 0.219038E+02 0.654904E+03 0.346032E-01 0.627974E+05 0.627974E+05 0.125568E+01

```

Off-design. - The following cases are off-design cases for the two-spool turbojet.

Case 1: The engine is operating at a reduced turbine inlet temperature at sea-level-static condition. The input change of T4 was required. The compressor shaft speed of both compressors is reduced as are the pressure ratios, airflows, etc., all of which are representative of part-power operation.

```

$D(1),T4=2660,*1 PART POWER ENGINE OPERATION
* RESULTS NORMAL OFF-DESIGN OUTPUT
OUTPUT          AM= 0.      ALTP= 0.      T4= 2660.00      ETAR= 1.0000
PCNF            CNF          ZF          PRF          WAF          WAF
0.836241E+02   0.836241E+00   0.849728E+00  0.175763E+01  0.503176E+03  0.503176E+C2
PCNC            CNC          ZC          PRC          WACC         WAC
0.904097E+02   0.925609E+00   0.709173E+00  0.324591E+01  0.314397E+03  0.503194E+C3
T2             P2          T21         P21          T3           P3
0.518668E+03   0.100000E+01   0.625500E+03  0.175763E+01  0.930191E+03  0.570511E+01
PCBLF          BLF          PCBLC       BLC          PCBLOB       BLOB
0.             0.           0.970000E-01  0.498098E+02  0.             0.
PCBLHP         BLHP        PCBLLP      BLLP         T4           P4
0.             0.           0.100000E+01  0.488098E+02  0.266000E+04  0.536548E+01
WA3            WFR         WG4         FAR4         ETAB         DPCDM
0.454384E+03   0.127365E+02   0.467120E+03  0.280302E-01  0.985000E+00  0.595301E-01
TFFHP         CNHP        DHTCHP      DHTC         T5           P5
0.305530E+03   0.175297E+01   0.613448E-01  0.797926E+02  0.239574E+04  0.320282E+1
TFFLP         CNLP        DHTCLP      DHTF         T55          P55
0.485755E+03   0.170349E+01   0.166673E-01  0.275883E+02  0.218600E+04  0.265090E+C1
PCBLDU        BLDU        T24         P24          T25          P25
0.             0.           0.             0.             0.625500E+03  0.175763E+01
WAD            WFD         WC24        FAR24        ETAD         DPDU
0.             0.           0.503176E+03  0.             0.             0.
ETAF           ETAC        ETATHP      ETATLP       AM55         AM25
0.848567E+00   0.809756E+00   0.897395E+00  0.906332E+00  0.238214E+00  0.
T6            P6          PS6         AM6          V6           WG6
0.218600E+04   0.265090E+01   0.255886E+01  0.238214E+00  0.525189E+03  0.515930E+C3
T7            WFA        WG7         FAR7         ETAA         DPAFT
0.218600E+04   0.             0.515930E+03  0.253113E-01  0.             0.483412E-01
PS8           AM8        V8          PS9          AM9          V9
0.138197E+01   0.100000E+01   0.206732E+04  0.138197E+01  0.100000E+01  0.206732E+04
PS28          AM28       V28         PS29         AM29         V29
0.             0.           0.             0.             0.             0.
BYPASS        HPEXT      WFT         WGT          VA           FRD
0.             0.           0.127365E+02  0.515913E+03  0.             0.
CVMNOZ        VJM        CVDNOZ      VJD          FGM          FGP
0.978239E+00   0.202233E+04   0.             0.             0.324293E+05  0.697137E+14
MAIN SCNIC CONVERGENT NOZZLE      FG= 39400.71      FN= 39400.71      SFC= 1.16372
DUCT SHOCK OUTSIDE C-D NOZZLE
CONVERGED AFTER 13 LOOPS

```

Case 2: The engine is now operating at a supersonic flight Mach number (2.5) at an altitude of 50 000 feet. The engine components are matched with an engine nozzle throat area A8 equal to the sea-level-static design value. This case is a setup for matching components only.

```

$D(1),T4=3260,AM=2.5,ALTP=50000,* SET UP FOR COMPONENT MATCHING ONLY
* RESULTS NORMAL OFF-DESIGN OUTPUT
OUTPUT          AM= 2.500    ALTP= 50000.    T4= 3260.00      ETAR= C.8703
PCNF            CNF          ZF          PRF          WAF          WAF
0.851964E+02   0.655689E+00   0.823071E+00  0.148416E+01  0.373082E+03  0.488914E+02
PCNC            CNC          ZC          PRC          WACC         WAC
0.102605E+03   0.832254E+00   0.556194E+00  0.245472E+01  0.268143E+03  0.488881E+03
T2             P2          T21         P21          T3           P3
0.875661E+03   0.170275E+01   0.996510E+03  0.252716E+01  0.135728E+04  0.520347E+01
PCBLF          BLF          PCBLC       BLC          PCBLOR       BLOB
0.             0.           0.970000E-01  0.474215E+02  0.             0.
PCBLHP         BLHP        PCBLLP      BLLP         T4           P4
0.             0.           0.100000E+01  0.474215E+02  0.326000E+04  0.580489E+01
WA3            WFR         WG4         FAR4         ETAB         DPCDM
0.441460E+03   0.145963E+02   0.456056E+03  0.330637E-01  0.985000E+00  0.642514E-C1
TFFHP         CNHP        DHTCHP      DHTC         T5           P5
0.305235E+03   0.179705E+01   0.618535E-01  0.982400E+02  0.294654E+04  0.346287E+01
TFFLP         CNLP        DHTCLP      DHTF         T55          P55
0.486451E+03   0.156951E+01   0.157361E-01  0.319779E+02  0.271686E+04  0.289862E+01
PCBLDU        BLDU        T24         P24          T25          P25
0.             0.           0.             0.             0.996510E+03  0.252716E+01
WAD            WFD         WC24        FAR24        ETAD         DPDU
0.             0.           0.488914E+03  0.             0.             0.
ETAF           ETAC        ETATHP      ETATLP       AM55         AM25
0.840470E+00   0.758931E+00   0.898441E+00  0.905158E+00  0.238434E+00  0.
T6            P6          PS6         AM6          V6           WG6
0.271686E+04   0.289862E+01   0.279887E+01  0.238434E+00  0.582650E+03  0.503478E+03
T7            WFA        WG7         FAR7         ETAA         DPAFT
0.271686E+04   0.             0.503478E+03  0.293565E-01  0.             0.480968E-01
PS8           AM8        V8          PS9          AM9          V9
0.151968E+01   0.100000E+01   0.229751E+04  0.151968E+01  0.100000E+01  0.229751E+C4
PS28          AM28       V28         PS29         AM29         V29
0.             0.           0.             0.             0.             0.
BYPASS        HPEXT      WFT         WGT          VA           FRD
0.             0.           0.145963E+02  0.503510E+03  0.242123E+04  0.367928E+05
CVMNOZ        VJM        CVDNOZ      VJD          FGM          FGP
0.839848E+00   0.192956E+04   0.             0.             0.301948E+05  0.256471E+C5
MAIN SCNIC CONVERGENT NOZZLE      FG= 55841.86      FN= 19049.05      SFC= 2.75849
DUCT SHOCK OUTSIDE C-D NOZZLE
CONVERGED AFTER 20 LOOPS

```

Case 3: The performance data for this case reflect an expansion to the maximum attainable area ratio for A9/A8, thus simulating a convergent-divergent nozzle. Performance is calculated such that the nozzle static pressure PS9 is equal to the free-stream static pressure PS0. This is accomplished by setting NOZFLT = 1.

```

SD(1),NOZFLT=1, * PERFORMANCE WITH MAX ATTAINABLE AREA RATIO
* RESULTS N O R M A L O F F - D E S I G N O U T P U T
NOZZLE DESIGN          A8= 0.86367638E+01    A8= 0.10000000E+01    A9= 0.28983159E+02    A9= 0.26768298E+01
ARATIO= 3.356      OUT OF RANGE,USE DATA FOR 2.150
OUTPUT
      AM= 2.500      ALTP= 50000.      T4= 3260.00      ETAR= 0.8703
      PCNF          CNF          ZF          PRF          WAF          WAF
0.851964E+02      0.655689E+00      0.823071E+00      0.148416E+01      0.373082E+03      0.488914E+03
      PCNC          CNC          ZC          PRC          WAC          WAC
0.102605E+03      0.832254E+00      0.556194E+00      0.245472E+01      0.268143E+03      0.488881E+03
      T2            P2            T21         P21          T3            P3
0.875661E+03      0.170275E+01      0.996510E+03      0.252716E+01      0.135728E+04      0.620347E+01
      PCBLF         BLF          PCRLC       BLC          PCBL0B       BLOB
0.            0.            0.970000E-01      0.474215E+02      0.            0.
      PCBLHP        BLHP         PCBLLP      BLLP         T4            P4
0.            0.            0.100000E+01      0.474215E+02      0.326000E+04      0.580489E+01
      NA3           WFR          WG4          FAR4         ETAB          DPCDM
0.441460E+03      0.145963E+02      0.456056E+03      0.330637E-01      0.985000E+00      0.642514E-01
      TFFHP         CNHP         DHTCHP      DHTC         T5            P5
0.305235E+03      0.179705E+01      0.618535E-01      0.982400E+02      0.294654E+04      0.346287E+01
      TFFLP         CNLP         DHTCLP      DHTF         T55          P55
0.486451E+03      0.156951E+01      0.157361E-01      0.313779E+02      0.271686E+04      0.289862E+01
      PCBLDU        BLDU         T24         P24          T25          P25
0.            0.            0.            0.            0.996510E+03      0.252716E+01
      WAD           WFD          WG24        FAR24        ETAD          DPOUC
0.            0.            0.488914E+03      0.            0.            0.
      ETAF          ETAC         ETATHP      ETATLP       AM55         AM25
0.840476E+00      0.758931E+00      0.898441E+00      0.906158E+00      0.238434E+00      0.
      T6            P6            PS6         AM6          V6            WG6
0.271686E+04      0.289862E+01      0.279887E+01      0.238434E+00      0.582650E+03      0.503478E+03
      T7            WFA          WG7         FAR7         ETAA         OPAFT
0.271686E+04      0.            0.503478E+03      0.298565E-01      0.            0.480968E-C1
      PS8           AM8          V8          PS9          AM9          V9
0.151752E+01      0.100000E+01      0.229751E+04      0.114456E+00      0.267683E+01      0.458357E+04
      PS28          AM28         V28         PS29         AM29         V29
0.            0.            0.            0.            0.            0.
      BYPASS        HPEXT       WFT         WGT          VA           FRD
0.            0.            0.145963E+02      0.503510E+03      0.242123E+04      0.367928E+05
      CVMNDZ        VJM         CVDNOZ      VJD          FGM          FGP
0.960271E+0C      0.440147E+04      0.            0.            0.688767E+05      0.
MAIN SHOCK OUTSIDE C-D NOZZLE      FG= 68876.71      FN= 32083.89      SFC= 1.62779
DUCT SHOCK OUTSIDE C-D NOZZLE
CONVFRGED AFTER 1 LOUPS

```

Case 4: The engine is operating in the supersonic flight condition with partial afterburning. No new setup case was required because the components are still matched from the previous two cases (2 and 3). IAF'TBN = 1 automatically expands the nozzle so PS9 = PS0. It also adjusts A8 for the new temperature and pressure condition such that the nozzle throat is sonic. Both A8 and A9 are printed out at the top of the listing.

```

SD(1),T7=2900,IAFTBN=1, * PARTIAL AFTERBURNING AT SUPERSONIC OPERATION
* RESULTS N O R M A L O F F - D E S I G N O U T P U T
NOZZLE DESIGN          A8= 0.89961808E+01    A8= 0.10000000E+01    A9= 0.30389738E+02    A9= 0.26712209E+01
ARATIO= 3.378      OUT OF RANGE,USE DATA FOR 2.150
OUTPUT
      AM= 2.500      ALTP= 50000.      T4= 3260.00      T7= 2900.0C      ETAR= 0.8703
      PCNF          CNF          ZF          PRF          WAF          WAF
0.851964E+02      0.655689E+00      0.823071E+00      0.148416E+01      0.373082E+03      0.488914E+03
      PCNC          CNC          ZC          PRC          WAC          WAC
0.102605E+03      0.832254E+00      0.556194E+00      0.245472E+01      0.268143E+03      0.488881E+03
      T2            P2            T21         P21          T3            P3
0.875661E+03      0.170275E+01      0.996510E+03      0.252716E+01      0.135728E+04      0.620347E+01
      PCBLF         BLF          PCRLC       BLC          PCBL0B       BLOB
0.            0.            0.970000E-01      0.474215E+02      0.            0.
      PCBLHP        BLHP         PCBLLP      BLLP         T4            P4
0.            0.            0.100000E+01      0.474215E+02      0.326000E+04      0.580489E+01

```

WA3	WFB	WG4	FAR4	ETAB	DPCOM
0.441460E+03	0.145963E+02	0.456056E+03	0.330637E-01	0.985000E+0C	0.642514E-01
TFHP	CNHP	DHTCHP	DHTC	T5	P5
0.305235E+03	0.179705E+01	0.618535E-01	0.982400E+02	0.294654E+04	0.346287E+01
TFFLP	CNLP	DHTCLP	DHTF	T55	P55
0.486451E+03	0.156951E+01	0.157361E-01	0.319779E+02	0.271686E+04	0.289862E+01
PCBLDU	BLDU	T24	P24	T25	P25
0.	0.	0.	0.	0.996510E+03	0.252716E+01
WAD	WFD	WG24	FAR24	ETAD	DPDUC
0.	0.	0.488914E+03	0.	0.	0.
ETAF	ETAC	ETATHP	ETATLP	AM55	AM25
0.840476E+00	0.758931E+00	0.898441E+00	0.905158E+00	0.238435E+00	0.
T6	P6	PS6	AM6	V6	WG6
0.271686E+04	0.289862E+01	0.279887E+01	0.238435E+00	0.582650E+03	0.503477E+03
T7	WFA	WG7	FAR7	ETAA	DPAFT
0.290000E+04	0.171925E+01	0.505197E+03	0.333732E-01	0.893552E+0C	0.480968E-01
PS8	AM8	V8	PS9	AM9	V9
0.151594E+01	0.100000E+01	0.237123E+04	0.114456E+00	0.267122E+01	0.474598E+04
PS28	AM28	V28	PS29	AM29	V29
0.	0.	0.	0.	0.	0.
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.	0.	0.163155E+02	0.505229E+03	0.242123E+04	0.367928E+05
CVMNDZ	VJM	CVDNDZ	VJD	FGM	FGP
0.960350E+00	0.455780E+04	0.	0.	0.715666E+05	0.

MAIN SHOCK OUTSIDE C-D NOZZLE FG= 71566.56 FN= 34773.74 SFC= 1.68909  
 DUCT SHOCK OUTSIDE C-D NOZZLE  
 CONVERGED AFTER 1 LOOPS

**Case 5: The engine is operating in the supersonic flight condition at design after-burner temperature.**

ID(1),T7=3500,IAFTBN=1,\* MAX AFTERBURNING AT SUPERSONIC OPERATION  
 \* RESULTS NORMAL OFF-DESIGN OUTPUT

NOZZLE DESIGN AB= 0.10154220E+02 AM8= 0.10000000E+01 A9= 0.35230915E+02 AM9= 0.266C7638E+01  
 ARATIO= 3.470 OUT OF RANGE,USE DATA FOR 2.150

OUTPUT AM= 2.500 ALTP= 50000. T4= 3260.00 T7= 3500.00 ETAR= C.8703  
 PCVF CNF ZF PRF WAF WAF WAF  
 0.851964E+02 0.655689E+00 0.823071E+00 0.148416E+01 0.373082E+03 0.488914E+03  
 PCNC CNC ZC PRC WACC WACC WACC  
 0.102605E+03 0.832254E+00 0.556194E+00 0.245472E+01 0.268143E+03 0.488881E+03  
 T2 P2 T21 P21 T3 P3  
 0.875661E+03 0.170275E+01 0.996510E+03 0.252716E+01 0.135728E+04 0.620347E+01  
 PCBLF BLF PCHLC RLC PCBLOR RLOB  
 0. 0. 0.970000E-01 0.474215E+02 0. 0.  
 PCBLHP BLHP PCHLLP BLLP T4 P4  
 0. 0. 0.100000E+01 0.474215E+02 0.326000E+04 0.580489E+01  
 WA3 WFR WG4 FAR4 ETAR DPCOM  
 0.441460E+03 0.145963E+02 0.456056E+03 0.330637E-01 0.985000E+00 0.642514E-01  
 TFHP CNHP DHTCHP DHTC T5 P5  
 0.305235E+03 0.179705E+01 0.618535E-01 0.982400E+02 0.294654E+04 0.346287E+01  
 TFFLP CNLP DHTCLP DHTF T55 P55  
 0.486451E+03 0.156951E+01 0.157361E-01 0.319779E+02 0.271686E+04 0.289862E+01  
 PCBLDU RLDU T24 P24 T25 P25  
 0. 0. 0. 0. 0.996510E+03 0.252716E+01  
 WAD WFD WG24 FAR24 ETAD DPDUC  
 0. 0. 0.488914E+03 0. 0. 0.  
 ETAF ETAC ETATHP ETATLP AM55 AM25  
 0.840476E+00 0.758931E+00 0.898441E+00 0.905158E+00 0.238435E+00 0.  
 T6 P6 PS6 AM6 V6 WG6  
 0.271686E+04 0.289862E+01 0.279887E+01 0.238435E+00 0.582650E+03 0.503477E+03  
 T7 WFA WG7 FAR7 ETAA DPAFT  
 0.350000E+04 0.724094E+01 0.510718E+03 0.446677E-01 0.919029E+0C 0.480968E-01  
 PS8 AM8 V8 PS9 AM9 V9  
 0.150725E+01 0.100000E+01 0.259758E+04 0.114456E+00 0.266076E+01 0.524976E+04  
 PS28 AM28 V28 PS29 AM29 V29  
 0. 0. 0. 0. 0. 0.  
 BYPASS HPEXT WFT WGT VA FRD  
 0. 0. 0.218372E+02 0.510751E+03 0.242123E+04 0.367928E+05  
 CVMNDZ VJM CVDNDZ VJD FGM FGP  
 0.960653E+00 0.504319E+04 0. 0. 0.800537E+05 0.

MAIN SHOCK OUTSIDE C-D NOZZLE FG= 80053.69 FN= 43260.88 SFC= 1.81721  
 DUCT SHOCK OUTSIDE C-D NOZZLE  
 CONVERGED AFTER 1 LOOPS

## Separate-Flow Turbofan

**Design point.** - This is the design case for the separate-flow turbofan. For the design point, the data that were changed are shown on the top of the following example.

The control card FAN = .TRUE. designates that there is now a bypass ratio. Thus the front compressor (fan) flow does not equal the inner compressor flow. Consequently, a value for the inner compressor flow must be input. This value determines the engine bypass ratio. Also, a duct Mach number AM23 must be input. If the design is to be a duct-burning turbofan engine, AM23 should remain low (i.e., similar to an afterburner inlet Mach number - between 0.15 and 0.25) to keep friction losses at a minimum in the burner. Also a computational problem could arise when the duct burner is operating if AM23 is too high. The large temperature increase due to fuel addition will cause choking at engine duct station 24, consequently a printout error would occur.

The values of AM and ALTP are input to zero which are the conditions designated for the engine design point. A duct pressure loss DPDUDS is also input.

In the output of the results, the duct parameters now have values associated with the engine design, which includes area (A28 and A29) and duct Mach numbers.

```

$D(1),IDES=1,
$D(1),IDES=1,
PRFDS=2.5,PRGDS=10,WAFCD=500,WACCD=100,ALTP=0,AM=0,IGASMX=0,AM23=.15,
FAN=.TRUE.,ISPOOL=2,DPDUDS=.02,
* SEPARATE FLOW TURBOFAN ENGINE--DESIGN CASE

FAN DESIGN          PRFCF= 0.37500000E+01 ETACF= 0.94204545E+00 WAFCF= 0.83333334E+00 T20S= 0.51866820E+03
COMPRESSOR DESIGN  PRCCF= 0.12857143E+01 ETACF= 0.98837209E+00 WACCF= 0.10000000E+01 T21DS= 0.70574329E+03
COMBUSTER DESIGN   WA3CDS= 0.19974841E+02 ETABCF= 0.98500000E+00 DTCCF= 0.10000000E+01
H.P. TURBINE DESIGN CNHPCF= 0.11419282E+01 TFHPCF= 0.15210490E+01 ETHPCF= 0.10000000E+01 DHHPCF= 0.10CE9C18E+01
L.P. TURBINE DESIGN CNLPCF= 0.11782237E+01 TFLPCF= 0.14620719E+01 ETLPCF= 0.10235344E+01 DHLPCF= 0.24967145E+01
DUCT NOZZLE DESIGN A28= 0.27472096E+01 AM28= 0.10000000E+01 A29= 0.27472096E+01 AM29= 0.100C000CE+01
TURBINE AREA DESIGN A55= 0.64546229E+01 AM55= 0.23825740E+00
AFTERBURNER DESIGN WG6CDS= 0.27906742E+04
NOZZLE DESIGN      AB= 0.26471373E+01 AM8= 0.10000000E+01 A9= 0.26471373E+01 AM9= 0.100C0000E+01

OUTPUT
AM= 0. ALTP= 0. T4= 3260.00 ETAR= 1.0000
PCNF 0.1000000E+03 CVF 0.1000000E+01 ZF 0.833333E+00 PRF 0.250000E+01 WAF 0.500000E+03 WAF
PCMC 0.1000000E+03 CMC 0.1000000E+01 ZC 0.814332E+00 PRC 0.100000E+02 WACC 0.100000E+03 WAC
T2 0.518668E+03 P2 0.100000E+01 T21 0.705743E+03 P21 0.250000E+01 T3 0.143796E+04 P3 0.250000E+02
PCBLF 0. BLF 0.970000E-01 PCBLC 0.207890E+02 BL 0. PCBL0B 0.
PCBLHP 0. BLHP 0.100000E+01 PCBLLP 0.207890E+02 BL 0. T4 P4
WA3 0.193530E+03 WFB 0.614714E+01 WG4 0.199678E+03 FAR4 0.317632E-01 ETAB 0.985000E+00 DPDCM 0.236000E+02
TFHP 0.328721E+02 CNHP 0.175142E+01 DHTCHP 0.600000E-01 DHTC 0.197341E+03 T5 0.262422E+04 P5 0.782808E+01
TFLP 0.889149E+02 CNLP 0.195209E+01 DHTCLP 0.171750E-01 DHTF 0.112529E+03 T55 0.218109E+04 P55 0.368952E+01
PCBLDU 0. BLDU 0.705743E+03 T24 0.245000E+01 P24 0.705743E+03 T25 0.245000E+01
WAD 0.285681E+03 WFD 0.285681E+03 WG24 0. FAR24 0. ETAD 0.200000E-01
ETAF 0.829000E+00 ETAC 0.850000E+00 ETATHP 0.900000E+00 ETATLP 0.903000E+00 AM55 0.238257E+00 AM25 0.157446E+00
T6 0.218109E+04 P6 0.355616E+01 PS6 0.238257E+00 AM6 0.524239E+03 V6 0.270466E+03
T7 0.218109E+04 WFA 0.220466E+03 WG7 0.285822E-01 FAR7 0. ETAA 0.480000E-01
PS8 0.192429E+01 AM8 0.100000E+01 V8 0.192429E+01 PS9 0. AM9 0.206441E+04
PS28 0.129756E+01 AM28 0.100000E+01 V28 0.129756E+01 PS29 0. AM29 0.118926E+04
BYPAS, 0.133297E+01 HPEXT 0.614714E+01 WFT 0.506147E+03 WGT 0. VA 0. FRD
CVNUZ 0.963650E+00 VJM 0.199971E+04 CVDVUZ 0.115401E+04 VJD 0. FGM 0.240381E+05 FGP
MAIN SCNIC CONVERGENT NOZZLE FG= 30945.83 FN= 30945.83 SFC= 0.71511
DUCT SCNIC CONVERGENT NOZZLE
CONVERGED AFTER 1 LOOPS

```

COMMON	0.833333E+00	0.100000E+03	-0.000000E-19	-0.000000E-19	0.814332E+00	0.100000E+03	0.326000E+04	0
0.100000E+03	0.100000E+03	0.326000E+04	0.100000E+03	0.	0.100000E+01	0.100000E+01	0.100000E+01	0.100000E+01
0.833333E+00	0.100000E+03	0.250000E+01	0.829000E+00	0.500000E+03	0.375000E+01	0.942045E+00	0.833333E+00	0.833333E+00
0.814332E+00	0.100000E+03	0.100000E+02	0.850000E+00	0.214319E+03	0.128571E+01	0.988372E+00	0.100000E+01	0.100000E+01
0.326000E+04	0.	0.182204E+04	0.985000E+00	0.199748E+02	0.560000E-01	0.100000E+01	0.985000E+00	0.985000E+00
0.500000E+02	0.200000E+01	0.900000E+00	0.152105E+01	0.114195E+01	0.100000E+01	0.100890E+01	0.518688E+03	0.518688E+03
0.130000E+03	0.230000E+01	0.903000E+00	0.146207E+01	0.117822E+01	0.102353E+01	0.249671E+01	0.705743E+03	0.705743E+03
0.	0.	0.	0.	0.303574E+04	0.200000E-01	0.	0.	0.
0.356000E+04	0.	0.	0.900000E+00	0.279067E+04	0.480000E-01	0.	0.	0.
0.64542E+01	0.	0.645462E+01	0.645462E+01	0.264714E+01	0.264714E+01	0.274721E+01	0.274721E+01	0.274721E+01
0.355616E+01	0.238257E+00	0.978768E+00	0.968658E+00	0.	0.	0.	0.	0.
0.518670E+03	0.100000E+01	0.123918E+03	0.159103E+01	0.518668E+03	0.100000E+01	0.123918E+03	0.159103E+01	0.159103E+01
0.705743E+03	0.250000E+01	0.168857E+03	0.160216E+01	0.143796E+04	0.250000E+02	0.352717E+03	0.162209E+01	0.162209E+01
0.326000E+04	0.236000E+02	0.907906E+03	0.187798E+01	0.262422E+04	0.782808E+01	0.710565E+03	0.188645E+01	0.188645E+01
0.218109E+04	0.368952E+01	0.574903E+03	0.188086E+01	0.	0.	0.207890E+02	0.	0.
0.100000E+01	0.250000E+01	0.829000E+00	0.500000E+03	0.500000E+03	0.193530E+03	0.199678E+03	0.317632E-01	0.
0.100000E+01	0.100000E+02	0.850000E+00	0.100000E+03	0.214319E+03	0.985000E+00	0.560000E-01	0.	0.
0.200000E+01	0.900000E+00	0.605341E-01	0.197341E+03	0.	0.199678E+03	0.317632E-01	0.111699E+04	0.111699E+04
0.230000E+01	0.903000E+00	0.428810E-01	0.112529E+03	0.207890E+02	0.220466E+03	0.286822E-01	0.	0.
0.	0.	0.100000E+01	0.833333E+00	0.100000E+03	0.814332E+00	0.100000E+03	0.614714E+01	0.614714E+01
0.500000E+02	0.130000E+03	0.	0.970000E-01	0.	0.	0.	0.100000E+01	0.100000E+01
0.100000E+01	0.500000E+03	0.214319E+03	0.	0.	0.352717E+03	0.	0.	0.
0.705743E+03	0.250000E+01	0.168857E+03	0.160216E+01	0.705743E+03	0.250000E+01	0.168857E+03	0.160216E+01	0.160216E+01
0.705743E+03	0.245000E+01	0.168857E+03	0.160355E+01	0.705743E+03	0.245000E+01	0.168857E+03	0.160355E+01	0.160355E+01
0.705743E+03	0.245000E+01	0.168857E+03	0.160355E+01	0.705743E+03	0.245000E+01	0.168857E+03	0.160355E+01	0.160355E+01
0.28561E+03	0.	0.285681E+03	0.	0.	0.200000E-01	0.133297E+01	0.	0.
0.58848E+03	0.129756E+01	0.118926E+04	0.100000E+01	0.58848E+03	0.129756E+01	0.118926E+04	0.100000E+01	0.100000E+01
0.218109E+04	0.368952E+01	0.574903E+03	0.188086E+01	0.705743E+03	0.245000E+01	0.168857E+03	0.160355E+01	0.160355E+01
0.614714E+01	0.220466E+03	0.286822E-01	0.	0.285681E+03	0.	0.100000E+01	0.	0.
0.218109E+04	0.368952E+01	0.574903E+03	0.188086E+01	0.218109E+04	0.351242E+01	0.574905E+03	0.188423E+01	0.188423E+01
0.218109E+04	0.351242E+01	0.574905E+03	0.188423E+01	0.218109E+04	0.351242E+01	0.574905E+03	0.188423E+01	0.188423E+01
0.220466E+03	0.	0.220466E+03	0.286822E-01	0.	0.480000E-01	0.524239E+03	0.	0.
0.355616E+01	0.524239E+03	0.238257E+00	0.216058E+04	0.337259E+01	0.552311E+03	0.251121E+00	0.157446E+00	0.157446E+00
0.188980E+04	0.192429E+01	0.206441E+04	0.100000E+01	0.188980E+04	0.192429E+01	0.206441E+04	0.100000E+01	0.100000E+01
0.	0.	0.116401E+04	0.103355E+05	0.199971E+04	0.137026E+05	0.172992E+04	0.517779E+04	0.517779E+04
0.240391E+05	0.690771E+04	0.614714E+01	0.506147E+03	0.122943E-01	0.309458E+05	0.309458E+05	0.715112E+00	0.715112E+00

Off-design. - The following are the off-design cases for the separate-flow turbofan engine.

Case 1: The engine is operating at Mach 2.5 at an altitude of 50 000 feet. The engine components are matched, with the duct and primary nozzle throat areas (A28 and A8) equal to the sea-level-static design value. This is a setup case for matching components only. However, note the very low thrust of the turbofan engine at this flight condition because the ram drag FRD is nearly equal to the gross thrust of the two streams FG.

```

SD(1),AM=2.5,ALTP=50000,      #1
* OFF DESIGN OPERATION AT MAX T4 (SETUP CASE)

```

OUTPUT	AM= 2.500	ALTP= 50000.	T4= 3260.00	ETAR= C.8703
PCNF	CNF	ZF	PRF	WAF
0.709714E+02	0.546211E+00	0.652696E+00	0.141563E+01	0.333068E+03
PCNC	CNC	ZC	PRC	WAC
0.792542E+02	0.670088E+00	0.657516E+00	0.479978E+01	0.100423E+03
T2	P2	T21	P21	T3
0.875661E+03	0.170275E+01	0.987251E+03	0.241047E+01	0.115697E+02
PCBLF	BLF	PCBLC	BLC	PCBLOB
0.	0.	0.970000E-01	0.974107E+01	0.
PCBLHP	BLHP	PCBLLP	BLLP	T4
0.	0.	0.100000E+01	0.974107E+01	0.326000E+04
WA3	WFB	WG4	FAR4	ETAB
0.906823E+02	0.248834E+01	0.931707E+02	0.274401E-01	0.985000E+00
TFFHP	CNHP	DHTCHP	DHTC	T5
0.333433E+02	0.138808E+01	0.607172E-01	0.199384E+03	0.261240E+04
TFFLP	CNLP	DHTCLP	DHTF	T55
0.975392E+02	0.138856E+01	0.150943E-01	0.984124E+02	0.223271E+04
PCBLDU	BLDU	T24	P24	T25
0.	0.	0.987251E+03	0.236231E+01	0.987251E+03
WAD	WFD	WG24	FAR24	ETAD
0.232645E+03	0.	0.232645E+03	0.	0.
ETAF	ETAC	ETATHP	ETATLP	AM55
0.796334E+00	0.715568E+00	0.855728E+00	0.912016E+00	0.238583E+00
T6	P6	P56	AM6	V6
0.223271E+04	0.173915E+01	0.167855E+01	0.238583E+00	0.531377E+03
T7	WFA	WG7	FAR7	ETAA
0.223271E+04	0.	0.102912E+03	0.247785E-01	0.
PS8	AM8	V8	PS9	AM9

0.903306E+00	0.100000E+01	0.208880E+04	0.908306E+00	0.100000E+01	0.208880E+04
PS28	AM28	V28	PS29	AM29	V29
0.125681E+01	0.100000E+01	0.140492E+04	0.125681E+01	0.100000E+01	0.140492E+04
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.231664E+01	0.	0.248834E+01	0.335557E+03	0.242123E+04	0.250648E+05
CVMNOZ	VJM	CVDNOZ	VJD	FGM	FGP
0.878349E+00	0.183469E+04	0.852815E+00	0.119813E+04	0.145319E+05	0.110884E+05

MAIN SCNIC CONVERGENT NOZZLE      FG= 25620.31      FN= 555.53      SFC=16.12516  
 DUCT SCNIC CONVERGENT NOZZLE  
 CONVERGED AFTER 34 LOOPS

Case 2: The same flight condition and engine operation is shown in this case as in case 1 because it is a continuation of the previous setup case. However, the engine is now duct burning. Therefore T24, IDBURN, and ETAD must be input. The duct burner does not have a performance map included in the subprogram as was the situation with the afterburner. Consequently, every duct-burning case requires the input of the burner efficiency (ETAD) and the control card (IDBURN = 1) in order to operate.

A new throat area (A28) is calculated reflecting the change in temperature, pressure, and mass flow of the gas in order that the nozzle is sonic. The value of A8 remains at the design value unless it is altered with a separate input. A permanent change in A8 would require a new setup case prior to running a duct-burning case.

\$D(11,T24=3460,ETAD=.90,IDBURN=1,  
 \* JFF DESIGN OPERATION WITH DUCTBURNING

DUCT NOZZLE DESIGN      A28= 0.58485843E+01      AM28= 0.10000000E+01      A29= 0.18051276E+02      AM29= 0.25474720E+01  
 ARATIO= 3.086      OUT OF RANGE, USE DATA FOR 2.150

INPUT      AM= 2.500      ALTP= 50000.      T4= 3250.00      T24= 3460.00      ETAR= 0.8703  
          PCMF      CNF      ZF      PRF      WAF      WAF  
 0.709714E+02      0.546211E+00      0.652696E+00      0.141563E+01      0.256159E+03      0.333068E+03  
          PCNC      CNC      ZC      PRC      WACC      WAC  
 0.792542E+02      0.670088E+00      0.657516E+00      0.479978E+01      0.574781E+02      0.100423E+03  
          T2      P2      T21      P21      T3      P3  
 0.875661E+03      0.170275E+01      0.987251E+03      0.241047E+01      0.170093E+04      0.115697E+02  
          PCBLF      RLF      PCBLC      BLC      PCRLOR      BLOB  
 0.      0.      0.970000E-01      0.974107E+01      0.      0.  
          PCBLHP      BLHP      PCBLLP      BLLP      T4      P4  
 0.      0.      0.100000E+01      0.974107E+01      0.326000E+04      0.108562E+07  
          WA3      WFR      WG4      FAR4      ETAB      OPCOM  
 0.906823E+02      0.248834E+01      0.931707E+02      0.274401E-01      0.985000E+00      0.616665E-01  
          TFFHP      CNHP      DHTCHP      DHTC      T5      P5  
 0.333433E+02      0.138808E+01      0.607172E-01      0.199384E+03      0.261240E+04      0.332216E+01  
          TEFLP      CNLP      DHTCLP      DHTF      T55      P55  
 0.975392E+02      0.138856E+01      0.150943E-01      0.984124E+02      0.223271E+04      0.173915E+01  
          PCRLDU      BLDU      T24      P24      T25      P25  
 0.      0.      0.346000E+04      0.224230E+01      0.346000E+04      0.224230E+01  
          WAD      WFD      WG24      FAR24      ETAD      DPDUK  
 0.232645E+03      0.109448E+02      0.243590E+03      0.470450E-01      0.900000E+00      0.199789E-01  
          ETAF      ETAC      ETATHP      ETATLP      AM55      AM25  
 0.796334E+00      0.715568E+00      0.855728E+00      0.912016E+00      0.238583E+00      0.268149E+07  
          T6      P6      PS6      AM6      V6      WG6  
 0.223271E+04      0.173915E+01      0.167855E+01      0.238583E+00      0.531377E+03      0.102912E+03  
          T7      WFA      WG7      FAR7      ETAA      DPAFT  
 0.223271E+04      0.      0.102912E+03      0.247785E-01      0.      0.480924E-01  
          PS8      AM8      VR      PS9      AM9      V9  
 0.908306E+00      0.100000E+01      0.208880E+04      0.908306E+00      0.100000E+01      0.208880E+04  
          PS28      AM28      V28      PS29      AM29      V29  
 0.124170E+01      0.100000E+01      0.258242E+04      0.114456E+00      0.254747E+01      0.510859E+04  
          BYPASS      HPEXT      WFT      WGT      VA      FRD  
 0.231664E+01      0.      0.134331E+02      0.346501E+03      0.242123E+04      0.250648E+05  
          CVMNOZ      VJM      CVDNOZ      VJD      FGM      FGP  
 0.878349E+00      0.183469E+04      0.965722E+00      0.493348E+04      0.432198E+05      0.444708E+04

MAIN SCNIC CONVERGENT NOZZLE      FG= 47666.88      FN= 22602.09      SFC= 2.13959  
 DUCT SHOCK OUTSIDE C-D NOZZLE  
 CONVERGED AFTER 1 LOOPS



## Mixed-Flow Turbofan

Design point. - A mixed-flow turbofan engine is designed by using the input IGASMX = 1 and resetting the AM and ALTP to the desired value (sea-level-static in the example). This output listing shows the rebalancing of the fan and inner compressor pressure ratio to a value such that the static pressures at stations 25 and 55 will match. The overall compressor pressure ratio remains fixed. The values of the combination are not necessarily the optimum values; but it does give an indication, for the selected design values, as to what combination can be considered. Once the static-pressure values are satisfied, the program operates in a normal design mode. The other input cards are those required for the afterburner design. These values of design efficiency and temperature must be input for the design case in order to scale the generalized afterburner map to the proper design conditions. The afterburning takes place in the completely mixed streams - similar to an afterburning turbojet. The mixed-flow turbofan engine uses a single exhaust nozzle. The design areas are then printed out for this type of operation.

```

SD(1),IDES=1, #1
SD(1),IDES=1,
AM=0,ALTP=0,IGASMX=1, # MIXED FLOW TURBOFAN - DESIGN CASE
T7DS=3460,ETAADS=.89,AM6=.15,DPAFDS=.05, # AFTERBURNER DESIGN VALUES
FAN DESIGN          PRFCF= 0.37500000E+01  ETAFCF= 0.94204545E+00  WAFCF= 0.83333334E+00  T2DS= 0.51866820E+03
COMPRESSOR DESIGN  PRCCF= 0.12857143E+01  ETACCF= 0.98837209E+00  WACCF= 0.10000000E+01  T21DS= 0.70574329E+03
COMBUSTER DESIGN   WA3CDS= 0.19974841E+02  ETABCF= 0.98500000E+00  DTCCF= 0.10000000E+01
H.P. TURBINE DESIGN CNHPCF= 0.11419282E+01  TFHPCF= 0.15210490E+01  ETHPCF= 0.10000000E+01  DHHPCF= 0.10089018E+01
L.P. TURBINE DESIGN CVLPCF= 0.11782237E+01  TFLPCF= 0.14620719E+01  ETLPCF= 0.10235344E+01  DHLPCF= 0.24967145E+C1
CHANGE PRFDS FROM  2.500 TO  3.701 AND PRCD5 FROM  10.000 TO  6.754
FAN DESIGN          PRFCF= 0.67532611E+01  ETAFCF= 0.94204545E+00  WAFCF= 0.83333334E+00  T2DS= 0.51866820E+03
COMPRESSOR DESIGN  PRCCF= 0.82205365E+00  ETACCF= 0.98837209E+00  WACCF= 0.10000000E+01  T21DS= 0.80123660E+03
COMBUSTER DESIGN   WA3CDS= 0.27828602E+02  ETABCF= 0.98500000E+00  DTCCF= 0.10000000E+01
H.P. TURBINE DESIGN CNHPCF= 0.11419282E+01  TFHPCF= 0.10948068E+01  ETHPCF= 0.10000000E+01  DHHPCF= 0.89296639E+00
L.P. TURBINE DESIGN CVLPCF= 0.11947035E+01  TFLPCF= 0.11955020E+01  ETLPCF= 0.10235344E+01  DHLPCF= 0.26482349E+C1
CHANGE PRFDS FROM  2.500 TO  4.057 AND PRCD5 FROM  10.000 TO  6.163
FAN DESIGN          PRFCF= 0.76415902E+01  ETAFCF= 0.94204545E+00  WAFCF= 0.83333334E+00  T2DS= 0.51866820E+03
COMPRESSOR DESIGN  PRCCF= 0.73753452E+00  ETACCF= 0.98837209E+00  WACCF= 0.10000000E+01  T21DS= 0.82498885E+03
COMBUSTER DESIGN   WA3CDS= 0.30070055E+02  ETABCF= 0.98500000E+00  DTCCF= 0.10000000E+01
H.P. TURBINE DESIGN CNHPCF= 0.11419282E+01  TFHPCF= 0.10136263E+01  ETHPCF= 0.10000000E+01  DHHPCF= 0.86289610E+00
L.P. TURBINE DESIGN CVLPCF= 0.11389344E+01  TFLPCF= 0.11433495E+01  ETLPCF= 0.10235344E+01  DHLPCF= 0.26415626E+01
CHANGE PRFDS FROM  2.500 TO  4.215 AND PRCD5 FROM  10.000 TO  5.931
FAN DESIGN          PRFCF= 0.80373937E+01  ETAFCF= 0.94204545E+00  WAFCF= 0.83333334E+00  T2DS= 0.51866820E+03
COMPRESSOR DESIGN  PRCCF= 0.70446542E+00  ETACCF= 0.98837209E+00  WACCF= 0.10000000E+01  T21DS= 0.83567622E+03
COMBUSTER DESIGN   WA3CDS= 0.31058852E+02  ETABCF= 0.98500000E+00  DTCCF= 0.10000000E+01
H.P. TURBINE DESIGN CNHPCF= 0.11419282E+01  TFHPCF= 0.98150524E+00  ETHPCF= 0.10000000E+01  DHHPCF= 0.84999365E+C0
L.P. TURBINE DESIGN CVLPCF= 0.12007443E+01  TFLPCF= 0.11225456E+01  ETLPCF= 0.10235344E+01  DHLPCF= 0.26381416E+01
TURBINE/DUCT AREA DESIGN  A55= 0.87603192E+01  AM55= 0.23825511E+00  A25= 0.10494306E+02  AM25= 0.57861657E+01
AFTERBURNER DESIGN  WG6CDS= 0.51521512E+04
NOZZLE DESIGN      Ad= 0.48660238E+01  AM8= 0.10000000E+01  A9= 0.48660238E+01  AM9= 0.10000000E+01

```

OUTPUT	AM= 0.	ALTP= 0.	0.	T4= 3260.00			ETAR= 1.0000
	PCAF	CNF	ZF	PRF	WAFIC	WAF	
	0.100000E+03	0.100000E+01	0.833333E+00	0.421496E+01	0.500000E+03	0.500000E+03	
	PCNL	CNL	ZC	PRC	WACC	KAC	
	0.100000E+03	0.100000E+01	0.814332E+00	0.593126E+01	0.100000E+03	0.332181E+03	
	T2	P2	T21	P21	T3	P3	
	0.514667E+03	0.100000E+01	0.835076E+03	0.421496E+01	0.144718E+04	0.250000E+03	
	PCALF	BLF	PCALC	BLC	PCBLOR	BLOR	
	0.	0.	0.970000E-01	0.322216E+02	0.	0.	
	PCRLHP	BLHP	PCALLP	RLLP	T4	P4	
	0.	0.	0.100000E+01	0.322216E+02	0.326000E+04	0.236000E+02	
	WA3	WFB	WG4	FAR4	ETAB	OPCOM	
	0.299959E+03	0.948284E+01	0.309442E+03	0.316137E-01	0.985000E+00	0.560000E-01	
	TFFHP	CNHP	DHTCHP	DHTC	T5	P5	
	0.509427E+02	0.175142E+01	0.600000E-01	0.165259E+03	0.272550E+04	0.949214E+01	
	TFFLP	CNLP	DHTCLP	DHTF	T55	P55	
	0.115803E+03	0.191548E+01	0.171750E-01	0.123352E+03	0.224356E+04	0.427559E+01	
	PCALDI	BLDU	T24	P24	T25	P25	
	0.	0.	0.835076E+03	0.413066E+01	0.835076E+03	0.413066E+01	
	WAD	WFD	WG24	FAR24	ETAD	DPDUC	
	0.167819E+03	0.	0.167819E+03	0.	0.	0.200000E-01	
	ETAF	ETAC	ETATHP	ETATLP	AM55	AM25	
	0.829000E+00	0.850000E+00	0.900000E+00	0.903000E+00	0.238255E+00	0.578617E-01	
	T6	P6	PS6	V6	AM6	WG6	
	0.181619E+04	0.421426E+01	0.415724E+01	0.143360E+00	0.290659E+03	0.509483E+03	
	T7	WFA	WG7	FAR7	ETAA	OPAFT	
	0.181619E+04	0.	0.509483E+03	0.189657E-01	0.	0.500000E-01	
	PS8	AM8	V8	PS9	AM9	V9	
	0.217954E+01	0.100000E+01	0.189038E+04	0.217954E+01	0.100000E+01	0.189038E+04	
	PS28	AM28	V28	PS29	AM29	V29	
	0.	0.	0.	0.	0.	0.	
	BYPASS	HPEXT	WFT	WGT	VA	FRD	
	0.505203E+00	0.	0.948284E+01	0.509483E+03	0.	0.	
	CVMNDZ	VJM	CVDNDZ	VJD	FGM	FGP	
	0.963154E+00	0.182074E+04	0.	0.	0.288318E+05	0.121464E+05	

MAIN SCNIC CONVERGENT NOZZLE FG= 40978.13 FN= 40978.13 SFC= 0.83308

CONVERGED AFTER 1 LOOPS

COMMON	0.833333E+00	0.100000E+03	-0.000000E-19	-0.000000E-19	0.814332E+00	0.100000E+03	0.326000E+04	0
0.100000E+03	0.100000E+03	0.326000E+04	0.100000E+03	0.	0.100000E+01	0.100000E+01	0.100000E+01	
0.833333E+00	0.100000E+03	0.421496E+01	0.829000E+00	0.500000E+03	0.803739E+01	0.942045E+00	0.833333E+00	
0.814332E+00	0.100000E+03	0.593126E+01	0.850000E+00	0.332181E+03	0.704465E+00	0.988372E+00	0.100000E+01	
0.326000E+04	0.	0.181282E+04	0.985000E+00	0.310899E+02	0.550000E-01	0.100000E+01	0.985000E+00	
0.500000E+02	0.200000E+01	0.900000E+00	0.981505E+00	0.114193E+01	0.100000E+01	0.849994E+00	0.518668E+03	
0.130000E+03	0.230000E+01	0.903000E+00	0.112255E+01	0.120074E+01	0.102353E+01	0.263514E+01	0.835076E+03	
0.	0.	0.	0.	0.115056E+04	0.200000E-01	0.	0.	
0.346000E+04	0.	0.	0.890000E+00	0.515215E+04	0.500000E-01	0.	0.	
0.876032E+01	0.104943E+02	0.192546E+02	0.192546E+02	0.486602E+01	0.486602E+01	0.	0.	
0.412124E+01	0.238255E+00	0.	0.963159E+00	0.	0.	0.	0.	
0.518670E+03	0.100000E+01	0.123918E+03	0.159103E+01	0.518668E+03	0.100000E+01	0.123918E+03	0.159103E+01	
0.835076E+03	0.421496E+01	0.200259E+03	0.160719E+01	0.144718E+04	0.250000E+02	0.355137E+03	0.162376E+01	
0.326000E+04	0.236000E+02	0.907719E+03	0.187788E+01	0.272550E+04	0.949214E+01	0.741460E+03	0.188473E+01	
0.224356E+04	0.427559E+01	0.593308E+03	0.187903E+01	0.	0.322216E+02	0.	0.	
0.100000E+01	0.421496E+01	0.829000E+00	0.500000E+03	0.500000E+03	0.299959E+03	0.309442E+03	0.316137E-01	
0.100000E+01	0.593126E+01	0.850000E+00	0.100000E+03	0.332181E+03	0.985000E+00	0.560000E-01	0.	
0.200000E+01	0.900000E+00	0.509996E-01	0.166259E+03	0.	0.309442E+03	0.316137E-01	0.111659E+04	
0.230000E+01	0.903000E+00	0.452585E-01	0.123352E+03	0.322216E+02	0.341664E+03	0.285472E-01	0.	
0.	0.	0.100000E+01	0.833333E+00	0.100000E+03	0.814332E+00	0.100000E+03	0.948284E+01	
0.500000E+02	0.130000E+03	0.	0.970000E-01	0.	0.	0.	0.100000E+01	
0.100000E+01	0.500000E+03	0.332181E+03	0.	0.	0.355137E+03	0.	0.	
0.835076E+03	0.421496E+01	0.200259E+03	0.160719E+01	0.835076E+03	0.421496E+01	0.200259E+03	0.160719E+01	
0.835076E+03	0.413066E+01	0.200259E+03	0.160858E+01	0.835076E+03	0.413066E+01	0.200259E+03	0.160858E+01	
0.	0.	0.	0.	0.	0.	0.	0.	
0.167819E+03	0.	0.167819E+03	0.	0.	0.200000E-01	0.505203E+00	0.	
0.	0.	0.	0.	0.	0.	0.	0.	
0.224356E+04	0.427559E+01	0.593308E+03	0.187903E+01	0.835076E+03	0.413066E+01	0.200259E+03	0.160858E+01	
0.948284E+01	0.341664E+03	0.285472E-01	0.	0.167819E+03	0.	0.100000E+01	0.	
0.181619E+04	0.421426E+01	0.463844E+03	0.181456E+01	0.181619E+04	0.400355E+01	0.463844E+03	0.181808E+01	
0.181619E+04	0.400355E+01	0.463844E+03	0.181808E+01	0.181619E+04	0.400355E+01	0.463844E+03	0.181808E+01	
0.509483E+03	0.	0.509483E+03	0.189657E-01	0.	0.500000E-01	0.531343E+03	0.816543E+02	
0.415724E+01	0.290659E+03	0.143360E+00	0.181019E+04	0.394954E+01	0.306344E+03	0.151095E+00	0.578617E-01	
0.155960E+04	0.217954E+01	0.189038E+04	0.100000E+01	0.155960E+04	0.217954E+01	0.189038E+04	0.100000E+01	
0.	0.	0.	0.	0.182074E+04	0.288318E+05	0.	0.121464E+05	
0.288318E+05	0.121464E+05	0.948284E+01	0.509483E+03	0.189657E-01	0.409781E+05	0.409781E+05	0.833084E+00	

**Off-design.** - The following are off-design cases for the mixed-flow turbofan.

Case 1: The off-design case was selected to be supersonic Mach number (2.5) at 50 000 feet altitude. Case 1 illustrates a setup case for the afterburning case (case 2) which follows immediately after. Since the engine is operating at the design T4, the only input cards required are the AM and ALTP cards.

```

SD(1),AM=2.5,ALTP=50000, #1
* OFF DESIGN OPERATION AT MAX T4 (SETUP CASE)

OUTPUT          AM= 2.500      ALTP= 50000.      T4= 3260.00      ETAR= 0.8703
PCNF           CNF           ZF           PRF           WAF           WAF
0.563944E+02  0.434023E+00  0.435777E+00  0.140745E+01  0.222610E+03  0.291724E+03
PCNC           CNC           ZC           PRC           WACC          WAC
0.856482E+02  0.786044E+00  0.696801E+00  0.370696E+01  0.583886E+02  0.118544E+03
T2            P2            T21          P21           T3            P3
0.875661E+03  0.170275E+01  0.991445E+03  0.239654E+01  0.153061E+04  0.888387E+01
PCBLF         BLF           PCRLC        RLC           PCBL0B        BL0B
0.           0.           0.970000E-01 0.114987E+02  0.           0.
PCBLHP       BLHP         PCRLLP       RLLP          T4            P4
0.           0.           0.100000E+01 0.114987E+02  0.326000E+04  0.837006E+01
WA3          WFA          WG4          FAR4          ETAB          DPCOM
0.107045E+03  0.323859E+01  0.110283E+03  0.302545E-01  0.985000E+00  0.578362E-01
TFFHP       CNHP         DHTCHP      DHTC          T5            P5
0.511908E+02  0.150006E+01  0.538222E-01  0.148746E+03  0.278122E+04  0.369848E+01
TFFLP       CNLP         DHTCLP      DHTF          T55           P55
0.107005E+03  0.106935E+01  0.103248E-01  0.755756E+02  0.244645E+04  0.231426E+01
PCBLDU      BLDU        T24          P24           T25           P25
0.           0.           0.991445E+03  0.230175E+01  0.991445E+03  0.230175E+01
WAD          WFD          WG24        FAR24         ETAD          DPDUC
0.173180E+03  0.173180E+03  0.173180E+03  0.           0.           0.395519E-01
ETAF        ETAC         ETATHP      ETATLP        AM55         AM25
0.753899E+00  0.773388E+00  0.889117E+00  0.889316E+00  0.161228E+00  0.117851E+00
T6          P6          P56         AM6           V6           W6
0.163458E+04  0.230603E+01  0.227469E+01  0.143137E+00  0.276685E+03  0.294962E+03
T7          WFA          WG7          FAR7          ETAA         DPAPT
0.163458E+04  0.           0.294962E+03  0.11016E-01  0.           0.501864E-01
PS8         AM8          V8           PS3           AM9          V9
0.118688E+01  0.100000E+01  0.179734E+04  0.118688E+01  0.100000E+01  0.179734E+04
PS28        AM28        V28         PS29         AM29        V29
0.           0.           0.           0.           0.           0.
BYPASS      HPEXT       WFT          WGT          VA           FRD
0.146090E+01  0.           0.323859E+01  0.294962E+03  0.242123E+04  0.219534E+05
CVMNOZ      VJM         CVDNOZ      VJD          FGM          FGP
0.858472E+00  0.154296E+04  0.           0.           0.141454E+05  0.110434E+05

MAIN SCNIC CONVERGENT NOZZLE      FG= 25188.82      FN= 3235.39      SFC= 3.6C356
CONVERGED AFTER 21 LOOPS

```

Case 2: The previous setup case is used here for an afterburning example. Values of afterburner efficiency ETAA, pressure loss DPAPT, and Mach number AM6 indicate values selected from the afterburner performance map at off-design. The expanded nozzle areas are shown; a printout indicating that a nozzle having a greater expansion ratio than the convergent-divergent nozzle map used would have been more optimum (instead of 2.150). Consequently, a slight loss in velocity coefficient due to an under-expanded nozzle occurred.

```

SD(1), T7=3460, IAFBN=1, # AFTERBURNING CASE
NOZZLE DESIGN      AB= 0.76637719E+01  AM8= 0.10000000E+01  A9= 0.23089376E+02  AM9= 0.25228365E+01
ARATIO= 3.013      OUT OF RANGE, USE DATA FOR 2.150

OUTPUT          AM= 2.500      ALTP= 50000.      T4= 3260.00      T7= 3460.00      ETAR= 0.8703
PCNF           CNF           ZF           PRF           WAF           WAF
0.563944E+02  0.434023E+00  0.435777E+00  0.140745E+01  0.222610E+03  0.291724E+03
PCNC           CNC           ZC           PRC           WACC          WAC
0.856482E+02  0.786044E+00  0.696801E+00  0.370696E+01  0.583886E+02  0.118544E+03
T2            P2            T21          P21           T3            P3
0.875661E+03  0.170275E+01  0.991445E+03  0.239654E+01  0.153061E+04  0.888387E+01
PCBLF         BLF           PCRLC        RLC           PCBL0B        BL0B
0.           0.           0.970000E-01 0.114987E+02  0.           0.
PCBLHP       BLHP         PCRLLP       RLLP          T4            P4
0.           0.           0.100000E+01 0.114987E+02  0.326000E+04  0.837006E+01
WA3          WFA          WG4          FAR4          ETAB          DPCOM
0.107045E+03  0.323859E+01  0.110283E+03  0.302545E-01  0.985000E+00  0.578362E-01
TFFHP       CNHP         DHTCHP      DHTC          T5            P5
0.511908E+02  0.150006E+01  0.538222E-01  0.148746E+03  0.278122E+04  0.369848E+01
TFFLP       CNLP         DHTCLP      DHTF          T55           P55
0.107005E+03  0.106935E+01  0.103248E-01  0.755756E+02  0.244645E+04  0.231426E+01
PCBLDU      BLDU        T24          P24           T25           P25
0.           0.           0.991445E+03  0.230175E+01  0.991445E+03  0.230175E+01
WAD          WFD          WG24        FAR24         ETAD          DPDUC
0.173180E+03  0.173180E+03  0.173180E+03  0.           0.           0.395519E-01
ETAF        ETAC         ETATHP      ETATLP        AM55         AM25
0.753899E+00  0.773388E+00  0.889117E+00  0.889316E+00  0.161228E+00  0.117851E+00

```

T6	P6	PS6	AM6	V6	WG6
0.163458E+04	0.230603E+01	0.227468E+01	0.143137E+00	0.276685E+03	0.294962E+C3
T7	WFA	WG7	FAR7	ETAA	UPAFT
0.346000E+04	0.109850E+02	0.305947E+03	0.487570E-01	0.774686E+00	0.501864E-01
PSR	AM8	V8	PS3	AM9	V9
0.119096E+01	0.100000E+01	0.258203E+04	0.114456E+00	0.252284E+01	0.508533E+04
PS28	AM28	V28	PS29	AM29	V29
0.	0.	0.	0.	0.	0.
BYPASS	HPEXT	WFT	WGT	VA	FRD
0.146090E+01	0.	0.142236E+02	0.305947E+03	0.242123E+04	0.219534E+C5
CVMNQZ	VJM	CVDNQZ	VJD	FGM	FGP
0.966656E+00	0.491576E+04	0.	0.	0.467446E+05	0.

MAIN SHOCK OUTSIDE C-D NOZZLE      FG= 46744.62      FN= 24791.19      SFC= 2.06544

CONVERGED AFTER 1 LOOPS

Design case without pressure ratio rematch. - The mixed-flow turbofan engine is shown again. However, the design fan and compressor pressure ratios were picked such that no rematch was required to balance the static pressures. This is another design case. As many design cases as desired can be run in succession. Care must always be taken to include the variables in input that need to be changed from the previous design case.

```

SD(1),IDES=1,
SD(1),IDES=1,
PRFDS=5.0,PRCDS=5.0,AM=0,ALTP=0,+ MIXED FLOW TURBOFAN WITHOUT REQUIRED P REMATCH
FAN DESIGN            PRFCF= 0.1000000E+02    ETACF= 0.9420454E+00    WACF= 0.8333333E+00    T2DS= 0.5186820E+03
COMPRESSOR DESIGN    PRCCF= 0.57142857E+00    ETACCF= 0.98837209F+00    WACCF= 0.1000000E+01    T2IDS= 0.88129464E+C3
COMBUSTER DESIGN    WA3CDS= 0.35981686E+02    ETABCF= 0.9850000E+00    DTCCCF= 0.1000000E+01
H.P. TURBINE DESIGN    CNHPCF= 0.11419282E+01    TFHPCF= 0.85001019E+00    ET+PCF= 0.1000000E+01    DHPCF= 0.78992675E+00
L.P. TURBINE DESIGN    CNLPCF= 0.12091271E+01    TFLPCF= 0.10362098E+01    ETLPCF= 0.10235344F+01    DHLPCF= 0.2584124E+01
TURBINE/DUCT AREA DESIGN    A55= 0.93469959E+01    AM55= 0.23825344E+00    A25= 0.10994153E+01    AM25= 0.35644379E+00
AFTERRURNER DESIGN    W6CDS= 0.48604900E+04
NOZZLE DESIGN            AR= 0.46035268E+01    AM8= 0.1000000E+01    A9= 0.46035268E+01    AM9= 0.1000000E+01
OUTPUT
AM= 0.            ALTP= 0.            T4= 3260.00            ETAR= 1.0000
PCNF            CNF            ZF            WAF            WAF
0.100000E+03    0.100000E+01    0.833333E+00    0.500000E+01    0.500000E+03    0.500000E+03
PCNC            CNC            ZC            WAC            WAC
0.100000E+03    0.100000E+01    0.814332E+00    0.500000E+01    0.100000E+03    0.383577E+C3
T2            P2            T21            P21            T3            P3
0.518668E+03    0.100000E+01    0.881300E+03    0.500000E+01    0.144858E+04    0.250000E+02
PCHLF            BLF            PCRLC            BLC            PCRL0B            RLOB
0.            0.            0.970000E-01    0.372070E+02    0.            0.
PCRLHP            RLHP            PCRLLP            BLLP            T4            P4
0.            0.            0.100000E+01    0.372070E+02    0.326000E+04    0.236000E+02
WA3            WFB            WG4            FAR4            ETAB            OPCOM
0.346370E+03    0.109422E+02    0.357312E+03    0.315911E-01    0.985000E+00    0.560000E-C1
TFFHP            CNHP            DHTCHP            DHTC            T5            P5
0.588227E+02    0.175142E+01    0.600000E-01    0.154510E+03    0.276368E+04    0.101882E+02
TFFLP            CNLP            DHTCLP            DHTF            T55            P55
0.125457E+03    0.190220E+01    0.171750E-01    0.122659E+03    0.228157E+04    0.466804E+C1
PCPLDU            BLDU            T24            P24            T25            P25
0.            0.            0.881300E+03    0.490000E+01    0.881300E+03    0.490000E+01
WAD            WFD            WG24            FAR24            ETAD            OPDUC
0.115423E+03    0.            0.116423E+03    0.            0.            0.200000E-01
ETAF            ETAC            ETATHP            ETATLP            AM55            AM25
0.929000E+00    0.850000E+00    0.900000E+00    0.903000E+00    0.238253E+00    0.356444E+00
T6            P6            PS6            AM6            V6            W66
0.199038E+04    0.468986E+01    0.449153E+01    0.256683E+00    0.541234E+03    0.510942E+03
T7            WFA            WG7            FAR7            ETAA            OPAFT
0.199038E+04    0.            0.510942E+03    0.218844E-01    0.            0.500000E-01
PS8            AM8            V8            PS9            AM9            V9
0.24324E+01    0.100000E+01    0.197586E+04    0.24324E+01    0.100000E+01    0.197586E+04
PS28            AM28            V28            PS29            AM29            V29
0.            0.            0.            0.            0.            0.
BYPASS            HPEXT            WFT            WGT            VA            FRD
0.30351E+00    0.            0.109422E+02    0.510942E+03    0.            0.
CVMNQZ            VJM            CVDNQZ            VJD            FGM            FGP
0.957521E+00    0.189192E+04    0.            0.            0.300448E+05    0.139553E+05
MAIN SCNIC CONVERGENT NOZZLE      FG= 44000.11      FN= 44000.11      SFC= 0.85527
CONVERGED AFTER 1 LOOPS

```

COMMON	0.833333E+00	0.100000E+03	-0.000000E-19	-0.000000E-19	0.814332E+00	0.100000E+03	0.326000E+04	0
0.100000E+03	0.100000E+03	0.326000E+04	0.100000E+03	0.	0.100000E+01	0.100000E+01	0.100000E+01	
0.933333E+00	0.100000E+03	0.500000E+01	0.829000E+00	0.500000E+03	0.100000E+02	0.942045E+00	0.833333E+00	
0.814332E+00	0.100000E+03	0.500000E+01	0.850000E+00	0.383577E+03	0.571429E+00	0.988372E+00	0.100000E+01	
0.326000E+04	0.	0.181142E+04	0.985000E+00	0.358817E+02	0.560000E-01	0.100000E+01	0.985000E+00	
0.500000E+02	0.200000E+01	0.900000E+00	0.850010E+00	0.114193E+01	0.100000E+01	0.789927E+00	0.518668E+03	
0.130000E+03	0.230000E+01	0.903000E+00	0.103621E+01	0.120913E+01	0.102353E+01	0.258413E+01	0.881300E+03	
0.	0.	0.	0.	0.691241E+03	0.200000E-01	0.	0.	
0.346000E+04	0.	0.	0.890000E+00	0.486049E+04	0.500000E-01	0.	0.	
0.934700E+01	0.109942E+01	0.104464E+02	0.104464E+02	0.460353E+01	0.460353E+01	0.	0.	
0.449964E+01	0.238253E+00	0.	0.957521E+00	0.	0.	0.	0.	
0.518670E+03	0.100000E+01	0.123918E+03	0.159103E+01	0.518668E+03	0.100000E+01	0.123918E+03	0.159103E+01	
0.941300E+03	0.500000E+01	0.211573E+03	0.160867E+01	0.144898E+04	0.250000E+02	0.355503E+03	0.162402E+01	
0.326000E+04	0.236000E+02	0.907691E+03	0.187786E+01	0.276368E+04	0.101882E+02	0.753181E+03	0.188413E+01	
0.228157E+04	0.466804E+01	0.604585E+03	0.187798E+01	0.	0.372070E+02	0.	0.	
0.100000E+01	0.500000E+01	0.829000E+00	0.500000E+03	0.500000E+03	0.346370E+03	0.357312E+03	0.315911E-01	
0.100000E+01	0.500000E+01	0.850000E+00	0.100000E+03	0.383577E+03	0.985000E+00	0.560000E-01	0.	
0.200000E+01	0.900000E+00	0.473956E-01	0.154510E+03	0.	0.357312E+03	0.315911E-01	0.111699E+04	
0.230000E+01	0.903000E+00	0.443824E-01	0.122659E+03	0.372070E+02	0.394519E+03	0.285267E-01	0.	
0.	0.	0.100000E+01	0.833333E+00	0.100000E+03	0.814332E+00	0.100000E+03	0.109422E+02	
0.500000E+02	0.130000E+03	0.	0.970000E-01	0.	0.	0.	0.100000E+01	
0.100000E+01	0.500000E+03	0.383577E+03	0.	0.	0.355503E+03	0.	0.	
0.881300E+03	0.500000E+01	0.211573E+03	0.160867E+01	0.881300E+03	0.500000E+01	0.211573E+03	0.160867E+01	
0.881300E+03	0.490000E+01	0.211573E+03	0.161005E+01	0.881300E+03	0.490000E+01	0.211573E+03	0.161005E+01	
0.	0.	0.	0.	0.	0.	0.	0.	
0.116423E+03	0.	0.116423E+03	0.	0.	0.200000E-01	0.303518E+00	0.	
0.	0.	0.	0.	0.	0.	0.	0.	
0.228157E+04	0.466804E+01	0.604585E+03	0.187798E+01	0.881300E+03	0.490000E+01	0.211573E+03	0.161005E+01	
0.109422E+02	0.394519E+03	0.285267E-01	0.	0.116423E+03	0.	0.100000E+01	0.	
0.199038E+04	0.468986E+01	0.515033E+03	0.183454E+01	0.199038E+04	0.445537E+01	0.515033E+03	0.183806E+01	
0.199038E+04	0.445537E+01	0.515033E+03	0.183806E+01	0.199038E+04	0.445537E+01	0.515033E+03	0.183806E+01	
0.510942E+03	0.	0.510942E+03	0.218844E-01	0.	0.500000E-01	0.535611E+03	0.510401E+03	
0.449153E+01	0.541234E+03	0.256683E+00	0.196783E+04	0.424783E+01	0.571843E+03	0.271337E+00	0.356444E+00	
0.171607E+04	0.243248E+01	0.197586E+04	0.100000E+01	0.171607E+04	0.243248E+01	0.197586E+04	0.100000E+01	
0.	0.	0.	0.	0.189192E+04	0.300448E+05	0.	0.139553E+05	
0.300448E+05	0.139553E+05	0.109422E+02	0.510942E+03	0.218844E-01	0.440001E+05	0.440001E+05	0.895270E+00	

## CONCLUDING REMARKS

The computer code (GENENG) presented herein has proven to be an indispensable tool for steady-state cycle analysis of various types of jet engines, namely two-spool mixed- or unmixed-flow turbofans and one- or two-spool turbojets.

This program has been found to be valuable for many applications because it has the capability of studying a broad range of engine types having different design characteristics, while it also has low-execution-time requirements. The laborious task of matching components manually at off-design operating has been eliminated.

The program has proven itself to be easy to use, especially in terms of input requirements. It is felt that with a minimum of effort the reader can become proficient in using the computer code.

The code is available to be reproduced on the requestor's tape upon application to the authors.

Lewis Research Center,

National Aeronautics and Space Administration,

Cleveland, Ohio, October 1, 1971,

132-15.

# APPENDIX A

## GENENG PROGRAM LISTING

```

$IBFTC GENENG DECK
COMMON /POINT/IDATPT
COMMON/LOOPPR/KKGO,PRFNEW,PRCNEW
DATAIII/O/
DIMENSION XLO(5),XHI(5)
EQUIVALENCE(XLO,DHLDSV),(XHI,DHHISV)
COMMON/ALL/X(28)/DESIGN/Y(80)/FRONT/Z(80)/SIDE/W(48)/BACK/V(72)
COMMON/DUMMYS/DUMMY(100)
EQUIVALENCE (P6DSAV,DUMMY(7)),(AM6DSV,DUMMY(8)),(ETAASV,DUMMY(9)),
1(FAR7SV,DUMMY(10))
LOGICAL ERRER,CLEAR
DATA CLEAR/.TRUE./
COMMON/ERER/ERRER
ERRER=.FALSE.
IF (.VDT.CLEAR) CALL ENGBAL
CLEAR=.FALSE.
DO 1 J=1,408
1 X(J)=0.
IF (III.EQ.0) KKGO=0
III=1
IDATPT=0
CALL CONDUT (1)
P6DSAV=1.
AM6DSV=1.
ETAASV=1.
FAR7SV=1.
DO 2 I=1,5
2 XLO(I)=100.
XHI(I)=100.
CALL ENGBAL
STOP
END

```

```

$IBFTC ENGBAL DECK
SUBROJTINE ENGBAL
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASMX,IDBURN,IAFTBN,IDCD ,IMCD ,IDSHJC,IMSHOC,NOZFLT,
3ITRYS ,LOOPER,NOMAP ,NUMMAP,MAPEDG,TDLALL,ERR(6)
COMMON /DESIGN/
1PCNFGJ,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC,
2ZFDS ,PCNFDS,PRFDS ,ETAFDS,WAFDS ,PRFCF ,ETAFCF,WAFCF ,
3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,
4T4DS ,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCOCF,ETABCF,
5TFHPDS ,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS ,
6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLPDF,ETLPCF,DHLPCF,T21DS ,
7T24DS ,WFDDS ,DTDUDS,ETADD,WA23DS,DPDUDS,DTDUCF,ETADCF,
8T7DS ,WFADS ,DTAFDS,ETAADS,WG6CDS,DPAFDS,DTAFCF,ETAACF,
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
$PS55 ,AM55 ,CVDNOZ,CVMNJZ,A8SAV ,A9SAV ,A28SAV,A29SAV

```

	COMMON / FRONT/	17
	1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,	18
	2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,	19
	3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,	20
	4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLOU ,BLOB ,	21
	5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,	22
	6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,	23
	7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,	24
	8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,	25
	9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,	26
	\$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLOB ,PCBLHP ,PCBLLP	27
	COMMON / SIDE/	28
	1XP1 ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3 ,DUMS1 ,DUMS2 ,	29
	2XT21 ,XP21 ,XH21 ,XS21 ,T23 ,P23 ,H23 ,S23 ,	30
	3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 ,	31
	4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 ,	32
	5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,JPDUC ,BYPASS ,DUMS3 ,	33
	6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29	34
	COMMON / BACK/	35
	1XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,	36
	2XWFB ,XWG55 ,XFAR55 ,XWFD ,XWG24 ,XFAR24 ,XFP1 ,DUMB ,	37
	3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,	38
	4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,	39
	5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,	40
	6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,	41
	7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,	42
	8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,	43
	9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC	44
	COMMON/DUMMYS/DUMMY(100)	45
	LOGICAL ERROR,FAN	46
	EQUIVALENCE (FAN,DUMMY(11)),(ISPOOL,DUMMY(6))	47
	COMMON/ERER/ERRER	48
	DIMENSION VAR(6),DEL(6),ERRB(6),DELVAR(6),EMAT(6,6),VMAT(6),	49
	IAMAT(5)	50
	DATA AWORD/6HENGBAL/	51
	DATA VDELTA,VLIM,VCHNGE,NOMISX/	52
	1 0.001,0.100,0.850,4/	53
	CALL PUTIN	54
	IF (INIT.EQ.1) GO TO 1	55
	TFFHP=TFFHPDS	56
	TFFLP=TFFLPDS	57
1	LOOPER=0	58
	NUMMAP=0	59
	NOMISS=0	60
2	LOOP=0	61
	MISMAT=0	62
	NOMAP=0	63
	IGO=2	64
	DO 3 I=1,6	65
	VMAT(I)=0.	66
	AMAT(I)=0.	67
	DELVAR(I)=0.	68
	DO 3 L=1,6	69
3	EMAT(I,L)=0.	70
4	LOOPER=LOOPER+1	71
	CALL CDFAN	72
	WORD=AWORD	73
	IF (LJOPER.GT.ITRYS) ERRER=.TRUE.	74
	IF (LJOPER.GT.ITRYS) GO TO 24	75
	IF (NOMAP.GT.0) GO TO 2	76
	NUMMAP=0	77

5	VAR(1)=ZF*100.	78
	IF (MJDE.NE.3) VAR(2)=PCNF	79
	IF (MJDE.EQ.3) VAR(2)=T4/10.	80
	VAR(3)=ZC*100.	81
	IF (MJDE.NE.1) VAR(4)=PCNC	82
	IF (MJDE.EQ.1) VAR(4)=T4/10.	83
	VAR(5)=TFFHP	84
	VAR(6)=TFFLP	85
	NMAX=5	85
	IF (FAN.OR.ISPOOL.EQ.2) GO TO 6	87
	NMAX=3	88
	VAR(3)=TFFLP	89
6	DO 7 I=1,NMAX	90
7	IF (ABS(ERR(I)).GT.TOLALL) GO TO 8	91
	CALL PERF	92
	CALL ERROR	93
8	IF (LJJP.GT.0) GO TO 10	94
	MAPEDG=0	95
	MAPSET=0	96
	DO 9 I=1,NMAX	97
	ERRB(I)=ERR(I)	98
9	DEL(I)=VDELTA*VAR(I)	99
	GO TO 13	100
10	IF (MISMAT.GT.0) GO TO 27	101
	IF (MAPEDG.EQ.0) GO TO 11	102
	MAPEDG=0	103
	MAPSET=1	104
	VAR(LJJP)=VAR(LOOP)+2.*DEL(LOOP)	105
	GO TO 14	106
11	IF (MAPSET.EQ.0) VAR(LOOP)=VAR(LOOP)+DEL(LOOP)	107
	IF (MAPSET.EQ.1) VAR(LOOP)=VAR(LOOP)-DEL(LOOP)	108
	MAPSET=0	109
	DO 12 I=1,NMAX	110
12	EMAT(I,LOOP)=(ERRB(I)-ERR(I))/DEL(LOOP)	111
13	LOOP=LOOP+1	112
	IF (LJJP.GT.NMAX) GO TO 15	113
	VAR(LJJP)=VAR(LOOP)-DEL(LOOP)	114
14	ZF=VAR(1)/100.	115
	IF (MJDE.NE.3) PCNF=VAR(2)	116
	IF (MJDE.EQ.3) T4=VAR(2)*10.	117
	ZC=VAR(3)/100.	118
	IF (MJDE.NE.1) PCNC=VAR(4)	119
	IF (MJDE.EQ.1) T4=VAR(4)*10.	120
	TFFHP=VAR(5)	121
	TFFLP=VAR(6)	122
	IF (ISPOOL.EQ.1) TFFLP=VAR(3)	123
	IF (ZF.LT.0.) ZF=0.05	124
	IF (ZC.LT.0.) ZC=0.05	125
	GO TO (2,4),IGO	126
15	DO 16 I=1,NMAX	127
16	AMAT(I)=-ERRB(I)	128
	DO 18 I=1,NMAX	129
	IZERO=0	130
	DO 17 LOOP=1,NMAX	131
17	IF (EMAT(I,LOOP).EQ.0.) IZERO=IZERO+1	132
	IF (IZERO.LT.NMAX) GO TO 18	133
	WRITE (6,30) I	134
	LOOPER=ITRYS+100	135
	GO TO 24	136
18	CONTINJE	137
	DO 20 LOOP=1,NMAX	138
	IZERO=0	139
	DO 19 I=1,NMAX	140



19	IF (EMAT(I,LOOP).EQ.0.) IZERO=IZERO+1	141
	IF (IZERO.LT.NMAX) GO TO 20	142
	WRITE (6,31) LOOP	143
	LOOPER=ITRYS+100	144
	GO TO 24	145
20	CONTINUE	146
21	CALL MATRIX (EMAT,VMAT,AMAT,NMAX)	147
	LBIG=0	148
	VARBIG=0.0	149
	DO 22 L=1,NMAX	150
	ABSVAR=ABS(VMAT(L))	151
	IF (ABSVAR.LE.VLIM*VAR(L)) GO TO 22	152
	IF (ABSVAR.LE.VARBIG) GO TO 22	153
	LBIG=L	154
	VARBIG=ABSVAR	155
22	CONTINUE	156
	VRATIO=1.0	157
	IF (LBIG.GT.0) VRATIO=VLIM*VAR(LBIG)/VARBIG	158
	ERRAVE=0.0	159
	VMTAVE=0.0	160
	DELAVE=0.0	161
	DO 23 L=1,NMAX	162
	DELVAR(L)=VRATIO*VMAT(L)	163
	ERRAVE=ERRAVE+ABS(AMAT(L))	164
	VMTAVE=VMTAVE+ABS(VMAT(L))	165
	DELAVE=DELAVE+ABS(DELVAR(L))	166
23	VAR(L)=VAR(L)+DELVAR(L)	167
	ERRAVE=ERRAVE/FLOAT(NMAX)	168
	VMTAVE=VMTAVE/FLOAT(NMAX)	169
	DELAVE=DELAVE/FLOAT(NMAX)	170
	IF (MISMAT.GT.0) GO TO 29	171
	IF (NOMISS.EQ.0) MISMAT=1	172
	IF (MISMAT.EQ.0) IGO=1	173
24	WRITE (8,32) LOOPER	174
	DO 25 I=1,NMAX	175
25	WRITE (8,33) AMAT(I),(EMAT(I,L),L=1,6),VMAT(I),DELVAR(I),VAR(I)	176
	WRITE (8,34) ERRAVE,VMTAVE,DELAVE	177
26	IF (LJPPER.LT.ITRYS) GO TO 14	178
	CALL ERROR	179
	RETURN	180
27	VMTAVX=VMTAVE	181
	DO 28 I=1,NMAX	182
28	AMAT(I)=-ERR(I)	183
	GO TO 21	184
29	WRITE (8,35) AMAT,ERRAVE,DELVAR,DELAVE,VMAT,VMTAVE,VAR	185
	MISMAT=MISMAT+1	186
	IF (VMTAVE.LT.VCHNGE*VMTAVX) GO TO 26	187
	WRITE (8,36)	188
	IF (MISMAT.LT.NOMISX) NOMISS=1	189
	MISMAT=0	190
	LOOP=J	191
	IGO=2	192
	GO TO 5	193
C		194
30	FORMAT (4HOROW,I2,16H IS ZERO IN EMAT)	195
31	FORMAT (7HOCOLUMN,I2,16H IS ZERO IN EMAT)	196
32	FORMAT (8H8  ERR8,28X23HERROR MATRIX AFTER LOOP,I4,29X4HVMTAT,6X5H	197
	1DELVAR,7X14HVARIABLE\$\$\$\$\$)	198
33	FORMAT (1H0,F8.4,8X6F10.4,10XF10.4,F11.4,4XF11.4,6H\$\$\$\$\$)	199
34	FORMAT (1H0,F8.4,32X14HAVERAGE VALUES,31X,2F11.4,6H\$\$\$\$\$)	200

35	FORMAT (12H----- AMAT,7F16.6,6H\$\$\$\$\$,/,12H -----DELVAR,7F16.6,6	201
	1H\$\$\$\$\$,/,12H ----- VMAT,7F16.6,6H\$\$\$\$\$,/,12H ----- VAR,6F16.6	202
	2,6H\$\$\$\$\$)	203
36	FORMAT (1H0,50X22HCHANGE TOD SMALL\$\$\$\$\$)	204
	END	205

\$IBFTC GUESS DECK

FUNCTION GUESS(M,T,TD,P,PD,W,WD,D,DD,VD)	1
IF (M.EQ.0) GUESS=VD*((T/TD)**1.60)*((DD/D)**0.50)	2
IF (M.EQ.1) GUESS=VD*((P/PD)**1.80)*((DD/D)**0.33)	3
IF (M.EQ.2) GUESS=VD*((W/WD)**0.33)*((DD/D)**1.00)	4
IF (M.EQ.3) GUESS=VD*((W/WD)**0.00)*((P/PD)**0.50)	5
IF (M.EQ.4) GUESS=VD*((W/WD)**0.00)*((P/PD)**0.50)	6
IF (M.EQ.5) GUESS=VD*((T/TD)**1.10)*((DD/D)**0.60)	7
IF (M.EQ.6) GUESS=VD*((P/PD)**1.00)*((D/DD)**0.25)	8
IF (M.EQ.7) GUESS=VD*((P/PD)**0.62)*((D/DD)**0.31)	9
RETURN	10
END	11

\$IBFTC MATRIX DECK

SUBROJTIME MATRIX (E,V,A,N)	1
DIMENSION E(6,6),V(6),A(6),PIV(7),T(6,7)	2
NN=N+1	3
NM=N-1	4
DO 1 I=1,N	5
T(I,NV)=A(I)	6
DO 1 J=1,N	7
1 T(I,J)=E(I,J)	8
DO 7 I=1,N	9
TEMP=0.	10
DO 2 J=I,N	11
IF (TEMP.GT.ABS(T(J,I))) GO TO 2	12
TEMP=ABS(T(J,I))	13
IPIV=J	14
2 CONTINUE	15
IP1=I+1	16
DO 3 J=IP1,NN	17
3' PIV(J)=T(IPIV,J)/T(IPIV,I)	18
IFROM=N	19
ITO=N	20
4 IF (IFROM.EQ.IPIV) GO TO 6	21
RM=-T(IFROM,I)	22
DO 5 J=IP1,NN	23
5 T(ITO,J)=T(IFROM,J)+RM*PIV(J)	24
ITO=ITO-1	25
6 IFROM=IFROM-1	26
IF (IFROM.GE.I) GO TO 4	27
DO 7 J=IP1,NN	28
7 T(I,J)=PIV(J)	29
DO 8 I=1,NM	30
J=NN-I	31
K=N-I	32
DO 8 L=J,N	33

```

8      T(K,NI)=T(K,NN)-T(K,L)*T(L,NN)
      DO 9 I=1,N
9      V(I)=T(I,NN)
      RETURN
      END

```

```

$IBFTC PUTIN  DECK
SUBROJTINE PUTIN
COMMON /POINT/IDATPT
COMMON /  ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASMX ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,
3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLALL ,ERR(6)
COMMON /DESIGN/
1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
4T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3CDS ,JPCODS ,DTCOCF ,ETABCF ,
5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,JHLPCF ,T21DS ,
7T24DS ,WFDDDS ,DTODDS ,ETADDS ,WA23DS ,DPDUDS ,DTDUCF ,ETADCF ,
8T7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
$PS55 ,AM55 ,CVDNOZ ,CVMNJZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON / FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,
7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,
8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLDB ,PCBLHP ,PCBLLP
COMMON / SIDE/
1XP1 ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3 ,DUMS1 ,DUMS2 ,
2XT21 ,XP21 ,XH21 ,XS21 ,T23 ,P23 ,H23 ,S23 ,
3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 ,
4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 ,
5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS ,DUMS3 ,
6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29
COMMON / BACK/
1XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,
2XWFB ,XWG55 ,XFAR55 ,XWFD ,XWG24 ,XFAR24 ,XP1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC
COMMON/DUMMYS/DUMMY(100)
COMMON/LOOPPR/KKGO,PRFNEW,PRCNEW
DIMENSION XSAVE(308),XFILL(1)
EQUIVALENCE (XFILL,WORD)
LOGICAL ERER
COMMON/ERER/ERRER

```

C ***	ITITLE=1	WILL READ IN TITLE	52
C ***	MODE =0	FOR CONSTANT T4	53
C ***	MODE =1	FOR CONSTANT PCNC	54
C ***	MODE =2	FOR CONSTANT WFB	55
C***	MODE=3	FOR CONSTANT PCNF	55
C ***	INIT =1	WILL NOT INITIALIZE POINT	57
C ***	IDUMP =1	WILL DUMP LOOPING WRITE-OUTS IF ERROR OCCURS	58
C ***	IDUMP =2	WILL DUMP LOOPING WRITE-OUTS AFTER EVERY POINT	59
C ***	IAMTP =0	WILL USE INPUT AM AND MIL SPEC ETAR	60
C ***	IAMTP =1	WILL USE INPUT AM AND INPUT ETAR	61
C ***	IAMTP =2	WILL USE T2 AS T1=T1+T2 AND STANDARD P1	62
C ***	IAMTP =3	WILL USE P2 AND STANDARD T1	63
C ***	IAMTP =4	WILL USE T2 AND P2	64
C ***	IAMTP =5	WILL USE RAM2 FOR SPECIAL RECOVERY	65
C ***	IGASMX=-1	SEPERATE FLOW, INPUT AM6	66
C ***	IGASMX=0	SEPARATE FLOW, A6=A55	67
C ***	IGASMX=1	WILL MIX DUCT AND MAIN STREAMS, A6=A25+A55	68
C ***	IGASMX=2	WILL MIX DUCT AND MAIN STREAMS, INPUT AM6	69
C ***	IDBURN=1	FOR DUCT BURNING, INPUT T24	70
C ***	IDBURN=2	FOR DUCT BURNING, INPUT WFD	71
C ***	IAFTBV=1	FOR AFTERBURNING, INPUT T7	72
C ***	IAFTBN=2	FOR AFTERBURNING, INPUT WFA	73
C ***	IDCD =1	DUCT NOZZLE WILL BE C-D	74
C ***	IMCD =1	MAIN NOZZLE WILL BE C-D	75
C ***	NOZFLT=1	FOR FLOATING MAIN NOZZLE	76
C ***	NOZFLT=2	FOR FLOATING DUCT NOZZLE	77
C ***	NOZFLT=3	FOR FLOATING MAIN AND DUCT NOZZLES	78
C ***	ITRYS =N	NUMBER OF PASSES THRU ENGINE BEFORE QUITTING	79
		DIMENSION ITABLE(400)	80
		DATA (ITABLE(I),I=1,3)/0,400,0/	81
1		CALL ZERO	82
		IF (K<GO.EQ.1) GO TO 5	83
		IDES=0	84
		CALL INPUT (5,6,1,WORD,ITABLE)	85
		IF (ERRER.AND.IAFTBN.GT.0) GO TO 1	86
		ERRER=.FALSE.	87
C		TABLE IS REFERENCED TO COMMON/ALL/FIRST ENTRY	88
		IF (IDES.EQ.0) GO TO 7	89
		IF (K<GO.NE.2) GO TO 3	90
		DO 2 I=1,308	91
2		XFILL(I)=XSAVE(I)	92
		CALL INPUT (5,6,1,WORD,ITABLE)	93
3		CONTINUE	94
C		SAVE INPUT IN CASE OF LOOP ON PRESSURE RATIOS	95
		DO 4 I=1,308	96
4		XSAVE(I)=XFILL(I)	97
		GO TO 7	98
5		DO 6 I=1,308	99
6		XFILL(I)=XSAVE(I)	100
		WRITE (6,8) PRFDS,PRFNEW,PRCDS,PRCNEW	101
		PRCDS=PRCNEW	102
		PRFDS=PRFNEW	103
7		CONTINUE	104
		KKGO=2	105
		IF (IAFTBN.GT.0.OR.IDBURN.GT.0) INIT=1	106
		IF (NOZFLT.GT.0) INIT=1	107
		IF (MODE.EQ.0) WRITE (8,9) IDES,AM,ALTP,T4,T24,T7	108
		IF (MODE.EQ.1) WRITE (8,10) IDES,AM,ALTP,PCNC,T24,T7	109
		IF (MODE.EQ.2) WRITE (8,11) IDES,AM,ALTP,WFB,T24,T7	110
		CALL COINLT	111
		RETURN	112
C			113

```

C
C
8   FORMAT (18HOCHANGE PRFDS FROM,F9.3,4H TO,F9.3,16H AND PRCD$ FROM
1,F10.3,4H TO,F10.3)
9   FORMAT (1HO,7H IDE$=,I3,10X7H      A4=,F7.3,6X7H  ALTP=,F7.0,6X7H
1   T4=,F8.2,5X7H  T24=,F8.2,5X7H    T7=,F8.2,6H$$$$$$)
10  FORMAT (1HO,7H IDE$=,I3,10X7H      AM=,F7.3,6X7H  ALTP=,F7.0,6X7H
1PCNC=,F8.3,5X7H  T24=,F8.2,5X7H    T7=,F8.2,6H$$$$$$)
11  FORMAT (1HO,7H IDE$=,I3,10X7H      AM=,F7.3,6X7H  ALTP=,F7.0,6X7H
1   WFB=,F8.4,5X7H  T24=,F8.2,5X7H    T7=,F8.2,6H$$$$$$)
END

```

```

$IBFTC ZERO   DECK
SUBROUTINE ZERO
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASM$,IDBURN,IAFTBV,IDCD ,IMCD ,IDSHOC,IMSHOC,VOZFLT,
3ITRYS ,LOOPER,NOMAP ,NUMMAP,MAPEDG,TOLALL,ERR(6)
COMMON/DESIGN/QXQ(80)
COMMON / FRONT/
1T1   ,P1   ,H1   ,S1   ,T2   ,P2   ,H2   ,S2   ,
2T21  ,P21  ,H21  ,S21  ,T3   ,P3   ,H3   ,S3   ,
3T4   ,P4   ,H4   ,S4   ,T5   ,P5   ,H5   ,S5   ,
4T55  ,P55  ,H55  ,S55  ,BLF  ,BLC  ,BLDU ,BLOB ,
5CNF  ,PRF  ,ETAF ,WAF  ,WAF  ,WA3  ,WG4  ,FAR4 ,
6CNC  ,PRC  ,ETAC ,WACC ,WAC  ,ETAB ,DPCOM ,DUMF ,
7CNHP ,ETATHP,DHTCHP,DHTC ,BLHP ,WG5  ,FAR5 ,CS   ,
8CNLP ,ETATLP,DHTCLP,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,
9AM   ,ALTP ,ETAR ,ZF   ,PCNF ,ZC   ,PCNC ,WFB  ,
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU,PCBLOB,PCBLHP,PCBLLP
COMMON / SIDE/
1XP1  ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3   ,DUMS1 ,DUMS2 ,
2XT21 ,XP21 ,XH21 ,XS21 ,T23   ,P23   ,H23   ,S23   ,
3T24  ,P24  ,H24  ,S24  ,T25   ,P25   ,H25   ,S25   ,
4T28  ,P28  ,H28  ,S28  ,T29   ,P29   ,H29   ,S29   ,
5WAD  ,WFD  ,WG24 ,FAR24 ,ETAD  ,DPDUC ,BYPASS,DUMS3 ,
6TS28 ,PS28 ,V28  ,AM28 ,TS29  ,PS29 ,V29   ,AM29
COMMON / BACK/
1XT55 ,XP55 ,XH55 ,XS55 ,XT25  ,XP25 ,XH25 ,XS25 ,
2XWFR ,XWG55 ,XFAR55,XWFD ,XWG24 ,XFAR24,XXP1 ,DUMB ,
3T6   ,P6   ,H6   ,S6   ,T7   ,P7   ,H7   ,S7   ,
4T8   ,P8   ,H8   ,S8   ,T9   ,P9   ,H9   ,S9   ,
5WG6  ,WFA  ,WG7  ,FAR7  ,ETAA  ,DPAFT ,V55  ,V25  ,
6PS6  ,V6   ,AM6  ,TS7  ,PS7  ,V7   ,AM7  ,AM25 ,
7TS8  ,PS8  ,V8   ,AM8  ,TS9  ,PS9  ,V9   ,AM9  ,
8VA   ,FRD  ,VJD  ,FGMD ,VJM  ,FGMM ,FGPD ,FGPM ,
9FGM  ,FGP  ,WFT  ,WGT  ,FART ,FG   ,FN   ,SFC
COMMON/DUMMYS/DUMMY(100)
DIMENSION Z1(63),Z2(48),Z3(72)
EQUIVALENCE (Z1,T1),(Z2,XP1),(Z3,XT55)
IDES=J
JDES=J
INIT=0
IDBURN=0
IAFTBV=0
IDSHOC=3
IMSHOC=3
T2Q=T2

```

	P2Q=P2	46
	T4Q=T4	47
	DO 1 I=1,63	48
1	Z1(I)=0.	49
	DO 2 I=1,48	50
2	Z2(I)=0.	51
	DO 3 I=1,72	52
3	Z3(I)=0.	53
	T2=T2Q	54
	P2=P2Q	55
	T4=T4Q	56
	CALL SYG (1)	57
	RETURN	58
	END	59

```

$IBFTC COINLT DECK
SUBROJTINE COINLT
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASM ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,
3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPELG ,TOLALL ,ERR(6)
COMMON /DESIGN/
1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
4T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3CDS ,DPCODS ,DTCOCF ,ETABCF ,
5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T21DS ,
7T24DS ,WFDDDS ,DTDUDS ,ETAADS ,WA23DS ,DPDUDS ,DTDUCF ,ETADCF ,
8T7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
$PS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON / FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,
7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,
8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLDB ,PCBLHP ,PCBLLP
COMMON/SIDE/ZYX(48)/BACK/YZX(72)
COMMON/DUMMYS/DUMMY(100)
DATA AWORD/6HCOINLT/
WORD=AWORD
AJ=773.26
G=32.174049
ALT=ALTP*2.0855531E+07/(2.0855531E+07-ALTP)
CALL ATMOS (ALT,T1,XX1,XX2,XX3,P1,CS,XX4,IIER)
IF (IAMTP.EQ.2) T1=T1+T2
IF (IAMTP.EQ.5) CALL RAM2 (AM,ETAR)
IF (IAMTP.NE.1.AND.IAMTP.NE.5) CALL RAM (AM,ETAR)
FAR=0.0
CALL PROCOM (FAR,T1,CS,XX2,XX3,R1,PHI1,H1)
S1=PHI1-R1*ALOG(P1)
H2=H1+(AM*CS)**2/(2.*AJ*G)

```

```

P2T=1.
DO 1 I=1,10
CALL THERMO (P2T,H2,T2T,S2T,AW,0,0.0,1)
IF (ABS(S2T-S1).LE.0.0001*S1) GO TO 2
1 P2T=P1*EXP((AW/1.986375)*((S2T-S1)+(1.986375/AW)*ALDG(P2T/P1)))
CALL ERROR
RETURN
2 IF (IAMTP.EQ.3.OR.IAMTP.EQ.4) ETAR=P2/P2T
P2=ETAR*P2T
IF (IAMTP.NE.4) CALL THERMO (P2,H2,T2,S2,XX5,0,0.0,1)
IF (IAMTP.EQ.4) CALL THERMO (P2,H2,T2,S2,XX5,0,0.0,0)
IF (INIT.EQ.1) GO TO 5
IF (IDES.EQ.1) GO TO 3
IF (MODE.EQ.3) GO TO 4
PCNF=3JESS(MODE,T4,T4DS,PCNC,PCNCDS,WFB,WFBDS,T2,T2DS,PCNFDS)
PCNFGJ=PCNF
GO TO 4
3 PCNF=PCNFDS
PCNFGJ=PCNF
T2DS=T2
4 ZF=ZFDS
5 RETURN
END

```

\$IBFTC ATMOS DECK

```

SUBROUTINE ATMOS (ZFT, TM, SIGMA, RHO, THETA, DELTA, CA, AMU, K)
C THIS IS A SUBROUTINE TO COMPUTE CERTAIN ELEMENTS OF THE 1962
C U.S. STANDARD ATMOSPHERE UP TO 90 KILOMETERS.
C CALLING SEQUENCE...
C
C CALL ATMOS (ZFT, TM, SIGMA, RHO, THETA, DELTA, CA, AMU, K)
C ZFT = GEOMETRIC ALTITUDE (FEET)
C TM = MOLECULAR SCALE TEMPERATURE (DEGREES RANKINE)
C SIGMA = RATIO OF DENSITY TO THAT AT SEA LEVEL
C RHO = DENSITY (LB-SEC**2-FT**(-4) OR SLUGS-FT**3)
C THETA = RATIO OF TEMPERATURE TO THAT AT SEA LEVEL
C DELTA = RATIO OF PRESSURE TO THAT AT SEA LEVEL
C CA = SPEED OF SOUND (FT/SEC)
C AMU = VISCOSITY COEFFICIENT (LB-SEC/FT**2)
C
C K = 1 NORMAL
C = 2 ALTITUDE LESS THAN -5000 METERS OR GREATER THAN 90 KM
C = 3 FLOATING POINT OVERFLOW
C
C ALL DATA AND FUNDAMENTAL CONSTANTS ARE IN THE METRIC SYSTEM AS
C THESE QJANTITIES ARE DEFINED AS EXACT IN THIS SYSTEM.
C
C THE RADIUS OF THE EARTH (REFT59) IS THE VALUE ASSOCIATED WITH THE
C 1959 ARDC ATMOSPHERE SO THAT PROGRAMS CURRENTLY USING THE LIBRARY
C ROUTINE WILL NOT REQUIRE ALTERATION TO USE THIS ROUTINE.
C DIMENSION HB(10),TMB(10),DELTAB(10),ALM(10)
C DATA(HB(I), TMB(I), DELTAB(I), ALM(I),I=1,10)/
1 -5.0, 320.65, 1.75363E 00, -6.5,
2 0.0, 288.15, 1.00000E 00, -6.5,
3 11.0, 216.65, 2.23361E-01, 0.0,
4 20.0, 216.65, 5.40328E-02, 1.0,
5 32.0, 228.65, 8.56663E-03, 2.8,
6 47.0, 270.65, 1.09455E-03, 0.0,

```

7	52.0,	270.65,	5.82289E-04,	-2.0,	34
8	61.0,	252.65,	1.79718E-04,	-4.0,	35
9	79.0,	180.65,	1.0241 E-05,	0.0,	36
\$	88.743,	180.65,	1.6223 E-06,	0.0/	37
	DATA	REFT59/2.0855531E 07/,	GZ	/9.80665/,	38
1	AMZ	/28.9644	/,	RSTAR /8.31432/,	39
2	FTTOKM/3.048E-04	/,	S	/110.4 /,	40
3	AMUZ /1.2024E-05	/,	CAZ	/1116.45/,	41
4	RHOZ /0.076474	/,	GZENG	/32.1741/	42
C	CONVERT GEOMETRIC ALTITUDE TO GEOPOTENTIAL ALTITUDE				43
	HFT=(REFT59/(REFT59+ZFT))*ZFT				44
C	CONVERT HFT AND ZFT TO KILOMETERS				45
	Z=FTTOKM*ZFT				46
	H=FTTOKM*HFT				47
	K=1				48
	TMZ=TMZ(2)				49
	IF (H.LT.-5.0.OR.Z.GT.90.0) GO TO 7				50
	DO 1 M=1,10				51
	IF (H-HB(M)) 2,3,1				52
1	CONTINUE				53
	GO TO 7				54
2	M=M-1				55
3	DELH=H-HB(M)				56
	IF (ALM(M).EQ.0.0) GO TO 4				57
	TMK=TMZ(M)+ALM(M)*DELH				58
C	GRADIENT IS NON ZERO, PAGE 10, EQUATION I.2.10-(3)				59
	DELTA=DELTAB(M)*((TMZ(M)/TMK)**(GZ*AMZ/(RSTAR*ALM(M))))				60
	GO TO 5				61
4	TMK=TMZ(M)				62
C	GRADIENT IS ZERO, PAGE 10, EQUATION I.2.10-(4)				63
	DELTA=DELTAB(M)*EXP(-GZ*AMZ*DELH/(RSTAR*TMZ(M)))				64
5	THETA=TMK/TMZ				65
	SIGMA=DELTA/THETA				66
	ALPHA=SQRT(THETA**3)*((TMZ+S)/(TMK+S))				67
C	CONVERSION TO ENGLISH UNITS				68
	TM=1.8*TMK				69
	RHO=RHOZ*SIGMA/GZENG				70
	CA=CAZ*SQRT(THETA)				71
	AMU=AMUZ*ALPHA/GZENG				72
	CALL JVERFL (J)				73
	GO TO (6,8),J				74
6	K=K+2				75
	GO TO 8				76
7	K=2				77
8	RETURN				78
	END				79

\$IBFTC RAMS	DECK				
	SUBROUTINE RAM (AM,ETAR)				1
	IF (AM.GT.1.) GO TO 2				2
	ETAR=1.				3
1	RETURN				4
2	IF (AM.GT.5.) GO TO 3				5
	ETAR=1.-0.075*((AM-1.)**1.35)				6
	GO TO 1				7
3	ETAR=900./((AM**4)+935.)				8
	GO TO 1				9
	END				10



```

$IBFTC RAMT#0 DECK
SUBROJTINE RAM2 (AM,ETAR)
DIMENSION PRINLT(15),FMN(15)
DIMENSION Y(3),X(3)
DATA FMN/0.,.1,.2,.3,.4,.5,.8,1.1,1.2,1.4,1.6,1.8,2.2,2.4,2.7/
DATA PRINLT/.9,.932,.95,.961,.968,.97,.9701,.97,.9681,.958,.94,
1.9181,.858,.8201,.75/
M=0
DO 1 J=1,15
1 IF (AM.GE.FMN(J)) M=J-1
IF (M.EQ.0) M=1
IF (M.GE.14) M=13
DO 2 I=1,3
MM=M-1+I
X(I)=FMN(MM)
2 Y(I)=PRINLT(MM)
CALL PARABO (X,Y,AM,ETAR)
RETURN
END

```

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$IBFTC COFAN DECK
SUBROJTINE COFAN
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASMK ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,
3ITRYS ,LOOPER ,NDMAP ,NUMMAP ,MAPELG ,TDLALL ,ERR(6)
COMMON /DESIGN/
1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELF6 ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETAFFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
4T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3CDS ,DPCDSD ,DTCOCF ,ETABCF ,
5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T21DS ,
7T24DS ,WFDDSD ,DTDUDSD ,ETADDSD ,WA23DS ,DPDUDSD ,DTDUCF ,ETADCF ,
8T7DS ,WFADSD ,DTAFDSD ,ETAADSD ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
$PS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON / FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,
5CNF ,PRF ,ETAFF ,WAFCF ,WAF ,WA3 ,WG4 ,FAR4 ,
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,
7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,
8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLOB ,PCBLHP ,PCBLLP
COMMON/SIDE/ZYX(48)/BACK/YZX(72)
COMMON/DUMMYS/DUMMY(100)
COMMON / FAN/CNX(15),PRX(15,15),WACX(15,15),ETAX(15,15),
1NCN,NPT(15)
DIMENSION WLH(2)
DATA A#WORD,WLH/6H COFAN,6H (LO) ,6H (HI) /
WORD=A#WORD
THETA=SQRT(T2/518.668)
IF (IDES.EQ.1) THETAD=THETA

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CNF=PCNF*THETAD/(100.*THETA) 37
IF (IDES.EQ.1) WAFDS=W AFC*P2/THETA 38
IF (ZF.LT.0.) ZF=0. 39
IF (ZF.GT.1.) ZF=1. 40
CNFS=CNF 41
CALL SEARCH (ZF,CNF,PRF,W AFC,ETAF,CNX(1),NCN,PRX(1,1),WACX(1,1),ET 42
LAX(1,1),NPT(1),15,15,IGD) 43
IF ((CNF-CNFS).GT.0.0005*CNF) MAPEDG=1 44
IF (IGD.EQ.1.OR.IGD.EQ.2) WRITE (8,11) CNFS,WLH(IGD) 45
WAF=W AFC*P2/THETA 46
IF (IDES.NE.1) GO TO 1 47
PRFCF=(PRFDS-1.)/(PRF-1.) 48
ETAFCF=ETA FDS/ETA F 49
W AFCF=W AFCDS/W AF 50
WRITE (6,12) PRFCF,ETAFCF,W AFCF,T2DS 51
1 PRF=PRFCF*(PRF-1.)+1. 52
ETA F=ETAFCF*ETA F 53
W AF=W AFCF*W AF 54
W AFC=W AFCF*W AFC 55
PCNF=100.*THETA*CNF/THETAD 56
DUMD1=PCNF 57
CALL THCOMP (PRF,ETA F,T2,H2,S2,P2,T21,H21,S21,P21) 58
IF (PCBLF.GT.0.) BLF=PCBLF*W AF 59
IF (JDES.EQ.1) GO TO 8 60
JDES=1 61
IF (INIT.EQ.1) GO TO 7 62
IF (IDES.EQ.1) GO TO 5 63
IF (MJDE.NE.2) GO TO 2 64
T4=GUESS(3,Y1,Y2,PCNF,PCNFDS,WFB,WFBDS,Y7,Y8,T4DS) 65
PCNC=GUESS(4,Y1,Y2,PCNF,PCNFDS,WFB,WFBDS,Y7,Y8,PCNCDS) 66
GO TO 6 67
2 IF (MJDE.EQ.1) GO TO 4 68
IF (MJDE.EQ.0) GO TO 3 69
T4=GUESS(7,Y1,Y2,PCNF,PCNFDS,Y5,Y6,T2,T2DS,T4DS) 70
3 PCNC=GUESS(5,T4,T4DS,Y3,Y4,Y5,Y6,T21,T21DS,PCNCDS) 71
GO TO 6 72
4 T4=GUESS(6,Y1,Y2,PCNC,PCNCDS,Y5,Y6,T21,T21DS,T4DS) 73
GO TO 5 74
5 PCNC=PCNCDS 75
T4=T4DS 76
WFB=WFBDS 77
T21DS=T21 78
6 ZC=ZCDS 79
PCNCGU=PCNC 80
T4GU=T4 81
7 INIT=0 82
8 IF (MJDE.NE.3) GO TO 9 83
IF (ABS(CNF-CNFS).LE.0.001*CNFS) GO TO 10 84
WRITE (8,13) CNFS,CNF 85
CALL ERROR 86
9 PCNF=100.*THETA*CNF/THETAD 87
10 CALL DCOMP 88
RETURN 89
C 90
C 91
11 FORMAT (19H0 * * CNF OFF MAP,F10.4,2X A6,11H * * $$$$) 92
12 FORMAT (11HOFAN DESIGN,13X8H PRFCF=,E15.8,8H ETAFCF=,E15.8,8H W 93
1FCF=,E15.8,8H T2DS=,E15.8) 94
13 FORMAT (10HOCNF WAS= ,E15.8,11H AND NOW= ,E15.8,24H CHECK PCNF I 95
INPUT$$$$) 96
END 97

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$IBFTC COCOMP DECK
SUBROJTINE COCOMP
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASMX ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHDC ,IMSHOC ,NOZFLT ,
3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLALL ,ERR(6)
COMMON /DESIGN/
1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
4T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3CDS ,DPCODS ,DTCOCF ,ETABCF ,
5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T21DS ,
7T24DS ,WFDS ,DTODS ,ETADS ,WA23DS ,DPDUDS ,DTDUCF ,ETADCF ,
8T7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
$P555 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON / FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,
7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,
8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLDB ,PCBLHP ,PCBLLP
COMMON/SIDE/ZYX(48)/BACK/YZX(72)
COMMON/DUMMYS/DUMMY(100)
LOGICAL ERRER ,FAN
EQUIVALENCE (FAN,DUMMY(11)),(ISPOOL,DUMMY(6))
COMMON / COMP/CNX(15),PRX(15,15),WACX(15,15),ETAX(15,15),
1NCN,NPT(15)
DIMENSION WLH(2)
DATA AWORD,WLH/6HCOCOMP,6H (LO) ,6H (HI) /
WORD=AWORD
THETA=SQRT(T21/518.668)
IF (IDES.EQ.1) THETAD=THETA
CNC=PCNC*THETAD/(100.*THETA)
IF (.NOT.FAN) WACC=(WAF-BLF)*THETA/P21
IF (IDES.EQ.1) WACDS=WACC*P21/THETA
IF (ZC.LT.0.) ZC=0.
IF (ZC.GT.1.) ZC=1.
CNCS=CNC
IF (ISPOOL.EQ.1) GO TO 1
CALL SEARCH (ZC,CNC,PRC,WACC,ETAC,CNX(1),NCN,PRX(1,1),WACX(1,1),ET
1AX(1,1),NPT(1),15,15,IGO)
GO TO 2
1 PRC=1.
ETAC=1.
CNC=1.
PRCCF=1.
2 CONTINUE
IF (MJDE.EQ.1) GO TO 3
IF ((CNC-CNCS).GT.0.0005*CNC) MAPEDG=1
3 IF (IGJ.EQ.1.OR.IGD.EQ.2) WRITE (8,7) CNCS,WLH(IGO)
WAC=WACC*P21/THETA
IF (IDES.NE.1) GO TO 4
IF (ISPOOL.EQ.2) PRCCF=(PRCDS-1.)/(PRC-1.)
ETACCF=ETACDS/ETAC
IF (ISPOOL.EQ.1) ETACCF=1.0

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	WACCF=WACDS/WAC	62
	WRITE (6,8) PRCCF,ETACCF,WACCF,T21DS	63
4	PRC=PRCCF*(PRC-1.)+1.	64
	ETAC=ETACCF*ETAC	65
	WAC=WACCF*WAC	66
	WACC=WACCF*WACC	67
	IF (.VDT.FAN) ERR(5)=(WAF-WAC-BLF)/WAC	68
	CALL TCOMP (PRC,ETAC,T21,H21,S21,P21,T3,H3,S3,P3)	69
	IF (PCBLC.GT.0.) BLC=PCBLC*WAC	70
	WA3=WAC-BLC	71
	BLDU=PCBLDU*BLC	72
	BLOB=PCBLOB*BLC	73
	BLHP=PCBLHP*BLC	74
	BLLP=PCBLLP*BLC	75
	IF (MJDE.NE.1) GO TO 5	76
	IF (ABS(CNC-CNCS).LE.0.001*CNCS) GO TO 6	77
	WRITE (8,9) CNCS,CNC	78
	CALL ERRDR	79
5	PCNC=100.*THETA*CNC/THETAD	80
6	CALL COCOMB	81
	RETURN	82
C		83
C		84
7	FORMAT (19H0* * * CNC OFF MAP,F10.4,2XA6,11H* * *\$\$\$\$\$)	85
8	FORMAT (18HOCOMPRESSOR DESIGN,6X8H PRCCF=,E15.8,8H ETACCF=,E15.8,	86
	18H WACCF=,E15.8,8H T21DS=,E15.8)	87
9	FORMAT (10HCNC WAS= ,E15.8,11H AND NOW= ,E15.8,24H CHECK PCNC I	88
	INPUT\$\$\$\$\$)	89
	END	90

\$IBFTC COCOMB DECK

	SUBROJTINE COCOMB	1
	COMMON / ALL/	2
	1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,	3
	2IGASMX ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,	4
	3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLALL ,ERR(6)	5
	COMMON /DESIGN/	5
	1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC ,	7
	2ZFDS ,PCNFDS ,PRFDS ,ETAFFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,	8
	3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,	9
	4T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3CDS ,DPCDSD ,DTCOCF ,ETABCF ,	10
	5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,	11
	6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T21DS ,	12
	7T24DS ,WFDDSD ,DTDUDSD ,ETADDSD ,WA23DS ,DPDUDSD ,DTDUCF ,ETADCF ,	13
	8T7DS ,WFADSD ,DTAFDSD ,ETAADSD ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,	14
	9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,	15
	\$PS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV	16
	COMMON / FRONT/	17
	1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,	18
	2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,	19
	3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,	20
	4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLDB ,	21
	5CNF ,PRF ,ETAFF ,WAFCF ,WAF ,WA3 ,WG4 ,FAR4 ,	22
	6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCDM ,DUMF ,	23
	7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,	24
	8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,	25
	9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,	26

	\$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU,PCBLOB,PCBLHP,PCBLLP	27
	COMMON/SIDE/ZYX(48)/BACK/YZX(72)	28
	COMMON/DUMMYS/DIMMY(100)	29
	LOGICAL ERRER,FAN	30
	EQUIVALENCE (FAN,DIMMY(11)),(ISPOOL,DIMMY(6))	31
	COMMON / COMB/PSI(15),DELT(15,15),ETA(15,15),NPS,NPT(15)	32
	DIMENSION Q(9),DUMBO(15,15)	33
	DATA AWORD/6HCOCOMB/	34
	WORD=AWORD	35
	Q(2)=0.	35
	Q(3)=0.	37
	P3PSI=14.696*P3	38
	WA3C=WA3*SQRT(T3)/P3PSI	39
	IF (IDES.EQ.1) WA3CDS=WA3C	40
	DPCOM=DPCODS*(WA3C/WA3CDS)	41
	IF (DPCOM.GT.1.) DPCOM=1.	42
	P4=P3*(1.-DPCOM)	43
1	IF (T4.GT.4000.) T4=4000.	44
	IF (T4.GE.1000.) GO TO 2	45
	T4=1000.	46
	IF (MJDE.EQ.1) MAPEDG=1	47
2	DTCO=T4-T3	48
	IF (IDES.NE.1) GO TO 3	49
	DTCODS=DTCO	50
	DTCOCF=DTCODS/DTCO	51
3	DTCO=DTCOCF*DTCO	52
	P3PSIN=P3PSI	53
	CALL SEARCH (-1.,P3PSIN,DTCO,ETAB,DUMMY,PSI(1),NPS,DELT(1,1),ETA(1	54
	1,1),DJMBO(1,1),NPT(1),15,15,IGO)	55
	IF (IGO.EQ.7) CALL ERROR	56
	IF (IDES.NE.1) GO TO 4	57
	ETABCF=ETABDS/ETAB	58
4	ETAB=ETABCF*ETAB	59
	HV=((((((-0.4594317E-19*T4)-.2034116E-15)*T4+.2783643E-11)*T4+.2051	60
	1501E-07)*T4-.2453116E-03)*T4-.9433296E-01)*T4+.1845537E+05	61
	CALL THERMO (P4,HA,T4,XX1,XX2,0,0,0,0)	62
	FAR4=(HA-H3)/(HV*ETAB)	63
	IF (FAR4.LT.0.) FAR4=0.	64
	WFBX=FAR4*WA3	65
	IF (MJDE.NE.2) GO TO 7	66
	ERRW=(WFB-WFBX)/WFB	67
	DIR=SQRT(WFB/WFBX)	68
	CALL AFQUIR (Q(1),T4,ERRW,0.,20.,0.0001,DIR,T4T,IGO)	69
	GO TO (5,8,6),IGO	70
5	T4=T4T	71
	GO TO 1	72
6	CALL ERROR	73
7	WFB=WFBX	74
8	CALL THERMO (P4,H4,T4,S4,XX2,1,FAR4,0)	75
	WG4=WFB+WA3	76
	IF (IDES.EQ.1) WRITE (6,10) WA3CDS,ETABCF,DTCOCF	77
	IF (ISPOOL.EQ.1) GO TO 9	78
	CALL CJHPTB	79
	RETURN	80
9	FAR5=FAR4	81
	T5=T4	82
	WG5=WG4	83
	S5=S4	84
	H5=H4	85
	P5=P4	86
	ETATHP=1.	87
	CNHP=1.	88

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DHHPCF=1. 89
ETHPCF=1. 90
TFHPCF=1. 91
THHCAL=0. 92
DHTCHP=0. 93
DHTC=0. 94
CALL COLPTB 95
RETURN 96
C 97
10 FORMAT (17HOCOMBUSTER DESIGN,7X8H WA3CDS=,E15.8,8H ETABCF=,E15.8,8 98
1H DTCJCF=,E15.8) 99
END 100

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\$IBFTC COHPTB DECK

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SUBROJTIME COHPTB 1
COMMON / ALL/ 2
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP , 3
2IGASMX ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT, 4
3ITRYS ,LODPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLALL ,ERR(6) 5
COMMON /DESIGN/ 6
1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC, 7
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF , 8
3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF , 9
4T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3CDS ,DPCODS ,DTCOCF ,ETABCF, 10
5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS , 11
6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T21DS , 12
7T24DS ,WFDDSD ,DTDUDS ,ETADDS ,WA23DS ,DPDUDS ,DTDUCF ,ETADCF, 13
8T7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF, 14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 , 15
$PS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV 16
COMMON / FRONT/ 17
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 , 18
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 , 19
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 , 20
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB , 21
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 , 22
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF , 23
7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS , 24
8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT , 25
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB , 26
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLOB ,PCBLHP ,PCBLLP 27
COMMON /SIDE/QXQ(48)/BACK/QWQ(72) 28
COMMON /DUMMYS/DUMMY(100) 29
EQUIVALENCE(TFFACT,DUMMY(12)),(CNACT,DUMMY(13)),(DHC ACT,DUMMY(14)) 30
COMMON /TERBHI/DHHISV,TFHISV,CNHISV,ETHISV,DHHPDS 31
COMMON /HTURB/TFFX(15),CNX(15,15),DHTCX(15,15),ETATX(15,15), 32
INTFFS,NPTFF(15) 33
DATA AWORD,WLO,WHI/6HCOHPTB,6H (LO) ,6H (HI) / 34
WORD=AWORD 35
IF (IDES.EQ.0) GO TO 1 36
CNHPCF=CNHPDS*SQRT(T4)/PCNC 37
CNHP=CNHPCF*PCNC/SQRT(T4) 38
CNHPS=CNHP 39
TFFHPS=TFFHP 40
CALL SEARCH (-1.,TFFHP,CNHP,DHTCHP,ETATHP,TFFX(1),NTFFS,CNX(1,1),D 41
1HTCX(1,1),ETATX(1,1),NPTFF(1),15,15,IGO) 42
IF (IGO.EQ.1.OR.IGO.EQ.11.OR.IGO.EQ.21) WRITE (8,8) TFFHPS,WLO 43
IF (IGO.EQ.2.OR.IGO.EQ.12.OR.IGO.EQ.22) WRITE (8,8) TFFHPS,WHI 44

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IF (IGJ.EQ.10.OR.IGJ.EQ.11.OR.IGJ.EQ.12) WRITE (8,9) CNHPS,WLD      45
IF (IGD.EQ.20.OR.IGD.EQ.21.OR.IGD.EQ.22) WRITE (8,9) CNHPS,WHI    46
IF (IGD.NE.7) GO TO 2                                               47
CALL ERROR                                                            48
RETURN                                                                49
2  MAPGD=0                                                            50
   IF (ABS(TFFHPS-TFFHP).LE.0.001*TFFHPS) GO TO 3                  51
   MAPGD=1                                                            52
   IF (ABS(CNHPS-CNHP).GT.0.001*CNHPS) MAPGD=3                     53
   GO TO 4                                                            54
3  IF (ABS(CNHPS-CNHP).GT.0.001*CNHPS) MAPGD=2                     55
4  IF (MAPGD.GT.0) CALL MAPBAC (1,MAPGD,TFFHPS,TFFHP,CNHPS,CNHP,PCNC, 56
1T4,MODE,NOMAP,NUMMAP)
   IF (NJMAP.GT.0) RETURN                                           58
   TFHCAL=WG4*SQRT(T4)/(14.696*P4)                                   59
   BTUEXT=0.706705*HPEXT                                           60
   DHTCC=(BTUEXT+WAC*(H3-H21))/(WG4*T4)                            61
   IF (IDES.EQ.0) GO TO 5                                           62
   TFHPCF=TFHPDS/TFHCAL                                           63
   DHHPCF=DHTCC/DHTCHP                                           64
   ETHPCF=ETHPDS/ETATHP                                           65
   WRITE (6,10) CNHPCF,TFHPCF,ETHPCF,DHHPCF                       66
5  TFHCAL=TFHPCF*TFHCAL                                           67
   DHTCHP=DHHPCF*DHTCHP                                           68
   ETATHP=ETHPCF*ETATHP                                           69
   DHTC=DHTCC*T4                                                  70
   TFFACT=TFHCAL/TFHPCF                                           71
   CNACT=CNHP/CNHPCF                                              72
   DHC ACT=DHTCHP/DHHPCF                                           73
   ERR(1)=(TFHCAL-TFFHP)/TFHCAL                                   74
   ERR(2)=(DHTCC-DHTCHP)/DHTCC                                   75
   CALL THTURB (DHTC,ETATHP,FAR4,H4,S4,P4,T5,H5,S5,P5)           75
   IF (BLHP.LE.0.) GO TO 6                                          77
   FAR5=WFBI/(WA3+BLHP)                                           78
   WG5=WG4+BLHP                                                   79
   H5=(BLHP*H3+WG4*H5)/WG5                                       80
   CALL THERMO (P5,H5,T5,S5,XX2,1,FAR5,1)                          81
   GO TO 7                                                          82
6  FAR5=FAR4                                                        83
   WG5=WG4                                                         84
7  CALL COLPTB                                                       85
   RETURN                                                           85
C                                                                    87
C                                                                    88
8  FORMAT (19H0*****TFFHP OFF MAP,F10.4,2XA6,11H*****$$$$$$)  89
9  FORMAT (19H0***** CNHP OFF MAP,F10.4,2XA6,11H*****$$$$$$)  90
10 FORMAT (20H0.P. TURBINE DESIGN,5X7HCNHPCF=,E15.8,8H TFHPCF=,E15.8  91
1,8H ETHPCF=,E15.8,8H DHHPCF=,E15.8)                             92
   END                                                              93

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\$IBFTC COLPTB DECK

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SUBROUTINE COLPTB                                                  1
COMMON / ALL/                                                    2
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,            3
2IGASM ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,    4
3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLLALL ,ERR(6)         5
COMMON /DESIGN/                                                  6
1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC ,    7

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2ZFDS ,PCNFDS,PRFDS ,ETAFDS,WAFDS ,PRFCF ,ETAFCF,WAFCF ,      8
3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,      9
4T4DS ,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCCCF,ETABCF,     10
5TFHPDS,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS ,     11
6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLPFC,ETLPCF,DHLPCF,T21DS ,    12
7T24DS ,WFDDS ,DTDUDS,ETADDS,WA23DS,DPDUDS,DTDUCF,ETADCF,    13
8T7DS ,WFADS ,DTAFDS,ETAADS,WG6CDS,DPAFDS,DTAFCF,ETAACF,     14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,                          15
$PS55 ,AM55 ,CVDNOZ,CVMNOZ,A8SAV ,A9SAV ,A28SAV,A29SAV        16
COMMON / FRONT/                                               17
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,                               18
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,                          19
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,                               20
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,                     21
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,                     22
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,                 23
7CNHP ,ETATHP,DHTCHP,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,               24
8CNLP ,ETATLP,DHTCLP,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,         25
9AM ,ALP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,                       26
$TFFHP ,TFFLP ,PCBLF ,PCBL ,PCBLDU,PCBLDB,PCBLHP,PCBLP        27
COMMON/SIDE/QXQ(48)/BACK/QWQ(72)                               28
COMMON/DUMMYS/DUMMY(100)                                       29
EQUIVALENCE(TFFACT,DUMMY(15)),(CNACT,DUMMY(16)),(DHCCT,DUMMY(17)) 30
LOGICAL ERROR,FAN                                              31
EQUIVALENCE (FAN,DUMMY(11)),(ISPOOL,DUMMY(6))                 32
COMMON/TERBLO/DHLOSV,TFLOSV,CNLOSV,ETLOSV,DHLPDS              33
COMMON /LTURB/TFFX(15),CNX(15,15),DHTCX(15,15),ETATX(15,15), 34
1NTFFS,NPTFFF(15)                                             35
DATA AWORD,WLD,WHI/6HCOLPTB,6H (LO) ,6H (HI) /                36
WORD=AWORD                                                     37
IF (IDES.EQ.0) GO TO 1                                          38
CNLPFC=CNLPDS*SQRT(T5)/PCNF                                    39
1 CNLP=CNLPFC*PCNF/SQRT(T5)                                     40
CNLPS=CNLP                                                       41
TFFLPS=TFFLP                                                    42
CALL SEARCH (-1.,TFFLP,CNLP,DHTCLP,ETATLP,TFFX(1),NTFFS,CNX(1,1),D 43
1HTCX(1,1),ETATX(1,1),NPTFFF(1),15,15,IGO)                    44
IF (IGD.EQ.1.OR.IGD.EQ.11.OR.IGD.EQ.21) WRITE (8,10) TFFLPS,WLD 45
IF (IGD.EQ.2.OR.IGD.EQ.12.OR.IGD.EQ.22) WRITE (8,10) TFFLPS,WHI 46
IF (IGJ.EQ.10.OR.IGJ.EQ.11.OR.IGJ.EQ.12) WRITE (8,11) CNLPS,WLD 47
IF (IGJ.EQ.20.OR.IGJ.EQ.21.OR.IGJ.EQ.22) WRITE (8,11) CNLPS,WHI 48
IF (IGJ.NE.7) GO TO 2                                           49
CALL ERROR                                                      50
RETURN                                                           51
2 MAPGO=0                                                         52
IF (ABS(TFFLPS-TFFLP).LE.0.001*TFFLPS) GO TO 3                 53
MAPGO=1                                                           54
IF (ABS(CNLPS-CNLP).GT.0.001*CNLPS) MAPGO=3                     55
GO TO 4                                                           56
3 IF (ABS(CNLPS-CNLP).GT.0.001*CNLPS) MAPGO=2                   57
4 IF (MAPGO.GT.0) CALL MAPBAC (2,MAPGO,TFFLPS,TFFLP,CNLPS,CNLP,PCNF, 58
1T4,MODE,NOMAP,NUMMAP)                                          59
IF (NOMAP.GT.0) RETJRN                                          60
TFLCAL=WG5*SQRT(T5)/(14.696*P5)                                  61
DHTCF=WAF*(H21-H2)/(WG5*T5)                                     62
IF (ISPOOL.EQ.2) GO TO 5                                         63
BTUEXT=0.706705*HPEXT                                           64
DHTCF=(BTUEXT+WAF*(H21-H2))/(WG5*T5)                            65
5 CONTINUE                                                       66
IF (IDES.EQ.0) GO TO 6                                           67
TFLPCF=TFLPDS/TFLCAL                                             68
DHLPCF=DHTCF/DHTCLP                                             69

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	ETLPCF=ETLPDS/ETATLP	70
	WRITE (6,12) CNLPCF,TFLPCF,ETLPCF,DHLPCF	71
6	TFLCAL=TFLPCF*TFLCAL	72
	DHTCLP=DHLPCF*DHTCLP	73
	ETATLP=ETLPCF*ETATLP	74
	DHTF=DHTCF*T5	75
	TFFACT=TFLCAL/TFLPCF	76
	CNACT=CNLP/CNLPCF	77
	DHCACT=DHTCLP/DHLPCF	78
	I1=3	79
	I2=4	80
	IF (ISPOOL.NE.1) GO TO 7	81
	I1=1	82
	I2=2	83
7	ERR(I1)=(TFLCAL-TFFLP)/TFLCAL	84
	ERR(I2)=(DHTCF-DHTCLP)/DHTCF	85
	CALL THTURB (DHTF,ETATLP,FAR5,H5,S5,P5,T55,H55,S55,P55)	86
	IF (BLLP.LE.0.) GO TO 8	87
	FAR55=WFB/(WA3+BLHP+BLLP)	88
	WG55=WG5+BLLP	89
	H55=(BLLP*H3+WG5*H55)/WG55	90
	CALL THERMO (P55,H55,T55,S55,XX2,1,FAR55,1)	91
	GO TO 9	92
8	FAR55=FAR5	93
	WG55=WG5	94
9	CALL FRTOSD	95
	RETURN	96
C		97
C		98
10	FORMAT (19H0*****TFFLP OFF MAP,F10.4,2XA6,11H*****\$\$\$\$\$\$)	99
11	FORMAT (19H0***** CNLP OFF MAP,F10.4,2XA6,11H*****\$\$\$\$\$\$)	100
12	FORMAT (20HOL.P. TURBINE DESIGN,5X7HCNLPCF=,E15.8,8H TFLPCF=,E15.8	101
	1,8H ETLPCF=,E15.8,8H DHLPCF=,E15.8)	102
	END	103

\$IBFTC CODUCT DECK

	SUBROJTINE CODUCT	1
	COMMON / ALL/	2
	1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,	3
	2IGASM ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,	4
	3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLALL ,ERR(6)	5
	COMMON /DESIGN/	6
	1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC ,	7
	2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,	8
	3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,	9
	4T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3CDS ,DPCODS ,DTCCOF ,ETABCF ,	10
	5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,	11
	6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T21DS ,	12
	7T24DS ,WFDDS ,DTDUDS ,ETAADS ,WA23DS ,DPDUDS ,DTDUCF ,ETADCF ,	13
	8T7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,	14
	9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,	15
	\$PS55 ,AM55 ,CVDNOZ ,CVMNJZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV	16
	COMMON /FRONT/XX(80)	17
	COMMON / SIDE/	18
	1P1 ,WAF ,WAC ,BLF ,BLDU ,H3 ,DUMS1 ,DUMS2 ,	19
	2T21 ,P21 ,H21 ,S21 ,T23 ,P23 ,H23 ,S23 ,	20
	3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 ,	21

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4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 , 22
5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS,DUMS3 , 23
6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29 24
COMMON/BACK/ZZ(72) 25
COMMON/DUMMYS/DUMMY(100) 26
EQUIVALENCE(A24,DUMMY(4)),(AM23,DUMMY(5)) 27
EQUIVALENCE (AM25,ZZ(48)) 28
DIMENSION Q(9) 29
DATA AWORD1,AWORD2/6HCONDUCT,6HDNOZZL/ 30
WORD=AWORD1 31
Q(2)=0. 32
Q(3)=0. 33
AJ=778.26 34
CAPSF=2116.2170 35
GOGO=0.0 36
G=32.174049 37
WAX=WAF-WAC-BLF 38
WAD=WAX+BLDU 39
P23=P21 40
C*** DRY LJSS 41
H23=(BLDU*H3+WAX*H21)/WAD 42
CALL THERMD (P23,H23,T23,S23,XX2,1,0.0,1) 43
WA23C=WAD*SQRT(T23)/P23 44
IF (IDES.EQ.1) WA23DS=WA23C 45
BYPASS=(WAF-WAC)/WAC 46
DPDUC=DPDUCS*(WA23C/WA23DS) 47
IF (DPDUC.GT.1.) DPDUC=1.0 48
P24=P23*(1.-DPDUC) 49
CALL PROCOM (0.,T23,XX1,XX2,XX3,XX4,PHI23,XX6) 50
IF (IGASM.GT.0) IDBURN=0 51
AM24=AM23 52
TS24=T23*0.875 53
1 DO 2 I=1,15 54
CALL PROCOM (0.,TS24,CS24,AK24,CP24,REX24,PHIS24,HS24) 55
V24=AM24*CS24 56
HSCAL=H23-V24**2/(2.*G*AJ) 57
DELHS=HSCAL-HS24 58
IF (ABS(DELHS).LE.0.001*HSCAL) GO TO 3 59
2 TS24=TS24+DELHS/CP24 60
GO TO 10 61
3 C1=P24*SQRT(G/(T23*AJ))*CAPSF 62
IF (IDES.NE.1) GO TO 4 63
IF (GOGO.GT.0.) GO TO 4 64
ASTOA=((AK24+1.)/2.)*((AK24+1.)/(2.*(AK24-1.)))*AM24*(1.+(((AK24-
11.)/2.)*AM24**2))*((AK24+1.)/(2.*(AK24-1.))) 65
EQWCR=SQRT(G*AK24/REX24/AJ)/(SQRT(518.69)/2116.2)*(2.0/(AK24+1.))* 66
1*((AK24+1.)/2./((AK24-1.))) 67
WA23CC=WA23C/SQRT(518.69) 68
A24=1./ASTOA*WA23CC/EQWCR 69
GOGO=1.0 70
4 WQA=WAD/A24 71
WQAT=C1*SQRT(AK24/REX24)*AM24/(1.+(AK24-1.)*AM24**2/2.)*((AK24+1.
1)/(2.*(AK24-1.))) 72
DIR=WQA/WQAT 73
EW=(WQA-WQAT)/WQA 74
CALL AFQUIR (Q(1),AM24,EW,0.,30.,0.001,DIR,AM24T,IGD) 75
GO TO (5,6,10),IGD 76
5 AM24=AM24T 77
IF (AM24.GT.1.0) AM24=0.5 78
GO TO 1 79
6 PS24=P24/EXP((PHI23-PHIS24)/REX24) 80
IF (IDBURN.GT.0) GO TO 7 81
82
83

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C*** NON-DJCT BURNING
T24=T23
WFD=0.
FAR24=0
GO TO 16
7 IF (IDBURN.EQ.2) T24=T23+2000.
8 IF (T24.GT.4000.) T24=4000.
IF (T24.LT.T23) T24=T23
C*** DUCT BURNING
RHO42=CAPSF*PS24/(AJ*REX24*TS24)
PS42=PS24
V42=V24
Q(2)=0.
Q(3)=0.
C *** IF DESIRED, ENTER CALCULATIONS FOR ETAD HERE
HV=((( (-.4594317E-19*T24)-.2034116E-15)*T24+.2783643E-11)*T24+.2
1051501E-07)*T24-.2453116E-03)*T24-.9433296E-01)*T24+.1845537E+05
CALL THERMO (P24,HA,T24,XX1,XX2,0,0,0,0)
FAR24=(HA-H23)/(HV*ETAD)
IF (FAR24.LT.0.) FAR24=0.
WFDX=FAR24*WAD
IF (IDBURN.NE.2) GO TO 11
ERRW=(WFD-WFDX)/WFD
DIR=SQRT(WFD/WFDX)
CALL AFQUIR (Q(1),T24,ERRW,0.,20.,0.0001,DIR,T24T,IGD)
GO TO (9,12,10),IGD
9 T24=T24T
GO TO 8
10 CALL ERROR
11 WFD=WFDX
12 CONTINUE
C*** MOMENT JM LOSS
WG24=WFD+WAD
CALL PROCOM (FAR24,T24,XX1,XX2,XX3,REX24,PHI24,H24)
RHO24=CAPSF*P24/(AJ*REX24*T24)
V24=WG24/(RHO24*A24)
Q(2)=0.
Q(3)=0.
PS24=PS42-0.01
13 RHO24=WG24/(V24*A24)
HS24=H24-V24**2/(2.*G*AJ)
CALL THERMO (1.0,HS24,TS24,PHIS24,XX2,1,FAR24,1)
IF (TS24.GE.301.) GO TO 14
CALL THERMO (1.0,HS24,400.,PHIS24,XX2,1,FAR24,1)
V24=SQRT(2.*G*AJ*(H24-HS24))
GO TO 13
14 PS24=RHO24*AJ*REX24*TS24/CAPSF
PS24A=PS42+(RHO42*V42**2-RHO24*V24**2)/(G*CAPSF)
DIR=SQRT(ABS(PS24/PS24A))
EP=(PS24-PS24A)/PS24
CALL AFQUIR (Q(1),V24,EP,0.,50.,0.001,DIR,V24T,IGD)
V24=V24T
IF (V24.LT.25.) V24=25.
GO TO (13,15,10),IGD
15 P24=PS24*EXP((PHI24-PHIS24)/REX24)
CALL PROCOM (FAR24,TS24,CS24,XX2,XX3,XX4,XX5,XX6)
AM24=V24*CS24
16 CALL THERMO (P24,H24,T24,S24,XX1,1,FAR24,0)
WG24=WFD+WAD
T25=T24
P25=P24
H25=H24

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	S25=S24	146
	AM25=AM24	147
	IF (IGASM <sub>X</sub> .GT.0) GO TO 20	148
	WORD=A#ORD2	149
	A28SAV=A28	150
	A29SAV=A29	151
	NOZD=0	152
	IDNOZ=0	153
	IF (NOZFLT.EQ.2.OR.NOZFLT.EQ.3) NOZD=1	154
	IF (IDES.EQ.1.OR.IDBURN.GT.0.OR.NOZD.EQ.1) IDNOZ=1	155
	IF (IDBURN.EQ.1.OR.NOZFLT.EQ.2.OR.NOZFLT.EQ.3) IDCD=1	156
	IF (IDBURN.EQ.0.AND.NOZFLT.EQ.0) IDCD=0	157
	IF (IDCD.EQ.1) GO TO 17	158
	CALL CONVRG (T25,H25,P25,S25,FAR24,WG24,P1,IDNOZ,A28,P25R,T28,H28,	159
	1P28,S28,TS28,PS28,V28,AM28,ICON)	160
	GO TO (18,18,18,10),ICON	161
17	CALL CJNDIV (T25,H25,P25,S25,FAR24,WG24,P1,IDNOZ,A28,A29,P25R,T28,	162
	1H28,P28,S28,T29,H29,P29,S29,TS28,TS29,PS28,PS29,V28,V29,AM28,AM29,	163
	2ICON)	164
	IDSHOC=ICON	165
	GO TO (19,19,19,10),ICON	166
18	T29=T28	167
	H29=H28	168
	P29=P28	169
	S29=S28	170
	TS29=TS28	171
	PS29=PS28	172
	V29=V28	173
	AM29=A#28	174
	A29=A28	175
	IDSHOC=ICON+3	176
19	ERR(5)=(P25R-P25)/P25R	177
	IF (IDNOZ.EQ.1) WRITE (6,21) A28,AM28,A29,AM29	178
20	CALL FASTBK	179
	RETURN	180
C		181
C		182
21	FORMAT (19HODUCT NOZZLE DESIGN,5X8H      A28=,E15.8,8H      AM28=,E15.8	183
	1,8H      A29=,E15.8,8H      AM29=,E15.8)	184
	END	185

\$IBFTC COMIX DECK

	SUBROUTINE COMIX	1
	COMMON / ALL/	2
	1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,	3
	2IGASM <sub>X</sub> ,IDBURN,IAFTBN,IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,	4
	3ITRYS ,LOOPER,NOMAP ,NUMMAP,MAPEDG,TOLALL,ERR(6)	5
	COMMON /DESIGN/	6
	1PCNFGJ,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC,	7
	2ZFDS ,PCNFDS,PRFDS ,ETAFDS,WAFDS ,PRFCF ,ETAFCF,WAFCF ,	8
	3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,	9
	4T4DS ,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCOCF,ETABCF,	10
	5TFHPDS,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS ,	11
	6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLPDF,ETLPCF,DHLPDF,T21DS ,	12
	7T24DS ,WFDS ,DTDUDS,ETADDS,WA23DS,DPDUDS,DTDUCF,ETADCF,	13
	8T7DS ,WFADS ,DTAFDS,ETAADS,WG6CDS,JPAFDS,DTAFCF,ETAACF,	14
	9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,	15

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$PS55 ,AM55 ,CVDNQZ,CVMNJZ,A8SAV ,A9SAV ,A28SAV,A29SAV      16
COMMON/FRONT/QZQ(80)/SIDE/QWQ(48)                             17
COMMON / BACK/                                                18
1T55 ,P55 ,H55 ,S55 ,T25 ,P25 ,H25 ,S25 ,                    19
2WFB ,WG55 ,FAR55 ,WFD ,WG24 ,FAR24 ,P1 ,DUMB ,              20
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,                            21
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,                            22
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,                23
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,                    24
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,                      25
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,                26
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC                       27
EQUIVALENCE(ZF,QZQ(68)),(PCNF,QZQ(69))                         28
COMMON/DUMMYS/DUMMY(100)                                       29
COMMON/LOOPPR/KKGD,PRFNEW,PRCNEW                               30
DATA AWORD/6H COMIX/                                           31
DIMENSION QQ(9)                                                 32
WORD=AWORD                                                       33
AJ=778.26                                                         34
CAPSF=2116.2170                                                 35
G=32.174049                                                       36
CALL PROCOM (FAR55,T55,XX1,XX2,XX3,XX4,PHI55,XX5)             37
CALL PROCOM (FAR24,T25,XX1,XX2,XX3,XX4,PHI25,XX5)             38
IF (IDES.EQ.0) GO TO 12                                          39
C *** CALCULATE A55 AND A25 WITH PS25=PS55                     40
IF (PS55.EQ.0.) GO TO 3                                          41
TS55=T55*(PS55/P55)**0.286                                       42
DO 1 I=1,15                                                       43
CALL PROCOM (FAR55,TS55,CS55,AK55,CP55,REX55,PHI55,HS55)     44
PHIS=PHI55-REX55*ALDG(P55/PS55)                                   45
DELPHI=PHIS-PHIS55                                               46
IF (ABS(DELPHI).LE.0.0001*PHIS) GO TO 6                          47
1 TS55=TS55*EXP(4.0*DELPHI)                                       48
2 CALL ERROR                                                       49
RETURN                                                            50
3 TS55=0.875*T55                                                 51
DO 4 I=1,15                                                       52
CALL PROCOM (FAR55,TS55,CS55,AK55,CP55,REX55,PHI55,HS55)     53
V55=AM55*CS55                                                    54
HSCAL=(55-V55)**2/(2.*G*AJ)                                       55
DELHS=HSCAL-HS55                                                 56
IF (ABS(DELHS).LE.0.0005*HSCAL) GO TO 5                          57
4 TS55=TS55+DELHS/CP55                                           58
GO TO 2                                                            59
5 PS55=P55/EXP((PHI55-PHIS55)/REX55)                             60
IF (PS55.GT.P25.AND.IDES.EQ.1.AND.IGASM.X.GT.0) GO TO 47       61
6 IF (H55.GT.HS55) GO TO 7                                        62
WRITE (8,48) P55,PS55,T55,TS55,H55,HS55                         63
CALL ERROR                                                       64
7 V55=SQRT(2.*G*AJ*(H55-HS55))                                    65
RHO=CAPSF*PS55/(AJ*REX55*TS55)                                   66
A55=WG55/(RHO*V55)                                               67
AM55=V55/CS55                                                    68
IF (IGASM.X.GT.0) GO TO 8                                         69
WRITE (6,49) A55,AM55                                           70
IF (IGASM.X.EQ.-1) GO TO 35                                       71
IF (IGASM.X.EQ.0) GO TO 43                                       72
8 PS25=PS55                                                       73
TS25=T25*(PS25/P25)**0.286                                       74
DO 9 I=1,15                                                       75
CALL PROCOM (FAR24,TS25,CS25,AK25,CP25,REX25,PHI25,HS25)     76
PHIS=PHI25-REX25*ALDG(P25/PS25)                                   77

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	GQ(2)=0.	141
	GQ(3)=0.	142
	AM25=0.25	143
	TS25=0.875*T25	144
23	DO 24 I=1,15	145
	CALL PROCOM (FAR24,TS25,CS25,AK25,CP25,REX25,PHIS25,HS25)	146
	V25=AM25*CS25	147
	HSCAL=H25-V25**2/(2.*G*AJ)	148
	DELHS=HSCAL-HS25	149
	IF (ABS(DELHS).LE.0.0005*HSCAL) GO TO 25	150
24	TS25=TS25+DELHS/CP25	151
	GO TO 2	152
25	WQAT=C1*SQRT(AK25/REX25)*AM25/(1.+(AK25-1.)*AM25**2/2.)*((AK25+1.	153
	I)/(2.*(AK25-1.))	154
	AMX=AM25	155
	IGGD=1	156
	GO TO 16	157
26	PS25=P25/EXP((PHI25-PHIS25)/REX25)	158
27	WG6=WG24+WG55	159
	ERR(5)=(PS25-PS55)/PS25	160
	WF6=WFD+WFB	161
	FAR6=WFD/(WG6-WF6)	162
	H6=(WG24*H25+WG55*H55)/WG6	163
	CALL THERMO (1.,H6,T6,PHI6,AMX,1,FAR6,1)	164
	C1=PS55*A55*(1.+AK55*AM55**2)+PS25*A25*(1.+AK25*AM25**2)	165
	TS6=0.833*T6	166
	DO 32 I=1,15	167
	CALL PROCOM (FAR6,TS6,CS6,AK6,CP6,REX6,PHIS6,HS6)	168
	C2=WG5*SQRT(AJ*REX6*T6/(AK6*G))	169
	C3=C2/(CAPSF*C1)	170
	C4=(AK5-1.)/2.-(C3*AK6)**2	171
	C5=1.-2.*AK6*C3**2	172
	C6=C5**2+4.*C4*C3**2	173
	IF (C5) 28,29,30	174
28	CALL ERROR	175
	RETURN	176
29	AM62G=-C5/(2.*C4)	177
	GO TO 31	178
30	AM62G=(SQRT(C6)-C5)/(2.*C4)	179
31	IF (AM62G.LE.0.) GO TO 28	180
	AM6G=SQRT(AM62G)	181
	V6=AM5G*CS6	182
	HSCAL=H6-V6**2/(2.*G*AJ)	183
	DELHS=HSCAL-HS6	184
	IF (ABS(DELHS).LE.0.0005*HSCAL) GO TO 33	185
32	TS6=TS5+DELHS/CP6	186
	GO TO 28	187
33	IF (IGASMX.GT.0) GO TO 34	188
34	A6G=A25+A55	189
	C7=SQRT(1.+(AK6-1.)*AM62G/2.)	190
	PS6=C2/(CAPSF*A6G*AM6G*C7)	191
	P6=PS6*EXP((PHI6-PHIS6)/REX6)	192
	CALL THERMO (P6,H6,T6,S6,XX1,1,FAR6,0)	193
	S6AVE=(WG24*S25+WG55*S55)/WG6	194
	IF (S6.GE.S6AVE) GO TO 35	195
	S6=S6AVE	196
	P6=EXP(AMX*(PHI6-S6)/1.986375)	197
35	IF (IGASMX.EQ.1) GO TO 45	198
	IF (IGASMX.EQ.-1) GO TO 36	199
	IF (IGASMX.EQ.2) GO TO 37	200
36	T6=T55	201
	P6=P55	202
	H6=H55	203

	S6=S55	204
	WG6=WG55	205
	PS6=PS55	206
	FAR6=FAR55	207
	AK6=AK55	208
37	IF (IDES.EQ.0) GO TO 40	209
C***	CALCULATES A6 AS A FUNCTION OF INPUT AM6	210
	TS6=T5/(1.0+(((AK6-1.0)/2.0)*AM6**2))	211
	DO 38 JJ=1,15	212
	AK6P=AK6	213
	CALL PROCOM (FAR6,TS6,CS6,AK6,CP6,REX6,PHIS6,HS6)	214
	V6=AM5*CS6	215
	DELA6=AK6P-AK6	216
	IF (ABS(DELA6).LE.0.0005*AK6) GO TO 39	217
38	TS6=T5/(1.0+(((AK6-1.0)/2.0)*AM6**2))	218
	GO TO 28	219
39	PS6=P5/((1.0+(((AK6-1.0)/2.0)*AM6**2))**((AK6/(AK6-1.0))))	220
	AM6ABD=AM6	221
	RHO=CAPSF*PS6/(AJ*REX6*TS6)	222
	A6=WG5/(RHO*V6)	223
	WRITE (6,54) A6	224
	GO TO 46	225
C	CALCULATES M6=F(A6DESIGN)	226
40	TS6P=T6/(1.0+(((AK6-1.0)/2.0)*AM6ABD**2))	227
	DO 41 I=1,15	228
	CALL PROCOM (FAR6,TS6P,CS6,AK6,CP6,REX6,PHIS6,HS6)	229
	PS6P=PS6*(TS6P/TS6)**(AK6/(AK6-1.0))	230
	RHO6=CAPSF*PS6P/(AJ*REX6*TS6P)	231
	V6=SQRT(2.*G*AJ*(H6-HS6))	232
	IF ((H6-HS6).LT.0.0) GO TO 44	233
	A6P=WG6/(RHO6*V6)	234
	DELA6=A6P-A6	235
	V6=WG5/(RHO6*A6)	236
	AM6=V5/CS6	237
	AM62=AM6**2	238
	IF (ABS(DELA6).LE.00.002*A6) GO TO 42	239
41	TS6P=T5/(1.0+(((AK6-1.0)/2.0)*AM62))	240
	GO TO 28	241
42	TS6=TS6P	242
	PS6=PS6P	243
	GO TO 46	244
43	T6=T55	245
	P6=P55	246
	H6=H55	247
	S6=S55	248
	WG6=WG55	249
	PS6=PS55	250
	V6=V55	251
	AM6=A55	252
	IF (IGASM.X.EQ.0) A6=A55	253
	GO TO 46	254
44	WRITE (6,55) H6,HS6	255
	GO TO 28	256
45	AM62=AM62G	257
	AM6=A6G	258
	A6=A25+A55	259
46	CALL CJAFBN	260
	RETURN	261
47	KKGO=1	262
	OPRDS=PRFDS*PRCDS	263
	PRFNEW=PRFDS*PS55/P25*1.02	264
	PRCNEW=OPRDS/PRFNEW	265

CALL ENGBAL	265
RETURN	267
C	268
C	269
48 FORMAT (22HOSQRT OF H55-HS55 NEG ,6E15.6,6H\$\$\$\$\$)	270
49 FORMAT (20HOTURBINE AREA DESIGN,6X6H A55=,E15.8,8H AM55=,E15.8)	271
50 FORMAT (22HOSQRT OF H25-HS25 NEG ,6E15.6,6H\$\$\$\$\$)	272
51 FORMAT (25HOTURBINE/DUCT AREA DESIGN,7H A55=,E15.8,8H AM55=,E15.8,8H A25=,E15.8,8H AM25=,E15.8)	273
52 FORMAT (12HOCOMIX PCNF=,F7.4,4H AM=,F8.6,5H P55=,F9.5,6H PS55=,F9.5,5H P25=,F9.5,6H PS25=,F9.5,6H\$\$\$\$\$)	275
53 FORMAT (10HOCOMIX ZF=,F8.5,4H AM=,F8.6,5H P55=,F9.5,6H PS55=,F9.5,15H P25=,F9.5,6H PS25=,F9.5,6H\$\$\$\$\$)	277
54 FORMAT (3X,27HAFTERBURNER DESIGN AREA A6 F8.3)	279
55 FORMAT (3X,18HNEG.HS6 FACTOR H6 F9.4,3X,4HHS6 F9.4)	280
END	281

\$IBFTC COAFBN DECK	
SUBROJTINE COAFBN	1
COMMON / ALL/	2
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,	3
2IGASMX, IDBURN, IAFTBN, IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,	4
3ITRYS ,LOOPER,NUMMAP ,NUMMAP,MAPEDG,TOLALL,ERR(6)	5
COMMON /DESIGN/	6
1PCNFGJ,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC,	7
2ZFDS ,PCNFDS,PRFDS ,ETAFDS,WAFDS ,PRFCF ,ETAFCF,WAFCF ,	8
3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF ,	9
4T4DS ,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCODS,DTCCOF,ETABCF,	10
5TFHPDS,CNHPDS,ETHPDS,TFHPCF,CNHPCF,ETHPCF,DHHPCF,T2DS ,	11
6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLPCF,ETLPCF,DHLPCF,T21DS ,	12
7T24DS ,WFDSD ,DTDUDS,ETADDS,WA23DS,DPDUDS,DTDUCF,ETADCF,	13
8T7DS ,WFADS ,DTAFDS,ETAADS,WG6CDS,JPAFDS,DTAFCF,ETAACF,	14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,	15
\$PS55 ,AM55 ,CVDNOZ,CVMNOZ,A8SAV ,A9SAV ,A28SAV,A29SAV	16
COMMON/FROnt/QXQ(80)/SIDE/QYQ(48)	17
COMMON / BACK/	18
1T55 ,P55 ,H55 ,S55 ,T25 ,P25 ,H25 ,S25 ,	19
2WFB ,WG55 ,FAR55 ,WFD ,WG24 ,FAR24 ,P1 ,DUMB ,	20
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,	21
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,	22
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,	23
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,	24
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,	25
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,	26
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC	27
COMMON/DUMMYS/DUMMY(100)	28
EQUIVALENCE (P6DSAV,DUMMY(7)),(AM6DSV,DUMMY(8)),(ETAASV,DUMMY(9)),	29
1(FAR7SV,DUMMY(10))	30
DIMENSION Q(9)	31
DATA AWORD/6HCDAFBN/	32
WORD=AWORD	33
Q(2)=J.	34
Q(3)=J.	35
AJ=779.26	35
CAPSF=2116.2170	37
G=32.174049	38
C*** P6DS AND AM6DS ARE SET FOR GENERALIZATION OF AFTERBURNER	39



```

C*** EFFICIENCY MAP GENERALIZATION                                40
IF (IDES.EQ.1) P6DS=P6*14.696                                     41
IF (IDES.EQ.1) AM6DS=AM6                                         42
WF6=WFB                                                           43
IF (I3ASMXX.GT.0) WF6=WF6+WFD                                    44
WA6=W36-WF6                                                       45
C *** DRY LOSS                                                    46
WG6C=W36*SQRT(T6)/P6                                             47
IF (IDES.EQ.1) WG6CDS=WG6C                                       48
DPAFT=DPAFDS*(WG6C/WG6CDS)                                       49
IF (DPAFT.GT.1.) DPAFT=1.                                         50
P7=P6*(1.-DPAFT)                                                 51
A7=A6                                                             52
FAR6=W6/WA6                                                       53
CALL PROCOM (FAR6,T6,XX1,XX2,XX3,XX4,PHI6,XX6)                   54
WQA=W35/A7                                                         55
C1=P7*SQRT(G/(T6*AJ))*CAPSF                                       56
AM7=AM6                                                             57
TS7=0.875*T6                                                       58
1 DO 2 I=1,15                                                       59
CALL PROCOM (FAR6,TS7,CS7,AK7,CP7,REX7,PHIS7,HS7)               60
V7=AM7*CS7                                                         61
HSCAL=-6-V7**2/(2.*G*AJ)                                          62
DELHS=HSCAL-HS7                                                   63
IF (ABS(DELHS).LE.0.0005*HSCAL) GO TO 3                           64
2 TS7=TS7+DELHS/CP7                                               65
GO TO 14                                                            66
3 WQAT=C1*SQRT(AK7/REX7)*AM7/(1.+(AK7-1.)*AM7**2/2.)*((AK7+1.)/(2.* 67
1(AK7-1.)))                                                       68
DIR=WQA/WQAT                                                       69
EW=(WQA-WQAT)/WQA                                                 70
CALL AFQUIR (Q(1),AM7,EW,0.,40.,.001,DIR,AM7T,IGO)              71
GO TO (4,5,14),IGO                                                72
4 AM7=AM7T                                                         73
IF (AM7.GE.1.0) AM7=0.9                                           74
GO TO 1                                                             75
5 PS7=P7/EXP((PHI6-PHIS7)/REX7)                                    76
IF (IAFTBN.GT.0) GO TO 7                                           77
C *** NON-AFTERBURNING                                           78
6 T7=T6                                                            79
WFA=0.0                                                            80
FAR7=FAR6                                                           81
WG7=W35                                                             82
IF (IDES.EQ.1.AND.T7DS.NE.0.) GO TO 5                             83
GO TO 20                                                            84
C *** AFTERBURNING                                               85
7 IF (IAFTBN.EQ.2) T7=T6+2000.                                     86
IF (IDES.EQ.1) T7=T7DS                                             87
IF (T7.LE.T6) GO TO 6                                             88
RHO65=CAPSF*PS7/(AJ*REX7*TS7)                                     89
PS65=PS7                                                            90
V65=V7                                                              91
Q(2)=0.                                                            92
Q(3)=0.                                                            93
8 IF (T7.GT.4000.) T7=4000.                                        94
HV=((((((-4594317E-19*T7)-.2034116E-15)*T7+.2783643E-11)*T7+.2051 95
1501E-07)*T4-.2453116E-03)*T4-.9433296E-01)*T4+.1845537E+05    96
CALL THERMO (P7,HA,T7,XX1,XX2,1,FAR6,0)                          97
C*** TO ALTER DESIGN ABETAA MAP FROM GENERAL TO SPECIFIC MAP    98
IF (IDES.NE.1) GO TO 9                                             99
FAR7DS=(HA-H6)/(HV*ETAADS)                                         100
CALL ETAAB (0.,0.,0.,0.,ETAADS,ETAASV,P6DS,P6DSAV,AM6DS,AM6DSV,IDE 101
1S,FAR7DS,FAR7SV)                                                102

```

	T7=T6	103
	GO TO 20	104
9	P6GS=25*14.696	105
	FAR7GS=(HA-H6)/(HV*ETAADS)	106
	DO 10 II=1,15	107
	CALL ETAAB (FAR7GS,AM6,P6GS,ETAA,ETAADS,ETAASV,P6DS,P6DSAV,AM6DS,A	108
	IM6DSV,IDES,FAR7DS,FAR7SV)	109
	FAR7=(HA-H6)/(HV*ETAA)	110
	DELFA7=ABS(FAR7-FAR7GS)	111
	IF (DELFA7.LE.0.01*FAR7) GO TO 11	112
10	FAR7GS=FAR7	113
11	CONTINUE	114
	IF (FAR7.GT.0.) GO TO 12	115
	CALL ERROR	116
12	WFAX=FAR7*WG6	117
	IF (IAFTBN.EQ.1) GO TO 15	118
	ERRW=(WFA-WFAX)/WFA	119
	DIR=SQRT(WFA/WFAX)	120
	CALL AFQUIR (Q(1),T7,ERRW,0.,30.,.0005,DIR,T7T,IGD)	121
	GO TO (13,16,14),IGD	122
13	T7=T7T	123
	GO TO 8	124
14	CALL ERROR	125
15	WFA=WFAX	126
16	FAR7=(WF6+WFA)/WA6	127
	WG7=WG6+WFA	128
C ***	MOMENTUM LOSS	129
	CALL PROCOM (FAR7,T7,XX1,XX2,XX3,REX7,PHI7,H7)	130
	RHO7=CAPSF*P7/(AJ*REX7*T7)	131
	V7=WG7/(RHO7*A7)	132
	Q(2)=0.	133
	Q(3)=0.	134
	PS7=PS65-0.01	135
17	RHO7=WG7/(V7*A7)	136
	HS7=H7-V7**2/(2.*G*AJ)	137
	CALL THERMO (1.0,HS7,TS7,PHIS7,XX2,1,FAR7,1)	138
	IF (TS7.GE.301.) GO TO 18	139
	CALL THERMO (1.0,HS7,400.,PHIS7,XX2,1,FAR7,0)	140
	V7=SQRT(2.*G*AJ*(H7-HS7))	141
	GO TO 17	142
18	PS7=RHO7*AJ*REX7*TS7/CAPSF	143
	PS7A=PS65+(RHO65*V65**2-RHO7*V7**2)/(G*CAPSF)	144
	DIR=SQRT(ABS(PS7/PS7A))	145
	EP=(PS7-PS7A)/PS7	146
	CALL AFQUIR (Q(1),V7,EP,0.,50.,.001,DIR,V7T,IGD)	147
	V7=V7T	148
	IF (V7.LT.100.) V7=100.	149
	GO TO (17,19,14),IGD	150
19	P7=PS7*EXP((PHI7-PHIS7)/REX7)	151
	CALL PROCOM (FAR7,TS7,CS7,XX2,XX3,XX4,XX5,XX6)	152
	AM7=V7/CS7	153
20	CALL THERMO (P7,H7,T7,S7,XX2,1,FAR7,0)	154
	IF (IDES.EQ.1) WRITE (6,21) WG6CDS	155
	CALL COMNOZ	156
	RETURN	157
C		158
C		159
C		160
21	FORMAT (19H0AFTERBURNER DESIGN,5X8H WG6CDS=,E15.8)	161
	END	162

```

$IBFTC FRTOSD DECK
SUBROJTINE FRTOSD
COMMON/ALL/XX(28)/DESIGN/YY(80)
COMMON / FRONT/
1T1      ,P1      ,H1      ,S1      ,T2      ,P2      ,H2      ,S2      ,
2T21    ,P21    ,H21    ,S21    ,T3      ,P3      ,H3      ,S3      ,
3T4     ,P4      ,H4      ,S4      ,T5      ,P5      ,H5      ,S5      ,
4T55    ,P55    ,H55    ,S55    ,BLF     ,BLC     ,BLDU    ,BLOB    ,
5CNF    ,PRF     ,ETAF    ,W AFC   ,WAF     ,WA3     ,WG4     ,FAR4    ,
6CNC    ,PRC     ,ETAC    ,WACC   ,WAC     ,ETAB    ,DPCOM   ,DUMF    ,
7CNHP   ,ETATHP ,DHTCHP ,DHTC   ,BLHP    ,WG5     ,FAR5    ,CS      ,
8CNLP   ,ETATLP ,DHTCLP ,DHTF   ,BLLP    ,WG55    ,FAR55   ,HPEXT   ,
9AM     ,ALTP    ,ETAR    ,ZF     ,PCNF    ,ZC      ,PCNC    ,WFB     ,
$TFFHP  ,TFFLP   ,PCBLF   ,PCBLC  ,PCBLDU ,PCBLOB ,PCBLHP ,PCBLLP ,
COMMON / SIDE/
1XP1    ,XWAF    ,XWAC    ,XBLF   ,XBLDU   ,XH3     ,DUMS1   ,DUMS2   ,
2XT21   ,XP21    ,XH21    ,XS21   ,T23     ,P23     ,H23     ,S23     ,
3T24    ,P24     ,H24     ,S24     ,T25     ,P25     ,H25     ,S25     ,
4T28    ,P28     ,H28     ,S28     ,T29     ,P29     ,H29     ,S29     ,
5WAD    ,WFD     ,WG24    ,FAR24  ,ETAD    ,DPDUC   ,BYPASS  ,DUMS3   ,
6TS28   ,PS28    ,V28     ,AM28   ,TS29    ,PS29    ,V29     ,AM29    ,
COMMON/BACK/ZZ(72)
COMMON/DUMMYS/DUMMY(100)
LOGICAL ERRER,FAN
EQUIVALENCE (FAN,DUMMY(11)),(ISPOOL,DUMMY(6))
XP1=P1
XWAF=WAF
XWAC=#AC
XBLF=BLF
XBLDU=BLDU
XH3=H3
XT21=T21
XP21=P21
XH21=H21
XS21=S21
IF (FAN) CALL CODUCT
IF (FAN) RETURN
CALL FASTBK
RETURN
END

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$IBFTC FASTBC DECK
SUBROJTINE FASTBK
COMMON/ALL/XX(28)/DESIGN/YY(80)
COMMON / FRONT/
1T1      ,P1      ,H1      ,S1      ,T2      ,P2      ,H2      ,S2      ,
2T21    ,P21    ,H21    ,S21    ,T3      ,P3      ,H3      ,S3      ,
3T4     ,P4      ,H4      ,S4      ,T5      ,P5      ,H5      ,S5      ,
4T55    ,P55    ,H55    ,S55    ,BLF     ,BLC     ,BLDU    ,BLOB    ,
5CNF    ,PRF     ,ETAF    ,W AFC   ,WAF     ,WA3     ,WG4     ,FAR4    ,
6CNC    ,PRC     ,ETAC    ,WACC   ,WAC     ,ETAB    ,DPCOM   ,DUMF    ,
7CNHP   ,ETATHP ,DHTCHP ,DHTC   ,BLHP    ,WG5     ,FAR5    ,CS      ,
8CNLP   ,ETATLP ,DHTCLP ,DHTF   ,BLLP    ,WG55    ,FAR55   ,HPEXT   ,
9AM     ,ALTP    ,ETAR    ,ZF     ,PCNF    ,ZC      ,PCNC    ,WFB     ,
$TFFHP  ,TFFLP   ,PCBLF   ,PCBLC  ,PCBLDU ,PCBLOB ,PCBLHP ,PCBLLP ,
COMMON / SIDE/
1XP1    ,XWAF    ,XWAC    ,XBLF   ,XBLDU   ,XH3     ,DUMS1   ,DUMS2   ,
2XT21   ,XP21    ,XH21    ,XS21   ,T23     ,P23     ,H23     ,S23     ,

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3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 , 17
4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 , 18
5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS ,DUMS3 , 19
6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29 , 20
COMMON / BACK/ 21
1XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 , 22
2XWFB ,XWG55 ,XFAR55 ,XWFD ,XWG24 ,XFAR24 ,XXP1 ,DUMB , 23
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 , 24
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 , 25
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 , 25
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 , 27
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 , 28
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM , 29
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC 30
COMMON/DUMMYS/DUMMY(100) 31
LOGICAL ERRER,FAN 32
EQUIVALENCE (FAN,DUMMY(11)),(ISPOOL,DUMMY(6)) 33
XT55=T55 34
XP55=P55 35
XH55=H55 36
XS55=S55 37
IF (FAN) GO TO 1 38
T25=T21 39
P25=P21 40
H25=H21 41
S25=S21 42
WG24=WAF-BLF 43
1 XT25=T25 44
XP25=P25 45
XH25=H25 46
XS25=S25 47
XWFB=WFB 48
XWG55=WG55 49
XFAR55=FAR55 50
XWFD=WFD 51
XWG24=WG24 52
XFAR24=FAR24 53
XXP1=P1 54
CALL COMIX 55
RETURN 56
END 57

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$IBFTC COMNOZ DECK
SUBROJTINE COMNOZ 1
COMMON / ALL/ 2
IWORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP , 3
ZIGASM ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT , 4
3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLALL ,ERR(6) 5
COMMON /DESIGN/ 6
1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC , 7
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF , 8
3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF , 9
4T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3CDS ,DPCJDS ,DTCCCF ,ETABCF , 10
5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS , 11
6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T21DS , 12
7T24DS ,WFDDS ,DTDUDS ,ETADDS ,WA23DS ,DPDUDS ,DTDUCF ,ETADCF , 13
8T7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF , 14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 , 15

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$PS55 ,AM55 ,CVDNOZ,CVMNOZ,A8SAV ,A9SAV ,A28SAV,A29SAV          15
COMMON/FRONT/QXQ(80)/SIDE/QYQ(48)                                17
EQUIVALENCE(AM,QXQ(65))                                         18
COMMON / BACK/                                                  19
1T55 ,P55 ,H55 ,S55 ,T25 ,P25 ,H25 ,S25 ,                      20
2WFB ,WG55 ,FAR55 ,WFD ,WG24 ,FAR24 ,P1 ,DUMB ,                21
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,                              22
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,                              23
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,                 24
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,                       25
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,                       26
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,                  27
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC                          28
COMMON/DUMMYS/DUMMY(100)                                        29
LOGICAL ERRER,FAN                                              30
EQUIVALENCE (FAN,DUMMY(11)),(ISPOOL,DUMMY(6))                 31
DATA A,ORD/6HMNOZZL/                                           32
WORD=4,WORD                                                    33
A8SAV=A8                                                        34
A9SAV=A9                                                        35
NOZM=0                                                          36
IMNOZ=0                                                         37
IF (NOZFLT.EQ.1.OR.NOZFLT.EQ.3) NOZM=1                          38
IF (IDES.EQ.1.OR.IAFTBN.GT.0.OR.NOZM.EQ.1) IMNOZ=1            39
IF (IAFTBN.EQ.0.AND.NOZFLT.EQ.0) IMCD=0                         40
IF (IAFTBN.EQ.1.OR.NOZFLT.EQ.1.OR.NOZFLT.EQ.3) IMCD=1         41
IF (IMCD.EQ.1) GO TO 1                                         42
CALL CONVRG (T7,H7,P7,S7,FAR7,WG7,P1,IMNOZ,A8,P7R,T8,H8,P8,S8, 43
1PS8,V9,AM8,ICON)                                             44
GO TO (3,3,3,2),ICON                                          45
1 CALL CONDIV (T7,H7,P7,S7,FAR7,WG7,P1,IMNOZ,A8,A9,P7R,T8,H8,P8,S8,T 46
19,H9,P9,S9,TS8,TS9,PS8,PS9,V8,V9,AM8,AM9,ICON)              47
IMSHOC=ICON                                                    48
GO TO (4,4,4,2),ICON                                          49
2 CALL ERROR                                                    50
3 T9=T8                                                         51
H9=H8                                                           52
P9=P8                                                           53
S9=S8                                                           54
TS9=TS8                                                         55
PS9=PS8                                                         56
V9=V8                                                           57
AM9=AM8                                                         58
A9=A8                                                           59
IMSHOC=ICON+3                                                  60
4 ERR(6)=(P7R-P7)/P7R                                          61
IF (ISPOOL.EQ.1) ERR(3)=ERR(6)                                  62
IF (IMNOZ.EQ.1) WRITE (6,5) A8,AM8,A9,AM9                      63
RETURN                                                         64
C                                                                 65
C                                                                 66
5 FORMAT (14HONNOZZLE DESIGN,10X8H A8=,E15.8,8H AM8=,E15.8,8H 67
1 A9=,E15.8,8H AM9=,E15.8)                                     68
END                                                             69

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5IBFTC ERROR DECK

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SUBROUTINE ERROR
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASMX ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHJC ,IMSHOC ,NOZFLT ,
3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLALL ,ERR(6)
COMMON /DESIGN/
1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
4T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3CDS ,DPCODS ,DTCOCF ,ETABCF ,
5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2OS ,
6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T2IDS ,
7T24DS ,WFDDSD ,DTDUDS ,ETADDS ,WA23DS ,DPDUDS ,DTDUCF ,ETADCF ,
8T7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
$PS55 ,AM55 ,CVDNDZ ,CVMNDZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON / FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,
7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,
8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU ,PCBLDB ,PCBLHP ,PCBLLP
COMMON / SIDE/
1XP1 ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3 ,DUMS1 ,DUMS2 ,
2XT21 ,XP21 ,XH21 ,XS21 ,T23 ,P23 ,H23 ,S23 ,
3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 ,
4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 ,
5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS ,DUMS3 ,
6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29
COMMON / BACK/
1XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,
2XWFB ,XWG55 ,XFAR55 ,XWFD ,XWG24 ,XFAR24 ,XXP1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC
COMMON/DUMMYS/DUMMY(100)
LOGICAL ERROR
COMMON/ERER/ERRER
DIMENSION TRASH1(80),TRASH2(80),TRASH3(48),TRASH4(72)
EQUIVALENCE (TRASH1,PCNFGU),(TRASH2,T1),(TRASH3,XP1),(TRASH4,XT55)
DATA A#WORD/6HCOMMON/
ERRER=.TRUE.
WRITE (6,2) WORD
WORD=A#WORD
WRITE (6,3) WORD,ZF,PCNF,ZC,PCNC,T4,MODE
WRITE (6,4)
WRITE (6,5) (TRASH1(I),I=1,80)
WRITE (6,6)
WRITE (6,5) (TRASH2(I),I=1,80)
WRITE (6,4)
WRITE (6,5) (TRASH3(I),I=1,48)
WRITE (6,4)

```

	WRITE (6,5) (TRASH4(I),I=1,72)	62
	WRITE (6,4)	63
	WRITE (6,7) LOOPER	64
	IF (IDJMP.EQ,0) GO TO 1	65
	WRITE (6,6)	66
	CALL SYG (2)	67
1	CALL ENGBAL	68
	RETURN	69
C		70
C		71
2	FORMAT (28HOAN ERROR HAS BEEN FOUND IN ,A6)	72
3	FORMAT (1H0,A6,9X,5E15.6,I4)	73
4	FORMAT (2H0 )	74
5	FORMAT (1H0,8E15.6)	75
6	FORMAT (1H1)	75
7	FORMAT (25H0FAILED TO CONVERGE AFTER,I4,6H LOOPS)	77
	END	78

	\$IBFTC SYGS DECK	
	SUBROUTINE SYG (ICON)	1
	DIMENSION WORD(132)	2
	DATA ONEDOL/6H\$ /	3
	GO TO (1,2),ICON	4
1	END FILE 8	5
	REWIND 8	6
	RETURN	7
C	TERMINATE THE FILE	8
2	WRITE (8,10)	9
	END FILE 8	10
	REWIND 8	11
C	READ RECORD	12
3	READ (8,11) (WORD(I),I=1,132)	13
C	CHECK FOR 12 LEADING DOLLAR SIGNS	14
	DO 4 I=1,12	15
	IF (WORD(I)-ONEDOL) 5,4,5	16
4	CONTINJE	17
	RETURN	18
C	CHECK FOR 6 TRAILING DOLLAR SIGNS	19
5	DO 8 I=1,132	20
	I=I	21
	IF (WORD(I)-ONEDOL) 8,6,8	22
6	K=I+5	23
	DO 7 J=I,K	24
	IF (WORD(J)-ONEDOL) 8,7,8	25
7	CONTINUE	26
	GO TO 9	27
8	CONTINJE	28
	WRITE (6,12)	29
	RETURN	30
C	PRINT LINE	31
9	I=I-1	32
	WRITE (6,11) (WORD(M),M=1,I)	33
	GO TO 3	34
C		35
C		36
10	FORMAT (12H\$\$\$\$\$\$\$\$\$\$\$\$)	37
11	FORMAT (132A1)	38
12	FORMAT (1H0,12HERROR IN SYG)	39
	END	40

```

$IBFTC PERFOR DECK
SUBROJTINE PERF 1
COMMON / ALL/ 2
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP , 3
2IGASMX ,IDBURN,IAFTBN, IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT, 4
3ITRYS ,LOOPER,NOMAP ,NUMMAP,MAPEDG,TOLALL,ERR(6) 5
COMMON /DESIGN/ 6
1PCNFGJ,PCNCGU,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC, 7
2ZFDS ,PCNFDS,PRFDS ,ETAFDS,WAFDS ,PRFCF ,ETAFCF,WAFCF , 8
3ZCDS ,PCNCDS,PRCDS ,ETACDS,WACDS ,PRCCF ,ETACCF,WACCF , 9
4T4DS ,WFBDS ,DTCODS,ETABDS,WA3CDS,DPCCDS,DTCOCF,ETABCF, 10
5TFHPDS,CNHPS,ETHPS,TFHPCF,CNHPCF,ETHPCF,DHHPHF,T2DS , 11
6TFLPDS,CNLPDS,ETLPDS,TFLPCF,CNLPFC,ETLPFC,DHLPFC,T21DS , 12
7T24DS ,WFDD ,DTDUDS,ETADDS,WA23DS,DPDUDS,DTDUCF,ETADCF, 13
8T7DS ,WFADS ,DTAFDS,ETAADS,WG6CDS,DPAFDS,DTAFCF,ETAACF, 14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 , 15
$PS55 ,AM55 ,CVDNOZ,CVMNOZ,A8SAV ,A9SAV ,A28SAV,A29SAV 16
COMMON / FRONT/ 17
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 , 18
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 , 19
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 , 20
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB , 21
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 , 22
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF , 23
7CNHP ,ETATHP,DHTCHP,DHTC ,BLHP ,WG5 ,FAR5 ,CS , 24
8CNLP ,ETATLP,DHTCLP,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT , 25
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB , 26
$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU,PCBLOB,PCBLHP,PCBLLP 27
COMMON / SIDE/ 28
1XP1 ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3 ,DUMS1 ,DUMS2 , 29
2XT21 ,XP21 ,XH21 ,XS21 ,T23 ,P23 ,H23 ,S23 , 30
3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 , 31
4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 , 32
5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS,DUMS3 , 33
6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29 34
COMMON / BACK/ 35
1XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 , 36
2XWFB ,XWG55 ,XFAR55,XWFD ,XWG24 ,XFAR24,XXP1 ,DUMB , 37
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 , 38
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 , 39
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 , 40
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 , 41
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 , 42
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM , 43
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC 44
COMMON/DUMMYS/DUMMY(100) 45
LOGICAL ERRER,FAN 46
EQUIVALENCE (FAN,DUMMY(11)),(ISPOOL,DUMMY(6)) 47
DATA AWORD/6H PERF/ 48
WORD=AWORD 49
G=32.174049 50
CAPSF=2116.2170 51
WFT=WFB+WFD+WFA 52
WAT=WAF-BLOB 53
WGT=WAT+WFT 54
FART=JFT/WAT 55
VA=AM*CS 56
FRD=VA*WAF/G 57
C TO REMOVE CD NOZ COEFF, CHANGE CALL FRASHO,CUMNOZ EQ. 58
C ALSO SUB CODUCT REMOVE IF (IDBURN.EQ.1) IDCD=1 59
C ALSO SJB COMNOZ REMOVE IF (IAFTBN.EQ.1) IMCD=1 60
ARATIO=A9/A8 61

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PTRATJ=P8/P1	62
PRATJ=PS9/P1	63
CALL PROCOM (FAR7,TS9,CS9,AK9,CP9,REX9,PHIS9,HS9)	64
CALL FRASHO (PTRATO,ARATIO,CUMNOZ)	65
CVMNOZ=CUMNOZ	66
VJM=CVMNOZ*V9	67
FGMM=VJM*WG7/G	68
FGPM=CAPSF*(PS9-P1)*A9	69
IF (IGASMX.GT.0.OR..NOT.FAN) GO TO 1	70
PRATJ=PS29/P1	71
ARATIO=A29/A28	72
PTRATJ=P28/P1	73
CALL PROCOM (FAR24,TS29,CS9,AK29,CP9,REX9,PHIS9,HS9)	74
CALL FRASHO (PTRATO,ARATIO,CUDNOZ)	75
CVDNOZ=CUDNOZ	76
VJD=CVDNOZ*V29	77
FGMD=VJD*WG24/G	78
FGPD=CAPSF*(PS29-P1)*A29	79
1 FGM=FGMM+FGMD	80
FGP=F3PM+FGPD	81
FG=FGM+FGP	82
FN=FG-FRD	83
SFC=3500.*WFT/FN	84
FG=DELFG*FG	85
FN=DELFN*FN	86
SFC=DELSFC*SFC	87
CALL JJTPUT	88
CALL ERROR	89
RETURN	90
END	91

\$IBFTC OUTPJT DECK

SUBROJTINE OUTPUT	1
COMMON / ALL/	2
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,	3
2IGASMX ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,	4
3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLALL ,ARR(6)	5
COMMON /DESIGN/	6
1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC ,	7
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,	8
3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,	9
4T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3CDS ,DPCODS ,DTCOCF ,ETABCF ,	10
5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,	11
6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T21DS ,	12
7T24DS ,WFDDSD ,DTDUDS ,ETAADS ,WA23DS ,DPDUDS ,DTDUCF ,ETAACF ,	13
8T7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,	14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,	15
\$PS55 ,AM55 ,CVDNOZ ,CVMNOZ ,ABSAV ,A9SAV ,A28SAV ,A29SAV	16
COMMON / FRONT/	17
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,	18
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,	19
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,	20
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,	21
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,	22
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,	23
7CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,	24
8CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,	25
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,	26

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$TFFHP ,TFFLP ,PCBLF ,PCBLC ,PCBLDU,PCBLOB,PCBLHP,PCBLLP      27
COMMON / SIDE/                                                  28
1XP1 ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3 ,DUMS1 ,DUMS2 ,           29
2XT21 ,XP21 ,XH21 ,XS21 ,T23 ,P23 ,H23 ,S23 ,               30
3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 ,                   31
4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 ,                   32
5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS,DUMS3 ,          33
6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29                34
COMMON / BACK/                                                  35
1XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,           36
2XWFB ,XWG55 ,XFAR55,XWFD ,XWG24 ,XFAR24,XXP1 ,DUMB ,         37
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,                           38
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,                           39
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,               40
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,                    41
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,                     42
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,                43
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC                       44
COMMON / DUMMYS / DUMMY(100)                                    45
EQUIVALENCE (T4PBL,DUMMY(2)),(T41,DUMMY(3))                   46
DIMENSION W(5,4),ANS1(80),ANS2(80),ANS3(48),ANS4(72)          47
EQUIVALENCE (ANS1,PCNFGU),(ANS2,T1),(ANS3,XP1),(ANS4,XT55)    48
DATA AWORD1,AWORD2/6HOUTPUT,6HCOMMON/                          49
DATA (W(1,I),I=1,4)/6HSUBSON,6HIC C-D,6H NDZZL,6HE /        50
DATA (W(2,I),I=1,4)/6HSHOCK ,6HINSIDE,6H C-D N,6HOZZLE /    51
DATA (W(3,I),I=1,4)/6HSHOCK ,6HOUTSID,6HE C-D ,6HNOZZLE/    52
DATA (W(4,I),I=1,4)/6HSUBSON,6HIC CON,6HVERG. ,6HNOZZLE/    53
DATA (W(5,I),I=1,4)/6HSONIC ,6HCONVER,6HGENT N,6HOZZLE /    54
WORD=AWORD1                                                     55
IF (IDBURN.GT.0) GO TO 2                                        56
IF (IAFTBN.GT.0) GO TO 1                                        57
WRITE (6,7) WORD,AM,ALTP,T4,ETAR                               58
GO TO 3                                                         59
1 WRITE (6,8) WORD,AM,ALTP,T4,T7,ETAR                           60
GO TO 3                                                         61
2 WRITE (6,9) WORD,AM,ALTP,T4,T24,ETAR                          62
3 CALL CONDOUT (2)                                              63
WRITE (6,10) (W(IMSHOC,I),I=1,4),FG,FN,SFC                    64
IF (IGASM.X.GT.0) GO TO 4                                       65
WRITE (6,11) (W(IDSHOC,I),I=1,4)                                66
4 WRITE (6,12) LOOPER                                           67
IF (IDES.NE.1) GO TO 5                                          68
WORD=AWORD2                                                     69
WRITE (6,13) WORD,ZF,PCNF,ZI,PCNI,ZC,PCNC,T4,MODE            70
WRITE (6,14)                                                    71
WRITE (6,15) (ANS1(I),I=1,80)                                    72
WRITE (6,14)                                                    73
WRITE (6,15) (ANS2(I),I=1,80)                                    74
WRITE (6,14)                                                    75
WRITE (6,15) (ANS3(I),I=1,48)                                    76
WRITE (6,14)                                                    77
WRITE (6,15) (ANS4(I),I=1,72)                                    78
WRITE (6,16)                                                    79
IF (IDES.EQ.1) GO TO 6                                          80
5 CONTINUE                                                       81
A8=ABSAV                                                         82
A9=A9SAV                                                         83
A28=A28SAV                                                       84
A29=A29SAV                                                       85
IF (IDUMP.NE.2) GO TO 6                                          86
WRITE (6,16)                                                    87
CALL SYG (2)                                                    88

```

6	CALL ENGBAL	89
	RETURN	90
C		91
C		92
C		93
7	FORMAT (1HB,A6,14X7H AM=,F7.3,6X7H ALTP=,F7.0,6X7H T4=,F8.2 1,25X7H ETAR=,F7.4)	94
8	FORMAT (1HB,A6,14X7H AM=,F7.3,6X7H ALTP=,F7.0,6X7H T4=,F8.2 1,5X7H T7=,F8.2,5X7H ETAR=,F7.4)	95
9	FORMAT (1HB,A6,14X7H AM=,F7.3,6X7H ALTP=,F7.0,6X7H T4=,F8.2 1,5X7H T24=,F8.2,5X7H ETAR=,F7.4)	96
10	FORMAT (6HOMAIN ,4A5,9X3HFG=,F9.2,18X3HFN=,F9.2,18X4HSFC=,F8.5)	97
11	FORMAT (6H DUCT ,4A5)	98
12	FORMAT (16HOCONVERGED AFTER,I4,6H LOOPS,/,1H1)	99
13	FORMAT (1H ,A6,9X,7E15.6,I4)	100
14	FORMAT (1H )	101
15	FORMAT (1H ,8E15.6)	102
16	FORMAT (1H1)	103
	END	104
		105
		106
		107

\$IBFTC CONOJT DECK

SUBROJTINE CONOUT (ICON)	1
COMMON / ALL/	2
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,	3
2IGASMX ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,	4
3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLALL ,ERR(6)	5
COMMON /DESIGN/	6
1PCNFGJ ,PCNCGU ,T4GU ,DUMD1 ,DUMD2 ,DELFG ,DELFN ,DELSFC ,	7
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,	8
3ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,	9
4T4DS ,WFBDS ,DTCDS ,ETABDS ,WA3CDS ,DPCDS ,DTCOCF ,ETABCF ,	10
5TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,	11
6TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T21DS ,	12
7T24DS ,WFDD ,DTDUDS ,ETADDS ,WA23DS ,DPDUDS ,DTDUCF ,ETADCF ,	13
8T7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,	14
9A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,	15
\$PS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV	16
COMMON / FRONT/	17
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,	18
2T21 ,P21 ,H21 ,S21 ,T3 ,P3 ,H3 ,S3 ,	19
3T4 ,P4 ,H4 ,S4 ,T5 ,P5 ,H5 ,S5 ,	20
4T55 ,P55 ,H55 ,S55 ,BLF ,BLC ,BLDU ,BLOB ,	21
5CNF ,PRF ,ETAF ,WAF ,WAF ,WA3 ,WG4 ,FAR4 ,	22
6CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,DUMF ,	23
7CNHPM ,ETATHP ,DHTCHM ,DHTC ,BLHP ,WG5 ,FAR5 ,CS ,	24
8CNLPM ,ETATLP ,DHTCLM ,DHTF ,BLLP ,WG55 ,FAR55 ,HPEXT ,	25
9AM ,ALTP ,ETAR ,ZF ,PCNF ,ZC ,PCNC ,WFB ,	26
\$TFFHPM ,TFFLPM ,PCBLF ,PCBLC ,PCBLDU ,PCBLOB ,PCBLHP ,PCBLLP	27
COMMON / SIDE/	28
1XP1 ,XWAF ,XWAC ,XBLF ,XBLDU ,XH3 ,DUMS1 ,DUMS2 ,	29
2XT21 ,XP21 ,XH21 ,XS21 ,T23 ,P23 ,H23 ,S23 ,	30
3T24 ,P24 ,H24 ,S24 ,T25 ,P25 ,H25 ,S25 ,	31
4T28 ,P28 ,H28 ,S28 ,T29 ,P29 ,H29 ,S29 ,	32
5WAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPDUC ,BYPASS ,DUMS3 ,	33
6TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29	34
COMMON / BACK/	35
1XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,	36

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2XWFB ,XWG55 ,XFAR55,XWFD ,XWG24 ,XFAR24,XXP1 ,DUMB , 37
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 , 38
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 , 39
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 , 40
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 , 41
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 , 42
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM , 43
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC 44
COMMON/DUMMYS/DUMMY(100) 45
EQUIVALENCE (TFFHP,DUMMY(12)),(CNHP,DUMMY(13)),(DHTCHP,DUMMY(14)) 46
EQUIVALENCE (TFFLP,DUMMY(15)),(CNLP,DUMMY(16)),(DHTCLP,DUMMY(17)) 47
DIMENSION PARAM(380),WORDY(380),IDUT(103),AOUT(6),WOUT(6) 48
EQUIVALENCE (PARAM,PCNFGU) 49
DATA (WORDY(I),I=1,98)/ 50
16HPCNFGU,6HPCNCGU,6HT4GU ,6HDUMD1 ,6HDUMD2 ,6HDELFG ,6HDELFM , 51
26HDELSFC,6HZFDS ,6HPCNFDS,6HPRFDS ,6HETAFDS,6HWAFDS ,6HPRFCF , 52
36HETAFCF,6HWAFCF ,6HZCDS ,6HPCNCDS,6HPRCDS ,6HETACDS,6HWACDS , 53
46HPRCCF ,6HETACCF,6HWACCF ,6HT4DS ,6HWFBD ,6HDTCCDS,6HETABDS, 54
56HWA3CDS,6HDPCCDS,6HDTCCCF,6HETABCF,6HTFHPDS,6HCNHPDS,6HETHPDS, 55
66HTFHPCF,6HCNHPCF,6HETHPCF,6HDHPCF,6HT2DS ,6HTFLPDS,6HCNLPDS, 56
76HETLPDS,6HTFLPCF,6HCNLPDF,6HETLPDF,6HDHLPDF,6HT21DS ,6HT24DS , 57
86HWFDD ,6HDTDUDS,6HETADDS,6HWA23DS,6HDPDUDS,6HDTDUCF,6HETADCF, 58
96HT7DS ,6HWFADS ,6HDTAFDS,6HETAADS,6HWG6CDS,6HDPAFDS,6HDTAFCF, 59
$6HETAACF,6HA55 ,6HA25 ,6HA6 ,6HA7 ,6HA8 ,6HA9 , 60
$6HA28 ,6HA29 ,6HPS55 ,6HAM55 ,6HCVDNOZ,6HCVMOZ,6HABSAV , 61
$6HA9SAV ,6HA28SAV,6HA29SAV,6HT1 ,6HP1 ,6H11 ,6HS1 , 62
$6HT2 ,6HP2 ,6HH2 ,6HS2 ,6HT21 ,6HP21 ,6HH21 , 63
$6HS21 ,6HT3 ,6HP3 ,6HH3 ,6HS3 ,6HT4 ,6HP4 / 64
DATA (WORDY(I),I=99,189)/ 65
16HH4 ,6HS4 ,6HT5 ,6HP5 ,6HH5 ,6HS5 ,6HT55 , 66
26HP55 ,6HH55 ,6HS55 ,6HBLF ,6HBLC ,6HBLDU ,6HBLOB , 67
36HCNF ,6HPRF ,6HETAF ,6HWAF ,6HWA3 ,6HWG4 , 68
46HFAR4 ,6HCNC ,6HPRC ,6HETAC ,6HWACC ,6HWAC ,6HETAB , 69
56HDPCCM ,6HDUMF ,6HCNHPM ,6HETATHP,6HDHTCHM,6HDHTC ,6HBLHP , 70
66HWG5 ,6HFAR5 ,6HCS ,6HCNLP ,6HETATLP,6HDHTCLM,6HDHTF , 71
76HBLLP ,6HWG55 ,6HFAR55 ,6HHPEXT ,6HAM ,6HALTP ,6HETAR , 72
86HZF ,6HPCNF ,6HZC ,6HPCNC ,6HWF ,6HTFFHPM,6HTFFLPM, 73
96HPCBLF ,6HPCBLC ,6HPCBLDU,6HPCBLOB,6HPCBLHP,6HPCBLLP,6HXP1 , 74
$6HXWAF ,6HXWAC ,6HXBLF ,6HXBLDU ,6HXH3 ,6HDUMS1 ,6HDUMS2 , 75
$6HXT21 ,6HXP21 ,6HXH21 ,6HXS21 ,6HT23 ,6HP23 ,6HH23 , 76
$6HS23 ,6HT24 ,6HP24 ,6HH24 ,6HS24 ,6HT25 ,6HP25 , 77
$6HH25 ,6HS25 ,6HT28 ,6HP28 ,6HH28 ,6HS28 ,6HT29 / 78
DATA (WORDY(I),I=190,280)/ 79
16HP29 ,6HH29 ,6HS29 ,6HWAD ,6HWF ,6HWG24 ,6HFAR24 , 80
26HETAD ,6HDPDUC ,6HBPASS,6HDUMS3 ,6HTS28 ,6HPS28 ,6HV28 , 81
36HAM23 ,6HTS29 ,6HPS29 ,6HV29 ,6HAM29 ,6HXT55 ,6HXP55 , 82
46HXH55 ,6HXS55 ,6HXT25 ,6HXP25 ,6HXH25 ,6HXS25 ,6HXWFB , 83
56HXWG55 ,6HXFAR55,6HXWFD ,6HXWG24 ,6HXFAR24,6HXXP1 ,6HDUMB , 84
66HT6 ,6HP6 ,6HH6 ,6HS6 ,6HT7 ,6HP7 ,6HH7 , 85
76HS7 ,6HT8 ,6HP8 ,6HH8 ,6HS8 ,6HT9 ,6HP9 , 86
86HH9 ,6HS9 ,6HWG6 ,6HWFA ,6HWG7 ,6HFAR7 ,6HETAA , 87
96HDPAFT ,6HV55 ,6HV25 ,6HPS6 ,6HV6 ,6HAM6 ,6HTS7 , 88
$6HPS7 ,6HV7 ,6HAM7 ,6HAM25 ,6HTS8 ,6HPS8 ,6HV8 , 89
$6HAM8 ,6HTS9 ,6HPS9 ,6HV9 ,6HAM9 ,6HVA ,6HFRD , 90
$6HVJD ,6HFGMD ,6HVJM ,6HFGMM ,6HFGPD ,6HFGPM ,6HFGM , 91
$6HFGP ,6HWFT ,6HWGT ,6HFART ,6HFG ,6HFN ,6HSFC / 92
DATA (WORDY(I),I=281,297)/ 93
111*0. ,6HTFFHP ,6HCNHP ,6HDHTCHP,6HTFFLP ,6HCNLP ,6HDHTCLP/ 94
DATA THEEND,BLANK,LIMIT/ 6HTHEEND,6H ,297/ 95
GO TO (1,6),ICON 96

```

C *** INPUT SECTION	97
1 DO 4 N=1,103	98
NUM=N	99
READ (5,11) AIN,CHANGE	100
IF (AIN.EQ.THEEND) GO TO 5	101
DO 2 J=1,LIMIT	102
JJ=J	103
IF (AIN.EQ.WORDY(J)) GO TO 3	104
2 CONTINUE	105
WRITE (6,12) AIN	106
GO TO 4	107
3 IOUT(NUM)=JJ	108
IF (CHANGE.NE.BLANK) WORDY(JJ)=CHANGE	109
4 CONTINUE	110
WRITE (6,13)	111
5 NUM=NUM-1	112
RETURN	113
C *** OUTPUT SECTION	114
6 IF (NUM.EQ.1) GO TO 10	115
N=NUM	116
J=6	117
DO 9 I=1,NUM,6	118
IF (N.GT.6) GO TO 7	119
J=N	120
7 N=N-6	121
DO 8 K=1,J	122
L=I+K-1	123
M=IOUT(L)	124
WOUT(K)=WORDY(M)	125
8 AOUT(K)=PARAM(M)	126
WRITE (6,14) (WOUT(K),K=1,J)	127
WRITE (6,15) (AOUT(K),K=1,J)	128
IF (N.LE.0) GO TO 10	129
9 CONTINUE	130
10 RETURN	131
C	132
C	133
11 FORMAT (A6,6X,A6)	134
12 FORMAT (10H0THE WORD ,A6,26H NOT FOUND IN COMMON ARRAY)	135
13 FORMAT (22H0ERROR IN CONOUT INPUT)	136
14 FORMAT (1H ,25XA6,5(9XA6))	137
15 FORMAT (1H ,20X6E15.6)	138
END	139

\$IBFTC THCOMP DECK	
SUBROJTINE THCOMP (PR,ETA,T,H,S,P,TQ,HO,SO,PO)	1
PO=P*PR	2
TP=T*PR**0.28572	3
DO 1 I=1,25	4
CALL THERMO (PO,HP,TP,SP,X1,0,X2,0)	5
DELS=SP-S	6
IF (ABS(DELS).LE.0.00005*S) GO TO 2	7
1 TP=TP/EXP(4.*DELS)	8
CALL ERROR	9
2 HO=H+((HP-H)/ETA)	10
CALL THERMO (PO,HO,TQ,SO,X1,0,X2,1)	11
RETURN	12
END	13

```

$IBFTC PROCJM  DECK
SUBROJTIME PRQCOM (FARX,TEX,CSEX,AKEX,CPEX,REX,PHI,HEX)
IF (FARX.LE.0.067623) GO TO 1
FARX=0.067623
1 IF (TEX.GE.300.) GO TO 2
TEX=300.
2 IF (TEX.LE.4000.) GO TO 3
TEX=4000.
3 IF (FARX.GE.0.0) GO TO 4
FARX=0.0
C AIR PATH
4 CPA=((((1.0115540E-25*TEX-1.4526770E-21)*TEX+7.6215767E-18)*TEX-
11.5128259E-14)*TEX-6.7178376E-12)*TEX+6.5519486E-08)*TEX-5.1536879
2E-05)*TEX+2.5020051E-01
HEA=((((1.2544425E-26*TEX-2.0752522E-22)*TEX+1.2702630E-18)*TEX
1-3.0256518E-15)*TEX-1.6794594E-12)*TEX+2.1839826E-08)*TEX-2.576844
20E-05)*TEX+2.5020051E-01)*TEX-1.7558886E+00
SEA=+2.5020051E-01*ALOG(TEX)+((((1.4450767E-26*TEX-2.4211288E-22
1)*TEX+1.5243153E-18)*TEX-3.7820648E-15)*TEX-2.2392790E-12)*TEX+3.2
2759743E-08)*TEX-5.1576879E-05)*TEX+4.5432300E-02
IF (FARX.LE.0.0) GO TO 5
C FUEL/AIR PATH
CPF=((((17.2678710E-25*TEX-1.3335668E-20)*TEX+1.0212913E-16)*TEX-
14.2051104E-13)*TEX+9.9686793E-10)*TEX-1.3771901E-06)*TEX+1.2258630
2E-03)*TEX+7.3816638E-02
HEF=((((19.0848388E-26*TEX-1.9050949E-21)*TEX+1.7021525E-17)*TEX
1-8.4102208E-14)*TEX+2.4921698E-10)*TEX-4.5906332E-07)*TEX+6.129315
20E-04)*TEX+7.3816638E-02)*TEX+3.0581530E+01
SEF=+7.3816638E-02*ALOG(TEX)+((((1.0382670E-25*TEX-2.2226118E-21
1)*TEX+2.0425826E-17)*TEX-1.0512776E-13)*TEX+3.3228928E-10)*TEX-6.8
2859505E-07)*TEX+1.2258630E-03)*TEX+6.483398E-01
5 CPEX=(CPA+FARX*CPF)/(1.+FARX)
HEX=(HEA+FARX*HEF)/(1.+FARX)
PHI=(SEA+FARX*SEF)/(1.+FARX)
AMW=28.97-.946186*FARX
REX=1.986375/AMW
AKEX=CPEX/(CPEX-REX)
CSEX=SQRT(AKEX*REX*TEX*25031.37)
RETURN
END

```

```

$IBFTC SERGH  DECK
SUBROJTIME SEARCH (P,A,B,C,D,AX,NA,BX,CX,DX,NO,NAM,NOM,NCODE)
DIMENSION AX(NAM),BX(NAM,NOM),CX(NAM,NOM),DX(NAM,NOM),NO(NAM),Q(9)
C *** NEEDS SUBROUTINE AFQUIR
C *** AX AND BX MUST BE STORED LO TO HI
C *** P=INPJT PROPORTION BETWEEN 0.0 AND 1.0
C IF NOT INPUT, P MUST EQUAL -1.
C *** NCODE=00 OK
C NCODE=01 A LO
C NCODE=02 A HI
C NCODE=07 ERROR
C NCODE=10 B LO
C NCODE=11 B HI
C NCODE=20 B HI
NCODE=0
C=0.
D=0.

```

C ***	FIND A	16
	DO 1 I=1,NA	17
	IH=I	18
	IF (A.LT.AX(I)) GO TO 2	19
1	CONTINUE	20
	IF (A.GT.AX(IH)) NCODE=2	21
	A=AX(IH)	22
	GO TO 3	23
2	IF (I+.GT.1) GO TO 3	24
	NCODE=1	25
	IH=2	26
	A=AX(1)	27
3	IL=IH-1	28
	LIMH=ND(IH)	29
	LIML=ND(IL)	30
C ***	FIND B	31
	PRM=(A-AX(IL))/(AX(IH)-AX(IL))	32
	PP=P	33
	IF (P.GE.0.) GO TO 6	34
	BL=BX(IL,1)+PRM*(BX(IH,1)-BX(IL,1))	35
	BH=BX(IL,LIML)+PRM*(BX(IH,LIMH)-BX(IL,LIML))	36
	IF (B.GE.BL) GO TO 4	37
	NCODE=NCODE+10	38
	B=BL	39
	GO TO 5	40
4	IF (B.LE.BH) GO TO 5	41
	NCODE=NCODE+20	42
	B=BH	43
5	PP=0.5	44
	Q(2)=0.	45
	Q(3)=0.	46
6	BH=PP*(BX(IH,LIMH)-BX(IH,1))+BX(IH,1)	47
	BL=PP*(BX(IL,LIML)-BX(IL,1))+BX(IL,1)	48
	DO 7 J=2,LIMH	49
	JH=J	50
	IF (B+.LT.BX(IH,J)) GO TO 8	51
7	CONTINUE	52
8	JL=JH-1	53
	DO 9 K=2,LIML	54
	KH=K	55
	IF (BL.LT.BX(IL,K)) GO TO 10	56
9	CONTINUE	57
10	KL=KH-1	58
	PR=(BX(IH,JL)-BH)/(BX(IH,JH)-BX(IH,JL))	59
	CH=CX(IH,JL)-PR*(CX(IH,JH)-CX(IH,JL))	60
	DH=DX(IH,JL)-PR*(DX(IH,JH)-DX(IH,JL))	61
	PR=(BX(IL,KL)-BL)/(BX(IL,KH)-BX(IL,KL))	62
	CL=CX(IL,KL)-PR*(CX(IL,KH)-CX(IL,KL))	63
	DL=DX(IL,KL)-PR*(DX(IL,KH)-DX(IL,KL))	64
	BT=BL+PRM*(BH-BL)	65
	CT=CL+PRM*(CH-CL)	66
	DT=DL+PRM*(DH-DL)	67
	IF (P.GE.0.) GO TO 13	68
	DIR=SQRT(B/BT)	69
	ERR=(B-BT)/B	70
	CALL AFQUIR(Q(1),PP,ERR,0.,25.,0.001,DIR,PT,ICON)	71
	GO TO (11,13,12),ICON	72
11	PP=PT	73
	IF (P>.LT.0.) PP=0.	74
	IF (P>.GT.1.) PP=1.	75
	GO TO 5	76
12	NCODE=7	77

13	B=BT	78
	C=CT	79
	D=DT	80
	RETURN	81
	END	82

```

$IBFTC MAPBAK DECK
SUBROJTIME MAPBAC (MAP,MAPGO,TFFS,TFF,CNS,CN,PCN,T,MODE,IGO,NUM)
DATA W4,WL,WT,WS/6H H.P.,6H L.P.,6H TFF ,6HSPEED /
IF (NJH.GT.0) GO TO 1
NUMH=0
NUML=0
1 IGO=MAPGO+3*(MAP-1)
GO TO (2,3,5,6,7,9),IGO
C *** HIGH PRESSURE TURBINE
2 TFF=TFF+0.1*(TFF-TFFS)
WRITE (8,10) WH,WT,TFFS,TFF
RETURN
3 CN=CN+0.05*(CN-CNS)
IF (MODE.NE.1) PCN=PCN*(CN/CNS)
IF (MODE.EQ.1) T=T*(CNS/CN)**2
WRITE (8,10) WH,WS,CNS,CN
IF (NJMH.GT.2) GO TO 4
NUM=1
NUMH=VJMH+1
RETURN
4 DELCN=CN-CNS
IF (DELCN.GE.0.) RETURN
TFF=TFF*(1.+DELCN/CN)
WRITE (8,11) WH,WT,TFFS,TFF
RETURN
5 TFF=TFF+0.1*(TFF-TFFS)
WRITE (8,10) WH,WT,TFFS,TFF
GO TO 3
C *** LOW PRESSURE TURBINE
6 TFF=TFF+0.1*(TFF-TFFS)
WRITE (8,10) WL,WT,TFFS,TFF
RETURN
7 CN=CN+0.05*(CN-CNS)
IF (MODE.NE.3) PCN=PCN*(CN/CNS)
IF (MODE.EQ.3) T=T*(CNS/CN)
WRITE (8,10) WL,WS,CNS,CN
IF (NJML.GT.2) GO TO 8
NUM=1
NUML=VJML+1
RETURN
8 DELCN=CN-CNS
IF (DELCN.GE.0.) RETURN
TFF=TFF*(1.+DELCN/CN)
WRITE (8,11) WL,WT,TFFS,TFF
RETURN
9 TFF=TFF+0.1*(TFF-TFFS)
WRITE (8,10) WL,WT,TFFS,TFF
GO TO 7
C
C
10 FORMAT (1H0,A6,12HTURBINE MAP ,A6,4HWAS=,E13.6,10H AND NOW=,E13.6
1,6H$$$$$)

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11  FORMAT (1HO,A6,A6,22HWAS ALSO CHANGED FROM ,E13.6,5H TO ,E13.6,6H      52
1$$$) 1$$$) 53
END 54

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$IBFTC CONVRG DECK
SUBROJTINE CONVRG (TI,HI,PI,SI,FAR,WG,PA,IDES,AO,PR,TO,HO,PO,SO,TS      1
1O,PSO,VO,AMO,ICON) 2
C ICON=1 SUBSONIC, COMPARE PI WITH PR 3
C ICON=2 SONIC, COMPARE PI WITH PR 4
C ICON=4 ERROR 5
AJ=778.26 6
CAPSF=2116.217 7
G=32.174049 8
CALL PROCOM (FAR,TI,XX1,XX2,XX3,XX4,PHII,XX6) 9
C *** SONIC CALCULATIONS 10
J=0 11
TSS=0.833*TI 12
1 J=J+1 13
CALL PROCOM (FAR,TSS,CSS,AKS,CP,REXS,PHISS,HSS) 14
HSCAL=HI-CSS**2/(2.*G*AJ) 15
DELHS=HSCAL-HSS 16
IF (ABS(DELHS)-0.0005*HSCAL) 4,4,2 17
2 TSS=TSS+DELHS/CP 18
IF (J-15) 1,1,3 19
3 ICON=4 20
RETURN 21
4 IF (IDES) 12,12,5 22
C *** ISENTROPIC EXPANSION CALCULATIONS 23
5 J=0 24
TSI=TI*(PA/PI)**0.286 25
6 J=J+1 26
CALL THERMO (PA,HSI,TSI,SSI,XX1,1,FAR,0) 27
IF (ABS(SSI-SI)-0.0001*SI) 8,8,7 28
7 TSI=TSI/EXP(4.*(SSI-SI)) 29
IF (J-30) 6,6,3 30
8 VIS=SQRT(2.*G*AJ*(HI-HSI)) 31
IF (VIS-CSS) 9,11,11 32
C *** SUBSONIC DESIGN, CALCULATE AO 33
9 VO=VIS 34
TSO=TSI 35
PSO=PA 36
CALL PROCOM (FAR,TSO,CSO,XX2,XX3,REX,PHISO,HSO) 37
RHO=CAPSF*PSO/(AJ*REX*TSO) 38
AO=WG/(RHO*VO) 39
AMO=VJ/CSO 40
PR=PI 41
ICON=1 42
10 TO=TI 43
HO=HI 44
PO=PI 45
SO=SI 46
RETURN 47
C *** SONIC DESIGN, CALCULATE AO 48
11 VO=CSS 49
TSO=TSS 50
PSO=PI*(TSO/TI)**(AKS/(AKS-1.)) 51
RHO=CAPSF*PSO/(AJ*REXS*TSO) 52
AO=WG/(RHO*VO) 53

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	AMO=1.0	54
	PR=PI	55
	ICON=2	56
	GO TO 10	57
C ***	NON-DESIGN, CALCULATE CRITICAL CONDITIONS	58
12	VO=CSS	59
	TSO=TSS	60
	PSO=PA	61
	RHO=CAPSF*PSO/(AJ*REXS*TSO)	62
	AOCRIT=WG/(RHO*VO)	63
	AMO=1.0	64
	PR=PSJ*(TI/TSO)**(AKS/(AKS-1.))	65
	IF (AJ-AOCRIT) 13,13,14	66
C ***	NON-DESIGN, CRITICAL AND SUPERCRITICAL CONDITIONS	67
13	PSO=PSJ*AOCRIT/AO	68
	PR=PR*AOCRIT/AO	69
	ICON=2	70
	GO TO 10	71
C ***	NON-DESIGN, SUBSONIC CALCULATIONS	72
14	PSO=PA	73
	J=0	74
	TSO=0.833*TSO	75
15	J=J+1	76
	CALL PROCOM (FAR,TSO,CSO,AKO,CP,REX,PHISO,HSS)	77
	RHO=CAPSF*PSO/(AJ*REX*TSO)	78
	VO=WG/(RHO*AO)	79
	HSCAL=HI-VO**2/(2.*G*AJ)	80
	DELHS=HSCAL-HSS	81
	IF (ABS(DELHS)-0.0005*HSCAL) 17,17,16	82
16	TSO=TSO+DELHS/CP	83
	IF (J-15) 15,15,3	84
17	AMO=VO/CSO	85
	PR=PSJ*(TI/TSO)**(AKO/(AKO-1.))	86
	ICON=1	87
	GO TO 10	88
	END	89

\$IBFTC CONDIV DECK		
	SUBROJTIME CONDIV (TI,HI,PI,SI,FAR,WG,PA,IDES,AT,AO,PIR,TT,HT,PT,S	1
	IT,TO,TO,PO,SO,TST,TSO,PST,PSO,VT,VO,AMT,AMO,ICON)	2
C	ICON=1 SUBSONIC, COMPARE PIR WITH PI	3
C	ICON=2 SONIC, SHOCK INSIDE NOZZLE, COMPARE PIR WITH PI	4
C	ICON=3 SONIC, SHOCK OUTSIDE NOZZLE, COMPARE PIR WITH PI	5
C	ICON=4 ERROR	6
	DIMENSION Q(9)	7
	Q(2)=0.	8
	Q(3)=0.	9
	AJ=779.26	10
	CAPSF=2116.2170	11
	G=32.174049	12
	CALL PROCOM (FAR,TI,XX1,XX2,XX3,XX4,PHII,XX6)	13
C ***	SONIC CALCULATIONS	14
	J=0	15
	TSS=0.833*TI	16
1	J=J+1	17
	CALL PROCOM (FAR,TSS,CSO,AK,CP,REXS,PHISS,HSS)	18
	HSCAL=HI-CSS**2/(2.*G*AJ)	19

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DELHS=HSCAL-HSS
IF (ABS(DELHS)-0.0005*HSCAL) 4,4,2
2 TSS=TSS+DELHS/CP
IF (J-15) 1,1,3
3 ICON=4
RETURN
4 IF (IDES) 11,11,5
C *** SONIC DESIGN, CALCULATE AT
5 VT=CSS
TST=TSS
PST=PI*(TST/TI)**(AK/(AK-1.))
RHO=CAPSF*PST/(AJ*REXS*TST)
AT=WG/(RHO*VT)
AMT=1.0
C *** IDEAL EXPANSION DESIGN, CALCULATE A0
PSD=PA
J=0
TSO=TI*(PSD/PI)**.286
6 J=J+1
CALL PROCOM (FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)
PHICAL=PHI I-REX*A LOG(P I/PSD)
DELPHI=PHICAL-PHISO
IF (ABS(DELPHI)-0.0001*PHICAL) 8,8,7
7 TSO=TSO*EXP(4.*DELPHI)
IF (J-15) 6,6,3
8 VO=SQRT(2.*G*AJ*(HI-HSO))
AMO=VJ/CSO
AO=(AT/AMO)*(2.*(1.+(AK-1.)*AMO**2/2.)/(AK+1.))**((AK+1.)/(2.*(AK-
11.)))
PIR=PI
ICON=3
9 TO=TI
HO=HI
PO=PI
SO=SI
10 TT=TI
HT=HI
PT=PI
ST=SI
RETURN
C *** ASSUME SONIC THROAT AND ISENTROPIC EXPANSION TO A0
11 VT=CSS
AMT=1.0
TST=TSS
RHO=WG/(AT*VT)
PST=RHO*AJ*REXS*TST/CAPSF
PIR=PST*(TI/TST)**(AK/(AK-1.))
IF (PST-PA) 12,27,27
12 TSO=0.95*TI
MAM=0
13 CALL PROCOM (FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)
AMO=SQRT(2.*((TI/TSO)-1.)/(AK-1.))
AOCAL=(AT/AMO)*(2.*(1.+(AK-1.)*AMO**2/2.)/(AK+1.))**((AK+1.)/(2.*(
1AK-1.)))
EA=(AJ-AOCAL)/AO
DIR=SQRT(AO/AOCAL)
CALL AFQUIR (Q(1),TSO,EA,0.,100.,0.0001,DIR,TSOT,JCON)
GO TO (14,18,3),JCON
14 TSO=TSOT
IF (TSO-TI) 15,13,16
15 TSC=2.*TI/(AK+1.)
IF (TSO.GT.TSC) GO TO 17

```

16	TSO=0.98*TI	82
	GO TO 13	83
17	IF (Q(2).LT.30.0.OR.AMO.LT.0.95.OR.MAM.EQ.1) GO TO 13	84
	TSO=2.*TI/(2.+0.98*(AK-1.))	85
	MAM=1	85
	GO TO 13	87
18	PSO=PIR*(TSO/TI)**(AK/(AK-1.))	88
	IF (PSO-PA) 20,19,27	89
C ***	CRITICAL FLOW, ISENTROPIC EXPANSION TO PA	90
19	VO=AMJ*CSO	91
	ICON=1	92
	GO TO 9	93
C ***	SUBSONIC FLOW	94
20	PSO=PA	95
	Q(2)=0.	96
	Q(3)=0.	97
	J=0	98
	TSO=0.833*TI	99
21	J=J+1	100
	CALL PROCOM (FAR,TSO,CSO,AK,CP,REX,PHISO,HSD)	101
	RHO=CAPSF*PSO/(AJ*REX*TSO)	102
	VO=VG/(RHO*AO)	103
	HSCAL=HI-VO**2/(2.*G*AJ)	104
	DELHS=HSCAL-HSD	105
	IF (ABS(DELHS)-0.0005*HSCAL) 23,23,22	106
22	TSO=TSJ+DELHS/CP	107
	IF (J-15) 21,21,3	108
23	AMO=VJ/CSO	109
	PIR=PSO*(TI/TSO)**(AK/(AK-1.))	110
	TST=TSO	111
24	CALL PROCOM (FAR,TST,CST,AK,CP,REX,PHIST,HST)	112
	PST=PIR*(TST/TI)**(AK/(AK-1.))	113
	RHO=PST*CAPSF/(AJ*REX*TST)	114
	VT=VG/(RHO*AT)	115
	HSCAL=HI-VT**2/(2.*G*AJ)	116
	EH=(HSCAL-HST)/HSCAL	117
	DIR=1.+(HSCAL-HST)/(CP*TST)	118
	CALL AFQUIR (Q(1),TST,EH,0.,20.,0.0005,DIR,TSTT,JCON)	119
	GO TO (25,26,3),JCON	120
25	TST=TSTT	121
	GO TO 24	122
26	AMT=VT/CST	123
	ICON=1	124
	GO TO 9	125
C ***	SUPERCRITICAL FLOW, ISENTROPIC EXPANSION TO PA	126
27	PSO=PA	127
	J=0	128
	TSO=TI*(PSO/PIR)**.286	129
28	J=J+1	130
	CALL PROCOM (FAR,TSO,CSO,AK,CP,REX,PHISO,HSD)	131
	PHICAL=PHII-REX*ALOG(PIR/PSO)	132
	DELPHI=PHICAL-PHISO	133
	IF (ABS(DELPHI)-0.0001*PHICAL) 30,30,29	134
29	TSO=TSJ*EXP(4.0*DELPHI)	135
	IF (J-15) 28,28,3	136
30	VO=SQRT(2.*G*AJ*(HI-HSD))	137
	AMO=VJ/CSO	138
	AOID=(AT/AMO)*(2.*(1.+(AK-1.)*AMO**2/2.)/(AK+1.))**((AK+1.)/(2.*(AK-1.)))	139
	ICON=3	140
	N=0	141
	IF (AO-AOID) 31,9,32	142
		143

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C *** SUPERCRITICAL FLOW, ISENTROPIC EXPANSION TO AO 144
31 N=1 145
32 TSO=0.833*TI 146
   J=0 147
33 J=J+1 148
   CALL PROCOM (FAR,TSO,CSO,AK,CP,REX,PHISO,HSO) 149
   AMO=SQRT(2.*((TI/TSO)-1.)/(AK-1.)) 150
   AOCAL=(AT/AMO)*(2.*(1.+(AK-1.)*AMO**2/2.)/(AK+1.))*((AK+1.)/(2.*(
1AK-1.))) 151
   DELA=AO-AOCAL 152
   IF (ABS(DELA)-0.0001*AO) 35,35,34 154
34 TSO=TSO*SQRT(AOCAL/AO) 155
   IF (J-50) 33,33,3 156
35 IF (N) 37,37,36 157
C *** UNDEREXPANDED, SHOCK OUTSIDE NOZZLE 158
36 PSO=PIR*(TSO/TI)**(AK/(AK-1.)) 159
   VO=AMJ*CSO 160
   GO TO 9 161
C *** OVEREXPANDED, FIND SHOCK POSITION 162
37 PSX=PIR*(TSO/TI)**(AK/(AK-1.)) 163
   PSY=PSX*(2.*AK*AMO**2/(AK+1.)-(AK-1.)/(AK+1.)) 164
   IF (PA-PSY) 38,39,39 165
C *** OVEREXPANDED, SHOCK OUTSIDE NOZZLE 166
38 PSO=PSX 167
   VO=AMJ*CSO 168
   GO TO 9 169
C *** OVEREXPANDED, SHOCK INSIDE NOZZLE 170
39 PSO=PA 171
   J=0 172
   TSO=0.833*TI 173
40 J=J+1 174
   CALL PROCOM (FAR,TSO,CSO,AK,CP,REX,PHISO,HSO) 175
   RHO=CAPSF*PSO/(AJ*REX*TSO) 176
   VO=WG/(RHO*AO) 177
   HSCAL=HI-VO**2/(2.*G*AJ) 178
   DELHS=HSCAL-HSO 179
   IF (ABS(DELHS)-0.0005*HSCAL) 42,42,41 180
41 TSO=TSO+DELHS/CP 181
   IF (J-15) 40,40,3 182
42 AMO=VO/CSO 183
   TO=TI 184
   HO=HI 185
   PO=PSJ*(TO/TSO)**(AK/(AK-1.)) 186
   SO=PHII-REX*ALOG(PO) 187
   ICON=2 188
   GO TO 10 189
   END 190

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$IBFTC THTERB DECK
SUBROJTINE THTURB (DH,ETA,FAR,H,S,P,TO,HO,SO,PO) 1
HO=H-J 2
HOP=H-DH/ETA 3
PT=P/2. 4
DO 1 I=1,25 5
CALL THERMO (PT,HOP,TT,ST,AMWT,1,FAR,1) 5
DELS=ST-S 7
IF (ABS(DELS).LE.0.00005*S) GO TO 2 8

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1	PT=P*EXP(DELS*AMWT/1.986375+ALOG(PT/P))	9
	CALL ERROR	10
2	PO=PT	11
	CALL THERMO (PO,HO,TO,SO,X1,1,FAR,1)	12
	RETURN	13
	END	14

\$IBFTC	THERMO DECK	
	SUBROUTINE THERMO (PX,HX,TX,SX,AMX,L,FAR,K)	1
	FX=0.	2
	IF (L.EQ.1) FX=FAR	3
	IF (K.EQ.1) GO TO 1	4
	CALL PROCOM (FX,TX,CS,AK,CP,R,PHI,HX)	5
	GO TO 3	6
1	TX=4.*HX	7
	DO 2 I=1,15	8
	CALL PROCOM (FX,TX,CS,AK,CP,R,PHI,H)	9
	DELH=HX-H	10
	IF (ABS(DELH).LE.0.00001*HX) GO TO 3	11
2	TX=TX+4.*DELH	12
	WRITE (8,4)	13
3	SX=PHI-R*ALOG(PX)	14
	AMX=1.986375/R	15
	RETURN	16
C		17
C		18
4	FORMAT (31HONO CONVERGENCE IN THERMO\$\$\$\$\$)	19
	END	20

\$IBFTC	AFQUER DECK	
	SUBROUTINE AFQUIR (X,AIND,DEPEND,ANS,AJ,TOL,DIR,ANEW,ICON)	1
	DIMENSION X(9)	2
C	X(1)=NAME OF ARRAY TO USE	3
C	AIND=INDEPENDANT VARIABLE	4
C	DEPEND=DEPENDANT VARIABLE	5
C	ANS=ANSWER UPON WHICH TO CONVERGE	6
C	AJ=MAX NUMBER OF TRYS	7
C	TOL=PERCENT TOLERANCE FOR CONVERGENCE	8
C	DIR=DIRECTION AND PERCENTAGE FOR FIRST GUESS	9
C	ANEW=CALCULATED VALUE OF NEXT TRY AT INDEPENDANT VARIABLE	10
C	ICON=CONTROL =1 GO THRU LOOP AGAIN	11
C	=2 YOU HAVE REACHED THE ANSWER	12
C	=3 COUNTER HAS HIT LIMITS	13
C	X(2)=COUNTER STORAGE	14
C	X(3)=CHOOSES METHOD OF CONVERGENCE	15
C	X(4)=THIRD DEPEND VAR	16
C	X(5)=THIRD IND VAR	17
C	X(6)=SECOND DEPEND VAR	18
C	X(7)=SECOND IND VAR	19
C	X(8)=FIRST DEPEND VAR	20
C	X(9)=FIRST IND VAR	21
C	X(3) MUST BE ZERO UPON FIRST ENTRY TO ROUTINE	22
	Y=0.	23
	IF (ANS) 1,2,1	24

1	DEP=DEPEND-ANS	25
	TOLANS=TOL*ANS	26
	GO TO 3	27
2	DEP=DEPEND	28
	TOLANS=TOL	29
3	IF (ABS(DEP)-TOLANS) 5,5,4	30
4	IF (X(2)-AJ) 8,8,7	31
5	ANEW=AIND	32
	X(2)=0.	33
	ICON=2	34
	RETURN	35
6	ANEW=Y	36
	X(2)=X(2)+1.	37
	ICON=1	38
	RETURN	39
7	ANEW=Y	40
	X(2)=0.	41
	ICON=3	42
	RETURN	43
8	IF (X(3)) 9,9,12	44
C ***	FIRST GUESS USING DIR	45
9	X(3)=1.	46
	X(8)=DEP	47
	X(9)=AIND	48
	IF (AIND) 10,11,10	49
10	Y=DIR*AIND	50
	GO TO 5	51
11	Y=DIR	52
	GO TO 6	53
12	IF (X(3)-1.) 13,13,16	54
C ***	LINEAR GUESS	55
13	X(3)=2.	56
	X(6)=DEP	57
	X(7)=AIND	58
	IF (X(8)-X(6)) 14,9,14	59
14	IF (X(9)-X(7)) 15,9,15	60
15	A=(X(9)-X(7))/(X(8)-X(6))	61
	Y=X(9)-A*X(8)	62
	IF (ABS(10.*X(9))-ABS(Y)) 9,9,6	63
C ***	QUADRATIC GUESS	64
16	X(4)=DEP	65
	X(5)=AIND	66
	IF (X(7)-X(5)) 18,17,18	67
17	IF (X(6)-X(4)) 13,9,13	68
18	IF (X(6)-X(4)) 19,13,19	69
19	IF (X(9)-X(5)) 23,20,23	70
20	IF (X(8)-X(4)) 21,22,21	71
21	X(9)=X(7)	72
	X(8)=X(6)	73
	GO TO 13	74
22	X(9)=X(7)	75
	X(8)=X(6)	76
	X(3)=1.	77
	IF (X(9)) 10,11,10	78
23	IF (X(8)-X(4)) 24,21,24	79
24	F=(X(5)-X(4))/(X(7)-X(5))	80
	A=(X(3)-X(4)-F*(X(9)-X(5)))/((X(9)-X(7))*(X(9)-X(5)))	81
	B=F-A*(X(5)+X(7))	82
	C=X(4)+X(5)*(A*X(7)-F)	83
	IF (A) 27,25,27	84
25	IF (B) 26,7,26	85

26	Y=-C/B	86
	GO TO 47	87
27	IF (B) 32,28,32	88
28	IF (C) 30,29,30	89
29	Y=0.	90
	GO TO 47	91
30	G=-C/A	92
	IF (G) 7,7,31	93
31	Y=SQRT(G)	94
	YY=-SQRT(G)	95
	GO TO 37	96
32	IF (C) 34,33,34	97
33	Y=-B/A	98
	YY=0.	99
	GO TO 37	100
34	D=4.*A*C/B**2	101
	IF (1.-D) 13,35,36	102
35	Y=-B/(2.*A)	103
	GO TO 47	104
36	E=SQRT(1.-D)	105
	Y=(-B/(2.*A))*(1.+E)	106
	YY=(-B/(2.*A))*(1.-E)	107
37	J=4	108
	DEPMIN=ABS(X(4))	109
	DO 39 I=6,8,2	110
	IF (DEPMIN-ABS(X(I))) 39,39,38	111
38	J=I	112
	DEPMIN=ABS(X(I))	113
39	CONTINUE	114
	K=J+1	115
	IF ((X(K)-Y)*(X(K)-YY)) 42,42,40	116
40	IF (ABS(X(K)-Y)-ABS(X(K)-YY)) 47,47,41	117
41	Y=YY	118
	GO TO 47	119
42	IF (J-6) 43,44,44	120
43	JJ=J+2	121
	KK=K+2	122
	GO TO 45	123
44	JJ=J-2	124
	KK=K-2	125
45	SLOPE=(X(KK)-X(K))/(X(JJ)-X(J))	126
	IF (SLOPE*X(J)*(X(K)-Y)) 46,46,47	127
46	Y=YY	128
47	X(9)=X(7)	129
	X(8)=X(6)	130
	X(7)=X(5)	131
	X(6)=X(4)	132
	GO TO 6	133
	END	134

\$IBFTC PARABO DECK

	SUBROJTIME PARABO (X,Y,XD,YANS)	1
	DIMENSION X(3),Y(3)	2
	A=((X(1)-X(2))*(Y(1)-Y(3))-(X(1)-X(3))*(Y(1)-Y(2)))/((X(1)-X(2))*(X(1)-X(3))*(X(3)-X(2)))	3
	B=((X(1)**2-X(2)**2)*(Y(1)-Y(3))-(X(1)**2-X(3)**2)*(Y(1)-Y(2)))/((X(1)-X(2))*(X(1)-X(3))*(X(2)-X(3)))	4
		5
		6



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D=(Y(1)*X(2)**2-Y(2)*X(1)**2-B*X(2)*X(1)*(X(2)-X(1)))/(X(2)**2-X(1)**2)
YANS=(A*XD+B)*XD+D
RETURN
END

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7  
8  
9  
10  
11

\$IBFTC BLKFAN DECK

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C THIS IS A GENERALIZED FAN MAP FOR UNREALISTIC SUPERSONIC ENGINE 1
BLOCK DATA 2
COMMON / FAN/CN(15),PR(15,15),WAC(15,15),ETA(15,15),N,NP(15) 3
DATA N,NP/10,6,3*7,5*10,8,5*0/ 4
DATA CN/0.3,0.4,0.5,0.6,0.7,0.8,0.9,1.0,1.1,1.2,5*0./ 5
DATA (PR( 1,J),WAC( 1,J),ETA( 1,J),J=1, 6)/ 6
1 1.00000, 243.600, 0.75592, 1.01200, 229.800, 0.76120, 7
2 1.02800, 199.800, 0.76648, 1.03840, 166.800, 0.75592, 8
3 1.04480, 133.200, 0.72512, 1.04800, 86.400, 0.64152/ 9
DATA (PR( 2,J),WAC( 2,J),ETA( 2,J),J=1, 7)/ 10
1 1.00000, 286.800, 0.75592, 1.02000, 270.000, 0.77616, 11
2 1.04000, 253.200, 0.79200, 1.05840, 233.400, 0.79728, 12
3 1.07520, 209.400, 0.80256, 1.09200, 183.600, 0.77616, 13
4 1.10000, 156.600, 0.74008/ 14
DATA (PR( 3,J),WAC( 3,J),ETA( 3,J),J=1, 7)/ 15
1 1.00000, 333.600, 0.75064, 1.02560, 322.800, 0.77616, 15
2 1.05120, 310.200, 0.80256, 1.08000, 291.600, 0.82808, 17
3 1.11600, 259.800, 0.84392, 1.13200, 240.000, 0.82808, 18
4 1.14800, 213.600, 0.77616/ 19
DATA (PR( 4,J),WAC( 4,J),ETA( 4,J),J=1, 7)/ 20
1 1.00000, 383.400, 0.74536, 1.03680, 376.200, 0.77616, 21
2 1.09800, 358.200, 0.82808, 1.12400, 340.200, 0.85448, 22
3 1.15000, 313.200, 0.88000, 1.18960, 276.600, 0.82808, 23
4 1.19520, 266.400, 0.80784/ 24
DATA (PR( 5,J),WAC( 5,J),ETA( 5,J),J=1,10)/ 25
1 1.00000, 439.800, 0.72512, 1.06400, 436.800, 0.77616, 26
2 1.11840, 428.400, 0.82808, 1.14800, 420.600, 0.85448, 27
3 1.13400, 406.800, 0.88000, 1.20960, 393.600, 0.90112, 28
4 1.21760, 388.200, 0.90376, 1.22400, 383.400, 0.90112, 29
5 1.24400, 368.400, 0.88000, 1.26720, 342.600, 0.82808/ 30
DATA (PR( 6,J),WAC( 6,J),ETA( 6,J),J=1,10)/ 31
1 1.00000, 499.800, 0.68816, 1.10000, 499.800, 0.77616, 32
2 1.15000, 493.200, 0.82808, 1.20000, 485.400, 0.85448, 33
3 1.22800, 476.400, 0.88000, 1.25520, 466.800, 0.90112, 34
4 1.27200, 456.600, 0.91080, 1.28640, 448.200, 0.90112, 35
5 1.30240, 433.200, 0.88000, 1.33200, 406.800, 0.82720/ 36
DATA (PR( 7,J),WAC( 7,J),ETA( 7,J),J=1,10)/ 37
1 1.00000, 566.400, 0.64152, 1.07600, 566.400, 0.72512, 38
2 1.15200, 566.400, 0.77616, 1.21920, 559.800, 0.82808, 39
3 1.25000, 553.200, 0.85888, 1.28960, 544.800, 0.88000, 40
4 1.33120, 528.600, 0.90112, 1.36160, 509.400, 0.88000, 41
5 1.39120, 483.600, 0.82808, 1.40000, 474.000, 0.81752/ 42
DATA (PR( 8,J),WAC( 8,J),ETA( 8,J),J=1,10)/ 43
1 1.00000, 633.600, 0.60016, 1.04400, 633.600, 0.64152, 44
2 1.13520, 633.600, 0.72512, 1.22080, 633.000, 0.77616, 45
3 1.23440, 625.800, 0.82808, 1.34000, 616.800, 0.85888, 45
4 1.40000, 600.000, 0.88000, 1.42800, 586.800, 0.85888, 47
5 1.44800, 576.600, 0.82808, 1.48000, 553.200, 0.78672/ 48
DATA (PR( 9,J),WAC( 9,J),ETA( 9,J),J=1,10)/ 49
1 1.00000, 700.200, 0.56936, 1.10400, 700.200, 0.64152, 50
2 1.22000, 700.200, 0.72512, 1.32400, 700.200, 0.77616, 51

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3	1.40000,	700.200,	0.80256,	1.44800,	698.400,	0.80784,	52
4	1.50000,	693.600,	0.80256,	1.53360,	683.400,	0.77616,	53
5	1.55800,	666.600,	0.74536,	1.58400,	656.400,	0.72512/	54
	DATA (PR(10,J),WAC(10,J),ETA(10,J),J=1, 8)/						55
1	1.00000,	750.000,	0.51744,	1.16320,	750.000,	0.64152,	56
2	1.31200,	750.000,	0.72512,	1.40000,	750.000,	0.75592,	57
3	1.48000,	750.000,	0.76120,	1.54000,	750.000,	0.75064,	58
4	1.58000,	749.400,	0.72512,	1.66000,	736.800,	0.64152/	59
	END						60

\$IBFTC BLKCP DECK

C	THIS IS GENERALIZED COMP. MAP FOR UNREALISTIC SUPERSONIC ENGINE						1
	BLOCK DATA						2
	COMMON / COMP/CN(15),PR(15,15),WAC(15,15),ETA(15,15),N,NP(15)						3
	DATA N,NP/10,2*6,2*8,4*10,2*8,5*0/						4
	DATA CN/.562,.674,.787,.899,1.,1.034,1.067,1.124,1.236,1.292,5*0./						5
	DATA (PR( 1,J),WAC( 1,J),ETA( 1,J),J=1, 6)/						6
1	1.00000,	51.000,	0.59082,	1.84000,	50.200,	0.62178,	7
2	2.42800,	49.500,	0.64242,	2.86900,	48.800,	0.65274,	8
3	3.83500,	46.700,	0.67338,	4.54900,	44.500,	0.64242/	9
	DATA (PR( 2,J),WAC( 2,J),ETA( 2,J),J=1, 6)/						10
1	1.00000,	59.300,	0.59082,	1.96600,	59.300,	0.64242,	11
2	3.09300,	58.800,	0.69402,	3.93300,	57.900,	0.72498,	12
3	4.68900,	56.700,	0.74562,	5.52900,	55.000,	0.72498/	13
	DATA (PR( 3,J),WAC( 3,J),ETA( 3,J),J=1, 8)/						14
1	1.00000,	70.000,	0.58566,	1.84000,	70.000,	0.64242,	15
2	2.63000,	70.000,	0.68370,	3.40800,	69.500,	0.72498,	16
3	4.52100,	68.800,	0.77744,	5.44500,	67.900,	0.79292,	17
4	6.31300,	66.400,	0.77744,	6.52300,	65.700,	0.76970/	18
	DATA (PR( 4,J),WAC( 4,J),ETA( 4,J),J=1, 8)/						19
1	1.00000,	84.800,	0.58050,	2.00800,	84.800,	0.64242,	20
2	3.42900,	84.800,	0.72498,	4.60500,	84.800,	0.77744,	21
3	5.69700,	84.000,	0.80840,	6.61400,	83.300,	0.82904,	22
4	7.53800,	81.700,	0.80840,	7.95800,	80.500,	0.79292/	23
	DATA (PR( 5,J),WAC( 5,J),ETA( 5,J),J=1,10)/						24
1	1.00000,	101.700,	0.57190,	2.51900,	101.700,	0.64242,	25
2	3.98200,	101.700,	0.72498,	5.27700,	101.700,	0.77744,	26
3	6.43300,	101.200,	0.80840,	7.20200,	101.000,	0.83936,	27
4	8.00000,	100.000,	0.86000,	8.56700,	99.500,	0.83936,	28
5	9.38500,	98.100,	0.80840,	9.59600,	97.400,	0.80582/	29
	DATA (PR( 6,J),WAC( 6,J),ETA( 6,J),J=1,10)/						30
1	1.00000,	108.100,	0.57018,	2.85500,	108.100,	0.64242,	31
2	4.29700,	108.100,	0.72498,	5.61300,	108.100,	0.77744,	32
3	6.93600,	107.600,	0.80840,	7.62200,	107.100,	0.83936,	33
4	8.54500,	106.700,	0.86000,	9.13400,	106.000,	0.83936,	34
5	9.92500,	104.500,	0.80840,	10.21900,	104.000,	0.80410/	35
	DATA (PR( 7,J),WAC( 7,J),ETA( 7,J),J=1,10)/						36
1	1.00000,	114.500,	0.55986,	3.26100,	114.500,	0.64242,	37
2	4.75900,	114.500,	0.72498,	6.11700,	114.500,	0.77744,	38
3	7.45400,	114.500,	0.80840,	8.30800,	114.300,	0.83936,	39
4	9.21800,	113.600,	0.84968,	9.63800,	113.300,	0.83936,	40
5	10.51300,	112.600,	0.80840,	10.99600,	112.400,	0.79808/	41
	DATA (PR( 8,J),WAC( 8,J),ETA( 8,J),J=1,10)/						42
1	1.00000,	122.900,	0.53922,	1.68600,	122.900,	0.57018,	43
2	3.84900,	122.900,	0.64242,	5.46600,	122.900,	0.72498,	44
3	6.85500,	122.900,	0.77744,	8.37100,	122.900,	0.80840,	45
4	8.95500,	122.600,	0.82388,	9.88300,	122.100,	0.83936,	46
5	10.91200,	121.700,	0.80840,	11.81500,	120.700,	0.77744/	47

	DATA (PR( 9,J),WAC( 9,J),ETA( 9,J),J=1, 8)/	48
1	1.00000, 139.800, 0.47644, 4.35300, 139.800, 0.60114,	49
2	7.62200, 139.800, 0.72498, 10.21900, 139.800, 0.77744,	50
3	11.05900, 139.800, 0.78260, 11.89900, 139.500, 0.77744,	51
4	13.15900, 139.300, 0.72498, 13.65600, 139.000, 0.69918/	52
	DATA (PR(10,J),WAC(10,J),ETA(10,J),J=1, 8)/	53
1	1.00000, 146.200, 0.46612, 3.76500, 146.200, 0.57018,	54
2	6.43100, 146.200, 0.64242, 9.17600, 146.200, 0.72498,	55
3	10.21900, 146.200, 0.75078, 11.47900, 146.200, 0.75078,	56
4	12.71100, 146.200, 0.72498, 14.41200, 146.200, 0.64242/	57
	END	58

\$IBFTC CMBDAT DECK

	BLOCK DATA	1
	COMMON / COMB/PSI(15),DELT(15,15),ETA(15,15),N,NP(15)	2
	DATA N,NP / 15,15*15 /	3
	DATA PSI/4.9116,9.8232,14.735,19.646,24.558,29.470,34.381,	4
	139.293,44.207,73.674,100.,200.,300.,400.,500./	5
	DATA DELT/15*200.,15*300.,15*400.,15*500.,15*600.,15*700.,15*800.,	6
	115*900.,15*1000.,15*1100.,15*1200.,15*1300.,15*1400.,15*1500.,	7
	215*1600./	8
	DATA ETA/	9
	1.600,.726,.777,.806,.826,.843,.855,.865,7*.870,	10
	2.758,.825,.858,.875,.888,.898,.906,.912,.914,6*.915,	11
	3.868,.893,.911,.925,.935,.942,.947,.951,7*.953,	12
	4.925,.936,.946,.955,.963,.969,.974,.977,.978,6*.979,	13
	5.960,.966,.972,.977,.982,.985,.990,.992,.993,6*.995,	14
	6.988,.991,.992,.994,.995,.997,.998,8*.999,	15
	78*1.00,7*.999,120*1.00/	15
	END	17

\$IBFTC HPTDAT DECK

	BLOCK DATA	1
	COMMON / HTURB/TFF(15),CN(15,15),DH(15,15),ETA(15,15),N,NP(15)	2
	DATA N,NP/10,9*15,12,5*0/	3
	DATA TFF / 39.670, 42.990, 47.460, 48.610, 49.175,	4
1	49.600, 50.000, 50.425, 50.920, 51.575, 5*0./	5
	DATA (CN( 1,J),DH( 1,J),ETA( 1,J),J=1,15)/	6
1	0.1872, 0.0032, 0.6219, 0.3372, 0.0057, 0.7078,	7
2	0.5156, 0.0084, 0.7868, 0.7128, 0.0108, 0.8090,	8
3	0.9382, 0.0133, 0.8090, 1.1442, 0.0152, 0.7963,	9
4	1.3138, 0.0164, 0.7779, 1.5382, 0.0174, 0.7422,	10
5	1.7264, 0.0179, 0.7078, 1.9324, 0.0176, 0.7635,	11
6	2.1500, 0.0167, 0.6068, 2.4058, 0.0144, 0.5309,	12
7	2.5892, 0.0120, 0.4773, 2.7862, 0.0082, 0.4045,	13
8	2.9460, 0.0034, 0.3034/	14
	DATA (CN( 2,J),DH( 2,J),ETA( 2,J),J=1,15)/	15
1	0.1872, 0.0038, 0.6068, 0.3942, 0.0080, 0.7078,	16
2	0.5814, 0.0113, 0.8090, 0.7128, 0.0136, 0.8292,	17
3	0.9442, 0.0156, 0.8363, 0.9804, 0.0176, 0.8393,	18
4	1.1068, 0.0192, 0.8368, 1.2754, 0.0212, 0.8302,	19
5	1.4450, 0.0228, 0.8254, 1.7068, 0.0248, 0.8090,	20

6	1.9596,	0.0260,	0.7696,	2.2706,	0.0261,	0.7078,	21	
7	2.5970,	0.0241,	0.6068,	3.0960,	0.0188,	0.5056,	22	
8	3.3774,	0.0128,	0.4197/				23	
	DATA (CN( 3,J),DH( 3,J),ETA( 3,J),J=1,15)/							24
1	0.1872,	0.0046,	0.5764,	0.4362,	0.0100,	0.7078,	25	
2	0.5568,	0.0144,	0.8090,	0.8726,	0.0184,	0.8494,	26	
3	1.0696,	0.0216,	0.8543,	1.2382,	0.0240,	0.8515,	27	
4	1.4538,	0.0268,	0.8494,	1.6882,	0.0292,	0.8409,	28	
5	1.9596,	0.0316,	0.8262,	2.2138,	0.0331,	0.8090,	29	
6	2.5520,	0.0344,	0.7579,	2.8050,	0.0346,	0.7078,	30	
7	3.0392,	0.0340,	0.6652,	3.2648,	0.0324,	0.6068,	31	
8	3.3774,	0.0312,	0.5865/				32	
	DATA (CN( 4,J),DH( 4,J),ETA( 4,J),J=1,15)/							33
1	0.1872,	0.0052,	0.5643,	0.2550,	0.0068,	0.6068,	34	
2	0.4784,	0.0120,	0.7078,	0.6942,	0.0164,	0.8090,	35	
3	0.9148,	0.0204,	0.8494,	1.1442,	0.0244,	0.8596,	36	
4	1.3882,	0.0280,	0.8596,	1.5618,	0.0304,	0.8575,	37	
5	1.8010,	0.0336,	0.8535,	1.9794,	0.0356,	0.8494,	38	
6	2.2794,	0.0388,	0.8363,	2.5138,	0.0412,	0.8262,	39	
7	2.8334,	0.0441,	0.8090,	3.1422,	0.0472,	0.7797,	40	
8	3.3774,	0.0494,	0.7584/				41	
	DATA (CN( 5,J),DH( 5,J),ETA( 5,J),J=1,15)/							42
1	0.1872,	0.0056,	0.5562,	0.3000,	0.0088,	0.6068,	43	
2	0.5254,	0.0144,	0.7078,	0.7500,	0.0192,	0.8090,	44	
3	0.9754,	0.0236,	0.8494,	1.2754,	0.0288,	0.8697,	45	
4	1.4824,	0.0321,	0.8696,	1.7638,	0.0360,	0.8662,	46	
5	2.0450,	0.0400,	0.8615,	2.3362,	0.0444,	0.8555,	47	
6	2.5450,	0.0496,	0.8520,	2.8706,	0.0540,	0.8494,	48	
7	3.0764,	0.0596,	0.8494,	3.1520,	0.0640,	0.8532,	49	
8	3.1518,	0.0661,	0.8570/				50	
	DATA (CN( 6,J),DH( 6,J),ETA( 6,J),J=1,15)/							51
1	0.1872,	0.0068,	0.5309,	0.3568,	0.0120,	0.6068,	52	
2	0.5196,	0.0192,	0.7078,	0.8628,	0.0252,	0.8090,	53	
3	1.0332,	0.0300,	0.8494,	1.2852,	0.0340,	0.8697,	54	
4	1.5010,	0.0384,	0.8819,	1.6882,	0.0421,	0.8899,	55	
5	1.9138,	0.0472,	0.8940,	2.1246,	0.0524,	0.8969,	56	
6	2.2706,	0.0564,	0.8975,	2.4226,	0.0612,	0.8976,	57	
7	2.4950,	0.0640,	0.8968,	2.5372,	0.0668,	0.8937,	58	
8	2.5558,	0.0698,	0.8896/				59	
	DATA (CN( 7,J),DH( 7,J),ETA( 7,J),J=1,15)/							60
1	0.1872,	0.0080,	0.5062,	0.4314,	0.0164,	0.6068,	61	
2	0.5844,	0.0236,	0.7078,	0.9568,	0.0308,	0.8090,	62	
3	1.2010,	0.0372,	0.8494,	1.3834,	0.0416,	0.8697,	63	
4	1.5108,	0.0448,	0.8797,	1.6186,	0.0476,	0.8899,	64	
5	1.7450,	0.0510,	0.8954,	1.8618,	0.0544,	0.9000,	65	
6	1.9558,	0.0576,	0.9010,	2.0000,	0.0600,	0.9000,	66	
7	2.0450,	0.0624,	0.8980,	2.0824,	0.0660,	0.8925,	67	
8	2.1010,	0.0700,	0.8793/				68	
	DATA (CN( 8,J),DH( 8,J),ETA( 8,J),J=1,15)/							69
1	0.1872,	0.0088,	0.5051,	0.4834,	0.0196,	0.6068,	70	
2	0.7314,	0.0272,	0.7078,	0.8814,	0.0316,	0.7665,	71	
3	1.0226,	0.0356,	0.8090,	1.1442,	0.0392,	0.8292,	72	
4	1.2804,	0.0432,	0.8494,	1.3696,	0.0460,	0.8596,	73	
5	1.4538,	0.0488,	0.8697,	1.5950,	0.0528,	0.8808,	74	
6	1.5746,	0.0560,	0.8848,	1.7450,	0.0596,	0.8848,	75	
7	1.6010,	0.0640,	0.8788,	1.8156,	0.0664,	0.8697,	76	
8	1.8196,	0.0693,	0.8590/				77	
	DATA (CN( 9,J),DH( 9,J),ETA( 9,J),J=1,15)/							78
1	0.1872,	0.0093,	0.4909,	0.3372,	0.0159,	0.5380,	79	
2	0.5344,	0.0232,	0.6068,	0.6754,	0.0284,	0.5573,	80	
3	0.8068,	0.0330,	0.7078,	0.9196,	0.0368,	0.7463,	81	
4	1.0128,	0.0400,	0.7776,	1.1254,	0.0442,	0.8090,	82	
5	1.2196,	0.0480,	0.8191,	1.3138,	0.0524,	0.8302,	83	

6	1.3696,	0.0556,	0.8347,	1.4068,	0.0580,	0.8363,	84	
7	1.4450,	0.0612,	0.8322,	1.4638,	0.0640,	0.8241,	85	
8	1.4676,	0.0668,	0.8090/				86	
	DATA (CN(10,J),DH(10,J),ETA(10,J),J=1,12)/							87
1	0.1872,	0.0132,	0.4257,	0.2814,	0.0180,	0.4747,	88	
2	0.3804,	0.0228,	0.5056,	0.4686,	0.0268,	0.5359,	89	
3	0.5528,	0.0314,	0.5683,	0.6382,	0.0352,	0.5941,	90	
4	0.5892,	0.0380,	0.6068,	0.7362,	0.0412,	0.6178,	91	
5	0.7696,	0.0440,	0.6240,	0.8068,	0.0476,	0.6310,	92	
6	0.8254,	0.0504,	0.6265,	0.8304,	0.0530,	0.6118/	93	
	END							94

\$IBFTC LPTDAT DECK

BLOCK DATA

	COMMON /	LTURB/	TFF(15),	CN(15,15),	DH(15,15),	ETA(15,15),	N,NP(15)	1
	DATA	N,NP/	11,9*15,	12,9,4*0/				2
	DATA	TFF /	88.470,	102.795,	116.835,	129.330,	141.045,	3
1	145.725,	150.000,	153.345,	156.405,	159.780,	163.170,	4*0./	4
	DATA (CN( 1,J),DH( 1,J),ETA( 1,J),J=1,15)/							5
1	0.3582,	0.0018,	0.7120,	0.5336,	0.0026,	0.7300,		6
2	0.7365,	0.0035,	0.7472,	0.9754,	0.0044,	0.7300,		7
3	1.2146,	0.0051,	0.7140,	1.4173,	0.0056,	0.7000,		8
4	1.5201,	0.0059,	0.6850,	1.7673,	0.0061,	0.6730,		9
5	2.0247,	0.0062,	0.6452,	2.2827,	0.0061,	0.6200,		10
6	2.4565,	0.0057,	0.6000,	2.6137,	0.0053,	0.5750,		11
7	2.3166,	0.0044,	0.5310,	2.9456,	0.0035,	0.5000,		12
8	3.3138,	0.0001,	0.3850/					13
	DATA (CN( 2,J),DH( 2,J),ETA( 2,J),J=1,15)/							14
1	0.3582,	0.0026,	0.8000,	0.5518,	0.0039,	0.8100,		15
2	0.7919,	0.0054,	0.8200,	1.0672,	0.0069,	0.8300,		16
3	1.2882,	0.0080,	0.8300,	1.4446,	0.0087,	0.8290,		17
4	1.5937,	0.0096,	0.8100,	1.8954,	0.0101,	0.8000,		18
5	2.0519,	0.0104,	0.7850,	2.2273,	0.0107,	0.7600,		19
6	2.3747,	0.0108,	0.7450,	2.6229,	0.0106,	0.7000,		20
7	2.8720,	0.0101,	0.6800,	3.0555,	0.0094,	0.6450,		21
8	3.3138,	0.0077,	0.5900/					22
	DATA (CN( 3,J),DH( 3,J),ETA( 3,J),J=1,15)/							23
1	0.3582,	0.0031,	0.8000,	0.5911,	0.0051,	0.8300,		24
2	0.8555,	0.0071,	0.8600,	1.0764,	0.0087,	0.8630,		25
3	1.2519,	0.0099,	0.8670,	1.4354,	0.0111,	0.8700,		26
4	1.5201,	0.0122,	0.8720,	1.8409,	0.0134,	0.8720,		27
5	2.0247,	0.0143,	0.8700,	2.2455,	0.0152,	0.8670,		28
6	2.4302,	0.0157,	0.8600,	2.5956,	0.0162,	0.8500,		29
7	2.7791,	0.0166,	0.8300,	3.0555,	0.0167,	0.8000,		30
8	3.3138,	0.0164,	0.7600/					31
	DATA (CN( 4,J),DH( 4,J),ETA( 4,J),J=1,15)/							32
1	0.3582,	0.0033,	0.7995,	0.4237,	0.0038,	0.8000,		33
2	0.5810,	0.0061,	0.8400,	0.8837,	0.0078,	0.8600,		34
3	1.1047,	0.0096,	0.8680,	1.2882,	0.0110,	0.8730,		35
4	1.5090,	0.0126,	0.8800,	1.7482,	0.0141,	0.8830,		36
5	2.0429,	0.0159,	0.8835,	2.2091,	0.0166,	0.8830,		37
6	2.3747,	0.0174,	0.8800,	2.6047,	0.0183,	0.8740,		38
7	2.3720,	0.0191,	0.8600,	3.1291,	0.0195,	0.8350,		39
8	3.3138,	0.0197,	0.8200/					40
	DATA (CN( 5,J),DH( 5,J),ETA( 5,J),J=1,15)/							41
1	0.3582,	0.0036,	0.7750,	0.5065,	0.0049,	0.8000,		42
2	0.7365,	0.0071,	0.8480,	0.9754,	0.0092,	0.8600,		43
3	1.2882,	0.0119,	0.8750,	1.5647,	0.0141,	0.8900,		44

4	1.7301,	0.0155,	0.8912,	1.9690,	0.0172,	0.8940,	46	
5	2.0983,	0.0181,	0.8955,	2.2637,	0.0192,	0.8970,	47	
6	2.4332,	0.0202,	0.8961,	2.6691,	0.0214,	0.8900,	48	
7	2.9456,	0.0226,	0.8790,	3.1846,	0.0235,	0.8671,	49	
8	3.3138,	0.0239,	0.8600/				50	
	DATA (CN( 6,J),DH( 6,J),ETA( 6,J),J=1,15)/							51
1	0.3582,	0.0038,	0.7600,	0.6164,	0.0064,	0.8000,	52	
2	0.8372,	0.0087,	0.8450,	1.1047,	0.0113,	0.8600,	53	
3	1.2882,	0.0130,	0.8730,	1.5283,	0.0152,	0.8900,	54	
4	1.7482,	0.0171,	0.8950,	1.9509,	0.0187,	0.9000,	55	
5	2.2133,	0.0209,	0.9005,	2.4302,	0.0226,	0.9010,	56	
6	2.5510,	0.0244,	0.9004,	2.8619,	0.0259,	0.9000,	57	
7	3.1384,	0.0286,	0.8900,	3.2584,	0.0303,	0.8800,	58	
8	3.3138,	0.0319,	0.8735/				59	
	DATA (CN( 7,J),DH( 7,J),ETA( 7,J),J=1,15)/							60
1	0.3582,	0.0044,	0.7310,	0.7728,	0.0089,	0.8000,	61	
2	1.0129,	0.0115,	0.8300,	1.2659,	0.0141,	0.8600,	62	
3	1.4729,	0.0162,	0.8750,	1.6785,	0.0181,	0.8900,	63	
4	1.8409,	0.0197,	0.8930,	2.0247,	0.0216,	0.8975,	64	
5	2.1901,	0.0235,	0.8999,	2.3000,	0.0250,	0.9000,	65	
6	2.3929,	0.0265,	0.8980,	2.5038,	0.0284,	0.8937,	66	
7	2.5583,	0.0296,	0.8900,	2.6137,	0.0314,	0.8799,	67	
8	2.5319,	0.0329,	0.8710/				68	
	DATA (CN( 8,J),DH( 8,J),ETA( 8,J),J=1,15)/							69
1	0.3582,	0.0048,	0.7100,	0.6072,	0.0078,	0.7450,	70	
2	0.7919,	0.0102,	0.7680,	0.9754,	0.0124,	0.8000,	71	
3	1.2337,	0.0153,	0.8380,	1.4548,	0.0177,	0.8600,	72	
4	1.5383,	0.0201,	0.8712,	1.8409,	0.0226,	0.8780,	73	
5	1.9509,	0.0242,	0.8800,	2.0801,	0.0261,	0.8775,	74	
6	2.1537,	0.0274,	0.8760,	2.2091,	0.0285,	0.8722,	75	
7	2.2537,	0.0299,	0.8660,	2.3009,	0.0314,	0.8600,	76	
8	2.3051,	0.0321,	0.8480/				77	
	DATA (CN( 9,J),DH( 9,J),ETA( 9,J),J=1,15)/							78
1	0.3582,	0.0054,	0.6780,	0.5518,	0.0080,	0.7000,	79	
2	0.5529,	0.0096,	0.7125,	0.8282,	0.0119,	0.7350,	80	
3	1.0129,	0.0141,	0.7690,	1.1691,	0.0150,	0.8000,	81	
4	1.2337,	0.0169,	0.8060,	1.3809,	0.0188,	0.8225,	82	
5	1.5283,	0.0209,	0.8395,	1.6201,	0.0223,	0.8450,	83	
6	1.7482,	0.0244,	0.8470,	1.8409,	0.0263,	0.8445,	84	
7	1.9954,	0.0279,	0.8330,	1.9147,	0.0289,	0.8235,	85	
8	1.9237,	0.0303,	0.8080/				86	
	DATA (CN(10,J),DH(10,J),ETA(10,J),J=1,12)/							87
1	0.3582,	0.0061,	0.6380,	0.4782,	0.0078,	0.6550,	88	
2	0.5447,	0.0104,	0.6700,	0.7546,	0.0122,	0.6850,	89	
3	0.8655,	0.0139,	0.7000,	0.9754,	0.0157,	0.7110,	90	
4	1.1047,	0.0181,	0.7180,	1.2015,	0.0201,	0.7180,	91	
5	1.2701,	0.0217,	0.7170,	1.3073,	0.0230,	0.7140,	92	
6	1.3365,	0.0244,	0.7000,	1.3407,	0.0251,	0.6890/	93	
	DATA (CN(11,J),DH(11,J),ETA(11,J),J=1, 9)/							94
1	0.3582,	0.0069,	0.6000,	0.4418,	0.0086,	0.6000,	95	
2	0.5518,	0.0106,	0.6120,	0.6447,	0.0123,	0.6170,	96	
3	0.7365,	0.0141,	0.6210,	0.8282,	0.0159,	0.6258,	97	
4	0.8837,	0.0172,	0.6250,	0.9391,	0.0186,	0.6230,	98	
5	0.9715,	0.0201,	0.6009/				99	
	END							100

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$IBFTC ETAABA DECK
SUBRODTINE ETAAB (FAR,EM6,P6,ETA,ETAADS,ETAASV,P6DS,P6DSAV,AM6DS,A      1
1M6DSV,IDES,FAR7DS,FAR7SV)                                             2
DIMENSION FART(25),ETABRT(25),EM6T(7),DELM6(7),P6T(14),DELP6(14)     3
DIMENSION X(3),Y(3)                                                     4
DATA FART/.0390,.0585,.0732,.0878,.0976,.1171,.1268,.1463,.1619,      5
1.1834,.1951,.2195,.2439,.2927,.3415,.4146,.4634,.5366,.5341,.7317,   6
2.8293,.9268,1.000,1.0634,1.7/                                         7
DATA ETABRT/.9400,.9887,1.0193,1.0306,1.0227,.9672,.9377,.9207,      8
1.9354,.9626,.9773,1.0193,1.0532,1.077,1.0781,1.077,1.0747,1.0668,   9
21.0573,1.0510,1.0374,1.0192,1.00,.9626,.9151/                          10
DATA EM6T/1.00,1.071,1.190,1.309,1.428,1.547,1.666/                    11
DATA DELM6/0.,.013,.041,.073,.110,.147,.187/                          12
DATA P6T/.220,.2267,.250,.300,.3333,.3767,.4167,.500,.5833,.6667,    13
1.75,.9333,.9167,1.0/                                                  14
DATA DELP6/-.142,-.125,-.10,-.075,-.062,-.05,-.041,-.027,-.019,     15
1-.013,-.008,-.004,-.0021,0./                                         16
IF (IDES.NE.1) GO TO 5                                                  17
DO 1 K=1,25                                                              18
1  ETABRT(K)=ETABRT(K)*ETAADS/ETAASV                                    19
DO 2 K=1,25                                                              20
2  FART(K)=FART(K)*FAR7DS/FAR7SV                                        21
DO 3 L=1,7                                                                22
3  EM6T(L)=EM6T(L)*AM6DS/AM6DSV                                        23
DO 4 M=1,14                                                                24
4  P6T(M)=P6T(M)*P6DS/P6DSAV                                        25
ETAASV=ETAADS                                                            26
P6DSAV=P6DS                                                              27
FAR7SV=FAR7DS                                                            28
AM6DSV=AM6DS                                                            29
RETURN                                                                    30
5  CONTINUE                                                            31
N=0                                                                      32
IF (FAR.GT.0.067) GO TO 8                                              33
DO 6 J=1,25                                                              34
6  IF (FAR.GE.FART(J)) N=J-1                                          35
IF (N.EQ.0) N=1                                                            36
IF (N.GE.24) N=23                                                        37
DO 7 I=1,3                                                                38
NN=N-1+I                                                                39
X(I)=FART(NN)                                                            40
7  Y(I)=ETABRT(NN)                                                    41
CALL PARABO (X,Y,FAR,ETA1)                                              42
GO TO 9                                                                    43
8  ETA1=-2.*FAR+.1948                                                44
9  M=0                                                                45
DO 10 J=1,7                                                                46
10 IF (EM6.GE.EM6T(J)) M=J-1                                          47
IF (M.EQ.0) M=1                                                            48
IF (M.GE.6) M=5                                                            49
DO 11 I=1,3                                                                50
MM=M-1+I                                                                51
X(I)=EM6T(MM)                                                            52
11 Y(I)=DELM6(MM)                                                       53
CALL PARABO (X,Y,EM6,COR1)                                              54
L=0                                                                        55
DO 12 J=1,14                                                                56
12 IF (P6.GE.P6T(J)) L=J-1                                          57
IF (L.EQ.0) L=1                                                            58
IF (L.GE.13) L=12                                                        59
DO 13 I=1,3                                                                60
LL=L-1+I                                                                61
X(I)=P6T(LL)                                                            62

```

```

13  Y(I)=JELP6(LL)
    CALL PARABO (X,Y,P6,COR2)
    ETA=ETA1*(1.-COR1)*(1.+COR2)
    RETURN
    END

```

63  
64  
65  
66  
67

\$IBFTC FRATIO DECK

```

SUBROUTINE FRASHO (PRATIO,ARATIO,CF)
DIMENSION PR(30),AR(10),CFR(30,10),XX(3),YY(3),ZZ(2)
DATA PR/1.0,1.25,1.5,1.75,2.0,2.25,2.5,2.75,3.0,3.25,3.5,3.75,4.0,
14.25,4.5,4.75,5.,5.5,6.,6.5,7.,8.,9.,11.,13.,15.,17.,19.,21.,23./
DATA AR/ 1.0,1.05,1.092,1.228,1.318,1.423,1.8,1.9,1.97,2.15/
DATA (CFR(I,1),I=1,180)/
1.971,.975,.978,.980,.981,.980,.9784,.9765,.9742,.9715,
2.9688,.9658,.9632,.960,.957,.9538,.9506,.9445,.938,.932,.927,.918,
3.910,.897,.8855,.875,.867,.859,.8515,.844,
4.963,.966,.970,.9743,.978,.9805,.982,.9805,.9786,.9762,.974,.9718,
5.9693,.967,.9645,.962,.9595,.954,.949,.944,.9395,.9308,.9238,
6.9117,.901,.892,.884,.8762,.869,.862,
7.956,.9605,.9638,.967,.970,.9731,.9754,.9775,.9795,.981,.980,.978,
8.9761,.9742,.9723,.9702,.968,.9637,.9594,.9554,.9517,.945,.9387,
9.9276,.9177,.9082,.8998,.891,.883,.8752,
$.947,.949,.9517,.954,.957,.9601,.9632,.9668,.971,.9737,.976,.9777,
$.978,.9772,.976,.9752,.974,.970,.9655,.9637,.9603,.955,.950,.9403,
$.9317,.9232,.915,.907,.8996,.892,
$.949,.9502,.9508,.9473,.9438,.941,.939,.940,.950,.956,.9605,.964,
$.967,.968,.9692,.970,.9703,.9698,.9696,.9684,.966,.960,.9444,
$.9438,.9347,.9265,.9193,.9128,.9067,.901,
$.948,.9485,.9487,.9484,.947,.9435,.936,.9275,.9185,.9155,.9155,
$.929,.941,.9455,.950,.953,.9558,.9591,.9608,.9609,.9601,.957,
$.9531,.9448,.9362,.9279,.9188,.910,.901,.8921/
DATA (CFR(I,1),I=181,300)/
1.808,.815,.8245,.833,.8418,.852,.861,.871,.8811,.892,.903,.9142,
2.924,.935,.9428,.9505,.958,.968,.9748,.9784,.9808,.982,.9816,
3.9778,.974,.9695,.9646,.9598,.9543,.9487,
4.796,.804,.8121,.82,.8291,.838,.8465,.856,.8654,.8763,.8862,
5.8975,.9085,.922,.930,.939,.946,.9575,.9655,.971,.975,.9783,
6.9792,.977,.974,.970,.9665,.9623,.958,.9538,
7.792,.7981,.804,.812,.8201,.828,.8357,.846,.8545,.864,.874,
8.8837,.894,.905,.9146,.9247,.933,.95,.9613,.966,.970,.9755,.9768,
9.9782,.9770,.9742,.970,.966,.961,.956,
$.774,.780,.7871,.794,.801,.8087,.8165,.8247,.833,.841,.850,.8583,
$.868,.878,.887,.8962,.907,.9255,.944,.953,.9592,.966,.9705,.9725,
$.9718,.97,.9685,.9664,.964,.9616/
Y(X1,X2,X,Y1,Y2)=(Y1-Y2)*(X-X1)/(X1-X2)+Y1
N=0
DO 1 J=1,30
1  IF (PRATIO.GE.PR(J)) N=J-1
    IF (N.EQ.0) N=1
    IF (N.GE.29) N=28
    M=0
    DO 2 K=1,10
2  IF (ARATIO.GE.AR(K)) M=K
    IF (M.EQ.10.OR.M.EQ.0) GO TO 5
    DO 4 L=1,2
4  LL=M+L-1
    DO 3 I=1,3
3  I=I+1

```



```

NN=N+I-1
XX(I)=PR(NN)
3 YY(I)=CFR(NN,LL)
CALL PARABO (XX,YY,PRATIO,ANS)
4 ZZ(L)=ANS
CF=Y(AR(M),AR(M+1),ARATIO,ZZ(1),ZZ(2))
RETURN
5 WRITE (6,7) ARATIO,AR(10)
DO 6 I=1,3
NN=N+I-1
XX(I)=PR(NN)
6 YY(I)=CFR(NN,10)
CALL PARABO (XX,YY,PRATIO,CF)
RETURN
C
C
7 FORMAT (8HOARATIO=F7.3,4X,26HOUT OF RANGE,USE DATA FOR ,F7.3)
END

```

```

$IBFTC INPUT DECK
SUBROJTIME INPUT(LIJNIT,LDUNIT,ID,D,IT)
DIMENSION D(1), IT(1)
C
C CONSTRUCTION OF THE TABLE
C
C BIT STANDARD RETURN RETURN RETURN RETURN RETURN
C NUMBER RETURN 1 2 3 4 5
C 0-2 $ A-Z 0-9( ) = OTH
C 3-4 JETPC RI OTH NLSF
C 5-6 JER TINL OTH PS CF
C 7-8 0-9.+$)OTH ,A-Z= -(*/
C 9 ,.)A-Z OTH
C 9-10 ., A-Z) 0-9+^ =$
C 10 OTH $A-Z*)=/(
C 10-11 0-9.,-OTH +^( $A-Z* )=/(
C 12 0-9 A-Z OTH
C 12-13 A-Z 0-9 OTH
C 13-14 +- OTH 0-9
C 15-16 +. DE- OTH
C 17-18 T F OTH
C 19-20 D E OTH
C 21 JTH .
C 21-22 A-Z OTH
C 23 ' OTH
C 24 $ OTH
C 25 = OTH
C 26-27 ( A-Z OTH
C 28-29 ) , OTH
C 30 ( OTH
C 31 ) OTH
C ERROR CODE DESIGNATIONS
C ROUTINE TYPE
C 10 IXQTI 100 ILLEGAL CHARACTER
C 20 ITABLI 200 NAME TOO LONG
C 30 INMBRI 300 TABLE FULL OR BAD
C 40 INAMEI 400 SCALING ERRJR
C 50 IARITI 500 NAME NOT IN TABLE
C 60 INAMEN 600 $DATA( ) INCOMPLETE

```

```

C      70 INPUT  700 FORMULA ILL FORMED                                39
C      800 FUNCTION UNDEFINED                                          40
C      MCNVRT = TYPE OF LEFT HAND VARIABLE                              41
C      KCNVRT = TYPE OF CURRENT VARIABLE                                42
C      ITYPE  MEANING                                                  43
C      1      REAL                                                    44
C      2      INTEGER                                                  45
C      3      DOUBLE PRECISION                                         46
C      4      TYPELESS OR NO CONVERSION                                47
C      5      SUBROUTINE                                               48
C      6      FUNCTION                                                 49
C      FORMAT OF TABLE                                               50
C      1ST WJRD  012 3456          7-31                                51
C      TYPE  NUMBER OF WORDS ADDRESS                                  52
C      NEXT 1 TO 15 WORDS  THE NAME, 4 CHARACTERS TO THE WORD        53
C      DIMENSION IFT( 31), IPTAB( 21), ITAB( 65)                       54
C      DIMENSION ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)          55
C      *      ,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)              56
C      *      *      *      *      *      *      *      *      *      *      57
C      COMMON                                                         58
C      ./ICOMVI/ VALUE      ,ICOMP      ,IFNTYP      ,IMAGE1      ,IRADIX      ,ISUB      59
C      *      ,KCH          ,KCNVRT      ,KCOUNT      ,KDIF          ,KFLD1      ,KFLD2      60
C      *      ,LCOMP        ,LCNVRT      ,LEVEL        ,LFRT          ,LOOK       61
C      *      ,MCNVRT      ,MDIF        ,MODALL        ,MSTOR         62
C      *      ,NAME         ,VERROR      ,NONEW         ,NDTARG        63
C      *      ,SMCHR        ,TEST        ,ERMARK        64
C      ./ICNSTI/ BLANK      ,BLANKS      ,DOLLAR        ,EDS           65
C      *      ,ICOMMA      ,IDOLAR      ,IFT           ,IPTAB         ,ITAB       66
C      *      ,KAM10       ,KBPC        ,KBPW          ,KCPCD         ,KERTYP      67
C      *      ,KZERO       ,NOPRNT      ,TAB1          68
C      ./IPARAM/ ABORT      ,<IUNIT      ,KOUNIT        ,LIMALF        ,LOCK        ,LDCX      69
C      *      ,NOLIST      ,NSTDIR      ,TRACE         70
C      ./ISTAKI/ STACK      ,ISTDIM      ,KSTACK        ,LEVLIM        71
C      *      *      *      *      *      *      *      *      *      *      72
C      INTEGER BLANK      ,BLANKS      ,EDS           ,IDDLAR        ,TAB1        ,TRACE      73
C      ,CONTYP      ,STORED                                             74
C      DOUBLE PRECISION STACK, VALUE                                    75
C      LOGICAL ABURT                                                  76
C      *      *      *      *      *      *      *      *      *      *      77
C      LOGICAL DOLLAR                                               78
C      *      *      *      *      *      *      *      *      *      *      79
C      LOGICAL ERMARK                                               80
C      *      *      *      *      *      *      *      *      *      *      81
C      LOGICAL LIMALF                                               82
C      *      *      *      *      *      *      *      *      *      *      83
C      LOGICAL LOCK                                                 84
C      *      *      *      *      *      *      *      *      *      *      85
C      LOGICAL NOLIST                                               86
C      *      *      *      *      *      *      *      *      *      *      87
C      LOGICAL NONEW                                                88
C      *      *      *      *      *      *      *      *      *      *      89
C      LOGICAL MODALL                                               90
C      *      *      *      *      *      *      *      *      *      *      91
C      *      *      *      *      *      *      *      *      *      *      92
C      LOGICAL SHORT                                               93
C      *      *      *      *      *      *      *      *      *      *      94
C      *      *      *      *      *      *      *      *      *      *      95
C      LOGICAL SMCHR                                               96
C      *      *      *      *      *      *      *      *      *      *      97
C      LOGICAL TEST                                                98
C      *      *      *      *      *      *      *      *      *      *      99
C      END      *****                                              100

```

	EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE),(NAME,ANAME)	101
	EQUIVALENCE (ICOMNI,ISUB),(IMAGE,IMAGE1),(IPARAM,ABORT)	102
	LOGICAL END2	103
C	* .TRUE. IF A \$END CARD HAS ALREADY BEEN FOUND	104
	KIUNIT=LIUNIT	105
	KOUNIT=LOUNIT	106
C	CALL DEBUGX	107
	LIMALF = .TRUE.	108
	DOLLAR = .FALSE.	109
	NOLIST = .FALSE.	110
	END2 = .FALSE.	111
	MODALL = .FALSE.	112
	ERMARK = .FALSE.	113
	LOCK = .FALSE.	114
	KERTYP = 0	115
	IFNTYP = -1	116
	LOCX=1	117
	L=LEVLIM+3	118
	DO 16 I=1,L	119
16	KSTACK(I)=0	120
	LEVEL=0	121
	LFRT=0	122
	NOTARG = ISTDIM	123
	LCNVRT = 3	124
C	* DOUBLE PRECISION, STANDARD	125
	MCNVRT = 1	126
	KDIF = 1	127
	MDIF = 1	128
1	NONEW = .FALSE.	129
	SMCHR = .TRUE.	130
	CALL ICHAR2(\$9470,24)	131
C	* ERROR IF \$ IS NOT 1ST CHARACTER READ	132
	LDOLLAR = KCOUNT	133
C	* CARD COLUMN OF LAST DOLLAR SIGN	134
	NONEW=.TRUE.	135
C	* \$DATA MUST BE ON A SINGLE CARD	136
C	D E OTH	137
	CALL ICHAR1(\$3,\$9470,19)	138
C	( A-Z OTH	139
4	CALL ICHAR1(\$4,\$9470,26)	140
	CALL ISUBI	141
	CALL ICHAR2(\$9470,31)	142
C	* ERRDR IF NO )	143
	NONEW=.FALSE.	144
	IT(1)=ISUB	145
C	CALL DEBUG 2 (5HSETNO,IT(1))	146
C	CALL DEBUB 2(6HKERSIN,KERSIN)	147
	IF (ID.NE. IT(1)) GO TO 99	148
C	* RETURN BECAUSE WRONG DATA SET	149
	IF (IT(2).LT. 0)GO TO 9370	150
C	* TABLE FULL OR BAD	151
	GO TO 19	152
3	IF (END2) GO TO 99	153
	END2 = .TRUE.	154
	NONEW = .FALSE.	155
5	CALL ICHAR2(\$1,9)	156
C	* PASS LETTERS AND ..)	157
	GO TO 5	158
9470	KERTYP = 470	159
	GO TO 999	160
99	CONTINUE	161

C	CALL DEBUG 2(6HSTATMT,99)	162
	KCH = IDOLAR	163
	KCOUNT = LDOLAR	164
C	* NEXT CALL BEGINS PROCESSING AT THIS CARD COLUMN	165
	IF(ERMARK) IT(1) = -IABS(IT(1))	166
C	* WARN PROGRAMER OF POSSIBLE ERRO	167
	IF(ERMARK.AND.ABORT) GO TO789	168
	RETURN	169
789	WRITE(6,787)	170
787	FORMAT(44H ERROR HAS OCCURED AND ABORT IS .TRUE., STOP)	171
	STOP	172
C	BEGIN TO INPUT THE DATA	173
C	STATEMENTS 19 AND 20 ARE THE SWITCHHOUSE	174
C	CONTROL COMES HERE FOR NEW DIRECTION.	175
C	LFRT = 0 INDICATES THAT THE PREVIOUS TASK WAS COMPLETED.	176
19	CONTINUE	177
C	CALL DEBUG2(6HSTATMT,19)	178
	TEST=.FALSE.	179
	GO TO 21	180
120	DOLLAR = .FALSE.	181
20	CONTINUE	182
C	CALL DEBUG 2(6HSTATMT,20)	183
	TEST=.TRUE.	184
C	ENTRY ICHAR4 LOADS LCOMP WITH TAB NO 10 AND RETURNS ON TAB NO 7	185
C	OTH , 'A-Z= -(*/	186
21	CALL ICHAR4(\$31, \$32, 7,10)	187
C NOW	WE TEST LCOMP ON EACH OF THE 3 POSSIBLE RETURNS. THIS IS A 12 WAY	188
C	0-9. + \$ )	189
	GO TO (203,510, 64,430),LCOMP	190
C	, ' A-Z =	191
31	GO TO (460,202,530,450),LCOMP	192
C	- ( * /	193
32	GO TO(520,420,440,441),LCOMP	194
64	IF(LFRT.NE.0) GO TO 9770	195
	LDOLAR = KCOUNT	196
C	* CARD COLUMN OF LAST DOLLAR SIGN	197
	DOLLAR = .TRUE.	198
C	CALL DEBUG 2 (6HSTATMT,64)	199
	TEST=.FALSE.	200
	CALL ICHAR4(\$9170,\$9170,3,5)	201
C	DE T P C	202
	GO TO(99,100,150,630),LCOMP	203
100	CALL ITABLI(IT)	204
	IF (KERTYP) 999,120,999	205
C		206
C		207
C	\$PARAMETER	208
C	( A-Z DTH	209
150	CALL ICHAR1(\$150,\$9170,26)	210
151	CALL ICHAR2(\$152,0)	211
C	* GET FIRST CHAR OF NAME ..DON'T BRAN	212
152	CALL INAMEN	213
	CALL ILOOKI(\$153,IPTAB)	214
9570	KERTYP = 570	215
	GO TO999	216
153	CALL ICHAR2(\$9170,29)	217
C	* ERROR IF NO =	218
	CALL ICHAR2(\$9170,10)	219
C	* GET CHARACTER,SEPERATE PART OF	220
	CALL INMBRI	221
	KCNVRT=IFLD(0,3,IPTAB(LOOK))	222
	CALL ICNVTI(3,KCNVRT)	223

```

C                                * CONVERT FROM DP TO TYPE OF VARIABLE          224
    IPT=IFLD(7,25,IPTAB(LOOK))      225
    IPARAM(IPT) = KVALUE             226
C                                ) , OTH                                     227
    CALL ICHARI($151,$9170,28)     228
C                                IF(TRACE.GT.0) CALL DEBUGX                   229
C                                * IF TRACE TURNED ON PUT ON HEAD             230
    GO TO 120                        231
C                                * GO TO SWITCH HOUSE TEST FOR INSERTIO      232
C                                233
C                                234
C    CONSTANTS = LOGICAL,NUMERIC,ALPHAMERIC,RADIX.                         235
201  ASSIGN 220 TO CONTYP           236
C                                * RADIX                                     237
    IFNTYP = -1                      238
    GO TO 210                          239
202  ASSIGN 230 TO CONTYP           240
C                                * *                                       241
    GO TO 210                          242
203  ASSIGN 250 TO CONTYP           243
C                                * NUMERIC,LOGICAL                          244
C                                245
C                                246
C    ALL CONSTANTS                210                                       247
210  LOP = 0                          248
    SHORT = .TRUE.                    249
    MSTOR = 0                          250
C    CALL DEBUG2(6HST CON,210)       251
    IF(LEVEL.EQ.0) LEVEL=3            252
    IF(KSTACK( LEVEL).EQ.1) GO TO 9770 253
C                                * TWO CONSTANTS IN ROW E.G.                 254
    GO TO CONTYP,(220,230,250)       255
C                                256
C                                257
C    RADIX CONSTANTS                220                                       258
C                                ( A-Z OTH                                     259
220  CALL ICHAR2($9170,30)           260
    CALL ISUBI                         261
    NAME(2) = 0                        262
221  CALL ICHAR2($2215,29)           263
C                                * ERROR IF NO COMMA AFTER BASE NUMBER      264
    GO TO 9170                          265
2215 CALL ICHAR2($9170,12)           266
C                                * ERROR IF NO NUMBER                         267
222  IRADIX = ISUB                    268
    SMCHR = .TRUE.                     269
223  CALL ISUBI                        270
    MSTOR = MSTOR +1                   271
    NAME(MSTOR) = ISUB                 272
    IF (MSTOR.GE.15) GO TO 225         273
C                                ) , OTH                                     274
224  CALL ICHARI($223,$9170,28)     275
    IRADIX = 10                        276
    GO TO 240                          277
225  ASSIGN 224 TO NEXT              278
    GO TO 241                          279
C                                280
C                                281
C    HOLLERITH CONSTANTS            230                                       282
230  MODALL = .TRUE.                  283
    TEST = .FALSE.                     284
    NAME(2) = BLANKS                   285

```

231	CALL INAMEN	286	
	IF(.NOT.MODALL)GO TO 240	287	
C		*END OF CONSTANT	288
	IF(L1MALF) GO TO 265	289	
C	LONG CONSTANT GOES TO 234	290	
234	ASSIGN 231 TO NEXT	291	
	GO TO 241	292	
C		293	
C		294	
C	STORE ALF + RADIX 240	295	
240	ASSIGN 260 TO NEXT	296	
	IF (MSTOR-2) 242,2405,241	297	
2405	IF (LFRT.EQ.1) GO TO 242	298	
	CALL ICHAR2(\$2406,29)	299	
	SMCHR = .TRUE.	300	
	GO TO 242	301	
2406	SMCHR = .TRUE.	302	
241	SHORT = .FALSE.	303	
	IF (LFRT.NE.0) GO TO 265	304	
C	CALL DEBUG2(6HST 241,0)	305	
242	KVALUE = NAME(1)	306	
	RVALUE(2) = ANAME(2)	307	
	IF (.NOT.ARG.GT.LEVEL) LCNVRT = 4	308	
C	CALL DEBUG2(6HST 242,0)	309	
C	CALL DEBUG2(6HSHORT ,SHORT)	310	
C	CALL DEBUG2(6HLFRT ,LFRT)	311	
243	IF (SHORT) GO TO 255	312	
	DO 245 I= 1,MSTOR	313	
	IF (LOCK) GO TO 245	314	
	D(LOCK) = ANAME(I)	315	
245	LOCK = LOCK + 1	316	
246	GO TO NEXT, (224,231,260)	317	
C	SEE NEXT AND MEANING	318	
C	250 * ALF OR ) AND RADIX	319	
C	224 MORE THAN 15 ELEMENTS IN RADIX FIELD	320	
C	231 MORE THAN 15 ELEMENTS IN ALF FIELD	321	
C		322	
C		323	
C	NUMERIC + LOGICAL 250	324	
250	TEST = .TRUE.	325	
	CALL INMBRI	326	
255	CALL IARITI	327	
	LFRT = 1	328	
	IF (KERTYP) 999,20,999	329	
C		330	
C		331	
C	RESET STACK BECAUSE IT WAS NOT USED	332	
260	LEVEL = 0	333	
	TEST = .TRUE.	334	
	CALL ICHAR2(\$19,29)	335	
C		* SKIP COMMA	336
	KERTYP = 171	337	
	GO TO 999	338	
265	KERTYP = 270	339	
	GO TO 999	340	
C		341	
C	TEST EMPTY PARENTHESES	342	
400	IF ( (KSTACK( LEVEL-1)+KSTACK( LEVEL-2) ).NE.0) GO TO 997	343	
C	CALL DEBUG 2(6HSTA ( ),400)	344	
C	CALL STACKP	345	
C		PRINT STACK	346
401	LEVEL=LEVEL-3	347	

C	EMPTY FUNCTION ARGUMENT IS NOT A CURRENT LEFT SIDE	348
	IF (KSTACK( LEVEL).LT.6) GO TO 404	349
405	VALUE=0.	350
	GO TO 403	351
404	RVALUE(1) = D (LOCK)	352
	RVALUE(2) = D(LOCK+1)	353
	CALL ICNVTI(MCNVRT,3)	354
403	CALL IARITI	355
	GO TO 20	356
C((((((((((((((((((((		357
420	IF (LEVEL.EQ.0) LEVEL=3	358
C	CALL DEBUG 2(6HSTAT (,420)	359
422	LOP=0	360
	LFRT = 1	361
	LEVEL=LEVEL+3	362
C	CALL STACKP	363
C	PRINT STACK	364
	IF (KSTACK( LEVEL-3)-1) 20, 997, 20	365
C))))))))))))))))))		366
430	IF (LEVEL.LT.6) GO TO 997	367
C	CALL DEBUG 2(6HSTAT ),430)	368
C	CALL STACKP	369
C	PRINT STACK	370
	LFRT = 1	371
431	IF (KSTACK( LEVEL)-1) 400, 432, 997	372
432	DO 433 I=1,3	373
	VALUE=STACK( LEVEL)	374
	KSTACK( LEVEL)=0	375
	LEVEL=LEVEL-1	376
433	CALL IARITI	377
	IF (KERTYP) 999, 20, 999	378
C*****KOP=2)		379
440	KOP=2	380
C	CALL DEBUG 2(6HSTAT *,440)	381
	GO TO 445	382
C//////////KOP=3)		383
441	KOP = 3	384
C		385
C	CALL DEBUG 2(6HSTAT /,441)	386
445	IF (LJP.NE.0) GO TO 997	387
C	CALL DEBUG 2(6HSTATMT,445)	388
C	CALL STACKP	389
C	PRINT STACK	390
	LOP = 1	391
	LFRT = 1	392
	IF (LEVEL.EQ.0) LEVEL=3	393
	IF (KSTACK( LEVEL).NE.1) GO TO 997	394
444	VALUE=STACK( LEVEL)	395
	KSTACK( LEVEL)=0	396
	LEVEL=LEVEL-1	397
	CALL IARITI	398
	IF (KERTYP.NE.0) GO TO 999	399
446	KSTACK( LEVEL)=KOP	400
	LEVEL=LEVEL+1	401
C	CALL STACKP	402
C	PRINT STACK	403
	GO TO 19	404
C=====		405
450	IF (LEVEL.NE.0) GO TO 997	406
C	CALL DEBUG 2(6HSTAT =,450)	407
451	LOP=0	408
	LOCK = LOOK	409

```

MCNVRT = KCNVRT                                410
MDIF = KDIF                                    411
LCNVRT = 3                                     412
C                                                413
      * DOUBLE PRECISION, STANDARD              414
LEVEL=3                                         415
LFRT=0                                          416
KSTACK( 3)=0                                   417
C CALL JDEBUG2(6HLOCX ,LOCX)                   418
C CALL DEBUG2(6HMCNVRT,MCNVRT)                 419
GO TO 19                                        420
C,,,,,
460 CONTINUE                                   421
C CALL DEBUG 2(6HSTAT ,,460)                   422
C CALL STACKP                                  423
C                                                424
      * PRINT STACK                             425
IF (LEVEL - 3) 461,463, 600                   426
461 IF (LEVEL.NE.0) GO TO 997                   427
462 LEVEL=3                                     428
463 IF(LFRT.EQ.0)GO TO 480                     429
C                                                430
      * INCREMENT LOCX WITHOUT STORING          431
C LFRT WILL BE ZERO AFTER $ EXPRESSION OR FOR CONSECUTIVE COMMAS 432
470 IF (KSTACK( 3).NE.1) GO TO 997             433
471 DO 475 I = 1,2                             434
      VALUE=STACK( LEVEL)                      435
      KSTACK( LEVEL)=0                        436
      LEVEL=LEVEL-1                           437
475 CALL IARITI                                 438
IF (KERTYP.NE.0) GO TO 999                     439
476 KSTACK( 1)=0                               440
CALL ICNVTI (LCNVRT,MCNVRT)                   441
474 IF(LOCK) GO TO 480                         442
D(LOCK) = RVALUE(1)                           443
IF (MCNVRT .EQ. 3) D(LOCX+1) = RVALUE(2)     444
480 LOP=0                                       445
C CALL STACKP                                  446
C                                                447
      * PRINT STACK                             448
LFRT=0                                          449
LOCX = LOCX + MDIF                             450
LEVEL=0                                         451
LCNVRT = 3                                     452
C                                                453
      * DOUBLE PRECISION, STANDARD              454
GO TO 19                                        455
C+++++(KOP=4)                                  456
510 KOP=4                                       457
C CALL JDEBUG 2(6HSTAT +,510)                  458
GO TO 521                                       459
C------(KOP=5)                               460
520 KOP=5                                       461
C CALL DEBUG 2(6HSTAT -,520)                   462
521 IF (LJP.NE.0) GO TO 997                   463
522 LOP=1                                       464
LFRT = 1                                       465
IF (LEVEL.EQ.0) LEVEL=3                       466
524 IF (KSTACK( LEVEL)-1) 525, 526, 997      467
525 STACK( LEVEL)=0. DO                       468
C CALL DEBUG 2(6HSTATMT,525)                  469
526 DO 528 I=1,2                               470
C CALL DEBUG 2(6HSTATMT,526)                  471
      VALUE=STACK( LEVEL)
      KSTACK( LEVEL)=0
      LEVEL=LEVEL-1

```



528	CALL IARITI	471
	IF (KERTYP.NE.0) GO TO 999	472
527	KSTACK( LEVEL)=KOP	473
	LEVEL=LEVEL+2	474
	GO TO 19	475
C	ABCDEFGHIJKLMN OPQRSTUVWXYZABCDEFGHIJKLMN OPQRS	475
530	LDP=0	477
C	CALL DEBUG 2(6HSTAT A,530)	478
	IF ( (LEVEL.NE.0).AND.(KSTACK( LEVEL).EQ.1) ) GO TO 997	479
535	TEST = .TRUE.	480
	CALL INAMEI(\$999,D,IT)	481
	IF (IFNTYP) 531,621,640	482
531	LFRT = 1	483
	IF(LEVEL.NE.0)GO TO 540	484
532	STACK( 3)=VALUE	485
C	CALL DEBUG 2(6HSTATMT,532)	486
	KSTACK( 3)=1	487
C	CALL STACKP	488
C	PRINT STACK	489
	GO TO 20	490
540	CONTINUE	491
C	CALL DEBEG 2(6HSTATMT,540)	492
	CALL IARITI	493
	IF(KERTYP) 999,20,999	494
C		495
C	PROCESS SUBROUTINES AND FUNCTIONS	496
C		497
C	COMMA SEPARATING FUNCTION ARGUMENTS	498
600	IF (KSTACK( LEVEL-3).LT.6) GO TO 997	499
C	CALL DEBUG 2(6HSTAT F,600)	500
601	IF (KSTACK( LEVEL)-1) 602, 603, 997	501
602	STACK( LEVEL)=0.	502
603	DO 610 I=1,2	503
	VALUE = STACK( LEVEL)	504
	KSTACK( LEVEL)=0	505
	LEVEL=LEVEL-1	506
610	CALL IARITI	507
	IF (KERTYP.NE.0) GO TO 999	508
611	KSTACK( LEVEL)=KSTACK( LEVEL-1)+1	509
	LEVEL=LEVEL+3	510
C	CALL STACKP	511
C	PRINT STACK	512
	GO TO 19	513
C	\$CALL	514
630	DO 631 I = 1,4	515
C	SKIP 'ALL' IN \$CALL	516
C	OTH OPERATORS	517
	CALL ICHAR2(\$9170,12)	518
631	CONTINJE	519
	DOLLAR = .FALSE.	520
	TEST = .TRUE.	521
	CALL INAMEI (\$999,D,IT)	522
	IF (IFNTYP .NE. 1) GO TO 640	523
	KVALUE = -KVALUE	524
C	* INDICATE THAT NO RESULT IS TO BE STORED FOR SJB	525
C	FUNCTION NAME	526
621	CONTINJE	527
C	* FUNCTIONS	528
C	CALL DEBUG2(6HST FNC,621)	529
C	CALL DEBEG2(6HIFNTYP,IFNTYP)	530

C	CALL DEBUG2(6HKVALUE,KVALUE)	531
	IFNTYP = -1	532
	IF (LEVEL.EQ.0) LEVEL = 3	533
	IF(NOTARG .GT. LEVEL) NOTARG = LEVEL	534
	LFRT = 1	535
	STACK(LEVEL)=KVALUE	536
	KSTACK( LEVEL)=6	537
	LEVEL = LEVEL + 3	538
C	CALL STACKP	539
C	PRINT STACK	540
C	( OTH	541
	CALL ICHAR2(\$622,30)	542
	GO TO 19	543
C	* IF THERE ARE ARGUMENTS	544
622	SMCHR = .TRUE.	545
	IF (KVALUE .LT. 0) LEVEL = LEVEL -3	546
	GO TO 405	547
640	IF (IFNTYP - 2) 9770,201,700	548
C	LOCX THE SUBROUTINE THAT STORES LOCX THE SUBSCRIPT OF D ARR	549
700	IF (IFNTYP .NE. 3) GO TO 9770	550
	IFNTYP = -1	551
	IF(LEVEL.NE.0) GO TO 9770	552
	LFRT = 0	553
	CALL ICHAR2(\$9170,30)	554
C	* ERROR IF NO ( AFTER LOCX	555
	ASSIGN 704 TO NEXT	556
C	A-Z OTH	557
702	CALL ICHAR2(\$9170,9)	558
	CALL INAMEI(\$999,D,IT)	559
	IF (IFNTYP .GT. -1) GO TO 9770	560
	GO TO NEXT,(704,708)	561
704	LOOKX = LOOK	562
C	OTH ,	563
	CALL ICHAR2(\$706,29)	564
C	* ERROR IF NO COMMA	565
	GO TO 9170	566
706	ASSIGN 708 TO NEXT	567
	GO TO 702	568
708	KVALUE = LOCX - LOOKX + 1	569
	CALL ICNVTI(2,KCNVRT)	570
C	* CONVERT FROM INT TO TYPE OF 2ND ARG	571
	IF(LOOK) GO TO 710	572
	D(LOOK) = RVALUE(1)	573
710	CALL ICHAR2(\$9170,31)	574
C	* ERROR IF NO )	575
	TEST = .TRUE.	576
C	OTH ,	577
	CALL ICHAR2(\$19,29)	578
C	* SKIP THE COMMA THAT MUST FOLLOW	579
9170	KERTYP=170	580
	GO TO 999	581
9370	KERTYP=370	582
	GO TO 999	583
997	CONTINJE	584
9770	KERTYP = 770	585
	GO TO 999	586
999	CALL IERORI	587
	LFRT = 0	588
	LOP = 0	589
	NAME (1)=0	590
	GO TO 19	591
C	* NOW GO TO SWITCH HOUSE .. GOOD	592
	END	593

```

$IBFTC BLOCK DECK
BLOCK DATA
C ICOMVI BLOCK DATA PROGRAM
DIMENSION IFT( 31), IPTAB( 21), ITAB( 65)
DIMENSION ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)
. ,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)
C
COMMON
./ICOMVI/ VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB
. ,KCH ,KCNVRT ,KCOUNT ,KDIF ,KFLD1 ,KFLD2
. ,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK
. ,MCNVRT ,MDIF ,MODALL ,MSTOR
. ,NAME ,NERROR ,NONEW ,NOTARG
. ,SMCHR ,TEST ,ERMARK
./ICNSTI/ BLANK ,BLANKS ,DOLLAR ,EOS
. ,ICOMMA ,IDOLLAR ,IFT ,IPTAB ,ITAB
. ,KAM10 ,KBPC ,KBPW ,KCPCD ,KERTYP
./IPARAM/ ABORT ,KIUNIT ,KOUNIT ,LIMALF ,LOCK ,LOCKX
. ,NOLIST ,NSTDIR ,TRACE
./ISTAKI/ STACK ,ISTDIM
C
INTEGER BLANK ,BLANKS ,EOS ,IDOLLAR ,TAB1 ,TRACE
DOUBLE PRECISION STACK, VALUE
LOGICAL ABORT,DOLLAR,ERMARK,LIMALF,LOCK,NOLIST,NONEW,MODALL,
.NSTDIR,SMCHR,TEST
DATA NERROR,TRACE/0,0/
DATA ITAB/ 65* 20572347824/
DATA ITAB( 49)/-11890146992/
. , ITAB( 44)/ 11923699376/, ITAB( 61)/ 29355228304/
. , ITAB( 29)/ 29053238816/, ITAB( 12)/ -3484760752/
. , ITAB( 28)/-11821997744/, ITAB( 17)/-11903778480/
. , ITAB( 33)/-12155961008/, ITAB( 45)/-12192136880/
. , ITAB( 50)/-12208914096/, ITAB( 13)/-12041137840/
. , ITAB( 60)/-11957255792/, ITAB( 11)/-11890146992/
. , ITAB( 1)/ 29063724720/, ITAB( 2)/ 29063724720/
. , ITAB( 3)/ 29063724720/, ITAB( 4)/ 29063724720/
. , ITAB( 5)/ 29063724720/, ITAB( 6)/ 29063724720/
. , ITAB( 7)/ 29063724720/, ITAB( 8)/ 29063724720/
. , ITAB( 9)/ 29063724720/, ITAB( 10)/ 29063724720/
. , ITAB( 18)/ 20572347824/, ITAB( 20)/ 18961735088/
. , ITAB( 21)/ 17350532528/, ITAB( 22)/ 17350565296/
. , ITAB( 23)/ 23256571312/, ITAB( 26)/ 20035476912/
. , ITAB( 36)/ 22182960560/, ITAB( 38)/ 22182960560/
. , ITAB( 40)/ 18424854176/, ITAB( 42)/ 19498606000/
. , ITAB( 51)/ 22719831472/, ITAB( 52)/ 17887731120/
DATA BLANK,BLANKS,EOS,ICOMMA,IDOLLAR,KAM10,KBPC,KBPW,KCH,KCPCD,
. KZERJ,NOPRNT,TAB1,IMAGE1(2),IMAGE1(3)/
. 48,-17997958192, 10, 59, 43,
. 7, 6, 36, 43, 80,
. 0, 11555507248, 64, 11555507248, 11555507248/
DATA ABORT,IMAGE1(1),IMAGE1(81),IRADIX,ISTDIM,
. KCOUNT,KDIF,KSTACK,LEVLIM,NSTDIR,STACK/
. .FALSE.,1HE,1H ,10,27,0,1,27*0,24,.TRUE.,27*0.D0/
DATA IFT/
. 0, 31,-17716740112, 21051935792,-17716740128,
. 23481748528,-17716740144, -3864988720,-17716740160,-19756682288,
. -17715740176,-20009401392,-17716740192, 19113857200,-17716740208,
. -8210767088,-17716740224, -8428915760,-17716740240, -8433110064,
. -17715740256, 18568907824,-17716740272, 21907455024, -9126805536,
. -9954237936, -9126805552, -3863968816, -9126805696, -3863883248,
. 0/

```

```

DATA IPTAB/
.      0,          21,  -536870928, 18565733616,  -536870976,      62
. -3650164950,  -536870992, -3863882800,  -536871024,  -6015524019,      63
.  -535871040, -6221022825, 17716740128, -2580698739, 17716740144,      64
. -2793802547, 17716740192, -3863968816, 17716740240, -21093496176,      65
.      0/
END

```

```

$IBFTC IARITI DECK
C      ARITHMETIC OPERATIONS FOR INPUT R. U. A. S. 2      1
      SUBROUTINE IARITI      2
C      CALLED FROM INPUT      3
      DIMENSION IFT( 31), IPTAB( 21), ITAB( 65)      4
      DIMENSION ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)      5
.      ,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)      6
C      COMMON      7
      /ICOMNI/ VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB      8
.      ,KCH ,KCNVRT ,KCOUNT ,KDIF ,KFLD1 ,KFLD2      9
.      ,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK      10
.      ,MCNVRT ,MDIF ,MODALL ,MSTOR      11
.      ,NAME ,NERROR ,NONEW ,NOTARG      12
.      ,SMCHR ,TEST ,ERMARK      13
.      ,ICNSTI/ BLANK ,BLANKS ,DOLLAR ,EOS      14
.      ,ICJMA ,IDOLAR ,IFT ,IPTAB ,ITAB      15
.      ,KAMIO ,KBPC ,KBPW ,KCPCD ,KERTYP      16
.      ,KZERO ,NOPRNT ,TAB1      17
.      /IPARAM/ ABORT ,KIUNIT ,KOUNIT ,LIMALF ,LOCK ,LOCX      18
.      ,NOLIST ,NSTDIR ,TRACE      19
.      /ISTACK/ STACK ,ISTDIM ,KSTACK ,LEVLIM      20
C      INTEGER BLANK ,BLANKS ,EOS ,IDOLAR ,TAB1 ,TRACE      21
      DOUBLE PRECISION STACK, VALUE      22
      LOGICAL ABORT,DOLLAR,ERMARK,LIMALF,LOCK,NOLIST,NONEW,MODALL,      23
.      NSTDIR,SMCHR,TEST      24
      EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE),(NAME,ANAME)      25
      EQUIVALENCE (ICOMNI,ISUB),(IMAGE,IMAGE1),(IPARAM,ABORT)      26
C      CALL DEBUGC(6HIARITI)      27
C      CALL STACKP      28
C      PRINT STACK      29
C      IF (KERTYP.NE.0) GO TO 100      30
      IF ( (LEVEL.LE.0).OR.(LEVEL.GT.LEVLIM) ) GO TO 120      31
C      BRANCH ON KSTACK( LEVEL)=-, 0, 1, 2, 3, 4, 5, 6, 7, 8 AND UP      32
      K=MAX(1,KSTACK( LEVEL)+2)      33
      IF(K.GE.8) GO TO 60      34
C      - 0 1 2 3 4 5 =KSTACK(LEVEL)      35
      3 GO TO (120,90, 120, 20, 30, 40, 50),K      36
      20 VALUE=STACK( LEVEL)*VALUE      37
      GO TO 90      38
      30 VALUE=STACK( LEVEL)/VALUE      39
      GO TO 90      40
      40 VALUE=STACK( LEVEL)+VALUE      41
      GO TO 90      42
      50 VALUE=STACK( LEVEL)-VALUE      43
      GO TO 90      44
      60 LEVEL1= LEVEL-K+9      45
      DO 61 I=LEVEL1,LEVEL      46

```

```

61  KSTACK( I)=0
    LEVEL = LEVEL1-1
C   CALL DEBUG 2(5HLEVEL,LEVEL)
C   CALL DEBUG 2(5HVALUE,VALUE)
    IF (LOCK) GO TO 62
    CALL IXQTI(VALUE,STACK( LEVEL))
62  CONTINUE
C   CALL DEBUG 2(5HVALUE,VALUE)
    IF (LEVEL .LE. NOTARG) NOTARG = ISTDIM
    IF (STACK(LEVEL).LT.0.DO) GO TO 110
    GO TO 90
90  STACK( LEVEL)=VALUE
    KSTACK( LEVEL)=1
100 CONTINUE
C   CALL STACKP
C
C           PRINT STACK
C   CALL DEBUGR
    RETURN
110 LFRT=0
C
C           * SPECIAL TREATMENT FOR SUBROUTINES
C   CALL DEBUG2(6HSTATMT,110)
    KSTACK(LEVEL)=0
    LEVEL=0
C
    OTH ,
    CALL ICHAR2($100,29)
    KERTYP = 150
    GO TO 100
120 KERTYP=750
    GO TO 100
    END

```

```

$IBFTC ICHAR4 DECK
    SUBROUTINE ICHAR4(*,*,LIST1,LIST2)
    DIMENSION IFT( 31), IPTAB( 21), ITAB( 65)
    DIMENSION ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)
    , KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)
C
    COMMON
    ,/ICOMVI/ VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB
    , ,KCH ,KCNVRT ,KCOUNT ,KDIF ,KFLD1 ,KFLD2
    , ,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK
    , ,MCNVRT ,MDIF ,MODALL ,MSTOR
    , ,NAME ,NERROR ,NONEW ,NOTARG
    , ,SMCHR ,TEST ,ERMARK
    ,/ICNSTI/ BLANK ,BLANKS ,DOLLAR ,EJS
    , ,ICOMMA ,IDOLLAR ,IFT ,IPTAB ,ITAB
    , ,KAMIO ,KBPC ,KBPW ,KCPCD ,KERTYP
    , ,KZERO ,NOPRNT ,TAB1
    ,/IPARAM/ ABORT ,KIUNIT ,KOUNIT ,LIMALF ,LOCK ,LOCX
    , ,NOLIST ,VSTDIR ,TRACE
    ,/ISTACK/ STACK ,ISTDIM ,KSTACK ,LEVLIM
C
    INTEGER BLANK ,BLANKS ,EOS ,IDOLLAR ,TAB1 ,TRACE
    ,COMPR
    DOUBLE PRECISION STACK, VALUE, IDEBUA, DEBGNA(2)
    DATA DEBGNA/6H1CHAR ,5H1CHR/
    LOGICAL ABORT,DOLLAR,ERMARK,LIMALF,LOCK,NOLIST,NONEW,MODALL,
    ,NSTDIR,SMCHR,TEST

```

	EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE),(NAME,ANAME)	27
	EQUIVALENCE (ICOMNI,ISUB),(IMAGE,IMAGE1),(IPARAM,ABORT)	28
	LOGICAL GOTCD	29
C	* TRUE IF NEW CARD WAS READ	30
	LOGICAL SMCHR	31
C	* RE-PROCESS THE SAME CHARACTER AS LAS	32
	KFLD2=LIST2	33
	IBITS=2	34
	ASSIGN 36 TO COMPR	35
	IDEBUN = 4	35
	GO TO 10	37
	ENTRY ICHARI(*,*,LIST1)	38
	IBITS = 2	39
	IDEBUN = 3	40
	GO TO 4	41
	ENTRY ICHAR2(*,LIST1)	42
	IBITS=1	43
	IDEBUN = 2	44
4	KFLD2 = -1	45
	ASSIGN 37 TO COMPR	46
	IF(MODALL)GO TO 40	47
10	ASSIGN 20 TO IGETR	48
12	KFLD1=LIST1	49
	IDEBUA = DEBGNA(1)	50
	ASSIGN 110 TO NEXT	51
	GOTCD=.FALSE.	52
	IF(SMCHR) GO TO 35	53
C	* SKIP GETTING NEW CHARACTER	54
	GO TO 200	55
C	* GET CHARACTER RETURN TO 20	56
20	IF(KCH.EQ.BLANK)GO TO 200	57
C	* BYPASS BLANKS	58
	IF (KCH.NE.EOS) GO TO 30	59
C	* EOS= END OF STATEMENT CHARACTER	60
	ASSIGN 200 TO INEWR	61
C	* PROVIDE TO GET FIRST CHARACTER FROM NEXT C	62
	GO TO 300	63
30	IF(TEST.AND.GOTCD)GO TO 100	64
35	ITEMP = ITAB(KCH+1)	65
	ICOMP=IFLD(LIST1,IBITS,ITEMP)	66
	GO TO COMPR,(36,37)	67
36	LCOMP=IFLD(LIST2,2,ITEMP)+1	68
37	CONTINUE	69
	IF(TRACE .LT.4) GO TO 38	70
C	CALL DEBUG2(IDEBUA,IDEBUN)	71
38	SMCHR = .FALSE.	72
	IF(ICJMP-1) 381,382,383	73
381	RETURN	74
382	RETURN1	75
383	RETURN2	76
40	ASSIGN 35 TO IGETR	77
	GO TO 12	78
C	- - - - - ITCR - - - - -	79
C	SPECIAL ROUTINE TO INSERT COMMA AT END OF CARD IF	80
C	THE NEXT CARD BEGINS WITH \$ OR A LEFT MEMBER	81
C	TO THIS ROUTINE, A LEFT MEMBER BEGINS WITH A-Z	82
C	FOLLOWED BY ANY OF 0-9()A-Z FOLLOWED BY =	83
100	IDEBUA = DEBGNA(2)	84
	ITEMP = ITAB(KCH+1)	85
	ICOMP=IFLD(0,3,ITEMP)	86
	GO TO NEXT,(110,120)	87

```

C          * 110 FOR 1ST CHR ON NEW CARD, 120 FOR FOLL      88
110  NONE#=.TRUE.                                           89
    ASSIGN 120 TO NEXT                                       90
C          $ A-Z 0-9() = OTH                                 91
    GO TO(130,200,140,140,140),ICOMP                       92
C                                                     93
C          $ A-Z 0-9() = OTH                                 94
120  GO TO(140,200,200,130,140),ICOMP                       95
130  KCOUNT=0                                              96
    KCH=ICOMMA                                              97
    NONE#=.FALSE.                                          98
    GOTCD = .FALSE.                                       99
    GO TO 35                                               100
C          * COMMA IS CHARACTER RETURNED                    101
140  KCOUNT=0                                              102
    NONE#=.FALSE.                                          103
    GOTCD = .FALSE.                                       104
    GO TO 200                                              105
C          * FIRST NON-BLANK CHARACTER ON CARD IS RETURNED 106
C - - - - - - - - - - - - - - - - - - - - - - - - - - - 107
C          ROUTINE TO GET NEXT CHARACTER                    108
C                                                     109
200  IF(KCOUNT.LT.KCPCD)GO TO 210                          110
    ASSIGN 210 TO INEWR                                     111
    GO TO 300                                               112
210  KCOUNT=KCPCD+1                                       113
C          * CARD COLUMN OF NEW CHARACTER                  114
    KCH=IFLD(0,KBPC,IMAGE(KCOUNT))                       115
    GO TO IGETR,(20,35)                                     116
C          * 35 IS USED ONLY FOR MODALL .TRU              117
C - - - - - - - - - - - - - - - - - - - - - - - - - - - 118
C          ROUTINE TO PRINT OLD CARD AND READ NEXT        119
C          IF NONEW IS TRUE, STORE TAB1 IN KCH AND RETURN 120
300  IF(NONEW)GO TO 310                                     121
    KCH=IMAGE(KCOUNT+1)                                    122
C          * GET CARRIAGE CONTROL (IMAGE(81) IS A BLANK) 123
    IF((KCH.EQ.NOPRNT).OR.NOLIST) GO TO 305                124
    WRITE(KOUNT,398)KCH,IMAGE                               125
398  FORMAT(1A1,5X,80A1)                                     126
305  READ(KIUNIT,399)IMAGE                                  127
399  FORMAT(80A1)                                           128
    GOTCD=.TRUE.                                           129
    KCOUNT=0                                              130
    GO TO INEWR,(200,210)                                    131
310  KCH=TAB1                                              132
    GO TO 30                                               133
C          * TAB1 IS RETURNED, INDICATES END OF C         134
    END                                                     135

```

```

$IBFTC ICNVTI DECK
SUBROJTINE ICNVTI(IFROM,ITO)                                1
DIMENSION IFT( 31), IPTAB( 21), ITAB( 65)                  2
DIMENSION ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)      3
.      ,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)          4
C                                                     5
COMMON                                                      6
./ICOMVI/ VALUE      ,ICOMP      ,IFNTYP      ,IMAGE1      ,IRADIX      ,ISUB      7
.      ,KCH          ,KCNVRT     ,KCOUNT     ,KDIF        ,KFLD1      ,KFLD2     8

```

•	,LCOMP	,LCNVRT	,LEVEL	,LFRT	,LOOK		9	
•	,MCNVRT	,MDIF	,MODALL	,MSTJR			10	
•	,NAME	,NERROR	,NONEW	,NOTARG			11	
•	,SMCHR	,TEST	,ERMARK				12	
•/ICNSTI/	BLANK	,BLANKS	,DOLLAR	,EOS			13	
•	,ICOMMA	,IDOLAR	,IFT	,IPTAB	,ITAB		14	
•	,KAM10	,KBPC	,KBPW	,KCPCD	,KERTYP		15	
•	,KZERO	,NDPRNT	,TAB1				16	
•/IPARAM/	ABORT	,KIUNIT	,KOUNIT	,LIMALF	,LOCK	,LOCX	17	
•	,NOLIST	,NSTDIR	,TRACE				18	
•/ISTACK/	STACK	,ISTDIM	,KSTACK	,LEVLIM			19	
C							20	
	INTEGER	BLANK	,BLANKS	,EOS	,IDOLAR	,TAB1	,TRACE	21
	DOUBLE PRECISION	STACK, VALUE						22
	LOGICAL	ABORT, DOLLAR, ERMARK,	LIMALF, LOCK, NOLIST, NONEW, MODALL,					23
	,NSTDIR, SMCHR, TEST							24
	EQUIVALENCE (STACK, ISTACK),	(VALUE, KVALUE, RVALUE),	(NAME, ANAME)					25
	EQUIVALENCE (ICOMNI, ISUB),	(IMAGE, IMAGE1), (IPARAM, ABORT)						26
	DIMENSION	NTYPE(4)						27
	DOUBLE PRECISION	NTYPE						28
	DATA	NTYPE / 6HREAL ,6HINT ,6HDP ,6HNOCONV/						29
	IF((IFROM.LE.0).OR.(ITO.LE.0))	GO TO 100						30
	IF (IFROM - 4) 1,99,100							31
1	IF (ITO - 4) 2,99,100							32
2	ITOM2 = ITO - 2							33
	IF (IFROM-2) 3,4,5							34
3	IF (ITOM2) 99,10,20							35
4	IF (ITOM2) 30,99,40							36
5	IF (ITOM2) 50,60,99							37
10	KVALUE = RVALUE(1)							38
	GO TO 99							39
20	VALUE = RVALUE(1)							40
	GO TO 99							41
30	RVALUE(1) = KVALUE							42
	GO TO 99							43
40	VALUE = KVALUE							44
	GO TO 99							45
50	RVALUE(1) = VALUE							46
	GO TO 99							47
60	KVALUE = VALUE							48
99	CONTINUE							49
C	CALL	DEBUG3 (6HICNVTI,0.DO,3)						50
C	CALL	DEBUG3(NTYPE(IFROM),NTYPE(ITO) ,3)						51
	RETURN							52
100	WRITE(⟨DUNIT,101) IFROM, ITO							53
101	FORMAT(35H ARGUMENTS OF ICNVTI BAD. IFROM = ,I13,8H, ITO = ,I13,							54
	. 41H(1 TO 4 ALLOWABLE). CHECK IPTAB IN COMNJ)							55
	GO TO 99							56
	END							57

\$IBFTC	IERORI	DECK						
	SUBROJTINE	IERORI						1
	DIMENSION	IFT( 31), IPTAB( 21), ITAB( 65)						2
	DIMENSION	ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)						3
•		,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)						4
C								5



```

COMMON
./ICOMNI/ VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB 5
. ,KCH ,KCNVRT ,KCOUNT ,KDIF ,KFLD1 ,KFLD2 7
. ,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK 8
. ,MCNVRT ,MDIF ,MODALL ,MSTOR 10
. ,NAME ,NERROR ,NONEW ,NOTARG 11
. ,SMCHR ,TEST ,ERMARK 12
./ICNSTI/ BLANK ,BLANKS ,DOLLAR ,EOS 13
. ,ICOMMA ,IDOLLAR ,IFT ,IPTAB ,ITAB 14
. ,KAM10 ,KBPC ,KBPW ,KCPCD ,KERTYP 15
. ,KZERO ,NOPRNT ,TAB1 16
./IPARAM/ ABORT ,KIUNIT ,KOUNTIT ,LIMALF ,LOCK ,LOCX 17
. ,NOLIST ,NSTDIR ,TRACE 18
./ISTACK/ STACK ,ISTDIM ,KSTACK ,LEVLIM 19
C 20
INTEGER BLANK ,BLANKS ,EOS ,IDOLLAR ,TAB1 ,TRACE 21
DOUBLE PRECISION STACK, VALUE 22
LOGICAL ABORT,DOLLAR,ERMARK,LIMALF,LOCK,NOLIST,NONEW,MODALL, 23
./NSTDIR,SMCHR,TEST 24
EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE),(NAME,ANAME) 25
EQUIVALENCE (ICOMNI,ISUB),(IMAGE,IMAGE1),(IPARAM,ABORT) 26
INTEGER H(4,7) 27
EQUIVALENCE (H(1,1),H1),(H(1,2),H2),(H(1,3),H3),(H(1,4),H4) 28
. ,(H(1,5),H5),(H(1,6),H6),(H(1,7),H7) 29
DATA KA/1H*/ 30
DATA <B/1H / 31
DATA <C/1H,/ 32
DATA KD/1H*/ 33
INTEGER MSGTYP(2,2), H1(4), H2(4), H3(4), H4(4), H5(4),H6(4),H7(4) 34
DATA MSGTYP(1,1)/24H*DIAGNOSTIC**ERROR** / 35
DATA +1(1) /24HINAPPROPRIATE CHARACTER / 36
DATA +2(1) /24HNAME TOO LONG / 37
DATA +3(1) /24HTABLE FULL OR DESTROYED / 38
DATA +4(1) /24HSD INCOMPLETE OR MISSING / 39
DATA +5(1) /24HNAME NOT IN TABLE / 40
DATA +6(1) /24HFUNCTN OR SUB NOT ABOARD / 41
DATA +7(1) /24HFORMULA ILL-FORMED / 42
C CALL DEBUGC(6HIERORI) 43
NONEW = .FALSE. 44
NERROR = NERROR + 1 45
C 46
ERMARK = .TRUE. * COUNT DIAGONISTICS 47
C 48
IOP = 0 * WARN PROGRAMER OF POSSIBLE ERRO 49
C 50
* COUNT OPERATORS +-,* 51
MODE = 1 51
IF(NOTARG.LT.LEVEL) MODE = 2 52
KC1=K3 53
IF(KCJUNT.EQ.0) KC1=KC 54
L = KCOUNT + 1 55
MODALL = .FALSE. 56
TEST = .TRUE. 57
M = IABS(KERTYP) 58
J = M/100 59
K = 2 60
IF(DOLLAR) MODE = 2 61
IF(KERTYP.GT.0) LOCK = .TRUE. 62
IF(KERTYP.LT.0) K=1 63
11 WRITE(<DUNIT,90)(MSGTYP(I,K),I=1,2),M,(H(I,J),I=1,4),KC1, 64
. IMAGE,(KB,I=1,L),KA,(KB,I=L,81),NERROR,LOCK 65
90 FORMAT(1H ,2A6,2H (,13,2H) ,4A6,2X,81A1/45X,83A1, 65
. /14X,11HERROR COUNT,I4,13H LOCK = ,(L1,1H.) 67
IF(NERROR.GT.64)STOP 68

```

```

IF(KERTYP.LT.0) GO TO 99
SMCHR = .TRUE.
30 CALL ICHAR4($31,$65,7,10)
C      0-9. + $ )
GO TO (30,65,39,80),LCOMP
C      , ' A-Z =
31 GO TO (60,65,30,60),LCOMP
39 SMCHR = .TRUE.
40 L = KCOUNT + 1
KC1 = KB
IF(KCJUNT.EQ.0)KC1=KC
WRITE(KOUNIT,91)KC1,IMAGE,(KB,I=1,L),KD,(KB,I=L,81)
91 FORMAT(14X,43HSKIP AHEAD AND GUESS AT NEXT GOOD STATEMENT,
. /14X,32HBEGIN PROCESSING AT SYMBOL ,81A1/45X,83A1
. /14X48HLOOK FOR DIAGNOSTICS BUT DON'T STORE ANYTHING.//)
IFNTYP = -1
DO 1 I=1,ISTDIM
1 KSTACK(I) = 0
LEVEL = 0
NOTARG = ISTDIM
99 CONTINUE
KERTYP = 0
CALL DEBUGR
RETURN
C ,=,=,=,=,=,=,=,=,=,=,=,=,=,=,=,=
60 IF(MODE.EQ.1) GO TO 40
C ALL OPERATORS
65 IOP = IOP + 1
IF(IOP.LE.16) GO TO 30
C IF YOU FIND 16 OPERATORS BEFORE ) GIVE UP
C
C ))))))))))))))))))))))))
80 MODE = 1
DOLLAR = .FALSE.
GO TO 30
C
C MODE ACTION
C
C 1 SKIP TO NEXT $ OR = OR , (UNCONDITIONAL)
C 2 SKIP TO NEXT $ (UNCONDITIONAL) OR TO = OR , AFTER )
END

```

```

$IBFTC ILOOKI DECK
SUBROUTINE ILOOKI(*,IT)
C CALLED FROM
C ITABLI
C INAMEI
DIMENSION IT(1)
DIMENSION IFT( 31), IPTAB( 21), ITAB( 65)
DIMENSION ANAME(15), IMAGE(80), IMAGE1(81), IPARAM(9)
. ,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)
C
COMMON
. /ICOMMI/ VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB
. ,KCH ,KCNVRT ,KCOUNT ,KDIF ,KFLD1 ,KFLD2
. ,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK
. ,MCNVRT ,MDIF ,MODALL ,MSTOR
. ,NAME ,VEERROR ,NONEW ,NOTARG

```

```

      .,SMCHR      ,TEST      ,ERMARK      16
./ICNSTI/ BLANK  ,BLANKS    ,DOLLAR      ,EOS      17
      .,ICOMMA    ,IDOLAR    ,IFT         ,IPTAB    ,ITAB      18
      .,KAM10     ,KBPC      ,KBPW       ,KCPCD    ,KERTYP    19
      .,KZERO     ,NOPRNT   ,TAB1       20
./IPARAM/ ABORT  ,KIUNIT   ,KOUNIT     ,LIMALF   ,LOCK      ,LOCKX    21
      .,NOLIST    ,NSTDIR   ,TRACE      22
./ISTA<I/ STACK ,ISTDIM   ,KSTACK     ,LEVLIM   23
C
      INTEGER BLANK  ,BLANKS    ,EOS         ,IDOLAR    ,TAB1      ,TRACE    24
      DOUBLE PRECISION STACK, VALUE  25
      LOGICAL ABORT,DOLLAR,ERMARK,LIMALF,LOCK,NOLIST,NONEW,MODALL, 27
      .NSTDIR,SMCHR,TEST  28
      EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE),(NAME,ANAME) 29
      EQUIVALENCE (ICOMNI,ISUB),(IMAGE,IMAGE1),(IPARAM,ABORT) 30
      LOOK=3 31
C
      * FIRST USABLE POSITION 32
1 IF(IT(LOOK) .EQ.0) GO TO 7 33
C
      * RETURN WITH NEXT AVAILABLE ENTR 34
      J=IFLD(3,4,IT(LOOK)) 35
C
      * MAX POSSIBLE J IS 15 36
      IF (J .NE.MSTOR) GO TO 8 37
C
      * 8 IF NAME IS WRONG LENGTH, TRY 38
      DO 4 <=1,J 39
C
      * CHECK NAME FOR MATCH 40
      LOOKK = LOOK + K 41
      IF (NAME(K).NE.IT(LJOKK)) GO TO 8 42
4 CONTINUE 43
C CALL DEBUG2(6H*LOOKF,LOOK) 44
C
      * ENTRY WAS FOUND 45
      RETURN 1 46
7 CONTINUE 47
C CALL DEBUG2(6H*LOOKN,LOOK) 48
C
      * NO ENTRY WAS FOUND 49
      RETURN 50
8 LOOK = LOOK+J+1 51
      52
      53
C
      * TRY NEXT ENTRY 54
      GO TO 1 55
      END 56

```

```

$IBFTC INAMEI DECK
      SUBROUTINE INAMEI(*,D,IT) 1
C      CALLED FROM INPUT 2
C      CALL DEBUGC(6HINAMEI) 3
C      SUBROUTINE TO OBTAIN LOCATION, MODE, AND CONTENTS OF A NAMED CELL 4
C
      DIMENSION IFT( 31), IPTAB( 21), ITAB( 65) 5
      DIMENSION ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9) 6
      .,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27) 7
C
      COMMON 8
      9
./ICOMNI/ VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB 10
      .,KCH ,KCNVRT ,KCOUNT ,KDIF ,KFLD1 ,KFLD2 11
      .,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK 12
      .,MCNVRT ,MDIF ,MODALL ,MSTOR 13
      .,NAME ,VERROR ,NONEW ,NOTARG 14
      .,SMCHR ,TEST ,ERMARK 15
      16

```

```

./ICNSTI/ BLANK      ,BLANKS      ,DOLLAR      ,EDS          17
.           ,ICOMMA      ,IDOLAR      ,IFT          ,IPTAB        ,ITAB          18
.           ,KAMIO      ,KBPC        ,KBPW        ,KCPCD        ,KERTYP       19
.           ,KZERO      ,NOPRNT      ,TAB1        ,LOCK         ,LOCX        20
./IPARAM/ ABORT      ,<IUNIT     ,KOUNIT      ,LIMALF      ,LOCK         ,LOCX        21
.           ,NOLIST     ,VSTDIR      ,TRACE       ,LEVLIM       22
./ISTAKI/ STACK     ,ISTDIM      ,KSTACK      ,LEVLIM      23
C
INTEGER BLANK      ,BLANKS      ,EDS          ,IDOLAR      ,TAB1        ,TRACE       24
DOUBLE PRECISION STACK, VALUE 25
LOGICAL ABORT,DOLLAR,ERMARK,LIMALF,LOCK,NOLIST,NONEW,MODALL, 27
.NSTDIR,SMCHR,TEST 28
EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE),(NAME,ANAME) 29
EQUIVALENCE (ICOMNI,ISUB),(IMAGE,IMAGE1),(IPARAM,ABORT) 30
DIMENSION IT(1) 31
C
C                                     * TABLE OF NAMES PROVIDED BY USER 32
DIMENSION D(1) 33
C
C                                     * USERS VARIABLES ARE IN D ARRAY 34
ISUB = 1 35
C
C                                     * UNDERSTOOD SUBSCRIPT 36
KDIF = 1 37
CALL INAMEN 38
CALL ILOOKI($1,IT) 39
CALL ILOOKI($64,IFT) 40
9540 KERTYP=540 41
C
C                                     * NAME NOT IN TABLE 42
GO TO 99 43
1 CONTINJE 44
C CALL DEBUG2(6HSTATMT,1) 45
ITL = IT(LOOK) 46
KCNVRT=IFLD(0,3,ITL) 47
C CALL DEBUG 2 (6HKCNVRT,KCNVRT) 48
C R I D T S F 49
GO TO(12,12,10,12,62,61),KCNVRT 50
10 KDIF = 2 51
C ( DTH 52
12 CALL ICHAR2($90,30) 53
C
C                                     * GO TO 90 FOR NO SUBSCRIPT 54
C A-Z 0-9 DTH 55
CALL ICHARI($82,$9140,12) 56
CALL INAMEN 57
CALL ILOOKI($83,IT) 58
GO TO 9540 59
82 SMCHR = .TRUE. 60
CALL ISUBI 61
GO TO 84 62
83 ITYPE=IFLD(0,3,IT(LOOK)) 63
LOC=IFLD(7,25,IT(LDDK)) 64
RVALUE(1) = D(LOC) 65
RVALUE(2) = D(LOC+1) 66
CALL ICNVTI (ITYPE,2) 67
ISUB = KVALUE 68
C ) DTH 69
84 CALL ICHAR2($9140,31) 70
C
C                                     * ERROR IF NO ) 71
GO TO 91 72
90 SMCHR = .TRUE. 73
91 LOC=IFLD(7,25,ITL) 74
C CALL DEBUG 2 (5HLOC 1,LOC) 75
LOC = LOC + (ISUB-1)*KDIF 76
50 RVALUE(1) = D(LOC) 77
RVALUE(2) = D(LOC+1) 78

```

C	CALL DEBUG 3(5HIT(L),ITL,2)	79
	CALL ICNVTI (KCNVRT,3)	80
52	LOOK = LOC	81
98	CONTINUE	82
C	CALL DEBUG 3(5HVALUE,VALUE,4)	83
C	CALL DEBUG 2 (6HLOOK ,LOOK)	84
C	CALL DEBUG 2 (5HLOC 2,LOC)	85
C	CALL DEBUG 2(6HKCNVRT,KCNVRT)	86
C	CALL DEBUGR	87
	RETURN	88
61	IFNTYP = 0	89
C	* FUNCTION	90
	GO TO 63	91
62	IFNTYP = 1	92
C	* SUBROUTINE	93
63	KVALJE=IFLD(7,25,ITL)	94
	GO TO 98	95
64	IFNTYP = 0	96
C	* LIBRARY FUNCTION OR SUBROUTINE	97
	KVALJE=IFLD(7,25,IFT(LOOK))	98
C	* PROGRAM NUMBER (USED BY IXQTI)	99
	MSTOR = 0	100
	KCNVRT=IFLD(0,3,IFT(LOOK))	101
C	* 5 FOR SUBROUTINES, 6 FOR FUNCTIONS	102
	IF(KCNVRT.EQ.5)IFNTYP=KVALUE	103
C	* PROVIDE FOR EXECUTION OF INPUT SUBRO	104
C	CALL DEBUG2(6HLIBF ,KVALUE)	105
C	MEANING OF IFNTYP	106
C	VALUE          NAME IS	107
C	-1(NORMAL) AN ORDINARY VARIABLE	108
C	0              FUNCTION (USER OR FORTRAN MATH)	109
C	1              USER SUBROUTINE	110
C	2              RADIX (INPUT FUNCTION)	111
C	3              LOCX (INPUT SUBROUTINE)	112
C	\$CALL LOCX(Y,I) CAUSES I TO BE SET SO THAT Y(I)REFERS TO THE CUR	113
C	LEFT SIDE.	114
	GO TO 98	115
9140	KERTYP =140	116
99	CONTINUE	117
C	CALL DEBUG2(6HRETURN,1)	118
C	CALL DEBUGR	119
C	* REGISTER RETURN	120
	RETURN 1	121
	END	122

\$IBFTC	INAMEN DECK	
	SUBROJTINE INAMEN	1
C	CALLED FROM	2
C	INAMEI	3
C	ITABLI	4
C	INPUT	5
C	CALL DEBUGC(6HINAMEN)	6
C		7
	DIMENSION IFT( 31), IPTAB( 21), ITAB( 65)	8
	DIMENSION ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)	9
	*                  ,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)	10
C		11

	COMMON						12	
	./ICOMNI/	VALUE	,ICOMP	,IFNTYP	,IMAGE1	,IRADIX	,ISUB	13
	.	,KCH	,KCNVRT	,KCOUNT	,KDIF	,KFLD1	,KFLD2	14
	.	,LCOMP	,LCNVRT	,LEVEL	,LFRT	,LOOK		15
	.	,MCNVRT	,MDIF	,MODALL	,MSTOR			16
	.	,NAME	,NERROR	,NONEW	,NOTARG			17
	.	,SMCHR	,TEST	,ERMARK				18
	./ICNSTI/	BLANK	,BLANKS	,DOLLAR	,EOS			19
	.	,ICOMMA	,IDOLLAR	,IFT	,IPTAB	,ITAB		20
	.	,KAM10	,KBPC	,KBPW	,KCPCD	,KERTYP		21
	.	,KZERO	,NDRNT	,TAB1				22
	./IPARAM/	ABORT	,KIUNIT	,KOUNIT	,LIMALF	,LOCK	,LOCX	23
	.	,NOLIST	,NSTDIR	,TRACE				24
	./ISTACKI/	STACK	,ISTDIM	,KSTACK	,LEVLIM			25
C								26
	INTEGER	BLANK	,BLANKS	,EOS	,IDOLLAR	,TAB1	,TRACE	27
	DOUBLE PRECISION	STACK,	VALUE,	DNAME				28
	LOGICAL	ABORT,	DOLLAR,	ERMARK,	LIMALF,	LOCK,	NOLIST,	29
		NONEW,	MODALL,					30
		NSTDIR,	SMCHR,	TEST				31
	EQUIVALENCE	(STACK,	ISTACK),	(VALUE,	KVALUE,	RVALUE),	(NAME,	32
		ANAME)						33
	EQUIVALENCE	(ICOMNI,	ISUB),	(IMAGE,	IMAGE1),	(IPARAM,	ABORT)	34
								35
	EQUIVALENCE	(NAME,	DNAME)					36
C	COLLECTS	NAME	(UP TO 15 WORDS)	TERMINATED	BY ANY SPECIAL	CHAR		37
	ASSIGN	6	TO NEXT					38
	IF	(MODALL)	ASSIGN	2	TO NEXT			39
	MSTOR	=	0					40
	J	=	<BPW					41
	NAME	(2)	=	BLANKS				42
1	GO	TO	NEXT,	(2,6,7)				43
C								44
								45
2	CALL	ICHAR2	(\$8	,23)				46
	TEST	=	.TRUE.					47
	MODALL	=	.FALSE.					48
	CALL	ICHAR2	(\$99,	23)				49
C								50
	TEST	=	.FALSE.					51
	MODALL	=	.TRUE.					52
	GO	TO	8					53
6	ASSIGN	7	TO NEXT					54
	GO	TO	8					55
C								56
	A-Z-0-9	OTH						57
7	CALL	ICHAR2	(\$99,	12)				58
8	IF	(J.LT.	KBPW)	GO	TO	9		59
	IF	(MSTOR.EQ.	15)	GO	TO	10		60
	MSTOR	=	MSTOR + 1					61
	NAME	(MSTOR)	=	BLANKS				62
	J	=	0					63
9	NAME	(MSTOR)	=	IFLD4	(KCH,	J,	KBPC,	64
								65
	J	=	J+KBPC					66
	GO	TO	1					67
10	IF	(MODALL)	GO	TO	99			68
	KERTYP	=	-260					69
	CALL	IERORI						70
95	CALL	ICHAR2	(\$99,	12)				71
C								72
								73
	GO	TO	95					74
99	CONTINUE							75
	SMCHR	=	.TRUE.					76
C	CALL	DEBUG3	(6HNAME	,DNAME,	3)			77
C	CALL	DEBUG2	(5HMSTOR,	MSTOR)				78
C	CALL	DEBUGR						79
	RETURN							80
	END							81

```

$IBFTC INMBRI DECK
C SUBROJTINE TO TRANSLATE A NUMERIC FIELD 1
SUBROJTINE INMBRI 2
C CALLED FROM INPUT 3
C INMBRI IS CALLED WITH FIRST DIGIT IN KCH 4
C CALL DEBUGC(6HINMBRI) 5
DIMENSION IFT( 31), IPTAB( 21), ITAB( 65) 6
DIMENSION ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9) 7
. ,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27) 8
C 9
COMMON 10
./ICOMVI/ VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB 11
. ,KCH ,KCNVRT ,KCOUNT ,KDIF ,KFLD1 ,KFLD2 12
. ,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK 13
. ,MCNVRT ,MDIF ,MODALL ,MSTOR 14
. ,NAME ,NERROR ,NONEW ,NOTARG 15
. ,SMCHR ,TEST ,ERMARK 16
./ICNSTI/ BLANK ,BLANKS ,EOS ,DOLLAR 17
. ,ICOMMA ,IDOLLAR ,IFT ,IPTAB ,ITAB 18
. ,KAM10 ,KBPC ,KBPW ,KCPCD ,KERTYP 19
. ,KZERO ,NOPRNT ,TAB1 20
./IPARAM/ ABORT ,KIUNIT ,KUNIT ,LIMALF ,LOCK ,LOCX 21
. ,NOLIST ,NSTDIR ,TRACE 22
./ISTAKI/ STACK ,ISTDIM ,KSTACK ,LEVLIM 23
C 24
INTEGER BLANK ,BLANKS ,EOS ,IDOLLAR ,TAB1 ,TRACE 25
DOUBLE PRECISION STACK, VALUE, DNBR, H, FD(4) 26
LOGICAL ABORT,DOLLAR,ERMARK,LIMALF,LOCK,NOLIST,NONEW,MODALL, 27
.NSTDIR,SMCHR,TEST 28
EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE),(NAME,ANAME) 29
EQUIVALENCE (ICOMNI,ISUB),(IMAGE,IMAGE1),(IPARAM,ABORT) 30
LOGICAL LVALUE 31
EQUIVALENCE (KVALUE,LVALUE) 32
LOGICAL SWITCH 33
DIMENSION LD(4) 34
DATA LD(1),LD(2),LD(3),LD(4)/8,4,2,1/ 35
DATA FD(1),FD(2),FD(3),FD(4)/1.0D8,1.0D4,1.0D2,10.0D/ 36
DNBR=0 37
C * THE NUMBER COLLECTED SO FAR 38
ICSC=0 39
C *THE CHARACTERISTIC SCALE FACTOR 40
IPF = 1 41
C * SIGN OF EXPONENT 42
IESE = 0 43
C * THE EXPONENT 44
ASSIGN 1 TO NEXT 45
SWITCH = .FALSE. 46
SMCHR = .TRUE. 47
CALL ICHAR2($70,21) 48
C * GO TO 70 FOR LOGICAL CONSTANTS 49
SMCHR = .TRUE. 50
C +- OTH 0-9 51
1 CALL ICHAR4($2, $3, 13,15) 52
GO TO 50 53
3 DNBR = DNBR*10.0D+FLOAT(KCH-KZERO) 54
GO TO NEXT,(1,15) 55
15 ICSC = ICSC - 1 56
GO TO 1 57
C . DE OTH 58
2 GO TO(20,30,50),LCOMP 59
19 SMCHR=.TRUE. 60
C *ENTER HERE FOR INITIAL DECIMAL POINT 61
20 IF (SWITCH) GO TO 9130 62

```

C	CALL DEBUG2(6HSTAT ,20)	63
	ASSIGN 15 TO NEXT	64
	SWITCH = .TRUE.	65
	GO TO 1	66
37	IPF=-1	67
	GO TO 36	68
30	CONTINUE	69
C	CALL DEBUG 2(6HSTAT E,30)	70
C	+ - OTH 0-9	71
	CALL ICHAR4(\$50,\$35, 13,15)	72
C	+ -	73
	GO TO(36,37),LCOMP	74
35	SMCHR=.TRUE.	75
36	CALL ISUBI	76
C		77
44	ICSC = ICSC+ISIGN(ISUB,IPF) * TEST WILL BE TRUE	78
C		79
50	CONTINUE * RESOLVE SCALE FACTORS	80
C	CALL DEBUG 2(6HSTAT ,50)	81
C	CALL DEBUG3 (5HDNBR1,DNBR ,4)	82
	H = 1.DO	83
	IESC=IABS(ICSC)	84
	DO 63 I=1,4	85
61	IF (IESC.LT. LD(I)) GO TO 63	86
62	IESC=IESC-LD(I)	87
	H=H*FD(I)	88
	GO TO 61	89
63	CONTINUE	90
	IF (ICSC.LT.0) GO TO 65	91
64	DNBR = DNBR*H	92
	GO TO 98	93
C	T F OTH	94
70	CALL ICHAR1(\$73,\$19,17)	95
	LVALJE=.TRUE.	96
71	IF(NOTARG.GT.LEVEL)LCNVRT=4	97
C		98
	RVALUE(2)=0. * LOGICAL CONSTANTS NOT CONVERTED	99
C	A-Z OTH .	100
72	CALL ICHAR1(\$9130,\$99,21)	101
C		102
	GO TO 72 *DISCARD REST OF WORD, MUST FIND	103
73	LVALUE=.FALSE.	104
	GO TO 71	105
65	DNBR = DNBR/H	106
98	VALUE = DNBR	107
	SMCHR = .TRUE.	108
99	CONTINUE	109
C	CALL DEBUG3(6HDNBR 2,DNBR,4)	110
C	CALL DEBUG3(5HVALUE,VALUE,4 )	111
C	CALL DEBUG2(6HICSC ,ICSC)	112
C	CALL DEBUG2(6HIPF ,IPF)	113
C	CALL DEBUG 2(6HIESC ,IESC)	114
C	CALL DEBUGR	115
	RETURN	116
9130	KERTYP = 130	117
	GO TO 99	118
	END	119



```

$IBFTC ITABLI DECK
C SUBROUTINE TO CONSTRUCT TABLE ENTRIES 1
  SUBROJTINE ITABLI(IT) 2
C CALLED FROM INPUT 3
C CALL DEBUGC(6HITABLI) 4
C 5
  DIMENSION IFT( 31), IPTAB( 21), ITAB( 65) 6
  DIMENSION ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9) 7
  . ,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27) 8
C 9
  COMMON 10
  ./ICOMNI/ VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB 11
  . ,KCH ,KCNVRT ,KCOUNT ,KDIF ,KFLD1 ,KFLD2 12
  . ,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK 13
  . ,MCNVRT ,MDIF ,MODALL ,MSTOR 14
  . ,NAME ,NERROR ,NONEW ,NOTARG 15
  . ,SMCHR ,TEST ,ERMARK 16
  ./ICNSTI/ BLANK ,BLANKS ,DOLLAR ,EOS 17
  . ,ICOMMA ,IDOLAR ,IFT ,IPTAB ,ITAB 18
  . ,KAM10 ,KBPC ,KBPW ,KCPCD ,KERTYP 19
  . ,KZERO ,NOPRNT ,TAB1 20
  ./IPARAM/ ABORT ,KIUNIT ,KOUNIT ,LIMALF ,LOCK ,LOCX 21
  . ,NOLIST ,NSTDIR ,TRACE 22
  ./ISTACK/ STACK ,ISTDIM ,KSTACK ,LEVLIM 23
C 24
  INTEGER BLANK ,BLANKS ,EOS ,IDOLAR ,TAB1 ,TRACE 25
  DOUBLE PRECISION STACK, VALUE 26
  LOGICAL ABORT,DOLLAR,ERMARK,LIMALF,LOCK,NOLIST,NONEW,MODALL, 27
  .NSTDIR,SMCHR,TEST 28
  EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE),(NAME,ANAME) 29
  EQUIVALENCE (ICOMNI,ISUB),(IMAGE,IMAGE1),(IPARAM,ABORT) 30
  DIMENSION IT (1) 31
C 32
  KDIF = 1 33
  ISUBX = 1 34
  ITYPE = 1 35
C * ALWAYS INITIALIZE TO .REAL. 36
C ( A-Z OTH 37
1 CALL ICHARI ($1,$9120,26) 38
C * SKIP 'ABLE' IN TABLE 39
3 CONTINUE 40
C CALL DEBUG2(6HSTATMT,3) 41
  CALL ICHAR4($4,$9120,7,9) 42
C . ) 0-9+ OTH 43
  GO TO (30 ,98 ,10 ,9120),LCOMP 44
C , A-Z OTH OTH 45
4 GO TO (3 ,20, 9120,9120),LCOMP 46
9120 KERTYP= 120 47
C , ILLEGAL CHARACTER 48
  GO TO 98 49
10 CONTINJE 50
C CALL DEBUG2(6HST 0-9 ,10) 51
  SMCHR = .TRUE. 52
  CALL ISUBI 53
  ISUBX = ISUB 54
  CALL ICHAR2($9120,25) 55
C *ERROR IF NO = 56
  GO TO 3 57
30 CONTINUE 58
C CALL DEBUG 2(6HSTATMT,30) 59
31 KDIF = 1 60

```

C	SPLIT TYPES (INT,REAL,DP,NO CONVERSION,FUNCTION ,SUBROUTINE)	61
	CALL ICHAR4(\$32,\$33,3,5)	62
	IF(LCOMP.GT.1) GO TO 9120	63
	KDIF = 2	64
	GO TO 33	65
32	IF (LCOMP.EQ.3) GO TO 9120	66
	ITYPE = LCOMP	67
	GO TO 34	68
33	ITYPE = LCOMP+2	69
C	A-Z OTH .	70
34	CALL ICHARI(\$9120,\$3,21)	71
	GO TO 34	72
20	CONTINUE	73
C	CALL DEBUG2(6HSTATMT,20)	74
	CALL INAMEN	75
50	CONTINUE	75
C	CALL DEBUG2(6HSTATMT,50)	77
	ITBUFF=IFLD4(ITYPE,0,3,ITBUFF)	78
	ITBUFF=IFLD4(MSTOR,3,4,ITBUFF)	79
	ITBUFF=IFLD4(ISUBX,7,25,ITBUFF)	80
C	CALL DEBUG2 (5HITYPE,ITYPE)	81
C	CALL DEBUG2 (5HMSTOR,MSTOR)	82
C	CALL DEBUG2 (5HISUBX,ISUBX)	83
C	CALL DEBUG4 (6HITBUFF,ITBUFF,2)	84
	ISUBX = ISUBX + KDIF	85
	CALL ILOOKI(\$56,IT)	86
	IF((IT(2).NE.0).AND.((LOOK+MSTOR +2).GT.IT(2))) GO TO 9320	87
	IT(LOJK)=ITBUFF	88
	DO 55 K=1,MSTOR	89
	LOOK=LOOK+1	90
55	IT(LOJK)=NAME(K)	91
C	CALL DEBUG2(6HSTATMT,55)	92
	LOOK=LOOK+1	93
	IT(LOJK)=0	94
	GO TO 3	95
56	IT(LOJK)=ITBUFF	96
C	CALL DEBUG2(6HSTATMT,56)	97
	GO TO 3	98
98	CONTINUE	99
C	CALL DEBUGR	100
	RETURN	101
9320	KERTYP =-320	102
	CALL IERORI	103
	GO TO 3	104
	END	105

\$IBFTC	ISUBI DECK	
	SUBROJTINE ISUBI	1
C	ISUBI FINDS SUBSCRIPTS AND INTEGER CONSTANTS	2
C	CALLED FROM	3
C	INAMEI	4
C	INMBRI	5
C	INPUT	6
C	ITABLI	7
C	ISJB BEGINS PROCESSING WITH THE NEXT CHARACTER.	8
C	CALL DEBUGC(5HISUBI)	9
C	COLLECTS INTEGER OF BASE IRADIX TERMINATED BY A SPECIAL CHARACTER	10
	DIMENSION IFT( 31), IPTAB( 21), ITAB( 65)	11

```

DIMENSION ANAME(15) , IMAGE(80) , IMAGE1(81) , IPARAM(9)      12
.      , KSTACK(27) , NAME(15) , RVALUE(2) , STACK(27)      13
C      COMMON      14
. /ICOMNI/ VALUE      , ICOMP      , IFNTYP      , IMAGE1      , IRADIX      , ISUB      15
.      , KCH      , KCNVRT      , KCOUNT      , KDIF      , KFLD1      , KFLD2      16
.      , LCJMP      , LCNVRT      , LEVEL      , LFRT      , LOOK      17
.      , MCNVRT      , MDIF      , MODALL      , MSTORE      18
.      , NAME      , NERROR      , NONEW      , NOTARG      19
.      , SMCHR      , TEST      , ERMARK      20
. /ICNSTI/ BLANK      , BLANKS      , DOLLAR      , EOS      21
.      , ICOMMA      , IDOLAR      , IFT      , IPTAB      , ITAB      22
.      , KAM10      , KBPC      , KBPW      , KCPCD      , KERTYP      23
.      , KZERO      , NOPRNT      , TAB1      24
. /IPARAM/ ABORT      , KIUNIT      , KUNIT      , LIMLAF      , LOCK      , LOCKX      25
.      , NOLIST      , NSTDIR      , TRACE      26
. /ISTAKI/ STACK      , ISTDIM      , KSTACK      , LEVLIM      27
C      INTEGER BLANK      , BLANKS      , EOS      , IDOLAR      , TAB1      , TRACE      28
DOUBLE PRECISION STACK, VALUE      29
LOGICAL ABORT, DOLLAR, ERMARK, LIMLAF, LOCK, NOLIST, NONEW, MODALL,
. NSTDIR, SMCHR, TEST      30
EQUIVALENCE (STACK, ISTACK), (VALUE, KVALUE, RVALUE), (NAME, ANAME)      31
EQUIVALENCE (ICOMNI, ISUB), (IMAGE, IMAGE1), (IPARAM, ABORT)      32
ISUB = 0      33
C      A-Z 0-9 OTH      34
80 CALL ICHARI($10,$99,12)      35
IDIGIT=KCH-KAM10      36
C      * VALUE OF LETTER USED AS DIGIT      37
30 IF(IDIGIT .GE. IRADIX) GO TO 99      38
35 ISUB = ISUB * IRADIX + IDIGIT      39
C      * ACCUM TOTAL. IRADIX      40
GO TO 80      41
99 SMCHR = .TRUE.      42
C      * ALLOW SAME CHARACTER TO BE READ      43
CALL DEBUG2(6H*ISUBI, ISUB)      44
CALL DEBUGR      45
RETURN      46
10 IDIGIT=KCH-KZERO      47
C      * VALUE OF DIGIT      48
GO TO 30      49
END      50
51
52
53

$IBFTC IXQTI DECK
SUBROJTIME IXQTI (ARGL, ARGS)      1
C USER MAY PUT HIS OWN COMMON STATEMENTS IN THIS ROUTINE AND      2
C USE THEM TO SUPPLY ARGUMENTS TO HIS CALLS IF HE DESIRES      3
COMMON      4
. /IPARAM/ ABORT , KIUNIT , KUNIT , LIMLAF , LOCK , LOCKX , NOLIST , NSTDIR      5
. , TRACE      6
DOUBLE PRECISION ARGS(27), ARGL, ARG22      7
DIMENSION ARG2(2)      8
EQUIVALENCE (ARG2, ARG22)      9
C      10
M = DABS(ARGS(1))      11
IF (M.LT.1 .OR. M.GT.16) GO TO 99      12
GO TO (1,2,3,4,5,6,7,8,9,10,11,12,13,14,15,16), M      13

```



```

$IBFTC DEBUGX DECK
SUBROUTINE DEBUGX
C
C          *INITIALIZATION
DIMENSION ISUBN(20)
C TRACE = 0 NO PRINTING
C TRACE = 1 PRINT DEBUG 2+3 CALLS ONLY
C TRACE = 2 PRINT DEBUG 2+3 CALLS ONLY
C TRACE = 3 PRINT DEBUG 2+3 AND STACK PRINT
C TRACE = 4 PRINT DEBUG 2+3 AND STACK PRINT AND CALLS FROM CHAR
DIMENSION IFT(27), IPTAB(21), ITAB(65)
DIMENSION ANAME(15), IMAGE(80), IMAGE1(81), IPARAM(9)
* ,KSTACK(27), NAME(15), RVALUE(2), STACK(27)
C
COMMON
./ICOMNI/ VALUE ,ICOMP ,IFNTYP ,IMAGE1 ,IRADIX ,ISUB
. ,KCH ,KCNVRT ,KCOUNT ,KDIF ,KFLD1 ,KFLD2
. ,LCOMP ,LCNVRT ,LEVEL ,LFRT ,LOOK
. ,MCNVRT ,MDIF ,MODALL ,MSTOR
. ,NAME ,NERROR ,NONEW ,NOTARG
. ,SMCHR ,TEST ,ERMARK
./ICNSTI/ BLANK ,BLANKS ,DOLLAR ,EOS
. ,ICOMMA ,IDOLLAR ,IFT ,IPTAB ,ITAB
. ,KAM10 ,KBPC ,KBPW ,KCPCD ,KERTYP
. ,KZERO ,NOPRNT ,TAB1
./IPARAM/ ABORT ,KIUNIT ,KOUNTIT ,LIMALF ,LOCK ,LOCX
. ,NOLIST ,NSTDIR ,TRACE
./ISTAKI/ STACK ,ISTDIM ,KSTACK ,LEVLIM
C
INTEGER BLANK ,BLANKS ,EOS ,IDOLLAR ,TAB1 ,TRACE
DATA BADCAL/6HBADCAL/
DOUBLE PRECISION ALFARG, ISUBN, STACK, VALUE, DBLANK, DARG
LOGICAL ABORT, DOLLAR, ERMARK, LIMALF, LOCK, NOLIST, NONEW, MODALL,
.NSTDIR, SMCHR, TEST
EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE), (NAME,ANAME)
EQUIVALENCE (ICOMNI,ISUB), (IMAGE,IMAGE1), (IPARAM,ABORT)
. , (DBLANK,BLANK)
ISUBC = 0
DO 10 I = 1,10
10 ISUBN(I) = DBLANK
IF(TRACE.EQ.0) GO TO 99
WRITE (KOUNTIT,410)
GO TO 99
ENTRY DEBUGC(ISUBNA)
DOUBLE PRECISION ISUBNA
C          * NEW SUBROUTINE CALLED
IF (ISUBC.GT.10) GO TO 98
ALFARG = DBLANK
NUMARG = -1
ISUBC = ISUBC+1
ISUBN(ISUBC) = ISUBNA
IF(TRACE .GE.4) GO TO 400
GO TO 99
C
98 ALFARG = BADCAL
C          * CALLS MIGHT GET OUT OF RANGE
C NUMARG = ISUBC
GO TO 400
C
ENTRY DEBUGR

```

C		* CALL AT RETURN	59
	IF (ISUBC.LT.1) GO TO 98		60
	ALFARG = DBLANK		61
	NUMARG = -1		62
	ISUBN(ISUBC) = DBLANK		63
	ISUBC = ISUBC - 1		64
	IF(TRACE .GE.4) GO TO 400		65
	GO TO 99		66
C			67
	ENTRY DEBUG2(ALFAR,NUMAR)		68
	DOUBLE PRECISION ALFAR		69
	NUMARG = NUMAR		70
	ALFARG = ALFAR		71
20	IF (TRACE .LT.1) GO TO 99		72
400	WRITE (KOUNIT,405) (ISUBN(I),I=1,4),ALFARG,NUMARG,		73
	.KCOUNT,SMCHR,KCH,KFLD1,ICOMP,KFLD2,LCOMP		74
99	RETURN		75
C			76
	ENTRY DEBUG3(ALFAR,DNUMAR,ITYPE)		77
	DOUBLE PRECISION DNUMAR		78
	ENTRY DEBUG4(ALFAR,DNUMAR,ITYPE)		79
	DARG = DNUMAR		80
	NUMARG=NUMAR		81
	ALFARG = ALFAR		82
	IF (TRACE .LT.1) GO TO 99		83
	GO TO (30,40,50,60), ITYPE		84
30	WRITE (KOUNIT,406) (ISUBN(I),I=1,4),ALFARG,NUMARG,		85
	.KCOUNT,SMCHR,KCH,KFLD1,ICOMP,KFLD2,LCOMP		86
	GO TO 99		87
40	WRITE (KOUNIT,407) (ISUBN(I),I=1,4),ALFARG,NUMARG,		88
	.KCOUNT,SMCHR,KCH,KFLD1,ICOMP,KFLD2,LCOMP		89
	GO TO 99		90
50	WRITE (KOUNIT,408) (ISUBN(I),I=1,4),ALFARG,NUMARG,		91
	.KCOUNT,SMCHR,KCH,KFLD1,ICOMP,KFLD2,LCOMP		92
	GO TO 99		93
60	WRITE (KOUNIT,409) (ISUBN(I),I=1,4),ALFARG,NUMARG,		94
	.KCOUNT,SMCHR,KCH,KFLD1,ICOMP,KFLD2,LCOMP		95
	GO TO 99		96
405	FORMAT(1H ,29X,5(A6,1X),I24,		97
	.1H(,I3,L2,1H),A6,2H (,I3,I2,1H),2H (,I3,I2,1H))		98
406	FORMAT(1H ,29X,5(A6,1X),E24.8,		99
	.1H(,I3,L2,1H),A6,2H (,I3,I2,1H),2H (,I3,I2,1H))		100
407	FORMAT(1H ,29X,5(A6,1X),11X,0I2,1X,		101
	.1H(,I3,L2,1H),A6,2H (,I3,I2,1H),2H (,I3,I2,1H))		102
408	FORMAT(1H ,29X,5(A6,1X),16X,A8,		103
	.1H(,I3,L2,1H),A6,2H (,I3,I2,1H),2H (,I3,I2,1H))		104
409	FORMAT(1H ,29X,5(A6,1X),D24.18,		105
	.1H(,I3,L2,1H),A6,2H (,I3,I2,1H),2H (,I3,I2,1H))		106
410	FORMAT(29X,11HDEBUG TRACE,52X,9HI--KCOUNT,5X,8HI--LIST1/		107
	.68X,9HD ARG OR,15X,9HI I-SMCHR,5X,10HI I--ICOMP /		108
	.29X,19HSUBROUTINES CALLED,11X,7HALF ARG,7X,11HNUMERIC ARG,9X,		109
	.3HI I,1X,5HKCH-I,5X,16HI I I--LIST2 /		110
	.92X,34I I,5X,1HI,5X,18HI I I I--LCOMP)		111
	END		112

```

$IBFTC STAC<P DECK
SUBROJTINE STACKP
DIMENSION IFT(27), IPTAB(21), ITAB(65)
DIMENSION ANAME(15), IMAGE(80), IMAGE1(81), IPARAM(9)
.      ,KSTACK(27),NAME(15),RVALUE(2),STACK(27)
C
COMMON
./ICOMNI/ VALUE      ,ICOMP      ,IFNTYP      ,IMAGE1      ,IRADIX      ,ISUB
.      ,KCH          ,KCNVRT      ,KCOUNT      ,KDIF          ,KFLD1      ,KFLD2
.      ,LCOMP        ,LCNVRT      ,LEVEL        ,LFRT          ,LOOK
.      ,MCONVRT      ,MDIF        ,MODALL        ,MSTOR
.      ,NAME          ,NERROR      ,NONEW         ,NOTARG
.      ,SMCHR        ,TEST        ,ERMARK
./ICNSTI/ BLANK      ,BLANKS      ,DOLLAR        ,EOS
.      ,ICOMMA      ,IDOLLAR      ,IFT           ,IPTAB        ,ITAB
.      ,KAM10        ,KBPC         ,KBPW          ,KCPCD        ,KERTYP
.      ,KZERO        ,NOPRNT      ,TAB1
./IPARAM/ ABORT      ,KIUNIT      ,KOUNIT        ,LIMALF      ,LOCK        ,LOCX
.      ,NOLIST      ,NSTDIR      ,TRACE
./ISTAKI/ STACK      ,ISTDIM      ,KSTACK        ,LEVLIM
C
INTEGER BLANK      ,BLANKS      ,EOS           ,IDOLLAR      ,TAB1        ,TRACE
DOUBLE PRECISION STACK, VALUE
LOGICAL ABORT,DOLLAR,ERMARK,LIMALF,LOCK,NOLIST,NONEW,MODALL,
.NSTDIR,SMCHR,TEST
EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE),(NAME,ANAME)
EQUIVALENCE (ICOMNI,ISUB),(IMAGE,IMAGE1),(IPARAM,ABORT)
IF(TRACE .LT. 3) GO TO 99
LEVELX = LEVEL + 3
WRITE(KOUNIT,85)LEVEL,VALUE,KVALUE,RVALUE,LOCX
WRITE(KOUNIT,89)(STACK(I),I=1,LEVELX)
WRITE(KOUNIT,86)(KSTACK(I),I=1,LEVELX)
89  FORMAT(11D12.4)
86  FORMAT(11I12)
85  FORMAT(7H LEVEL=I3,7H VALUE=D25.17,8H KVALUE=I3,3H ,0,2(1X,D12),
.6H LOCX=I6)
99  RETURN
END

```

```

$IBMAP IFLD DECK
TTL IFLD AND IFLD4 PROGRAM
ENTRY IFLD
ENTRY IFLD4
* FUNCTION (I,N,DATA)
* START EXTRACTION WITH BIT I+1
* EXTRACT N BITS FROM DATA AND RETURN IN REGISTER 1
IFLD SAVE
CAL* 3,4 I
STA LGL1
CAL* 4,4 N
STA LGL2
LDQ* 5,4 DATA
LGL1 LGL ** **=I
ZAC

```

```

$IBFTC STACKP  DECK
SUBROJTINE STACKP
DIMENSION IFT(27), IPTAB(21), ITAB(65)
DIMENSION ANAME(15) ,IMAGE(80) ,IMAGE1(81),IPARAM(9)
.      ,KSTACK(27),NAME(15) ,RVALUE(2) ,STACK(27)
C
COMMON
./ICOMNI/ VALUE      ,ICOMP      ,IFNTYP      ,IMAGE1      ,IRADIX      ,ISUB
.      ,KCH          ,KCNVRT      ,KCOUNT      ,KDIF        ,KFLD1       ,KFLD2
.      ,LCDMP        ,LCNVRT      ,LEVEL        ,LFRT        ,LOOK
.      ,MCNVRT      ,MDIF        ,MODALL       ,MSTOR
.      ,NAME         ,NERROR      ,NONEW        ,NOTARG
.      ,SMCHR        ,TEST        ,ERMARK
./ICNSTI/ BLANK      ,BLANKS      ,DOLLAR      ,EOS
.      ,ICOMMA      ,IDOLAR      ,IFT          ,IPTAB       ,ITAB
.      ,KAM10       ,KBPC        ,KBPW        ,KCPCD       ,KERTYP
.      ,KZERO        ,NOPRNT      ,TAB1
./IPARAM/ ABORT      ,KIUNIT      ,KOUNIT      ,LIMALF     ,LOCK        ,LOCX
.      ,NOLIST      ,NSTDIR      ,TRACE
./ISTACK/ STACK     ,ISTDIM      ,KSTACK      ,LEVLIM
C
INTEGER BLANK      ,BLANKS      ,EOS          ,IDOLAR     ,TAB1        ,TRACE
DOUBLE PRECISION STACK, VALUE
LOGICAL ABORT,DOLLAR,ERMARK,LIMALF,LOCK,NOLIST,NONEW,MODALL,
.NSTDIR,SMCHR,TEST
EQUIVALENCE (STACK,ISTACK), (VALUE,KVALUE,RVALUE),(NAME,ANAME)
EQUIVALENCE (ICOMNI,ISUB),(IMAGE,IMAGE1),(IPARAM,ABORT)
IF(TRACE .LT. 3) GO TO 99
LEVELX = LEVEL + 3
WRITE(KOUNIT,85)(LEVEL,VALUE,KVALUE,RVALUE,LOCX
WRITE(KOUNIT,89)(STACK(I),I=1,LEVELX)
WRITE(KOUNIT,86)(KSTACK(I),I=1,LEVELX)
89  FORMAT(11D12.4)
86  FORMAT(11I12)
85  FORMAT(7H LEVEL=I3,7H VALUE=D25.17,8H KVALUE=I3,3H .D,2(1X,D12),
.6H LOCX=I6)
99  RETURN
END

```

```

$IBMAP IFLD      DECK
TTL             IFLD AND IFLD4 PROGRAM
ENTRY          IFLD
ENTRY          IFLD4
*              FUNCTION (I,N,DATA)
*              START EXTRACTION WITH BIT I+1
*              EXTRACT N BITS FROM DATA AND RETURN IN REGISTER 1
IFLD          SAVE
CAL*          3,4          I
STA           LGL1
CAL*          4,4          N
STA           LGL2
LDQ*          5,4          DATA
LGL1          LGL         **          **=I
ZAC

```



LGL2	LGL	**	***=N		15
	XCL				16
	XCA				17
	RETURN	IFLD			18
	SPACE	2			19
*	FUNCTION OR SUBROUTINE IFLD4(NUDATA,I,N,DATA)				20
*	PUT LOW ORDER N BITS OF NUDATA IN DATA				21
*	STARTING WITH THE ITH+1 BIT.				22
*	THE MODIFIED DATA IS LEFT IN REG 0 FOR FUNCTION TYPE USAGE				23
*	I.E. IX=IFLD4(NUDATA),I,N,DATA)				24
*	DATA IS UNCHANGED IX CONTAINS THE CHANGED DATA				25
IFLD4	SAVE				26
	CAL*	4,4	I		27
	STA	I1			28
	STA	I2			29
	CAL*	5,4	N		30
	STA	N1			31
	STA	N2			32
	ZAC				33
	LDQ*	6,4	DATA		34
I1	LGL	**	***=I		35
	SLW	TEMP			36
N1	LGL	**	***=N		37
	CAL*	3,4	Z		38
N2	LGR	**	***=N		39
	CAL	TEMP			40
I2	LGR	**	***=I		41
	XCA				42
	RETURN	IFLD4			43
TEMP	PZE				44
	END				45

## APPENDIX B

### HUFF INPUT ROUTINE

The first version of the Huff Input Routine was reported in reference 6. The Huff Input Routine provides more versatility in reading input data into the computer than the NAMELIST feature in FORTRAN. The Huff Input Routine has the ability to make simple arithmetic manipulations (such as conversion of units) during loading and to load alpha-numeric data. While not an indispensable feature, it has been found to be quite convenient. The Huff Input Routine also allows for the automatic printout of data cards at execution time.

The following sections contain a general description of the Huff Input Routine and its usage.

#### Usage

The programmer transfers control to the INPUT routine with a standard FORTRAN IV call

CALL INPUT (5, 6, 1, X, ITABLE)

Argument 1 is the system input tape number (5 on the Lewis system). Argument 2 is the system output tape number (6 on the Lewis system). Argument 3 is the identifying number of a data group. This value is compared with an identification number occurring on the input card (\$DATA card). If the values agree, the data are processed until another \$DATA or end-of-data (\$END) card is encountered. If the values do not agree, no data are processed and control is returned to the calling program. Argument 4 is the Array X, which serves as a reference point for the storing of input data. Since all data are stored relative to X, the programmer must provide fixed relations between the location of X and other locations to be loaded (e. g., through the use of common blocks and/or equivalence statements). In this case, X is "WORD," the first name in the labeled common block "ALL." Common blocks ALL, DESIGN, FRONT, SIDE, BACK, and DUMMY are in all routines and hence are loaded sequentially so that the location of all variables is known. Argument 5 is the array ITABLE, which contains the names of the variables used on the cards and their subscript location relative to X. Sufficient space must be provided in the calling program for storing the table of names. This is done by a DIMENSION statement. The dimension of ITABLE(3) must initially be zero.

## Types of Input Statements

\$DATA statement. - The \$DATA statement identifies a group of data with an identification number. It must be the first statement on a card. For example, \$DATA(1) or \$D(1) on the first card of a data group causes the value 1 to be compared with argument 3 in the calling sequence. If unequal, control is returned to the calling program. If equal, data are loaded until the next \$DATA or \$END statement is reached.

\$TABLE statement. - The \$TABLE statement makes a list of names needed for loading data. Consider for example that the real variable names VELOCITY, MASS, and RADIUS are to be assigned to memory locations X(1), X(2), and X(3), respectively. The card would be punched \$TABLE (.REAL., 1 = VELOCITY, 2 = MASS, 3 = RADIUS). These variables will be treated as real in any subsequent loading of data. A limit of 15 computer words is placed on the length of a name. Since .REAL. is what designates the mode of a name, a name may begin with any alphabetic letter. For example, the statement \$TABLE (.INTEGER., 20 = INDEX, SUBSCRIPT, I) will place these names in the table and any values subsequently loaded will be stored in X(20), X(21), and X(22), respectively, as integers. In a similar manner \$TABLE (.DOUBLE PRECISION., 10 = RADIUS DOUBLE, .LOGICAL., 12 = SWITCH1) causes the name RADIUS DOUBLE to be stored in the table as a double-precision variable equivalent to X(10) and X(11), and the logical variables SWITCH 1 and SWITCH 2 will be equivalent to X(12) and X(13).

Note that \$TABLE statements are loaded as Number = Name to avoid confusion with loading statements.

## Loading Statement

The loading statement loads data by taking the name of a variable previously appearing in a \$TABLE statement and setting it equal to a value which may be of several forms.

Numeric values. - Standard FORTRAN language is used; for example, VELOCITY = 3.4, MASS = 32 (no decimal point is needed and MASS will have the REAL value 32), RADIUS = 4E21, and INDEX = 3. Data can be continued from one card to another; for example, SUBSCRIPT may appear at the end of one card and = 47 on the next card.

Subscripts may be used. Since 3 = RADIUS, RADIUS(2) = 6, 10, 12, , 14 will put real numbers in X(4), X(5), X(6), and X(8) and leave X(7) unchanged because of the double comma. If new values are assigned to a variable before the next \$D(1) card, the new value will override the previous one. For example, RADIUS(2) = 8 will override the RADIUS(2) = 6 card.

Internally addressed values. - An internally addressed value is one that refers to the contents of memory by name. RADIUS(7) = RADIUS(3); RADIUS(INDEX) causes RADIUS(7) to be replaced by the value of RADIUS(3) and RADIUS(8) to also be replaced

by RADIUS(3) since INDEX = 3.

The statement RADIUS(7) = RADIUS(INDEX + 1), however, is ILLEGAL.

Arithmetic expressions. - Provisions have been made to allow arithmetic operations to be performed on data at execution time. The operations + (addition), - (subtraction), \*(multiplication), and /(division) and the functions, included among which are SQRT, EXP, SIN, COS, and PWR(x,y)(=x\*\*y), may be used with name or numbers (or any expression that has a value) to compute the value of an arithmetic expression. Parentheses may be used to indicate the order of performing the operations. The computations are analyzed from left to right and any intermediate results are stored in up to 24 locations in the core (the stack) which is sufficient for fairly complex expressions. All numeric operations are carried out in double-precision floating-point FORTRAN arithmetic. As an example, RADIUS(2) = RADIUS(2)\*SQRT(RADIUS(2)) or RADIUS(2) = PWR(RADIUS(2), 1.5) will set RADIUS(2) =  $8^{3/2}$ .

Alphanumeric expressions. - Alphanumeric data may be entered by placing the variable name in the "REAL" list and then setting the variable equal to the data by first enclosing in parentheses the length of the word to be read in. As an example,

Q = (A39)THIS IS AN EXAMPLE OF ALPHANUMERIC DATA

The (A39) gives the length of the data including imbedded blanks. Of course, since on the IBM 7094 there are six characters per word, Q must internally be dimensioned to at least 7.

## Printing Input Cards

Each input card processed will normally be written on the tape specified by the second argument of the calling sequence. An end-of-statement symbol read on the card will cause interpretation of the card to stop at that point and permit comments to be placed on the remainder of the card to be printed with the output. In order to avoid printing the card at all, the nonprint character is placed in the next column following the end-of-statement character. The developers of the routine selected the sign ‡ for both characters. This is punched as a colon on an IBM Model 29 Key punch and corresponds to a 2-8 punch.

If the character following the end-of-statement symbol is other than a nonprint character, it is inserted as the printer control character in the first position of the output format before the card is written on the output tape. If no end-of-statement character occurs on the card, a blank printer control character is used. Comment cards having the end-of-statement character as the first nonblank character will be printed and may be placed anywhere except in a continued alphabetic field.

In summary, the end-of-statement character has the effect of moving the end of the card forward to the column ahead of the end-of-statement character. The column following it is printer control.

If the control parameter `NOLIST` is true, printing is suppressed for all cards.

## APPENDIX C

### SYMBOLS

#### General Symbols Internal to Program

Variables in program are formed by combining these symbols.

#### Station Numbers

See figures 1 to 4 for each type of engine.

#### Thermodynamic Properties

AM	Mach number
FAR	fuel-air ratio, f/a
H	enthalpy, Btu/lbm
P	total pressure, atm
PS	static pressure, atm
S	entropy, Btu/ <sup>o</sup> R/lbm
T	total temperature, <sup>o</sup> R
TS	static temperature, <sup>o</sup> R
V	velocity, ft/sec

#### Component Symbols

A,AFT	afterburner
B	burner
C	inner compressor
COM	combustor
D	fan duct
F	first or fan compressor

M	core nozzle
NOZ	nozzle
OB	overboard
T	total
THP	inner (high pressure) turbine
TLP	outer (low pressure) turbine

### Engine Symbols

BL	bleed, lbm/sec
CN	ratio of corrected speed to design corrected speed
DHT	turbine delta enthalpy, Btu/lbm
DHTC	turbine delta enthalpy (temperature corrected), $(H_{in} - H_{out})/T_{in}$ , Btu/ $^{\circ}$ R/lbm
DP	pressure drop, $\Delta P/P$
DT	temperature change, $^{\circ}$ R
ETA	efficiency
ETAR	ram recovery, $P_2/P_1$
HPEXT	horsepower extracted
PCBL	fractional bleed
PCN	percent of design shaft speed
PR	pressure ratio
TFF	turbine flow function, $\text{lbm}\sqrt{^{\circ}\text{R}}/(\text{psia})(\text{sec})$
WA	airflow, lbm/sec
WF	fuel flow, lbm/sec
WG	gas flow, lbm/sec
Z	ratio of pressure ratios

### Miscellaneous Symbols

A	area, ft
ALTP	altitude, ft

AM	Mach number of aircraft
BYPASS	bypass ratio (fan duct air/air entering intermediate compressor)
C	when following component symbol, signifies "corrected"
CF	correction factor, when used following component symbol
CS	ambient speed of sound, ft/sec
CV	nozzle velocity coefficient
DEL	delta degradation coefficient
DS	design value
DUM	dummy value
FG	gross thrust, lbf
FGM	momentum thrust, lbf
FGP	pressure thrust, lbf
FN	net thrust, lbf
FRD	ram drag, lbf
GU	initial or guessed values
ITRYS	number of loops through engine before quitting
LOOP	variable counter
LOOPER	number of loops through engine counter
SFC	specific fuel consumption, lbm/lbf/hr
TOLALL	tolerance on convergence
VA	velocity of aircraft, ft/sec
VJ	jet velocity, ft/sec

### Input Symbols

ALTP	altitude, ft
AM	Mach number of aircraft
AM6	design afterburner entrance Mach number
AM23	design duct-burner entrance Mach number



AM55 design low-pressure-turbine exit Mach number  
 A6 area at afterburner entrance (calculated from AM6), ft<sup>2</sup>  
 A8 main nozzle throat area (can be changed at off-design), ft<sup>2</sup>  
 A28 fan duct nozzle throat area (see A8), ft<sup>2</sup>  
 CNHPDS design corrected speed - inner turbine  
 CNLPDS design corrected speed - outer turbine  
 CVDNOZ nozzle thrust coefficient (tabular lookup)  
 CVMNOZ nozzle thrust coefficient (tabular lookup)  
 DELFG gross-thrust delta degradation multiplier  
 DELFN net-thrust delta degradation multiplier  
 DELSFC specific-fuel-consumption delta degradation multiplier  
 DPAFDS afterburner design pressure drop,  $\Delta P/P$   
 DPCODS combustor design pressure drop,  $\Delta P/P$   
 DPDUDS duct design pressure drop,  $\Delta P/P$   
 DTCODS combustor design temperature increase (automatically set to T4 - T3), °R  
 ETAA afterburner efficiency (not required)  
 ETAADS afterburner efficiency at design  
 ETABDS combustor efficiency at design  
 ETACDS inner compressor adiabatic efficiency at design  
 ETAD duct-burner combustion efficiency  
 ETAFDS front (outer) compressor adiabatic efficiency at design  
 ETAR inlet pressure recovery (ram recovery)  
 ETHPDS high-pressure-(inner) turbine design adiabatic efficiency  
 ETLPDS low-pressure-(outer) turbine design adiabatic efficiency  
 FAN logical control for fan and turbojet engines  
 HPEXT horsepower extraction  
 IAFTBN index on afterburning desired  
 IAMTP index on ram or inlet operation desired  
 IDBURN index on duct burning desired  
 IDCD duct nozzle convergent-divergent when IDCD equals 1 (design or off-design)

**IDES** index for design point; must be set equal to 1 to design engine; zeroed automatically  
**IDUMP** index for dumping of error matrix  
**IGASMIX** index for mixed-flow or non-mixed-flow turbofan  
**IMCD** main nozzle convergent-divergent when IMCD equals 1 (design or off-design)  
**ISPOOL** index for number of compressors for turbojet engines  
**ITRYS** index for maximum number of iterations  
**MODE** independent variable designator for engine operation  
**NOZFLT** index for floating main or duct nozzle  
**PCBLC** ratio of compressor bleed to turbines to compressor airflow  
**PCBLDU** ratio of compressor bleed leaked into fan duct to total compressor bleed flow  
**PCBLF** ratio of bleed from outer compressor to fan airflow dumped overboard (i. e., leakage)  
**PCBLHP** fraction of PCBLC used for high-pressure (inner) turbine (cooling)  
**PCBLLP** fraction of PCBLC used for low-pressure (outer) turbine (cooling)  
**PCBLOB** ratio of inner compressor bleed to compressor airflow (overboard for customer use)  
**PCNC** inner-compressor shaft speed as a percent of design shaft speed  
**PCNCDS** design inner-compressor shaft speed  
**PCNF** outer-compressor shaft speed as percent of design  
**PCNFDS** design outer-compressor shaft speed as percent of design shaft speed  
**PRCDS** design inner-compressor pressure ratio  
**PRFDS** design outer-compressor pressure ratio  
**PS55** static pressure at low-pressure-turbine exit  
**P2** compressor-face total pressure (for nonstandard days only), atm  
**TFHPDS** design inner-turbine flow function  
**TFLPDS** design outer-turbine flow function  
**TOLALL** tolerance on error matrix  
**T2** compressor-face total temperature (for nonstandard days only), T1 + T2  
**T24** duct-burner exit temperature, °R

T4	combustor exit temperature, °R
T4DS	design combustor exit temperature, °R
T7	afterburner exit temperature, °R
T7DS	design afterburner exit temperature, °R
WACCDS	design inner-compressor corrected airflow (turbofans only), lbm/sec
WAFCDs	design outer-compressor corrected airflow, lbm/sec
WFA	fuel flow rate to afterburner (IAFTBN = 2 only), lbm/sec
WFB	fuel flow rate to main burner (MODE = 2 only), lbm/sec
WFBDS	design fuel flow rate to main burner (MODE = 2 only), lbm/sec
ZCDS, ZFDS	design ratio of inner compressor and fan compressor pressure ratios, respectively; equals pressure ratio at design point on design speed line minus value of pressure ratio of lowest point on speed line divided by high (surge) value minus low value of pressure ratio on the design speed line

## Output Symbols<sup>1</sup>

A	area, ft <sup>2</sup>
ALTP	altitude, ft
AM	Mach number
BLC	bleed flow out of compressor, lbm/sec
BLF	bleed flow out of fan (dumped overboard), lbm/sec
BLHP	bleed flow into high-pressure turbine, lbm/sec
BYPASS	ratio of airflow into fan duct to airflow into inner compressor
CNC	corrected shaft speed - inner compressor
CNF	corrected shaft speed - fan
CNHP	corrected shaft speed - high-pressure turbine, $PCNC/\sqrt{T_{in}}$
CNHPCF	corrected speed - high-pressure-turbine correction factor
CNLP	corrected speed - low-pressure turbine, $PCNF/\sqrt{T_{in}}$

<sup>1</sup>Some symbols, such as T4, are followed by station numbers, see appropriate figure for each engine to determine station numbers.

CNLPCF	corrected speed - low-pressure-turbine correction factor
CVDNOZ	velocity coefficient of fan nozzle
CVMNOZ	velocity coefficient of core nozzle
DHHPCF	high-pressure-turbine delta enthalpy correction factor
DHLPCF	low-pressure-turbine delta enthalpy correction factor
DHTC	work done by high-pressure turbine, Btu/lbm
DHTCHP	enthalpy change temperature corrected - high-pressure turbine, Btu/ <sup>o</sup> R/atm/lbm
DHTCLP	enthalpy change temperature corrected - low-pressure turbine, Btu/ <sup>o</sup> R/atm/lbm
DHTF	work done by low-pressure turbine, Btu/lbm
DPCOM	$(\Delta P/P)_{\text{combustor}}$
DPDUC	$(\Delta P/P)_{\text{fan duct}}$
DTCOCF	temperature-rise-across-combustor correction factor
ETAB	combustor efficiency
ETABCF	combustor efficiency correction factor
ETAC	inner-compressor adiabatic efficiency
ETACCF	inner-compressor efficiency correction factor
ETAD	duct-burner efficiency
ETAF	fan adiabatic efficiency
ETAFCF	fan efficiency correction factor
ETATHP	high-pressure-turbine adiabatic efficiency
ETATLP	low-pressure-turbine adiabatic efficiency
ETHPCF	high-pressure-turbine efficiency correction factor
ETLPCF	low-pressure-turbine efficiency correction factor
FAR	fuel-air ratio, f/a
FG	gross thrust, lbf
FGM	momentum thrust, lbf
FGP	pressure thrust, lbf
FN	net thrust, lbf

FRD	ram drag, lbf
HPEXT	horsepower extracted, hp
P	pressure, atm
PCBLC	fraction of compressor exit air bled for cooling or lost to cycle
PCBLDU	fraction of bled air out of compressor which leaks into fan duct
PCBLF	fraction of fan exit airflow lost overboard
PCBLHP	fraction of compressor bleed air put into high-pressure turbine
PCBLLP	fraction of compressor bleed air put into low-pressure turbine
PCNC	inner compressor shaft speed as fraction of design
PCNF	fan compressor shaft speed as fraction of design
PRC	pressure ratio of inner compressor
PRCCF	pressure-ratio-of-inner-compressor correction factor
PRF	pressure ratio of fan or outer compressor
PRFCF	pressure-ratio-of-fan-or-outer-compressor correction factor
PS	static pressure, atm
SFC	specific fuel consumption, lbm/(lbm/hr)
T	temperature, °R
TFFHP	high-pressure-turbine flow function, $(\text{lbm})\left(\sqrt{\text{°R}}\right)(\text{in.}^2)/(\text{sec})(\text{lbf})$
TFFLP	low-pressure-turbine flow function, $(\text{lbm})\left(\sqrt{\text{°R}}\right)(\text{in.}^2)/(\text{sec})(\text{lbf})$
TFHPCF	high-pressure-turbine flow function correction factor
TFLPCF	low-pressure-turbine flow function correction factor
T2DS	design exit temperature of fan, °R
T21DS	design exit temperature of inner compressor, °R
V	velocity, ft/sec
VA	velocity of aircraft, ft/sec
VJD	fan duct exhaust velocity, ft/sec
VJM	core exhaust velocity, ft/sec
WA	airflow, lbm/sec
WA3CDS	corrected airflow in combustor at design, lbm/sec
WAC	inner-compressor airflow, lbm/sec

WACC	inner-compressor corrected airflow, lbm/sec
WACCF	inner-compressor corrected airflow correction factor
WAD	fan duct airflow, lbm/sec
WAF	fan airflow, lbm/sec
W AFC	fan corrected airflow, lbm/sec
WAFCF	fan corrected airflow correction factor
WFA	fuel flow rate to afterburner, lbm/sec
WFB	fuel flow rate to combustor, lbm/sec
WFD	fuel flow rate to duct burner, lbm/sec
WFT	total fuel flow rate, lbm/sec
WG	gas flow rate, lbm/sec
WGT	total gas flow rate, lbm/sec
ZC	ratio of inner compressor pressure ratios
ZF	ratio of fan pressure ratios

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TABLE I. - VARIABLES AND ERRORS

	Two-spool turbofan	Mixed-flow turbofan	Two-spool turbojet	One-spool turbojet
Variable 1	ZF	ZF	ZF	ZF
Variable 2	PCNF	PCNF	PCNF	PCNF
Variable 3	ZC	ZC	ZC	TFFLP
Variable 4	PCNC	PCNC	PCNC	-----
Variable 5	TFFHP	TFFHP	TFFHP	-----
Variable 6	TFFLP	TFFLP	TFFLP	-----
Error 1	<u>TFHCAL - TFFHP</u>	<u>TFHCAL - TFFHP</u>	<u>TFHCAL - TFFHP</u>	<u>TFLCAL - TFFLP</u>
	TFHCAL	TFHCAL	TFHCAL	TFLCAL
Error 2	<u>DHTCC - DHTCHP</u>	<u>DHTCC - DHTCHP</u>	<u>DHTCC - DHTCHP</u>	<u>DHTCF - DHTCLP</u>
	DHTCC	DHTCC	DHTCC	DHTCF
Error 3	<u>TFLCAL - TFFLP</u>	<u>TFLCAL - TFFLP</u>	<u>TFLCAL - TFFLP</u>	<u>P7R - P7</u>
	TFLCAL	TFLCAL	TFLCAL	P7R
Error 4	<u>DHTCF - DHTCLP</u>	<u>DHTCF - DHTCLP</u>	<u>DHTCF - DHTCLP</u>	-----
	DHTCF	DHTCF	DHTCF	-----
Error 5	<u>P25R - P25</u>	<u>PS25 - PS55</u>	<u>WAF - WAC - BLF</u>	-----
	P25R	PS25	WAC	-----
Error 6	<u>P7R - P7</u>	<u>P7R - P7</u>	<u>P7R - P7</u>	-----
	P7R	P7R	P7R	-----
Matrix size	6 × 6	6 × 6	6 × 6	3 × 3



TABLE II. - INPUTS REQUIRED FOR BASIC CYCLES

Variable	Units	Definition	Two-spool turbofan	Mixed-flow turbofan	Two-spool turbojet	One-spool turbojet
PRFDS	-----	Fan pressure ratio	Yes	Yes	Yes	Yes
WAFCDs	lb/sec	Fan corrected airflow	↓	↓	↓	↓
ETAfDS	-----	Fan efficiency	↓	↓	↓	↓
ZFDS	-----	Design Z of fan	↓	↓	↓	↓
PCNFDS	-----	Corrected speed of fan	↓	↓	↓	↓
PRCDS	-----	Compressor pressure ratio	↓	↓	↓	No
WACCDS	lb/sec	Compressor corrected airflow	↓	↓	No	↓
ETACDS	-----	Compressor efficiency	↓	↓	Yes	↓
ZCDS	-----	Design Z of compressor	↓	↓	↓	↓
PCNCDs	-----	Corrected speed of compressor	↓	↓	↓	↓
ETABDS	-----	Combustor efficiency	↓	↓	↓	Yes
DPCODS	-----	Combustor pressure drop, $\Delta P/P$	↓	↓	↓	Yes
T4DS	$^{\circ}R$	Turbine inlet temperature	↓	↓	↓	Yes
TFHPDS	$\frac{lb\sqrt{^{\circ}R}}{(sec)(psia)}$	High-pressure-turbine flow function	↓	↓	↓	No
CNHPDS	-----	Corrected speed - high-pressure turbine	↓	↓	↓	No
ETHPDS	-----	Efficiency - high-pressure turbine	↓	↓	↓	No
TFLPDS	$\frac{lb\sqrt{^{\circ}R}}{(sec)(psia)}$	Low-pressure-turbine flow function	↓	↓	↓	Yes
CNLPDS	-----	Corrected speed - low-pressure turbine	↓	↓	↓	Yes
ETLPDS	-----	Efficiency - low-pressure turbine	↓	↓	↓	Yes
DPDUDS	-----	$\Delta P/P$ of fan duct	↓	↓	No	No
DPAFDS	-----	$\Delta P/P$ of afterburner	↓	↓	Yes	Yes
FAN	-----	Logical variable	. TRUE.	. TRUE.	. FALSE.	. FALSE.
ISPOOL	-----	Number of spools	2	2	2	1

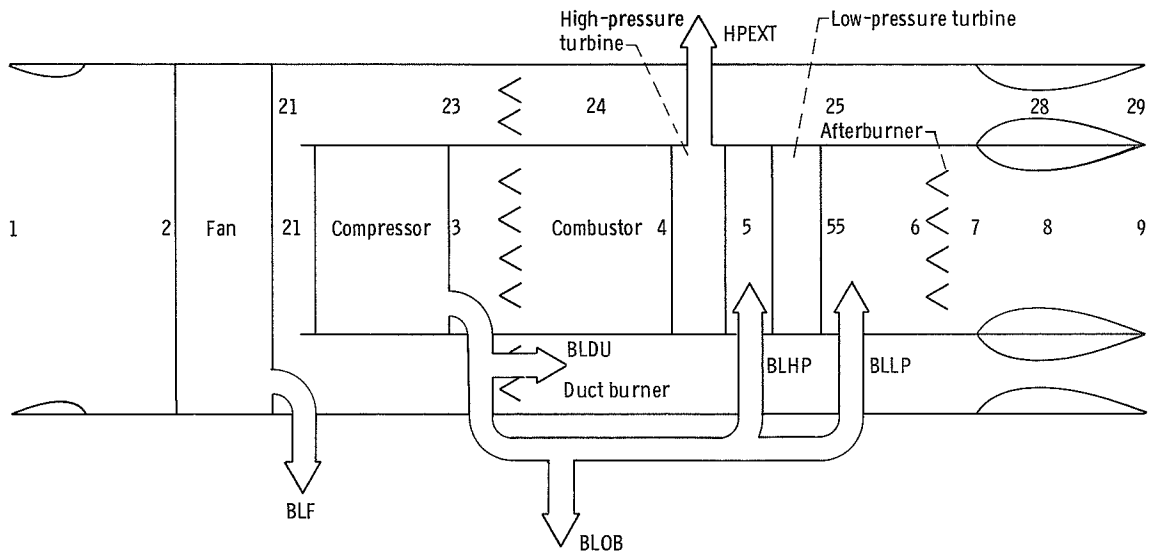


Figure 1. - Schematic of non-mixed-flow duct-burning and/or afterburning turbofan.

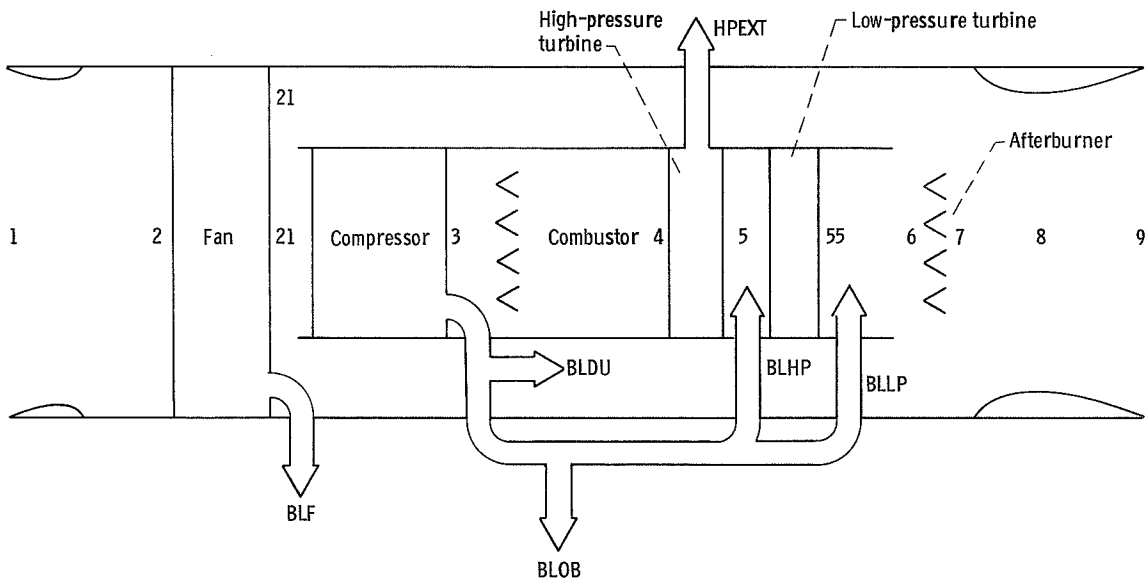


Figure 2. - Schematic of mixed-flow afterburning turbofan.

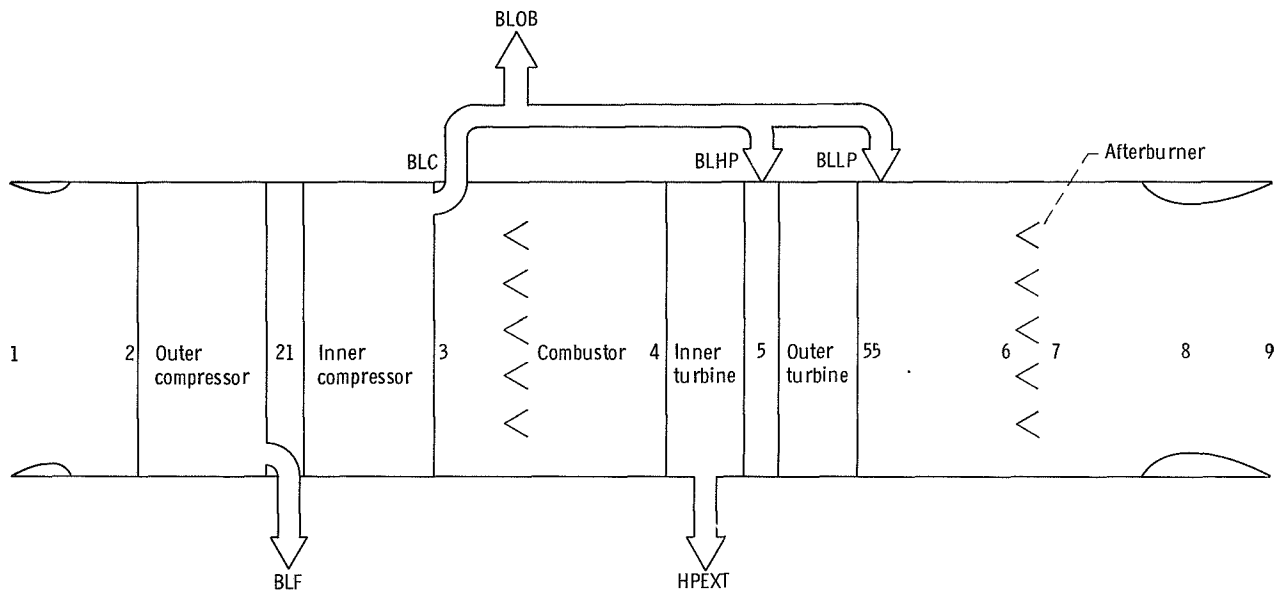


Figure 3. - Schematic of two-spool turbojet.

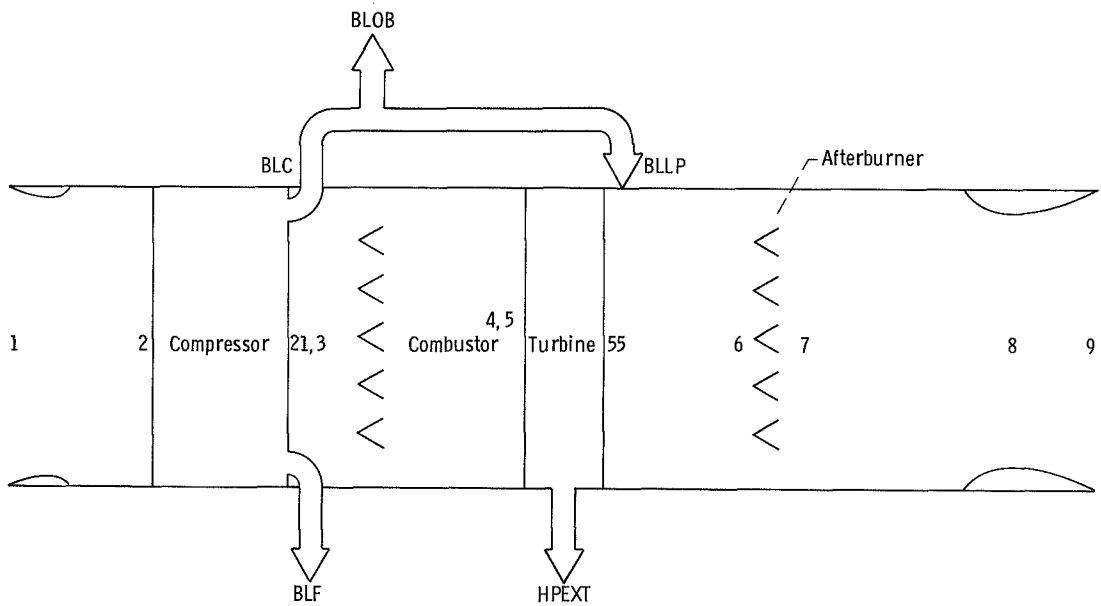


Figure 4. - Schematic of one-spool turbojet.



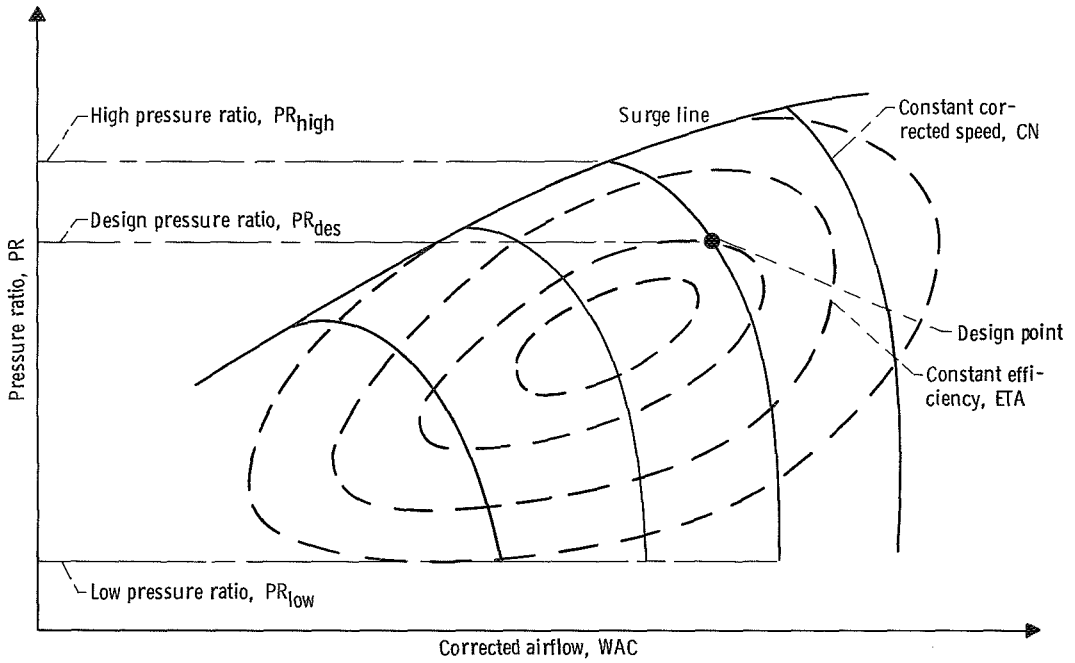


Figure 6. - Example of a specific fan-compressor map.  $Z = (PR_x - PR_{low}) / (PR_{high} - PR_{low})$ .

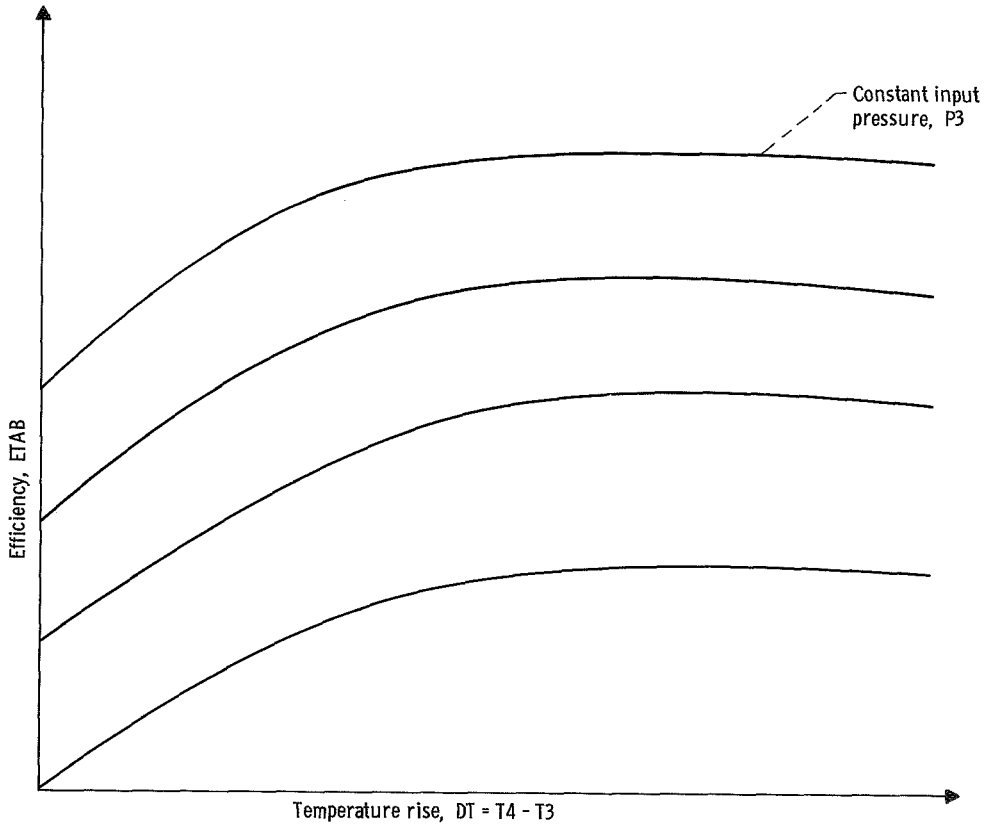


Figure 7. - Example of combustor map.

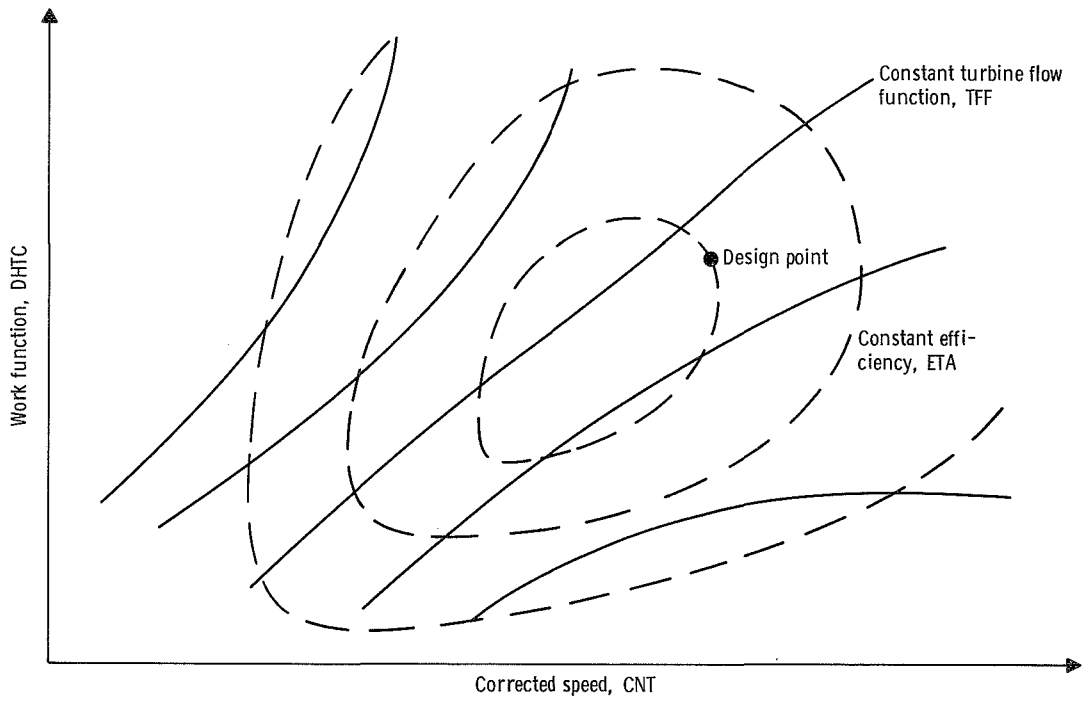
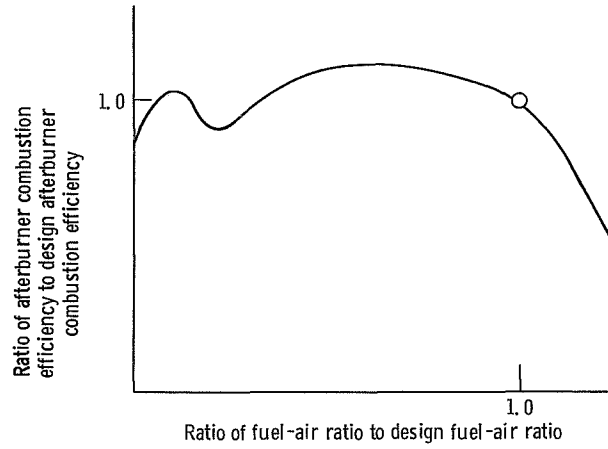
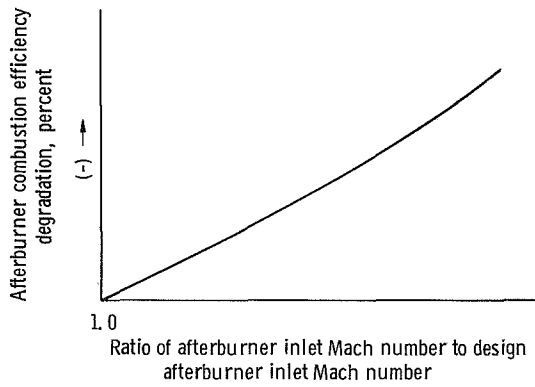


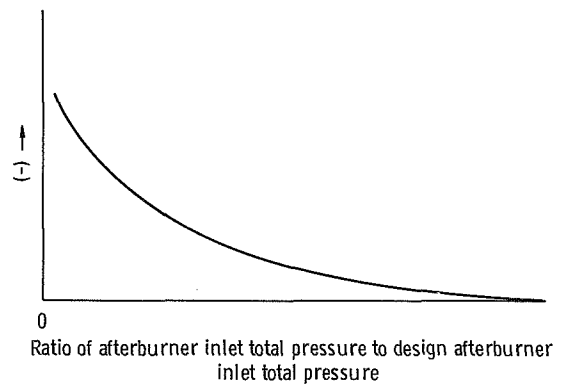
Figure 8. - Example of specific turbine map.



(a) Generalized afterburner combustion efficiency as function of fuel-air ratio.



(b) Efficiency correction factor as function of afterburner inlet Mach number.



(c) Efficiency correction factor as function of afterburner inlet total pressure ratio.

Figure 9. - Example of a generalized afterburner combustion efficiency performance map.

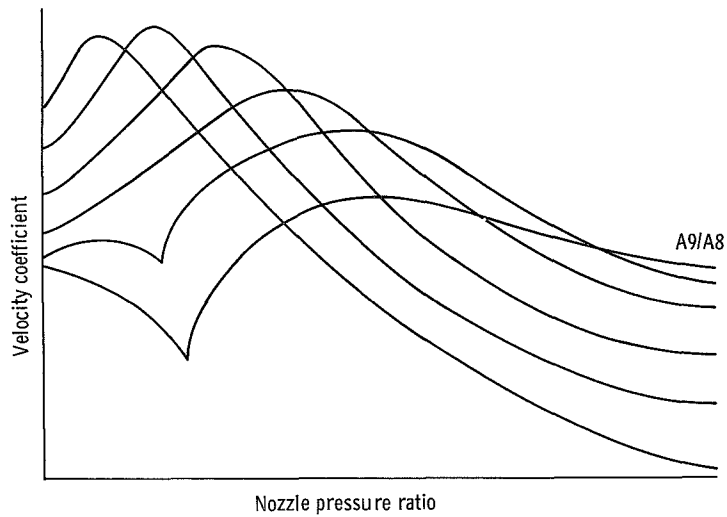


Figure 10. - Performance map for nozzle, giving velocity coefficient.



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